AOA System Owner's Manual





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This manual reflects the operation of System Software version 2.10 for the GI 260, or later. Some differences in operation may be observed when Comparing the information in this manual to later software versions.

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To obtain warranty service, contact your local Garmin Authorized Service Center. For assistance in locating a Service Center near you, visit the Garmin web site at http://www.garmin.com or contact Garmin Customer Service at 866-739-5687.



WARNING: For safety reasons, this AOA System's operational procedures must be learned on the ground.



CAUTION: This AOA System does not contain any user-serviceable parts. Repairs should only be made by an authorized Garmin service center. Unauthorized repairs or modifications could void both the warranty and the pilot's authority to operate this device under FAA regulations.



NOTE: All visual depictions contained within this document, including screen images of the GI 260 displays, are subject to change and may not reflect the most current GI 260 software. Depictions of equipment may differ slightly from the actual equipment.



NOTE: This product, its packaging, and its Components contain chemicals known to the State of California to cause cancer, birth defects, or reproductive harm. This notice is being provided in accordance with California's Proposition 65. If you have any questions or would like additional information, please refer to our web site at www.garmin.com/prop65.



NOTE: This AOA System is non-required and is to be used only as supplemental information to the pilot. This AOA System is not to be used or substituted for a certified stall warning system. No operational credit may be taken for reduced approach speed and shorted landing distances.



NOTE: The approved Pilot's Operating Handbook (POH) or Airplane Flight Manual (AFM) always supersedes this Owner's Manual.



NOTE: Refer to the AOA System Installation Manual for calibration instructions.



NOTE: This AOA system is designed to be accurate in the calibrated aerodynamic configuration. When the aircraft experiences changes in the airfoil shape (e.g., flap extension or icing accumulation), the AOA indicator may no longer accurately represent the angle of attack.

Record of Revisions			
Part Number	Revision	Date	Description
190-01773-00	А	09/2014	Initial release



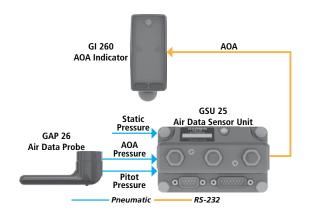
OVERVIEW

The Garmin AOA (Angle of Attack) System is designed to improve the pilot's awareness of the approximate available remaining lift of the aircraft's wings. The system calculates the approximate AOA (acute angle between the wing chord line and the relative wind) using pitot, AOA, and static air pressure inputs.

When correctly calibrated, the system provides a visual approach AOA reference, as well as increasing caution and warning annunciations as the AOA approaches the wing's maximum coefficient of lift (CLmax).

SYSTEM DESCRIPTION

The Garmin AOA System is comprised of three components; the GI 260 Indicator, the GAP 26 Probe, and the GSU 25 Air Data Computer. The GAP 26 sends pitot and AOA air pressure to the GSU 25. The GSU 25 measures the air pressure inputs from the probe and from an independent static source. The GSU 25 then calculates the AOA information and sends it to the GI 260. The GI 260 displays the AOA information to the pilot via ten color-coded LED annunciators. When calibrated correctly (refer to the AOA System Installation Manual for calibration instructions), the system indicates AOA during critical phases of flight. The system also provides awareness of AOA trends toward the target AOA for an approach, as well as visual alerting of critical AOA. When connected to an audio panel or compatible audio system, the GI 260 issues aural alerts of increasing frequency when the system approaches the critical angle of attack.





GI 260 CONTROLS



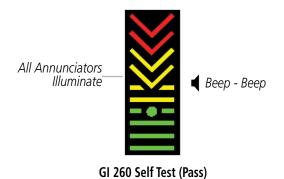
NOTE: The **TEST** and **MUTE** functions of the GI 260 are described in this manual. Refer to the AOA System Installation Manual for information on the **CAL** (calibration) and **SET** (alert volume) functions of the GI 260.



GI 260 Controls

SELF TEST

Press the **TEST** (CAL) button at any time to perform a self test of the GI 260. Upon successful completion of the self test, all annunciators illuminate and a "Beep - Beep" audible alert is played.





A failure is indicated by alternating flashing red chevrons and a "Beep - Beep" audible alert.



MUTE

START-UP

The system is muted by default upon start-up. The system remains muted until 15 seconds has elapsed since start-up and the Angle of Attack reaches the upper yellow chevron. The first slow audible "Beep-Beep" alert is heard when the upper yellow chevron annunciator illuminates

NORMAL OPERATION

Press the **MUTE** (SET) button to mute the GI 260 audible alert. When muted manually during normal operation, the audio alert is muted for at least 15 seconds and remains muted until the upper green bar illuminates for at least five seconds.



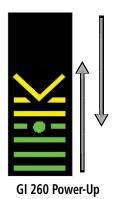
NORMAL OPERATION



NOTE: Refer to the approved Pilot's Operating Handbook (POH) or Airplane Flight Manual (AFM) for recommended operational procedures.

POWER-UP

The AOA system is ON by default during power-up. During power-up, the unit cycles its annunciators from bottom to top and back to the bottom. For the unit to cycle its annunciators it must receive valid AOA data from the GSU 25, determine it has a valid calibration, and pass the unit self test.



ARMING OF THE AOA SYSTEM

The AOA system arms automatically when the indicated airspeed exceeds 50 knots. Visual annunciations occur immediately upon arming (if applicable). Aural annunciations are delayed for 15 seconds after the system is armed.



CRUISE CONFIGURATION ANNUNCIATIONS

During flight at low angles of attack (cruise configuration), typically 0-1 annunciators may be illuminated.

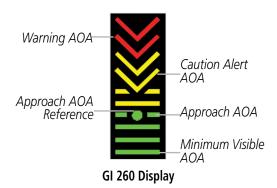


GI 260 (Cruise Configuration)

APPROACH CONFIGURATION ANNUNCIATIONS

As the angle of attack increases beyond the cruise configuration, the Approach AOA Reference (green circle) annunciator and the green bar annunciators begin to illuminate. The Approach AOA is reached when the green bar annunciators adjacent to the Approach AOA Reference illuminate.

The first slow audible "Beep-Beep" alert is heard when the upper yellow chevron annunciator illuminates. A fast audible "Beep-Beep" alert is heard when the red Warning AOA annunciator illuminates.





APPROACH AOA REFERENCE



NOTE: For information on calibrating the Approach AOA, refer to the calibration instructions in Appendix C of the AOA System Installation Manual.

The Approach AOA Reference (green circle) and the adjacent green bar annunciators should be calibrated to coincide with the published approach speed (if provided), or the speed upon crossing the runway threshold that is required in order to achieve calculated (or desired) aircraft landing performance. This speed is typically equal to or greater than 1.3 times the published stall speed in the landing configuration (Vso)

LOW AOA WITH APPROACH AOA REFERENCE

A low AOA approach is indicated by the illumination of less than four green bar annunciators and the Approach AOA Reference (green circle) annunciator.



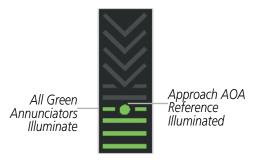
GI 260 (Low AOA with Approach AOA Reference)



APPROACH AOA

The Approach AOA is intended to align with the landing reference speed or threshold crossing speed. In some aircraft the Approach AOA may be consistent with Vref and equal to 1.3~x Vso (stall speed in the landing configuration).

The Approach AOA is indicated by the illumination of all green bar annunciators and the Approach AOA Reference (green circle) annunciator.



GI 260 (Approach AOA)

HIGH AOA WITH AUDIBLE ALERT

A high AOA approach is indicated by the illumination of the yellow Caution Alert AOA bar/chevron annunciators. A slow audible "Beep-Beep" alert coincides with illumination of the upper yellow chevron.





WARNING AOA WITH AUDIBLE ALERT



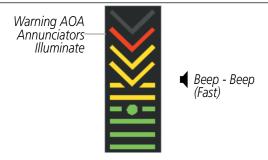
NOTE: For information on calibrating the Warning AOA, refer to the calibration instructions in Appendix C of the AOA System Installation Manual.

The first Warning AOA chevron is intended to coincide with the calibrated flap configuration (typically the landing configuration).

The Warning AOA is indicated by the illumination of the red Warning AOA annunciators and a fast audible "Beep-Beep" alert.



NOTE: During the landing flare, the AOA and stall warning may not coincide precisely due to ground effect, mounting of probe, etc.



GI 260 (Warning AOA)

EXAMPLE APPROACH

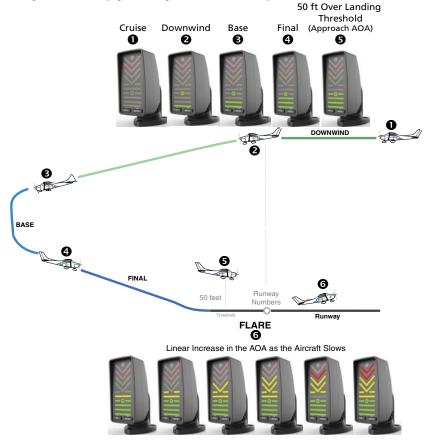


WARNING: This AOA System is non-required and is to be used only as supplemental information to the pilot, and may not be used as a substitution for the certified aircraft stall warning system.

A correctly calibrated AOA System will provide a linear increase in the AOA indication as the aircraft slows. The bottom red Warning AOA chevron is intended to coincide with the calibrated flap configuration (typically the landing configuration). It is recommended to simulate an approach to landing at a safe altitude to ensure that the lower Warning AOA chevron illuminates concurrently or prior to the first indication of the certified stall warning horn in the landing configuration.



The Approach AOA should be calibrated (refer to Appendix C of the AOA System Installation Manual) at an acceptable margin above CLmax to fly an approach. As a starting point, use the aircraft manual to determine the stall speed of the aircraft at the **actual gross weight** in the landing configuration. Multiply the calibrated airspeed by 1.3, then convert from calibrated airspeed to indicated airspeed (if necessary). Once the AOA angles have been calibrated, they will be accurate in the calibrated flap configuration, at any gross weight or altitude, every time.



Example Indications for a Typical Decelerating Approach and Flare



ABNORMAL OPERATION

AOA SYSTEM FAILURE

In the event that the GI 260 or the AOA system is malfunctioning, the unit can be powered-down by pulling the associated circuit breaker.

NUISANCE ALERTS

In the event that the AOA System is providing nuisance alerts, press the **MUTE** (SET) button to mute the GI 260 audible alerts. The audio remains muted for at least 15 seconds and remains muted until the upper green bar illuminates for at least five seconds.



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