

GARMIN®

G500(H)/G600/G700 TXi™

Pilot's Guide



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SOFTWARE VERSION

This manual reflects the operation of system software v3.86. Some differences in operation may be observed when comparing the information in this manual to later software versions.

INFORMATION & SUPPORT

For information regarding the [Aviation Limited Warranty](#), refer to Garmin's website.

For aviation product support, visit flyGarmin.com.

**WARNING**

Do not use terrain avoidance displays as the sole source of information for maintaining separation from terrain and obstacles. Garmin obtains terrain and obstacle data from third-party sources and cannot independently verify the accuracy of the information.

**WARNING**

Always refer to current aeronautical charts and NOTAMs for verification of displayed aeronautical information. Displayed aeronautical data may not incorporate the latest NOTAM information.

**WARNING**

Do not use geometric altitude for compliance with air traffic control altitude requirements. The primary barometric altimeter must be used for compliance with all air traffic control altitude regulations, requirements, instructions, and clearances.

**WARNING**

Do not use basemap information (land and water data) as the sole means of navigation. Basemap data is intended only to supplement other approved navigation data sources and should be considered only an aid to enhance situational awareness.

**WARNING**

Do not rely solely upon the display of traffic information to accurately depict all of the traffic within range of the aircraft. Due to lack of equipment, poor signal reception, and/or inaccurate information from aircraft or ground stations, traffic may be present that is not represented on the display.

**WARNING**

Do not use datalink weather information for maneuvering in, near, or around areas of hazardous weather. Information contained within datalink weather products may not accurately depict current weather conditions.

**WARNING**

Do not use the indicated datalink weather product age to determine the age of the weather information shown by the datalink weather product. Due to time delays inherent in gathering and processing weather data for datalink transmission, the weather information shown by the datalink weather product may be older than the indicated weather product age.



WARNING

Always refer to current aeronautical charts for appropriate minimum clearance altitudes. The displayed MSAs are only advisory in nature and should not be relied upon as the sole source of obstacle and terrain avoidance information.



WARNING

Do not use GPS to navigate to any active waypoint identified as a "NON WGS84 WPT" by a system message. "NON WGS84 WPT" waypoints are derived from an unknown map reference datum that may be incompatible with the map reference datum used by GPS (known as WGS84) and may be positioned in error as displayed.



WARNING

Do not rely solely upon the display of traffic information for collision avoidance maneuvering. The traffic display does not provide collision avoidance resolution advisories and does not under any circumstances or conditions relieve the pilot's responsibility to see and avoid other aircraft.



WARNING

Do not rely on the accuracy of attitude and heading indications in geographic areas where variation in the earth's magnetic field exists. This includes: North of 72° North latitude at all longitudes; South of 70° South latitude at all longitudes; North of 65° North latitude between longitude 75° W. and 120° W. (Northern Canada); North of 70° North latitude between longitude 70° W. and 128° W. (Northern Canada); North of 70° North latitude between longitude 85° E. and 114° E. (Northern Russia); South of 55° South latitude between longitude 120° E. and 165° E. (Region south of Australia and New Zealand). Aircraft equipped with GRS 7800 and configured for directional gyro (DG) free flight operations are only prohibited in the following regions due to unsuitability of the magnetic fields near the Earth's poles: North of 84° North latitude at all longitudes; South of 70° South latitude at all longitudes; South of 55° South latitude between longitude 120° E. and 165° E. (Region south of Australia and New Zealand).



WARNING

Do not rely solely upon datalink services to provide TFR information. Always confirm TFR information through official sources such as Flight Service Stations or Air Traffic Control.



WARNING

Do not rely on information from a lightning detection system display as the sole basis for hazardous weather avoidance. Range limitations and interference may cause the system to display inaccurate or incomplete information. For detailed information about the system, consult documentation from the lightning detection system manufacturer.

**WARNING**

Use appropriate primary systems for navigation, and for terrain, obstacle, and traffic avoidance. Garmin SVT is intended as an aid to situational awareness only and may not provide either the accuracy or reliability upon which to solely base decisions and/or plan maneuvers to avoid terrain, obstacles, or traffic.

**WARNING**

Do not use the Garmin SVT runway depiction as the sole means for determining the proximity of the aircraft to the runway or for maintaining the proper approach path angle during landing.

**WARNING**

Do not use TAWS information for primary terrain or obstacle avoidance. TAWS is intended only to enhance situational awareness.

**WARNING**

Do not rely solely on VNAV guidance when navigating horizontally and vertically around user-defined airports. It is the pilot's responsibility to ensure separation from terrain and obstacles during an approach to a user-defined airport.

**WARNING**

Be aware that this product, its packaging, and its components contain chemicals known to the State of California to cause cancer, birth defects, or reproductive harm. This notice is being provided in accordance with California's Proposition 65. If you have any questions or would like additional information, please refer to our website at www.garmin.com/prop65.

**CAUTION**

Ensure that any unit Repairs are made by an authorized Garmin service center. Unauthorized repairs or modifications could void both the warranty and affect the airworthiness of the aircraft.

**CAUTION**

Do not clean display surfaces with abrasive cloths or cleaners containing ammonia. They will harm the anti-reflective coating.



NOTE

Foreign materials (i.e., moisture, objects) that come into contact with the touchscreen can prevent the unit from responding to touch commands. To restore functionality, wipe the screen dry and/or remove interfering objects.



NOTE

Intruder aircraft at or below 500 ft AGL may not appear on the Garmin SVT display or may appear as a partial symbol.



NOTE

All visual depictions contained within this document, including screen images of the system panel and displays, are subject to change and may not reflect the most current system and aviation databases. Depictions of equipment may differ slightly from the actual equipment.



NOTE

The United States government operates the Global Positioning System and is solely responsible for its accuracy and maintenance. The GPS system is subject to changes which could affect the accuracy and performance of all GPS equipment. Portions of the system utilize GPS as a precision electronic NAVAID. Therefore, as with all NAVAIDs, information presented by the system can be misused or misinterpreted and, therefore, become unsafe.



NOTE

This device complies with part 15 of the FCC Rules. Operation is subject to the following two conditions: (1) this device may not cause harmful interference, and (2) this device must accept any interference received, including interference that may cause undesired operation.



NOTE

Interference from GPS repeaters operating inside nearby hangars can cause an intermittent loss of attitude and heading displays while the aircraft is on the ground. Moving the aircraft more than 100 yards away from the source of the interference should alleviate the condition.



NOTE

Use of polarized eye wear may cause the flight displays to appear dim or blank.

**NOTE**

Operating the system in the vicinity of metal buildings, metal structures, or electromagnetic fields can cause sensor differences that may result in nuisance miscompare annunciations during start up, shut down, or while taxiing. If one or more of the sensed values are unavailable, the annunciation indicates no comparison is possible.

**NOTE**

The system responds to a terminal procedure based on data coded within that procedure in the Navigation Database. Differences in system operation may be observed among similar types of procedures due to differences in the Navigation Database coding specific to each procedure.

**NOTE**

The FAA has asked Garmin to remind pilots who fly with Garmin database-dependent avionics of the following:

- *It is the pilot's responsibility to remain familiar with all FAA regulatory and advisory guidance and information related to the use of databases in the National Airspace System.*
- *Garmin equipment will only recognize and use databases that are obtained from Garmin or Jeppesen. Databases obtained from Garmin or Jeppesen that have a Type 2 LOA from the FAA are assured compliance with all data quality requirements (DQRs). A copy of the Type 2 LOA is available for each applicable database and can be viewed at flyGarmin.com by selecting "Aviation Database Declarations."*
- *Use of a current Garmin or Jeppesen database in your Garmin equipment is required for compliance with established FAA regulatory guidance, but does not constitute authorization to fly any and all terminal procedures that may be presented by the system. It is the pilot's responsibility to operate in accordance with established AFM(S) and regulatory guidance or limitations as applicable to the pilot, the aircraft, and installed equipment.*

**NOTE**

The pilot/operator must review and be familiar with Garmin's database exclusion list as discussed in SAIB CE-14-04 to determine what data may be incomplete. The database exclusion list can be viewed at flyGarmin.com by selecting "Database Exclusions List."

**NOTE**

The pilot/operator must have access to Garmin and Jeppesen database alerts and consider their impact on the intended aircraft operation. The database alerts can be viewed at flyGarmin.com by selecting "Aviation Database Alerts."

**NOTE**

If the pilot/operator wants or needs to adjust the database, contact Garmin Product Support.



NOTE

Garmin requests the flight crew report any observed discrepancies related to database information. These discrepancies could come in the form of an incorrect procedure; incorrectly identified terrain, obstacles and fixes; or any other displayed item used for navigation or communication in the air or on the ground. Visit flyGarmin.com and select "Aviation Data Error Report."



NOTE

When using autopilot to fly an approach with vertical guidance, the autopilot will not level the aircraft at the MDA/DH even if the MDA/DH is set in the altitude preselect.



NOTE

Information obtained solely through FIS-B does not replace a thorough preflight briefing. NOTAMs received via FIS-B uplink may not be a complete listing. Active NOTAMs are removed from the FIS-B data stream 30 days after issuance. FIS-B uplink is not an FAA approved source for NOTAMs.



NOTE

When using Stormscope, there are several atmospheric phenomena in addition to nearby thunderstorms that can cause isolated discharge points in the strike display mode. However, clusters of two or more discharge points in the strike display mode do indicate thunderstorm activity if these points reappear after the screen has been cleared.



NOTE

The navigation databases used in Garmin navigation systems contain Special Procedures. Prior to flying these procedures, pilots must have specific FAA authorization, training, and possession of the corresponding current, and legitimately-sourced chart (approach plate, etc.). Inclusion of the Special Procedure in the navigation database DOES NOT imply specific FAA authorization to fly the procedure.



NOTE

When operating the system with the magnetic sensor uncoupled from the AHRS, the displayed heading and heading information used by some system components (e.g., AFCS, traffic system, weather radar) will be different from the heading calculated by the AHRS. The difference is an amount equal to the difference between the current Magnetic Field Variation Database (MV DB) value and the MV DB value when the magnetic sensor was uncoupled. Due to the convergence of isogonic lines, this condition is most noticeable at or near the north and south magnetic poles. If the displayed heading is adjusted after decoupling, additional errors in system displayed headings and relative bearings may occur.

Record of Revision

REVISION	DATE	CHANGE DESCRIPTION
U	01.16.26	Updates for software v3.86.
T	09.03.25	Updates for software v3.85.
S	07.14.25	Addition of GDU 1210.
R	06.12.25	Updates for software v3.80.
Q	01.08.25	Updates for software v3.70.
P	04.01.24	Updates for software v3.61.
N	01.04.23	Updates for software v3.50.
M	01.25.22	Updates for software v3.40.
L	07.23.21	Updates for software v3.30.
K	06.21.21	Updates for software v3.21.
J	10.15.20	Updates for software v3.12.
H	05.21.19	Updates for software v3.01.
G	03.14.19	Updates for software v3.00.
F	10.16.18	Updates for software v2.30.
E	08.17.18	Updates for software v2.20.
D	01.24.18	Minor edits.
C	01.18.18	Minor edits.
B	01.16.18	Production Release.
A	05.19.17	Initial Release.

Available for Download

The following documents are available for viewing on your computer or portable device. Go to garmin.com/manuals and enter the product name or serial number.

Electronic Pilot's Guide

A version of this guide saved in Adobe Acrobat.

Upgrade Supplement

Details document changes for software enhancements.

Layout

SECTION	TITLE
1	System at a Glance
2	Get Started
3	Primary Flight Display
4	Advanced Features
5	Multi-Function Display
6	Weather Awareness
7	Traffic Awareness
8	Terrain Awareness
9	Fuel & Engine Indication System
10	Abnormal Operations
11	Messages
12	Qualification
13	Glossary
14	Regulatory Information

The layout and design of this guide is intended to provide clear, concise sections written in the logical order of a pilot’s flight instrument and systems scan.

Product Descriptions

This guide covers the operation of the following Garmin display models.

GDU 700L
 GDU 700P
 GDU 1060
 GDU 1210

When you see	It means
a product name in bold (e.g., GDU 1060)	the information pertains to that specific model only
a reference to GDU 700()	the information pertains to both GDU 700 display models (i.e., GDU 700L and GDU 700P)
product names separated by a forward slash (e.g., GDU 700P/1060/1210)	the information pertains to all listed displays

Feature Depictions

All graphical representations are provided to expand and enhance the text. Hardware and software feature depictions are not to scale.

Except for cases when operational differences dictate, the GDU 1060 display model is shown as typical.

Special Notices

This guide uses the following message types when applicable.



WARNING

Indicates when serious injury or death will occur.



CAUTION

Indicates when equipment damage is possible.



NOTE

Emphasizes a point about a specific feature, function, or operation.

Information Insets

Gray boxes highlight information about a particular feature or application function.

Green boxes may include references to additional information or a useful pilot's tip.

Blue boxes address common questions about Garmin product features and services.

Interactive Icons

The electronic version of this guide uses interactive links throughout to improve the reading experience.



When you see this icon in the upper right corner, it means that objects on the page are selectable. Tapping an object takes you to the associated material.

Reference Manuals

DOCUMENT	P/N
<i>GDL 69/69A SiriusXM Satellite Radio Activation Instructions</i>	190-00355-04
<i>G500(H)/G600/G700 TXi Pilot's Guide Addendum for GDL 60</i>	190-01717-40
<i>GTN Xi Series Pilot's Guide</i>	190-02327-03

Reference Websites

WEBSITE	QR CODE (SCAN/GO)
ADS-B Academy https://www.garmin.com/en-US/aviation/adsb/	
Aviation Limited Warranty https://www.garmin.com/en-US/legal/aviation-limited-warranty	
Connex Worldwide Weather https://fly.garmin.com/fly-garmin/connex/worldwide-weather/	
Database Concierge Go to http://www.flygarmin.com/support and select Database Management.	
FAA Dynamic Regulatory System https://drs.faa.gov	
Garmin Aviation Weather Radar 2.0 eLearning Course https://buy.garmin.com/en-US/US/p/pn/AVNE-GMWXOP-A1	
TXi Essentials 2.0 eLearning Course https://buy.garmin.com/en-US/US/p/pn/AVNE-GM650P-B1	

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Overview

G500(H)/G600/G700 TXi is a display and sensor system available in four display options.

GDU 1210 12" display¹

GDU 1060 10" display

GDU 700P 7" portrait display

GDU 700L 7" landscape display

GDU 1210



GDU 1060



GDU 700P



GDU 700L



Garmin Display Units

¹ Fixed-wing aircraft only.

Unit Configurations

Depending on system specifics, one or more of the following functions may apply.

- *Primary Flight Display (PFD)*: provides attitude, heading, air data, and navigation information to the pilot
- *Multi-Function Display (MFD)*: provides pilot awareness of factors that may affect the overall conduct of a flight
- *Engine Indication System (EIS)*: provides engine and airframe operating parameters to the pilot

GDU 700L PFD, EIS, or MFD/EIS

The GDU 700L provides a single PFD, piston EIS, or in single-engine piston EIS, MFD/EIS combined function.

When configured as an MFD/EIS display, engine instruments are dedicated to 40% of the screen. The remaining screen portion displays all configured MFD options using a pilot-selectable menu.



Piston EIS Only



PFD Only



Single Engine Piston EIS/MFD

GDU 700P PFD, MFD, EIS, MFD/EIS

GDU 700P provides a single PFD, MFD, EIS, or, in single-engine piston EIS, a combined MFD/EIS function. In some installations, it provides backup PFD or EIS information in the event that the primary PFD or EIS display fails or malfunctions.

When configured as an MFD/EIS display, engine instruments are dedicated to 40% of the screen. The remaining screen portion displays all configured MFD options using a pilot-selectable menu.



EIS Only



PFD Only



MFD Only



EIS/MFD¹

¹ Single-engine piston EIS only.

GDU 1060 PFD/MFD, EIS Option

GDU 1060 provides an MFD or PFD/MFD display with an optional EIS function. Each function is designated to a portion of the display.

When configured as an MFD, the system can display two MFDs, or you can expand one MFD to fill the screen.

When configured as a PFD/MFD, the MFD can be selectively hidden or shown. When hidden, the PFD expands into the MFD space.

If configured as a PFD/MFD or MFD with EIS, the EIS function is always visible on either the left or right edge of the display.



PFD Only



PFD/EIS



MFD (40%)/PFD (60%)



MFD/PFD/EIS



MFD/MFD



MFD/MFD/EIS

Display Features

PRIMARY FLIGHT DISPLAY		
GDU 700()/1060/1210 ⁸		
• Attitude • Airspeed • Altitude • Vertical Speed • Turn Coordinator • G-meter⁵ • HSI • HSI Map³ • Clock	• Lateral and Vertical Deviation Indicators • Datalink Weather Display¹ • Radar Altimeter¹ • Autopilot Annunciations¹ • Flight Director¹ • Synthetic Vision¹	• Flight Path Marker¹ • System Advisories • Safety Monitors¹ • GPS NAV Status • Display Backup^{1, 3} • Terrain Avoidance¹ • Smart Glide^{4, 6} • Stabilized Approach Alerts⁷
MULTI-FUNCTION DISPLAY		
GDU 700P/1060/1210 ^{2, 8}		
• Navigation Map • Traffic¹ • Terrain • Charts • Flight Plan • Weather¹	• Waypoint Info • Music Services¹ • Terrain Avoidance¹ • Engine Data¹ • Remote Database Confirmation^{4, 6}	• Remote Radio Tuning^{4, 5} • Smart Glide^{4, 6} • System Advisories • Video¹

¹ Function availability dependent upon aircraft interfaces or enablement.
² GDU 700() MFD/EIS provides the same MFD and EIS functionality listed with the exception of weather radar and multi-engine. ³ Not available for GDU 700L.
⁴ Feature requires a GTN Xi series navigator. ⁵ Available with TXi software v3.21 and later.
⁶ Available with TXi software v3.30 and later. ⁷ Available with TXi software v3.61 and later.
⁸ GDU 1210 is for fixed-wing aircraft only.

ENGINE INDICATION SYSTEM¹

GDU 700()/1060/1210^{2, 4, 7}

- | | | |
|-----------------------------------|---|------------------------------------|
| • Fuel Qty
(Main, Aux, etc.) | • Fuel Flow | • Amps/Volts |
| • RPM/Tach | • Fuel Pressure | • N1/N2 ⁶ |
| • Propeller Sync
Display | • Fuel Calculations | • N1 Bug ⁶ |
| • Automatic Starting ⁵ | • Fuel Temperature ⁶ | • User Selectable Fields |
| • Automatic Ignition | • Cylinder Operating
Temperatures
(CHT, EGT) | • User Adjustable
Advisories |
| • Manifold Pressure | • TIT | • Torque ³ |
| • Oil Pressure | • Lean Assist Mode | • Torque Bug ⁶ |
| • Oil Temperature | • Carburetor Air
Temperature | • Turbine Temperature ³ |
| • Shaft Horsepower ³ | • Intercooler
Temperatures
(IAT, CDT, Difference) | • Vacuum/Pressure |
| • Percent Power | | • Flight Control Trim
Position |
| • OAT ⁵ | | |

¹ Displayed engine operating parameters dependent upon configuration.

² GDU 700() MFD/EIS provides the same MFD and EIS functionality listed with the exception of weather radar and multi-engine. ³ Turboprop only.

⁴ Twin turbine aircraft: full-time EIS functions are available on GDU 700P only. When configured, the MFD Engine page provides engine data for the aircraft.

⁵ Available with TXi software v3.21 and later. ⁶ Available with TXi software v3.61 and later.

⁷ GDU 1210 is for fixed-wing aircraft only.

Pilot Interface



Bezel

The bezel provides a combination of mechanical controls and features. Ledges provide hand stability when performing data entry and making selections on the multi-touch display.

Power Key¹

Powers the unit on or off and allows manual transition into display backup mode.

Photocell

Measures the cockpit ambient light level to automatically adjust display brightness for day and night.

SD Card Slots

Each unit has two SD card slots:

- Top and bottom on GDU 700P/1060
- Left and right on GDU 700L/1210

The purpose of each slot depends on the display type.

Inner & Outer Knobs

All TXi units have a multi-purpose dual concentric knob near the bottom of the bezel. GDU 1060 and GDU 1210 have two, one on each side. Inner and outer knob functionality differs according to display features.

To view the specific mechanical features for your display type, refer to the appropriate illustration in this section.



GDU 700L



GDU 700P



GDU 1060



GDU 1210

¹ GDU 700L / 1060 only.



1	Bezel	4	SD Card Slots
2	Power Key	5	Inner & Outer Knobs
3	Photocell		

GDU 700P



1	Bezel	4	SD Card Slots
2	Power Key	5	Inner & Outer Knobs
3	Photocell		



1	Bezel	4	SD Card Slots
2	Power Key	5	Inner & Outer Knobs
3	Photocell		

GDU 1210



1	Bezel	3	SD Card Slots
2	Photocell	4	Inner & Outer Knobs

SD Card Slot



NOTE

Do not remove or insert an SD card while in flight. Always verify the system is powered off before inserting or removing an SD card.

Determine which card slot is appropriate for the task.

TASK	SD CARD SLOT	
	GDU 700P/1060	GDU 700L/1210
Exporting data logs	Top	Left
Saving system configurations		
Transferring crew profiles		
Capturing screen images ¹		
Enabling Flight Stream 510 connectivity ²	Bottom	Right
Updating databases	Top or bottom	Left or right

¹ Screen images save to the bottom/right card when the top/left card is unavailable.
² Install Flight Stream 510 in a GTN, if available.

Formatting the SD Card or Flight Stream 510

In the event that there is a file corruption or formatting issue with the SD card, or the Flight Stream 510 wireless transceiver when used as a database storage device, it may be necessary to reformat.

Formatting the SD card, or the wireless transceiver, improperly can cause it to not be recognized by the avionics system. To avoid this, and to ensure maximum compatibility with Garmin avionics and your personal computer, format the SD card, or wireless transceiver, with the Master Boot Record (MBR) partition scheme and the FAT32 (MS-DOS) file system format. Garmin recommends using the SD Memory Card Formatter application available as a download from [SDcard.org](https://www.sdcard.org). When running the application, use the Quick Format option.

For more information, visit the Aviation Product Support page at flyGarmin.com.

INSERT AN SD CARD

When inserting an SD card:

1. Verify unit power is off and the slot is empty.
2. Hold card in the proper orientation.
 - *GDU 700P/1060*: label faces left edge of display screen
 - *GDU 700L/1210*: label faces top edge of display screen
3. Ensure back edge of card is flush with the display bezel after insertion.

EJECT AN SD CARD

1. Remove power from the unit.
2. Release the spring latch by pressing lightly on exposed edge of card.

SD Card Full



NOTE

SD card full advisories may not occur with 32 GB cards.

Pilot PFD bottom data card full

The system issues an advisory message when it detects that the SD card is approaching full. TXi stops writing data to the card once it becomes full.

To continue writing data:

1. Remove the SD card from the indicated slot once the aircraft is on-ground and the unit is powered off.
2. Insert a card with sufficient free space and power on the unit.

Touchscreen

A multi-touch color display provides on-screen controls for unit operation.

GESTURES

TAP



Touching the screen briefly with a single finger.

Use this gesture for:

- Opening a page or menu
- Activating a command key or data entry field
- Displaying map feature information
- Selecting an option within an application

TAP AND HOLD



Certain momentary controls (e.g., directional arrow keys) provide a secondary tap and hold function. Tap the key and hold your finger in place until the desired action occurs.

Use this gesture for:

- Scrolling with arrow keys
- Enabling Caps Lock
- Increasing/decreasing values continuously

SWIPE



A smooth motion that involves touching an object, then sliding your finger across the screen and lifting up.

Use this gesture for:

- Viewing and scrolling lists
- Panning across a map or chart display
- Zooming HSI map or rotorcraft PFD VFR map display (upward and downward swipes only)

FLICK



Swiping the screen in a quick upward or downward motion. Information moves at a fast speed (faster than by holding the arrow key), then slows to a stop.

Use this gesture for:

- Scrolling an item list

PINCH & STRETCH



Touch any map or chart with two fingers at the same time, then bring the fingers close together (pinch) or spread them apart (stretch). Just remember: stretch to zoom in and pinch to zoom out.

Use this gesture for:

- Magnifying map features
- Identifying multiple objects in close proximity
- Reading details on a chart
- Magnifying a video image

Keys

COMMON COMMANDS



Displays Advisories page. Blinking icon indicates unread system advisories.



Inputs specified value.



Displays context menus.



Scrolls up.



Returns to previous page.



Scrolls down.



Cancels current function without inputting data.



Indicates corresponding item is selected (e.g., database update).



Adjusts MFD and PFD display sizes between 40% and 60%.¹



Switches display to full screen.¹



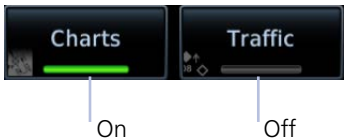
Switches from full PFD to split MFD and PFD.¹



Switches from full MFD to split dual MFDs.¹

¹ GDU 1060/1210 only.

FUNCTION KEYS



Toggle keys turn a specific function on or off. The current state of the function is indicated below the key label.

Menus

Menu group related controls into an expandable pane, allowing access to multiple functions on a single page. Depending on the number of available functions, a menu may comprise more than one pane.

SLIDE OVER MENUS

These menus slide out from the bottom or sides of the display when an object or menu item is selected on the underlying page.

Underlying
Page
Hidden



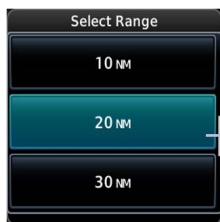
Tapping the underlying page closes the menu.

LISTS



Scrollable lists group control keys related to a single function (e.g., selectable range options). When scrolling, all keys in the list are inactive.

Scroll Bar



Popup lists open to the default or previously selected value.

Default or Current Setting

Tabs

Inactive
Tabs



Tabs group information into individual panes. Content includes scrolling lists, data fields, function keys, or a combination of controls.

Tabs are located along the left side or bottom edge of a pane.

Toggle
Keys



Lists containing a combination of controls typically appear within a tab and open to the first list item.

Toggle keys either enable or disable list items. In some cases, **Settings** or **Range** keys provide access to selectable setting options.

Keypads



Numeric



Alphanumeric

The GDU employs multiple keypad types to serve specific settings and functions.

Tapping **Layout** changes the format between ABCDE and QWERTY.

QWERTY

Control Knobs



Inner and outer knobs offer an alternative method for selecting and modifying data without the use of touch keys.

GDU 1060/1210: The dual-concentric knobs control the adjacent PFD or MFD display by default.

PFD KNOB FUNCTIONS

Outer Knob

- Selecting reference controls
- Cursor placement and initial field/page selections
- Moving cursor forward or backward within data field

**Inner Knob
(Turn)**

- Selecting reference values
- Inputting data
- Modifying individual characters in data entry field
- Performing coarse or fine altimeter bug adjustments¹

**Inner Knob
(Push)**

- Entering current or specified numerical value
- Synchronizing PFD reference to its current value
- Alternating between standard and pilot set barometric pressure

**Inner Knob
(Push and Turn)**

- Performing fine altimeter bug adjustments¹

Dual-knob PFD Control

For convenience, you may use both knobs to control the PFD. Pushing and holding the inner knob adjacent to the MFD toggles focus between MFD and PFD control functions. MFD control via touchscreen is still available. This feature is particularly useful if your cockpit configuration makes it difficult to access the PFD knob.

¹ Dependent upon installer configuration and altitude knob pilot setting (if applicable).

MFD KNOB FUNCTIONS

Outer Knob	• Selecting a page shortcut • Cursor placement and initial field/page selections • Moving cursor forward or backward within a data field
Inner Knob (Turn)	• Inputting data • Modifying individual characters in data entry field • Zooming • Controlling airborne weather radar and Stormscope lightning display range
Inner Knob (Push)	• Entering a specified numerical value
Inner Knob (Push and Hold)	• Switching between MFD and PFD control functions

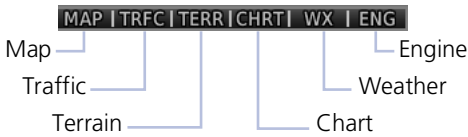
EIS KNOB FUNCTIONS

Outer Knob	• Cursor placement and initial field/page selections • Moving cursor forward or backward within data field • Selecting bug operating modes
Inner Knob (Turn)	• Inputting data • Modifying individual characters in data entry field • Adjusting bug value and indicator position • Switching between CHT/EGT modes
Inner Knob (Push)	• Entering a specified numerical value • Resetting lean/normalize value • Toggling the selected operating mode on or off

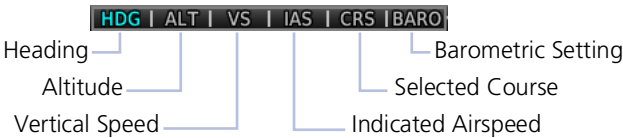
Knob Function Indicators

A locator bar works in conjunction with the outer knob providing quick access to the indicated display feature: MFD page, PFD bug, EIS mode. Selectable options are dependent upon configuration.

MFD



PFD



EIS

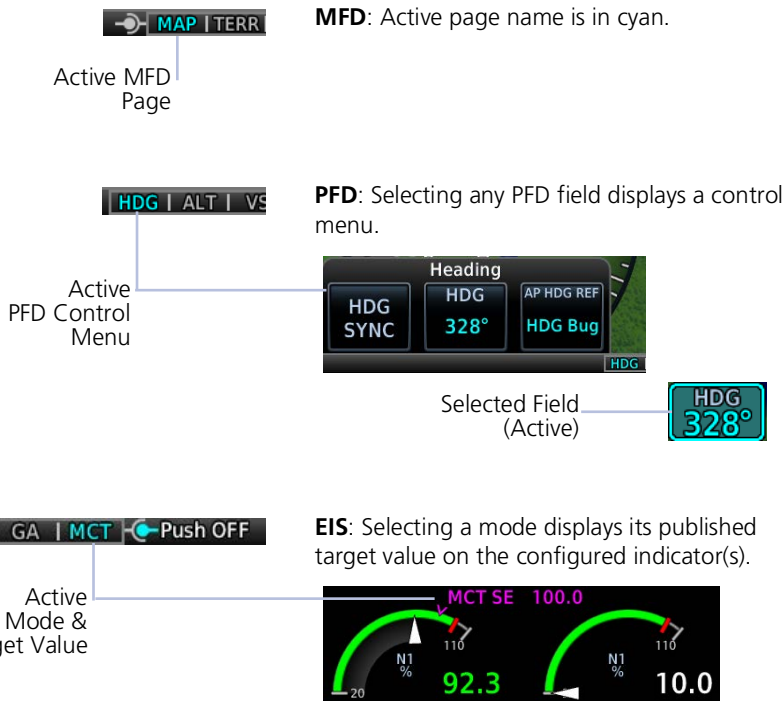


Additional EIS modes may be available depending on configuration.

System at a Glance

Turning the outer knob clockwise or counter-clockwise moves the locator through displayed menu options. Changes in the current screen configuration may result in a change of menu options. When the knob adjacent to the MFD is controlling the PFD, its locator bar displays PFD menu options.

Additional icons located to the left or right of the bar indicate available knob functions for the associated display.



Screenshots

FEATURE REQUIREMENTS

- SD card in the FAT32 format, with memory capacity between 8 GB and 32 GB

Save images to an SD card at any time using Screenshot. Images automatically save to the “print” folder in the SD card root directory.

1. Verify that an SD card is present in the top or left slot.
2. Go to the page of interest.
3. Push the appropriate bezel controls to take a screenshot.
 - *GDU 700()/1060*: Push and hold the right inner knob. With the knob depressed, push and release the **Power** key.
 - *GDU 1210*: Push and hold the right inner knob. With the knob depressed, push and release the left inner knob.



A camera icon momentarily shows in the annunciator bar indicating a successful screenshot. To view saved images, remove the SD card and open the “print” folder on a computer.

For more about proper SD card installation, read *SD Card Slot*.

Color Conventions



Red

- Warning conditions
- Operating limits



Yellow

- Cautionary conditions
- Conditional operating ranges



Green

- Safe operating conditions
- Normal operating ranges/status
- Active mode and function indications
- VOR/localizer/DME data
- Engaged autopilot modes



White

- Scales and markings
- Current data and values
- Armed autopilot modes



Magenta

- Active flight plan legs
- Computed bugs
- GPS data
- VNAV data



Cyan

- Pilot-selectable controls (e.g., references, AFCS alert annunciations)
- Glide range and best glide airport indications
- User mag var updates



Gray

- Missing or expired data
- Product unavailable



Blue

- Sky
- Water



Brown

- Ground

Compatible Equipment

Line Replaceable Units

SYSTEM REQUIRED LRUs (PFD)
ADAHRS or ADC with AHRS
GMU 44/44B
Garmin GPS navigator
Temperature probe
SYSTEM OPTIONAL LRUs
Backup GPS antenna
GAD 43/43e adapter
GBB 54 battery ¹
GCU 485 PFD controller
GEA 110 engine airframe interface ²
GEA 71 engine airframe interface ³
OPTIONAL INTERFACES
ADF
AOA computer
Airborne weather radar
Autopilot/flight director
DME
G5
GCO 14
GDL 69/69A SiriusXM datalink
GDL 88/GTX 345 ADS-B transceiver
GI 275
GSR 56 satellite datalink
GTX 330 or GTX 335 mode S transponder (TIS-A)
Radar altimeter
Stormscope
TAS/TCAS/TCAS II
VHF NAV and glideslope receiver

The TXi system consists of multiple LRUs, which are installed behind the instrument panel or in a separate avionics bay. Their modular design aids system maintenance and unit replacement.

Optional LRUs may include compatible equipment from either Garmin or a third-party manufacturer.

Some LRUs provide features that require registration and/or enablement prior to activation.

Check unit software version for compatibility.

¹ GBB 54 is only an option for GDU 700().

² Piston aircraft only.

³ Turboprop aircraft only.

LRU Status

Devices that interface to the system via Ethernet display product information on the System Status page. Some interfaced equipment provide controls for product registration, enablement or viewing detailed status information. For additional information, tap **More Info**.



The device is configured and communicating properly.



The device is not available and is not configured or it is not communicating properly.

Integrated Standby Instrument

PISTON AIRCRAFT ONLY

An integrated standby instrument configuration consists of two GDU 700Ps, each with separate AHRS/ADC sensors. One display serves as the PFD, the other as EIS, MFD, or MFD/EIS. When activated, the display backup switch forces both displays into their display backup mode of operation.

The GBB 54 backup battery provides power to the PFD and sensors in the event of primary aircraft power failure. Both displays monitor and compare independent attitude, altitude, and airspeed data. The EIS, MFD, or MFD/EIS automatically displays its backup primary flight information when:

1. Either display detects a miscompare between attitude, altitude, or airspeed parameters.
2. Communication between displays is lost.
3. The display backup toggle switch is active.

This configuration provides the following functionality, which satisfies the backup instrumentation requirements.

The MFD, EIS, or MFD/EIS behaves as follows:

- An MFD will transition to a full-screen PFD display
- An EIS will transition to a split-screen PFD/EIS display
- An MFD/EIS will transition to a split-screen PFD/EIS display

Standby Instruments

Except for installations limited to VFR, PFD installations require standby attitude, altitude, and airspeed instruments. Several types of standby instruments are acceptable, including individual analog instruments and certain electronic standby indicators.

G5 STANDBY ATTITUDE INSTRUMENT

FEATURE LIMITATIONS

- Available functionality dependent upon installation and configuration settings
- G5 standby (not part of GFC 500 system) synchronization functionality (an optional capability) requires installation of GAD 29B adapter

G5 is an optional electronic flight instrument capable of operating as a standalone flight display or as a fully integrated standby instrument.

GFC 500	G5 FUNCTION
Present	• Backup autopilot control and GPS coupling • Standby attitude display
Not present	• Standby attitude display

G5 Standby Instrument Features

- Mirrors GDU control settings for Barometric Correction and Selected Course

G5 Features when Installed with GFC 500 system

- Automatic synchronization of the following PFD controls: Selected Airspeed, Selected Altitude, Selected Heading, and Selected Vertical Speed
- Mirrors GDU control settings for Barometric Correction and Selected Course
- Displays CDI navigation source selection data received from the GDU

Unit Synchronization

To initiate synchronization, set the G5 barometric pressure setting to match the corresponding control setting on GDU. Both units remain in sync until the barometric pressure setting is manually adjusted on G5.

Synchronization occurs after a short delay. The duration of this delay varies depending on which unit is attempting to sync.

CONDITION	SYNCING OCCURS...
Pilot adjusts barometric setting on G5 to match corresponding correction control setting on GDU.	After 5 seconds
Pilot adjusts barometric correction control setting on GDU to match corresponding setting on G5.	After 1 minute

GI 275 STANDBY ATTITUDE INSTRUMENT

FEATURE REQUIREMENTS

- *GI 275 must be configured as a Full Time ADI, Standby ADI, or Standby ADI as Primary HSI*

FEATURE LIMITATIONS

- *Available functionality dependent upon installation and configuration settings*

GI 275 is a multifunction indicator capable of operating as a PFD, HSI, EIS, MFD, radar altimeter display, and/or autopilot interface.

When a GI 275 configured as a standby attitude instrument (ADI) detects a fault, the unit exclusively behaves as a primary attitude instrument until the fault is resolved. Alternatively, the crew can select reversionary operation modes via panel-mounted switches.

ADI CONFIGURATIONS & FUNCTIONS

Basic	• Standby attitude
3-in-1	• Standby attitude • Standby airspeed • Standby altitude
4-in-1	• Standby attitude • Standby airspeed • Standby altitude • Magnetic heading

Depending on configuration, the GI 275 ADI may have functions in addition to the basic standby attitude display.

GI 275 Standby Instrument Features

- May be configured to mirror GDU control settings for Barometric Correction and Selected Course.

PFD Controller

FEATURE LIMITATIONS

- *Applicable to PFD only*
- *Selecting a data field does not automatically open its control menu*
- *Available control options and pushbutton configuration dependent upon the interfacing autopilot*

GCU 485 is an optional PFD controller providing dedicated knobs and keys for setting reference bugs and executing autopilot commands. Lighting may be controlled by the aircraft lighting or the internal photocell (dependent upon configuration).

If configured, the PFD controller is used to control a Standby PFD.

Although control of PFD functions is available using the GDU knobs and touchscreen, GCU 485 offers an alternative means of data entry from a panel-mounted device. For more about setting reference bugs using the controls on GDU, read *Reference Bugs and Controls*.

MODELS -01, -02, -03, -04 & -05: ARM/GPSS/ENG/CDI

FEATURE LIMITATIONS

- Available functions dependent upon model type

Knob Functions:

- Altitude • Heading • Airspeed • Vertical Speed • BARO Setting

Key Functions:

- CDI Source Selection • GPSS Emulation Mode • Altitude Capture Arming
- Vertical Speed Mode Engagement

GCU 485-01, ARM/GPSS/ENG/CDI



CONTROL OPTIONS	
ALT	Sets the selected altitude bug.
HDG	Sets the selected heading bug.
BARO	Sets the selected barometric pressure value. Pushing the knob reverts the setting to standard.
IAS	Sets the selected indicated airspeed.
VS	Sets the selected vertical speed.
ARM	Arms the autopilot altitude capture function.
CDI	Changes the NAV source for the CDI.
ENG	Engages the autopilot vertical speed mode.
GPSS	Engages or disengages GPSS mode.

MODELS -06 & -16: CDI/XFR/BANK

FEATURE LIMITATIONS

- Available only for certain aircraft with GFC 600 autopilot

Knob Functions:

- Airspeed • Vertical Speed • Heading • Altitude

Key Functions:

- CDI Source Selection • PFD Control Selection (pilot or copilot)
- Low Bank Mode Engagement¹

GCU 485-06, CDI/XFR/BANK



CONTROL OPTIONS	
VS	Sets the selected vertical speed.
SPD	Sets the selected airspeed reference.
HDG	Sets the selected heading bug.
ALT	Sets the selected altitude bug.
CDI	Changes the NAV source for the CDI. ¹
XFR	Selects between pilot and copilot PFD.
BANK	Toggles Low Bank mode on or off. ²

¹ For systems with a single PFD Controller, pushing the **CDI** key changes the CDI NAV source for the selected PFD (pilot or copilot) only. ² GFC 600 only.

Backup Battery

FEATURE LIMITATIONS

- *GBB 54 is only an option for GDU 700()*

An optional GBB 54 backup battery provides emergency power to GDU 700(), the integrated ADAHRS, and a single GEA 110.

The backup battery is mounted remotely and provides power when aircraft power is unavailable. This allows for continued PFD, MFD, and EIS functionality when aircraft electrical power is lost.

LRU	DISPLAY	FUNCTION
GBB 54 Alternate emergency power source for GDU 700().	PFD	Battery availability and charge/discharge status
	MFD	
	EIS	

Cabin CO Monitor

The GCO 14 aircraft carbon monoxide sensor provides CO level information to all configured Garmin avionics. GDU uses this data to alert the flight crew of potentially hazardous levels of CO in the cabin.

Upon detecting heightened levels of CO, GCO 14 issues a visual alert annunciation on all configured TXi displays.

LRU	DISPLAY	FUNCTION
GCO 14 Aircraft carbon monoxide sensor.	PFD	Cabin CO level caution alert annunciation and inhibit option
	MFD	
	EIS	

ADC & AHRS

The ADC and AHRS sense air data and aircraft attitude for display and use by other systems. AHRS units have a magnetometer interface for determining magnetic heading. ADC units have an OAT probe interface for measuring outside air temperature.

LRU	DISPLAY	FUNCTION
GDC 72 GDC 74 GDC 7400 Third-party ADC	PFD	• Air temperature • Airspeed • Altitude • Vertical speed
GSU 75/75B Integrated ADAHRS ¹	PFD	ADC • Air temperature • Airspeed • Altitude • Vertical speed AHRS • Attitude • Heading • Rate of turn • Slip/skid/yaw
GRS 77 GRS 79 GRS 7800	PFD	• Attitude • Heading • Rate of turn • Slip/skid

¹ Integrated ADAHRS is only an option for GDU 700()/1060.

PFD Adapter

AUTOPILOT INTERFACE

LRU	DISPLAY	FUNCTION
GAD 43 Adapter	PFD	Analog attitude, heading, and yaw data for certain autopilots ¹
GAD 43e Enhanced Adapter	PFD	- Altitude preselect - VS control - Yaw damper - DME - NAV - ADF - RAD ALT

¹ Requires GRS 77/GRS 79/GSU 75 or internal ADAHRS.

Autopilot

AUTOPILOT ALTITUDE PRESELECT/VERTICAL SPEED

LRU	DISPLAY	FUNCTION
Collins: APS-65()	PFD	Selected altitude sync
Honeywell (BendixKing): KAP 100/150 KFC 150 KFC 200/250 KFC 275/325		- Altitude capture armed annunciation - Altitude capture engaged annunciation - Selected altitude sync - Selected VS sync¹ - VS engaged annunciation¹
S-TEC: 55X 60-2 65 60 PSS		- Selected altitude sync - Selected VS sync

¹ KAP 150 and KFC 150/275/325 only.

AUTOPILOT EXTERNAL FLIGHT DIRECTOR

LRU	DISPLAY	FUNCTION
Bendix: M-4D	PFD	External flight director
Century: IV 41 2000		
Cessna: 300B IFCS/400B IFCS/800B IFCS 1000 IFCS		
Collins: APS-65 AP-106/107		
Honeywell (BendixKing): KFC 150/200/250 KFC 225/275/325 KFC 300		
S-TEC: 55 55X 60-2/65 1500/2100		
Sperry: SPZ-200/500		

FLIGHT CONTROL SYSTEM

LRU	DISPLAY	FUNCTION
Avidyne: DFC90	PFD	• Mode annunciations • Bug sync • Flight Director
Garmin: GFC 500 GFC 600	PFD	• Electronic Stability and Protection annunciations^{1, 2} • Overspeed/underspeed protection annunciations³ • Mode annunciations⁴ • Bug sync⁵ • Flight Director

¹ ESP roll engagement limits display differently between autopilots.
² Available only when ESP is enabled and AP is disengaged.
³ Available only when AP is engaged and aircraft exceeds minimum or maximum engagement speed.
⁴ Track mode annunciation available only on GFC 500.
⁵ Includes value display and control for ALT, HDG, IAS, and VS. BARO sync, CDI source selection, and Selected Course sync output to primary attitude indicator when interfaced with G5 Electronic Flight Instrument.

Weather & Music

LRU	DISPLAY	FUNCTION
GDL 69/69A SXM Garmin SiriusXM receiver for weather. Overlays weather products on MFD and HSI Map.	PFD	Weather Products - NEXRAD - SXM Lightning - TFRs
	MFD	Weather Products¹ Map Page: - NEXRAD - Cloud Tops - Echo Tops - SXM Lightning - METARs - Storm Cells - TFRs Weather Page: - AIREP - NEXRAD - City Forecast - Cloud Tops - County Warnings - Cyclone - Echo Tops - Winds Aloft - Surface Pressure - SXM Lightning - Storm Cells - METARs - AIRMETs - SIGMETs - TFRs - PIREPs - Freezing Levels - Turbulence Forecast - Icing Forecast
GDL 69A SXM Garmin SiriusXM receiver for entertainment.	MFD	Entertainment Services SiriusXM Radio

LRU	DISPLAY	FUNCTION
GDL 88 GTX 345 GNX 375 Datalink traffic and weather.	PFD	Traffic Services • ADS-B • TIS-B Weather Services • FIS-B
	MFD	
GSR 56 Garmin (Iridium) Satellite Receiver for Connex weather. Overlays weather products on MFD and HSI Map.	PFD	Weather Products • Lightning • Precip • TFRs
	MFD	Weather Products Map Page: • Precip • IR Satellite • Lightning • METARs Weather Page: • Precip • IR Satellite • Winds Aloft • Lightning • METARs/TAFs • AIRMETs • SIGMETs • TFRs • PIREPs
WX-500 WX-1000E Depicts Stormscope data on MFD and HSI Map.	PFD	Lightning strikes and cells (areas of lightning activity)
	MFD	

LRU	DISPLAY	FUNCTION
Garmin: GWX 68 GWX 70 GWX 75 GWX 8000 Provides airborne weather radar information.⁶	MFD	• Ground, standby, test, and weather modes • Horizontal and vertical profiles • Roll/trim • Stabilization • Sector scan • Altitude compensated tilt^{2, 3} • High resolution color scale³ • Tilt limit (vertical sector scan) • WATCH shading • Target alert • Range • Tilt angle • Bearing angle • Gain GWX 70/75/8000 Features⁴ • Turbulence Detection • Ground Clutter Suppression GWX 8000 Features⁴ • Automatic scan mode • Hail Prediction • Lightning Prediction
Collins: RTA-800		• Ground, standby, test, and weather modes • Horizontal profile • Stabilization • Range • Tilt angle • Gain (ground mode)

LRU	DISPLAY	FUNCTION
Honeywell (BendixKing): ⁴ RDS 81 (RS 811A) RDS 82 (RS 181A) RDR 2000 (ART 2000) RDR 2100 (ART 2100) Third-party weather radar. ⁶		• Ground, standby, test, and weather modes • Horizontal profile • Vertical profile⁵ • Roll/trim • Stabilization • Range • Tilt angle • Bearing angle⁵ • Gain (ground mode)

¹ Product availability dependent upon subscription. ² GWX 70 only. ³ GWX 75 only.

⁴ Requires purchase of an enablement card. ⁵ Feature dependent on radar.

⁶ Not available on GDU 700() EIS/MFD configurations.

Engine Monitoring

RECIPROCATING ENGINES

LRU	DISPLAY	FUNCTION
GEA 110 Garmin Engine Adapter. Monitors engine, fuel, and electrical systems for piston engine aircraft.	EIS	• Manifold Pressure • RPM • Percent Power • Fuel Flow • OAT • Oil Pressure • Oil Temperature • EGT • CHT • TIT • IAT • Fuel Quantity • Volts • Amps • Engine Data • Vacuum/Pressure • Flight Control Trim Position

TURBINE ENGINES

LRU	DISPLAY	FUNCTION
GEA 71B GEA 71B Enhanced GEA 71C Garmin Engine Adapter. Monitors engine, fuel, and electrical systems for turbine aircraft.	EIS	• Propeller RPM • Fan Speed • Turbine Speed • Gas Producer RPM • Torque • Shaft Horsepower • Percent Power • Engine Temperature • Fuel Flow • Fuel Pressure • Fuel Temperature • Fuel Quantity • IAT • OAT • Oil Pressure • Oil Temperature • Volts • Amps • Engine Data • Vacuum/Pressure • Flight Control Trim Position

Magnetometer

LRU	DISPLAY	FUNCTION
GMU 44/44B Provides magnetic information to the AHRS.	PFD	• Heading • Compass

Navigation/FMS

LRU	DISPLAY	FUNCTION
GPS 175 GPS 400W GPS 500W GNC 355 GNC 420W GNS 430W GNS 480 GNS 530W GNX 375 GTN 625 GTN 625Xi GTN 635 GTN 635Xi GTN 650 GTN 650Xi GTN 725 GTN 725Xi GTN 750 GTN 750Xi	PFD	• ILS/VOR¹ • LOC¹ • GPS position • GS¹
	MFD	• GPS position

¹ Feature requires a NAV radio source.

Temperature Probe

LRU	DISPLAY	FUNCTION
GTP 59	PFD	Outside air temperature
	EIS	
Airframe Temperature Probe	PFD	Outside air temperature used for setting engine power
	EIS	

Radar Altimeter

LRU	DISPLAY	FUNCTION
Garmin: GRA 55/5500	PFD	Radar altitude (next to altitude tape)
Collins: RAC-870 ALT-50A ALT-55B ALT-4000		
FreeFlight: RA-4500		
Honeywell (BendixKing): KRA 10/10A KRA 405/405B		
Sperry: AA-100 AA-100A AA-200		

Traffic

LRU	DISPLAY	FUNCTION	
GTX 33 GTX 330 GTX 335 GTX 345	PFD	TIS-A	
	MFD		
Garmin: GTS 800 GTS 820 GTS 825 GTS 850 GTS 855	PFD		TAS/TCAS I
GTS 8000	PFD/MFD		TCAS II
Avidyne (Ryan): TAS 6XX (9900BX)	MFD	TAS/TCAS I	
Honeywell (BendixKing): KTA 810 KMH 820 KTA 910 KMH 920			
L3 Communications (Goodrich): SKY497 SKY899			
GDL 88 GNX 375 GTX 345			
	PFD	ADS-B	
	MFD		
Third-party TCAS II	PFD/MFD	TCAS II	

2 Get Started

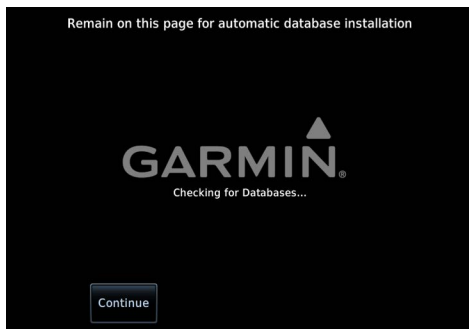
POWER UP	2-2
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Power Up

The GDU receives power directly from the aircraft's electrical system. To ensure safe operation, continuous built-in test features exercise the unit's processor, memory, external inputs, and outputs.

Upon power-up, the bezel key backlight momentarily illuminates. System failure annunciations typically disappear within the first 30 seconds after power-up.

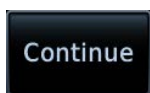
Splash Page, GDU 1060



"Checking for Databases..." annunciates as the unit checks for available database updates.

Databases automatically update during power up when newer database versions are available.

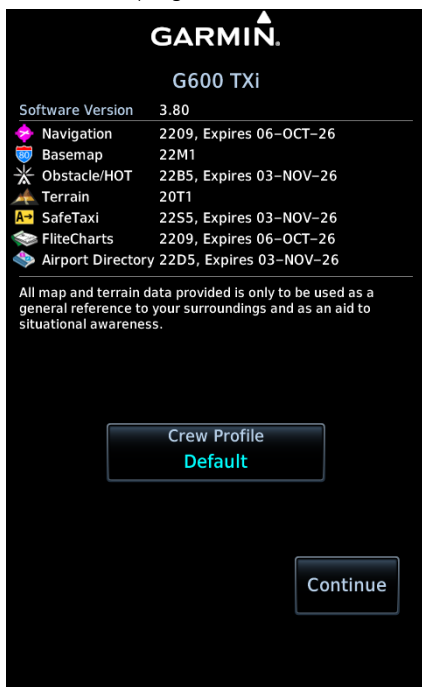
The splash page shows the transfer status of each new database for the unit. Remaining on this page allows the system to automatically restart and complete database activation.



You may advance to another page during the update process by tapping **Continue**. However, a manual system restart will be required to activate databases at a later time.

GDU TXi allows multiple methods for updating databases. For more about the update methods available with your system, read *Database Updates*.

Database Start-up Page, GDU 700P MFD



For installations with a GTN Xi series navigator, the database list does not appear. Refer to the primary GTN Xi for a status summary of all databases in the system.

On the MFD:

The database start-up page presents the unit software version, the name and status of all installed databases, and controls for selecting a crew profile (if multiple profiles are available).

From here you may:

- *Select a crew profile:*
Tap **Crew Profile**.
- *Advance to the next page:*
Tap **Continue**.

Read about crew profile selection and activation in *Crew Profiles*.

¹ Requires GDU TXi software v3.30 or later with GTN Xi v20.20 or later.

Crew Profile Options, GDU 700P MFD



Installations with a GTN Xi Series Navigator:

Crew profile options display if multiple profiles exist.

Choose a profile from the list and then tap **Continue** to advance to the next page. Tapping **Continue** without making a selection activates the default profile.

Profile options do not display if only one profile exists and database confirmation is performed remotely via the primary GTN Xi.¹

Read about crew profile selection and activation in *Crew Profiles*. Read about database confirmation for installations with a GTN Xi series navigator in *Remote Database Confirmation*.

¹ Requires GDU TXi software v3.30 or later with GTN Xi v20.20 or later.

Remote Database Confirmation

FEATURE REQUIREMENTS

- *TXi software v3.30 or later*
- *GTN Xi series navigator with software v20.20 or later*
- *SD cards containing databases must be removed from the unit*

When configured with a GTN Xi series navigator, database information for each configured GDU in the system is sent to the primary GTN Xi for pilot acknowledgment. Synchronization occurs once it is determined that no database issues exist. Transfer status indications appear on the splash page of each display. Once synchronization is complete, each MFD automatically skips its database start-up page.

- *If a database is corrupt or missing:* The unit will display its database list and indicate the database in question. Confirmation via the primary GTN Xi will still occur for all other configured LRUs if their databases are present and not corrupt.
- *If an SD card containing databases is installed in the unit:* Remote database confirmation is not available. Tap **Continue** to advance to the next page.

Remote database confirmation is a function of GTN Xi. For more information regarding functionality, consult *GTN Xi Series Pilot's Guide*.

Power Off Options

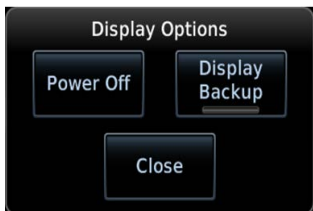


WARNING

Never attempt to power off the unit while airborne unless operational procedures dictate.

FEATURE LIMITATIONS

Manual Power Off function available only with GDU 700()/1060. GDU 1210 is powered off by removing electrical power to the unit.



Pushing the **Power** key once opens a pop-up menu. Selectable display options allow you to:

- Power off unit
- Close the menu (cancels shutdown)
- Change operating mode to display backup (if available)



Pushing and holding the **Power** key for 4 seconds powers off the unit without safeguards. Shutdown occurs once the timer reaches zero.

Databases

Databases are stored in the internal memory of each display. To view update cycles, or to purchase individual databases or database packages, go to flyGarmin.com.

SUPPORTED DATABASES	
Airport Directory	Airport facility and FBO information
Basemap	Bodies of water, geopolitical boundary, and road information
ChartView¹	Jeppesen terminal procedures
FliteCharts	AeroNav terminal procedures
IGRF^{2, 3}	Internal ADAHRS and external AHRS correct for variations in the earth's magnetic field by applying calculations derived from the IGRF database
Navigation Data	Airport, NAVAID, waypoint, and airspace information (Garmin or Jeppesen)
Obstacles	Obstacle and wire data
SafeTaxi	Airport surface diagrams
Terrain	Terrain elevation data

¹ Optional third-party database. Requires one-time feature enablement for use.

² When updated, the database is packaged with the Navigation database. Not available for individual download.

³ For magnetic field model version and part number information, refer to the AHRS section of the External LRUs page.

For information regarding third-party databases, go to jeppesen.com.

Database Effective Cycles

FEATURE REQUIREMENTS

- External or internal GPS navigator for system to determine database effectiveness

OR

- Flight Stream 510 wireless transceiver
- Garmin Pilot app on a portable electronic device

FEATURE LIMITATIONS

- EIS units do not support database functionality

Most databases expire at regular intervals. Exceptions include Basemap and Terrain, which neither expire nor update on a regular schedule. IGRF updates occur approximately every five years. Failure to update a database can lead to errors in heading information.

DATABASE EFFECTIVE STATUS

Databases with no effective date

- Effective upon release
- Includes Basemap and Terrain

Databases with specified effective dates

- Effective during a specific period
- Unit determines database status using the current date and time from GPS
- Automatic activation occurs on the effective date

	Navigation	2209, Expires 06-OCT-22
	Basemap	22M1
	Obstacle/HOT	22B5, Expires 03-NOV-22
	Terrain	20T1
	SafeTaxi	22S5, Expires 03-NOV-22
	FliteCharts	2209, Expires 06-OCT-22
	Airport Directory	22D5, Expires 03-NOV-22

	SafeTaxi	16S4, Eff. 21-JUL-16 to 15-SEP-16
	FliteCharts	1606, Eff. 26-MAY-16 to 23-JUN-16
	Airport Directory	16D4, Eff. 21-JUL-16 to 15-SEP-16

	Terrain	Database Not Found
---	---------	--------------------

On the MFD: The start-up page lists all currently installed databases. Review this list for current database types, cycle numbers, and expiration dates.

Yellow text denotes when a database is:

- Not available
- Installed before its effective date
- Missing date information
- Past its expiration date

Pilot PFD Navigation database expires on 15-SEP-16

Expired database notifications indicate as system advisories.

Database Conflicts

FEATURE LIMITATIONS

- *Applicable to GTN Xi/GDU TXi installations only*

Conflicts occur when a database is corrupt or missing. In such cases, remote database confirmation via the primary GTN Xi is no longer available for the unit.

When this happens:

- The database list displays on the MFD database start-up page¹
- The database name appears in yellow on both GDU and the primary GTN Xi

Resolve database conflicts when they occur. Expired or mismatched databases are managed from GTN Xi series navigator.

Active and Standby Databases

GDU uses two types of databases.

Active Databases

- Databases currently in use by the system

Standby Databases

- Databases that have not reached the effective date
- Databases that were installed prior to the last update

Database Information

During normal operation, you can view information about all active and standby databases from the associated tab.

From the MFD Home page:

Tap **System** > **System Status**.

From the GDU 700() PFD:

Tap **Menu** > **System** > **Databases**.

¹ Under normal operating conditions, the MFD database start-up page may appear only briefly or not at all.

ACTIVE DATABASE LIST

Navigation Database			
Region:	AMR-ADB2	Cycle:	2206
Effective:	16-JUN-22	Expires:	14-JUL-22
Basemap Database			
Cycle:	22M1	Version:	5.20
Obstacle Database			
Region:	Worldwide Fixed-Wing		
Cycle:	22B3	Type:	Obstacle/HOT
Effective:	19-MAY-22	Expires:	14-JUL-22
Terrain Database			
Region:	Worldwide-9		

View Copyrights

Selecting **Active** displays information about the databases currently in use. You can also access copyright information from here.

Tapping **View Copyrights** displays copyright information for all installed databases.

STANDBY DATABASE LIST

Airport Directory			
Source:	Internal		
Cycle:	22D2	Effective:	24-MAR-22
Navigation Database			
Source:	Internal		
Cycle:	2206	Effective:	16-JUN-22
FliteCharts Database			
Source:	Internal		
Cycle:	2204	Effective:	21-APR-22

No standby databases found

Selecting **Standby** displays information about databases that are not yet effective.

Previously installed databases also appear here once an update completes.¹

TXi notifies you when no standby databases are available.

¹ Applicable only to TXi software v3.50 and later.

Database Updates



NOTE

GDU supports SD cards in the FAT32 format only, with capacities ranging between 8 GB and 32 GB.

Databases are stored in the internal memory of each display. To view update cycles, or to purchase individual databases or database packages, go to flyGarmin.com.

The TXi system offers multiple methods for loading and updating databases. Do not attempt any of these while in flight (on ground only).

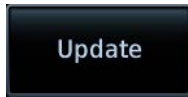
- **Load databases via SD card.** Once loading completes, you may power off the unit and remove the card.
- **Transfer databases from a portable electronic device using Database Concierge.** This method requires the Garmin Pilot app on a portable electronic device and Flight Stream 510.
- **Transfer databases from a GTN or another GDU using Database SYNC.** If enabled, databases synchronize across all capable Garmin avionics.

GDU prioritizes database transfers via SD card over all other available means of transfer. Updates via Database Concierge have priority over database synchronization with capable Garmin avionics (e.g., GTN, a second GDU).

Manual Updates

FEATURE LIMITATIONS

The Database Update page is available only when the aircraft is on ground.



The Database Update page presents a list of all available databases.

While on ground, you can access this page at any time for review purposes or to perform manual database transfers.

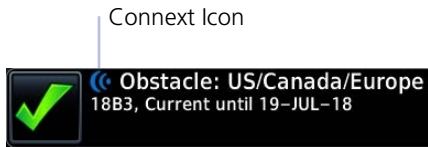
From the MFD Home page:

Tap **System** > **System Status** > **Menu** > **Update**.

From the GDU 700() PFD:

Tap **Menu** > **System** > **Databases** > **Update**.

DATABASE SOURCE INDICATION



Connex icon

A Connex icon indicates when a database is from Garmin Pilot via wireless transfer.

No indication means the database is either from an SD card or the unit's internal Standby storage.

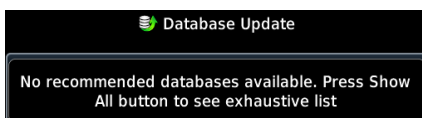
SELECT ALL DATABASES



Select applicable database(s) for transfer. If all listed databases require updating, choose **Select All**.

Tapping **Select None** deselects all databases.

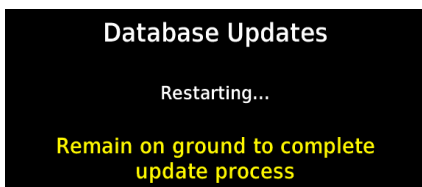
By default, this page displays only the databases recommended for update.



A message informs you when no recommended databases are available.



After all selections are made, initiate the transfer process by tapping **Start**.



Once the transfer process is complete, the unit will automatically restart if the aircraft is still on ground. A system restart is required to complete database installation.

SHOW ALL DATABASES



Tapping **Show All** displays a complete list of all databases, including ones that are not yet effective or that may be older than the currently active database(s).

For more details about a specific database, tap **Error Info**.

REVERT TO PREVIOUS DATABASE VERSION

FEATURE LIMITATIONS

- Available with GDU TXi software v3.50 and later

Updating a database does not overwrite its previous version. Once the update process completes, the previously installed version of the database transfers to the internal Standby storage. In the event that an automatic update is undesirable, you can revert to the previous version using the manual update feature.

	Airport Directory: US 22D2, Expired 19-MAY-22
	Navigation: AMR-ADB2 2206, Expired 14-JUL-22
	FliteCharts 2207, Expired 11-AUG-22
	Obstacle: Worldwide Fixed-Wing 22B2, Expired 19-MAY-22
	SafeTaxi: FullCoverage 22S2, Expired 19-MAY-22

Previous database versions appear as non-recommended databases in the standby list. They remain available for manual install until a newer version transfers to the internal Standby storage.

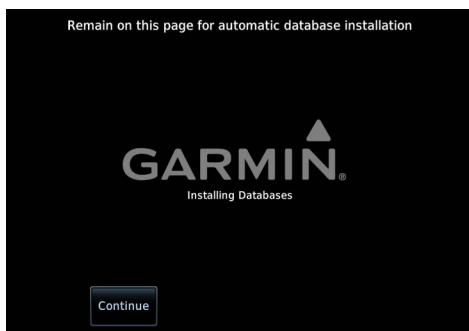
To revert to a previous database version:

1. Open the Database Update page
(Home > **System** > **System Status** > **Menu** > **Update**).
2. Tap **Show All**.
3. Select the previous database version.
4. Tap **Start**.

The unit restores the previous database version and transfers the newer database to the internal Standby storage. The unit automatically restarts once the swap is complete.

Automatic Updates

Updates occur during power up when a newer version of a database is available. The same process occurs when you install a new database for the first time.



Tapping **Continue** advances to the next page. Databases will continue to transfer in the background as you use other features.



Indications show when an automatic database transfer is in progress. Message text varies according to installation method.

Automatic updates occur when:

- A newer database is detected on the SD card or in the unit's internal Standby storage
- A newer database is within its effective dates
- A recommended database is available for transfer from a capable LRU via the Database SYNC function (e.g., GTN Xi)
- The aircraft is on ground

If enabled, the Database SYNC function synchronizes databases across all capable Garmin avionics. A coordinated restart of each unit completes the installation. Read more about database synchronization in *Synchronize Databases Across Multiple Units*.

DATABASE TRANSFER SEQUENCE

Databases transfer in three major steps:

Step 1: Small Databases	Step 2: Charts	Step 3: Terrain
Includes: • Navigation • Basemap • Obstacle • SafeTaxi • Airport Directory Restart required to complete installation.	Available ChartView or FliteCharts database. No restart required.	Available Terrain database. Installation occurs during the next unit power cycle.

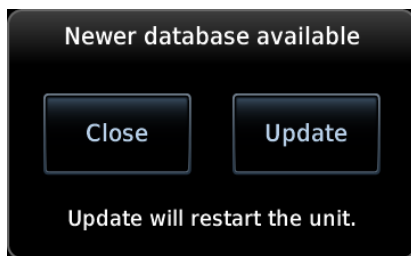
SYSTEM RESTART OPTIONS

A system restart is required to complete installation of all small databases and Terrain. This occurs automatically when you remain on the splash page during the transfer sequence.

If you choose to advance to another page, confirmation is needed once database transfer is complete.

- *For systems with a GTN Xi series navigator:* Confirm the restart request on the primary GTN Xi to complete installation for all capable Garmin avionics.¹
- *For all other systems:* A pop-up informs you when newer databases are available.

Database Update Pop-up, GDU TXi



Tapping **Update** opens the Database Update page. Review the list of recommended updates and tap **Start**. Confirm the restart request on each LRU if necessary.

Updates are indicated in the list of currently installed databases. In the event that an automatic database update is undesirable, you can revert to the previous database version from the Database Update page.²

¹ Requires GDU TXi software v3.30 or later with GTN Xi v20.20 or later.

² Available only with GDU TXi software v3.50 and later.

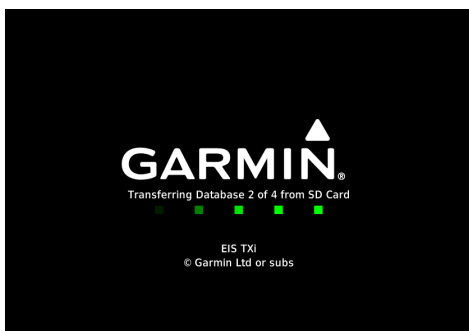
Load Databases from an SD Card

The unit stores all databases it receives in its internal storage. Once the update is complete, the SD card is no longer required.

Install or update a database using an SD card:

1. Download a database onto an SD card.
2. Insert the SD card with the most recent database(s) into the top/right card slot.
3. Power on the GDU.

The unit detects when an SD card with recommended databases is present in the slot. Updates occur automatically at power up.



A counter on the splash page shows the total number of available databases and the number of databases received by the unit.

Terrain databases may require up to 5 minutes for transfer. Total transfer time depends on the SD card type.

Tap **Continue** at any time to advance to the next page.



A system restart is required to complete installation.

- *If you remain on the splash page:* No action is necessary. The system automatically restarts once the update is complete.
- *If you advance to another page during the transfer process:* A pop-up informs you when newer databases are available. Tap **Update** and review the list of recommended updates, then tap **Start**.¹

Once installation is complete, you may power off the unit and remove the SD card.

Overwriting SD card database files

When database files are loaded to the SD card, any previously loaded database files of the same type residing on the SD card will be overwritten. This includes loading a database of a different coverage area or data cycle than that currently residing on the SD card.

You can transfer databases across all capable Garmin avionics via the Database SYNC function. Read more about database synchronization in *Synchronize Databases Across Multiple Units*.

¹ Installations with GDU TXi v3.50 or later and GTN Xi v20.30 or later: Confirmation is requested on the primary GTN Xi.

Transfer Databases Using Database Concierge



Database Concierge allows wireless transfer of databases from a portable electronic device while the aircraft is on ground.

FEATURE REQUIREMENTS

- *The aircraft is on ground and the avionics are powered on*
- *Garmin Pilot app on a portable electronic device*
- *Active Garmin Pilot subscription (for enabling Database Concierge)*
- *Flight Stream 510 wireless transceiver*

A pilot selects and downloads databases inside the Garmin Pilot app. Transfers occur once the Garmin Connex device establishes a wireless connection inside the aircraft. Prompts to connect to a Wi-Fi network and database transfer indications appear on both Garmin Pilot and GTN Xi.

Database Concierge Transfer Function

- Provides automatic updates for databases with effective dates
- Preloads databases that are not yet effective by placing them in the unit's internal Standby storage
- Supports Database SYNC with capable Garmin avionics
- Displays database type, cycle, effective date, and transfer progress
- Allows you to initiate transfer from the Database Update page via the **Start** key
- Requires pilot confirmation

TXi Installations with GTN Xi & GDL 60 Wi-Fi/LTE Datalink

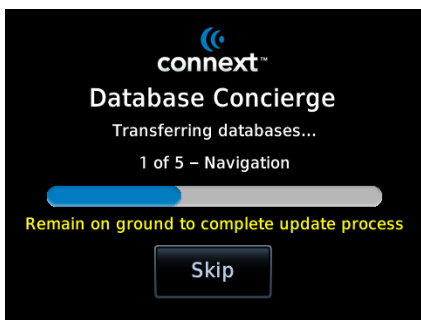
For information about transferring databases using Database Concierge, consult G500(H)/G600/G700 TXi Pilot's Guide Addendum for GDL 60.

TRANSFER DATABASES VIA FLIGHT STREAM 510



Transfers occur once Flight Stream 510 establishes a wireless connection inside the aircraft.

1. Purchase database(s) from flyGarmin.com.
2. Open Garmin Pilot and follow the download instructions.
3. Install Flight Stream 510 into the GDU. If the system contains a GTN, install the wireless transceiver into that unit instead.
4. Power on the unit.
5. Connect to Wi-Fi.
6. Follow the on-screen prompts.



Database Concierge transfers databases from the app to Flight Stream 510. A progress bar shows when this process is complete.

GDU either updates or preloads databases based on their effective date. A second progress bar indicates upload status.

A system restart is required to complete installation. Database updates are intended to occur while the aircraft is on-ground.

- *For systems with a GTN Xi series navigator:* Confirm the restart request on the primary GTN Xi to complete installation for all capable Garmin avionics.¹
- *For all other systems:* A pop-up informs you when newer databases are available. Tap **Update** and review the list of recommended updates, then tap **Start**. Confirm the restart request on each LRU if necessary.

Skip Database Transfer

Tapping **Skip** cancels any unfinished wireless transfers and initiates the update process.

GDU activates any databases that completed transfer before the interruption. Previously selected databases on an SD card or in the internal standby update as well.

The message "Transfers interrupted" displays if no databases are available.

¹ Requires GDU TXi software v3.50 or later with GTN Xi v20.30 or later.

Synchronize Databases Across Multiple Units



The Database SYNC function minimizes database maintenance by synchronizing active and standby databases across all capable Garmin avionics.

FEATURE REQUIREMENTS

- Database SYNC function enabled on all participating LRUs
- For background updates and coordinated LRU restart: GDU TXi software v3.50 or later & GTN Xi v20.30 or later

FEATURE LIMITATIONS

- Functionality not available for EIS-only configurations

Database SYNC Transfer Function

- Enables automatic database synchronization across all capable Garmin avionics¹
- Background updates allow the use of features without having to wait for individual LRUs to complete the update process²
- Coordinated automatic restart of all capable LRUs to complete installation (initiated via the aircraft's primary GTN Xi series navigator)³
- Available for all supported databases⁴
- Includes active and standby databases
- Informs you that enabling Database SYNC may overwrite any databases currently in standby
- Prompts unit restart if a new database is effective and the aircraft is on ground

¹ Database transfers to G500/G600 flight displays with software v7.00 and later available with GDU TXi software v3.50 and later. Transfers from these units to TXi not available.

² Applicable only to GDU TXi software v3.50 and later.

³ Requires GTN Xi v20.30 or later. If required, system restart confirmation is via the primary GTN Xi.

⁴ Terrain database synchronization not available for units with GDU TXi software earlier than v3.50.

For information regarding database packages, and individual database purchases, visit flyGarmin.com.

ENABLE DATABASE SYNC

From the 700P/1060/1210 MFD Home page:

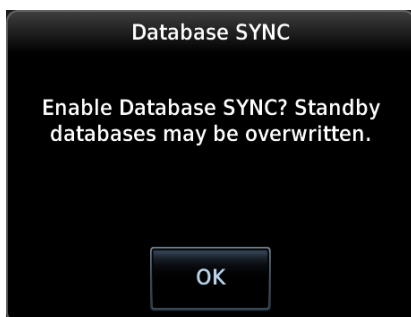
Tap **System** > **System Status** > **Menu** > **Database SYNC**.

From the 700L MFD Home page:

Tap **System** > **System Status** > **Database Information** > **Menu** > **Database SYNC**.

From the 700() PFD:

Tap **Menu** > **System** > **PFD Setup** > **Database SYNC**.



A pop-up informs you that enabling Database SYNC may overwrite any databases currently in standby.

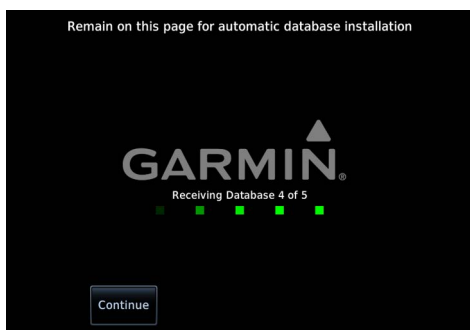
Tapping **OK** confirms the request.

Tapping **Cancel** closes the pop-up and aborts the request.

Toggling **Database SYNC** off disables the Chart Streaming function (if enabled).

SYNCHRONIZATION STATUS INDICATIONS

Synchronization occurs automatically at power up.



As each database uploads to the unit, a counter informs you of transfer status. Total number of available databases excludes chart databases.

All sync-enabled displays pause momentarily on the splash page until the transfer process is complete.

Tap **Continue** at any time to advance to the next page.



DATABASE ACTIVATION

A restart of all capable Garmin avionics is required to complete installation.

- *If you remain on the splash page:* No action is necessary. All capable LRUs automatically restart once the update is complete.
- *If you advance to another page during the transfer process:* A pop-up requests a system restart once the update is complete. Tap **Yes** and confirm the request.
 - *For systems with a GTN Xi series navigator:* Confirm the restart request on the primary GTN Xi to complete installation for all capable LRUs.¹
 - *For all other systems:* Confirm the update request on each LRU.

Once installation is complete, all previously installed databases transfer to each unit's internal Standby storage. You may revert to the previous database version at any time from the Database Update page on each unit.

Updates are indicated in the unit's list of currently installed databases.

Systems with a GTN Xi Series Navigator

During power-up, the primary GTN Xi provides a summary of active databases across all capable Garmin avionics. For more information about coordinated restart functionality, consult *GTN Xi Series Pilot's Guide*.

¹ Requires GDU TXi software v3.50 or later with GTN Xi v20.30 or later.

SYNCHRONIZED DATABASE LIST

FEATURE LIMITATIONS

- Available only with TXi software v3.50 and later

You can view a list of the databases currently synchronizing with other LRUs from the associated information tab.

From the MFD Home page:

Tap **System > System Status > DB SYNC**.

From the GDU 700() PFD:

Tap **Menu > System > Databases > DB SYNC**.



A progress bar shows when a database transfer is complete.

Status messages inform when:

- The unit is transmitting or receiving databases
- The Database SYNC function is disabled
- The LRU is offline
- A database is not authorized for synchronization
- A unit restart is required
- Database transfer is complete

Database SYNC inactive

A message informs you when database synchronization is not in progress.

If a particular LRU is not accepting a database, refer to the Database SYNC tab to determine the possible cause.

Stream Charts



Chart Streaming allows streaming of individual charts on an as-needed basis until database sync is complete. Toggling the function off has no effect on Database SYNC.

FEATURE LIMITATIONS

- Available only when Database SYNC is active and a current chart database is available

Chart Streaming Function

- Enables automatic streaming of individual charts from the newest chart database
- LRUs with chart streaming enabled display the most current chart information
- Current charts display on MFD Charts page; chart overlays are available for display on Map page
- Chart database effective date displays on Charts page if database has expired
- Previously installed chart database remains available in the unit's internal Standby storage¹
- Available for both ChartView and FliteCharts
- No restart required

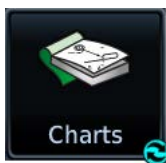
¹ Applicable only to TXi software v3.50 and later.

A typical chart database may take up to one hour to synchronize across multiple LRUs. For 100 Mb/s enabled LRUs (e.g., GDU TXi, GI 275, GTN Xi), synchronization may take up to 10 minutes.¹

¹ Available only with TXi software v3.50 and later.

CHART STREAMING STATUS ICONS & NOTATIONS

The following indications appear during the update process.



A spinning arrow appears on the Charts Home page icon when a new chart database is streaming.

 **FliteCharts**  **2204, Expired 19-MAY-22**

Chart database update status announces on the start-up page. Spinning arrow icon appears until any mismatches are resolved and synchronization is complete.

Getting Newer Charts. Expired 19-MAY-22

Textual announcements at the bottom of the Charts display inform you of update status.

For information about Charts page features, read *Charts* in section 5.

Connectivity

Connex works via wireless connectivity links to provide up-to-date, wireless information to and from the cockpit. The Connex interface allows wireless communication with Garmin Pilot from a portable electronic device.



Once installed in the SD card slot, the Flight Stream 510 wireless transceiver allows data streaming via Wi-Fi and Bluetooth wireless connectivity.

Remote Features

- ADS-B In Traffic Data
- AHRS Data from Compatible Display
- Connex Weather
- Database Concierge
- FIS-B Weather and Flight Information
- Flight Data Log Streaming
- Flight Plan Transfer
- GPS/WAAS Position, Velocity & Time Information
- GSR 56 Phone/SMS Text
- Primary Flight Data
- SiriusXM Audio Remote Control
- SiriusXM Weather

Available remote features and setup options are dependent upon aircraft interfaces or enablement.

Flight Stream 510 Setup



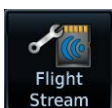
When Flight Stream 510 is present, GDU TXi supports wireless pairing with up to 13 portable electronic devices via the Garmin Pilot app.

FEATURE REQUIREMENTS

- *Flight Stream 510 wireless transceiver (installed in GTN, if available)*
- *Garmin Pilot app on a portable electronic device*

FEATURE LIMITATIONS

- *GDU allows pairing of up to 13 Bluetooth enabled devices, with two simultaneous device connections*
- *Auto reconnect function is not available for Android devices*
- *GDU does not support Flight Stream 510 Phone/SMS and flight plan transfer functionality*



Device information and pairing mode status display on the Flight Stream 510 Setup page.



Refer here when contacting customer service regarding Flight Stream 510.

Product information includes:

- Software version
- Part number

Setup page features allow you to:

- View Flight Stream 510 product information
- Enable database updates
- Pair and manage Bluetooth enabled devices
- View and edit Bluetooth enabled device name and Wi-Fi information

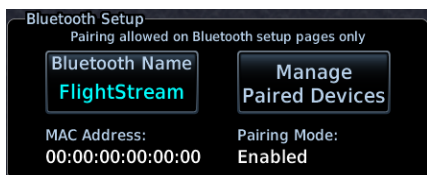
Data logs transfer via Bluetooth wireless technology. Databases transfer over Wi-Fi.

DATABASES

This feature allows automatic import of available database updates via Database Concierge.

Set Up Bluetooth

Depending on display type, Bluetooth device management options may reside on the Flight Stream page or on a dedicated setup page. Pairing occurs only when the Bluetooth Setup menu is opened.



Bluetooth Setup information includes:

- Current Bluetooth device name
- MAC address
- Pairing mode status

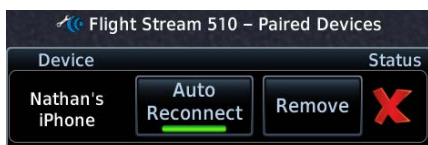
BLUETOOTH NAME

This key allows you to enter the name of the Bluetooth enabled device.

PFD only displays: Use the control knobs to enter the device name. Keypad entry is not available.

EIS only displays: The device name is set up using Garmin Pilot. Manual data entry is not required.

MANAGING PAIRED DEVICES



To view a list of all paired devices and their connection status, tap **Manage Paired Devices**.

AUTO RECONNECT

This key enables automatic connection between the GDU and the paired device when the two are within range.

REMOVE

Removing a device from the list means it is no longer paired with the GDU. This action requires pilot confirmation.

Be sure to remove pairing on both devices before attempting to pair them again.

Set Up Wi-Fi



Wi-Fi setup information includes:

- Wi-Fi SSID
- Wi-Fi password
- MAC address of both Wi-Fi module and connected device (available only after launching Garmin Pilot)

CONNECT TO WI-FI

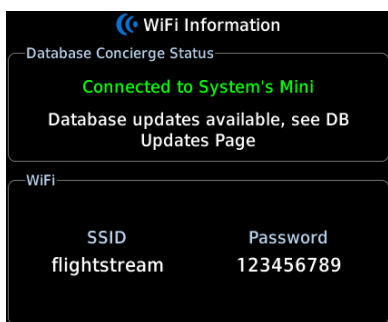
Enter the required SSID and password using the provided setup keys.

PFD only units: Use control knobs for SSID and password entry. Keypad entry is not available.

VIEW WI-FI INFORMATION



Tapping **WiFi Info** opens an information page.



This page is accessible from the Database Update and start-up pages. Information includes:

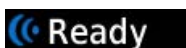
- Database Concierge connection status
- Connected device name
- Database update availability and instructions
- Wi-Fi SSID and password

WI-FI INFO KEY STATUS ANNUNCIATIONS

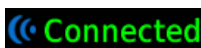
Wi-Fi connection status annunciates on the key label when this page is not active.



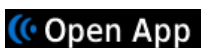
Flight Stream 510 requires power up.



Wi-Fi is active, but the GDU is waiting to connect with a paired device.



System detects a connection between the GDU and a paired device.



Flight Stream 510 requires Garmin Pilot to be opened in order for database transfer to commence.

Pilot Settings

Display Brightness Control

Turboprop Aircraft

Depending on installation, display brightness control may transition between lighting bus and photocell during engine start.

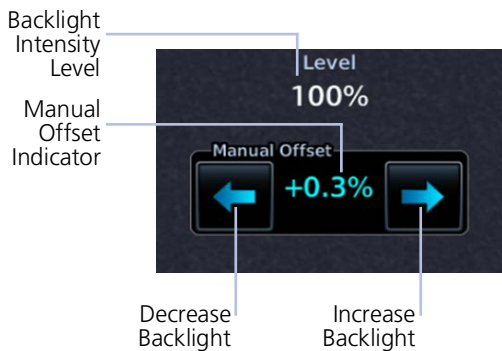
When you engage the start switch, display brightness control switches to photocell. When you disengage the switch, or after 60 seconds, brightness control returns to the dimmer bus.

Depending on configuration, display brightness is controlled using inputs from the built in photocell, aircraft dimmer bus, or both. Installer configured curves determine the amount of change in brightness that occurs in response to a control adjustment

If brightness control is not satisfactory, contact a Garmin dealer to adjust the lighting curves.

Automatic Brightness Control

Dimming is limited to prevent on screen indications from becoming unreadable. The built in photocell automatically controls display brightness based on ambient light levels.



During automatic control, the pilot may still adjust brightness using the manual offset controls in the System Backlight page.

The GDU retains manual offset settings through power cycles.

Manual Brightness Control

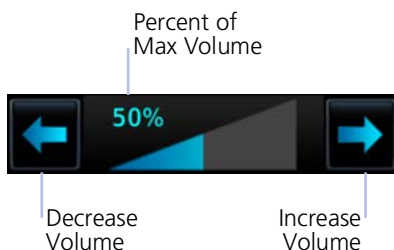
Optionally, the TXi system is configurable to use an aircraft dimming bus for display brightness control. Upon reaching minimum input level, display brightness reverts to the photocell. This prevents the display from going black in the event of a dimmer input failure.

System Status Page

The System Status page displays information specific to the GDU and its databases. Refer to this page when contacting customer service. Information includes:

- Serial number
- System ID
- Main software version
- Database information

Click Volume



If the GDU is wired for audio output, set the click volume to the preferred level.

Volume displays as a percentage of the maximum volume, with 0% being muted and 100% being maximum volume.

Clock

Specify the time format and local offset.

If a 12 hour or 24 hour clock is selected, tap **Local Offset**, then specify the appropriate offset value from UTC.

Options include 12 hour, 24 hour, and UTC.

Weather Display Shortcut

A knob shortcut option is available when there are two or more active weather sources. Select a weather service and verify shortcut operation. Depending on configuration, available shortcut options may include:

- Connex Weather
- FIS-B Weather
- Radar
- SiriusXM Weather
- Stormscope



Selecting **Radar** changes the weather shortcut indicator to reflect the weather radar option is active.

Unit Selections



NOTE

Engine gauge, airspeed, and altimeter units are not pilot selectable. Interfaced equipment may contain unit settings independent of the TXi system.

The System Units page displays a list of TXi system wide unit settings. Selections are synchronized across all configured GDUs.

SPECIFY UNIT TYPE

1. Review the current unit selections.
2. Tap the applicable parameter key.
3. Select a unit type.

DISPLAY SETTINGS		
PFD	MFD	EIS ¹
• BARO Pressure • Distance • NAV Angle³ • Temperature • Wind Speed	• Altitude • Distance • NAV Angle³ • Temperature	• Distance • Fuel Computer² • Temperature

¹ Engine gauge units are not adjustable. ² These units are not synchronized with other displays.

³ TXi software v3.70 and later: NAV Angle unit settings synchronize with GTN Xi v20.43 and later.

Crew Profiles

FEATURE REQUIREMENTS

- MFD for creating crew profiles, modifications, and deletions

The Manage Crew Profile option allows you to:

- Save TXi system wide and display specific settings
- Create individual profiles for aircraft operated by multiple pilots
- Transfer profiles between aircraft

Crew Profile Settings

Crew profile settings include both system wide and display specific settings. Specific display profile settings include the pilot adjustable parameters for each configured display.

SYSTEM PROFILE SETTINGS	
• BARO SYNC • Click volume • Database SYNC • EIS Lean Assist mode (ROP/LOP/EGT/TIT) ¹ • Engine advisories (on/off states, thresholds)	• Nearest Airport Criteria • Recently used waypoint lists • Units

¹ Lean mode is not applicable to turboprop installations.

DISPLAY PROFILE SETTINGS		
DISPLAY	FUNCTION	SETTINGS
PFD	Synthetic Vision	• All
	HSI	• HSI Map • Map overlays • Bearing pointers
	Setup	• Aircraft Symbol/Flight Director¹ • Airspeed reference bug status • AGL Field • Altitude Knob mode² • CDI/VDI Preview • Clock/Timer • G-Meter Display & Style • GPS Roll Indicator • LOC CDI Prompting • Menu Timeout • Wind Field
	System	• Backlight manual offset

¹ Fixed-wing aircraft only. Option dependent upon configuration.

² If altitude knob is configured as "Pilot Control." Applicable to TXi software v3.61 and later.

DISPLAY PROFILE SETTINGS		
DISPLAY	FUNCTION	SETTINGS
MFD	Map	• All
	Traffic	• Motion vector • Altitude filter (ABV/BLW/NRM/UNR)
	Terrain	• View • Layers
	Flight Plan	• Column data fields
	Datalink weather products (SXM, FIS-B, Connex)	• Map orientation • Layers • Legend on/off • FIS-B on/off • Map view configurations
	Stormscope	• View • Mode
	Music	• Preset group selection • Music category selection
	System	• Backlight manual offset • Chart Streaming • Startup page selections • Time format (UTC/local) • Weather knob shortcut

Crew Profile Management



Profile management functions are provided in a fly-out menu on the Manage Crew Profiles page. Available options are based on current profile selection.

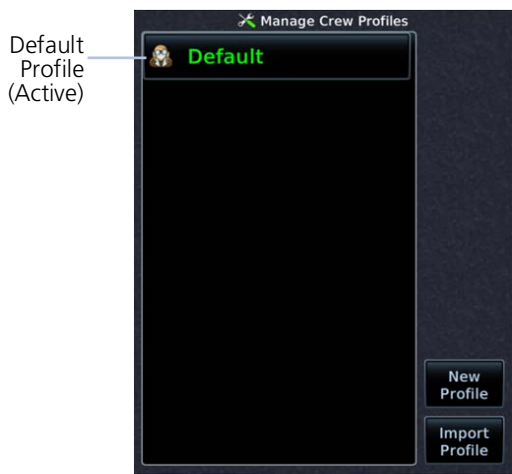
FEATURE LIMITATIONS

The system stores a maximum of ten profiles. This includes nine user generated profiles and one default profile. Both New Profile and Import Profile functions are unavailable when this limit is reached. It is recommended to have all GDUs online when managing crew profiles. Profile synchronization is delayed for any LRU not online at the time a profile change occurs.

From the MFD Home page:

Tap **System** > **Crew Profile**.

Setting modifications are automatically stored within the active crew profile. If no user generated profile is active, all adjustments are saved to the Default profile.

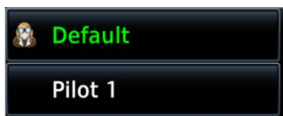


The Default option is automatically selected when no user-generated crew profile is imported or defined. New and existing profiles are listed in the Manage Crew Profiles page.

Create a New Profile

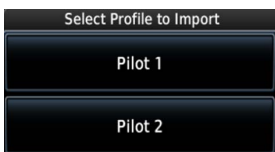
A profile name is required to complete operation.

1. Tap **New Profile**, enter desired profile name.
2. Tapping **Create and Activate** activates new profile and adds it to list.

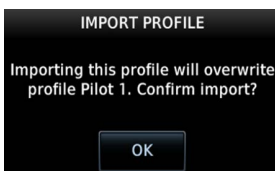


User-generated profile

Import a Profile



1. Insert an SD card containing a new profile.
2. Tap **Import Profile**.
3. Select profile to import.



If an imported profile has the same name as the existing entry, it is possible to overwrite the existing entry or cancel request.



If the imported profile has the same name as the active profile, the overwrite option is not given. Acknowledge the message, then activate a different profile and try again.



If the SD card does not contain importable profiles, acknowledge the message and replace the SD card.

Export a Profile

The export function writes a selected profile onto an SD card. This function overwrites any profile on the SD card with the same name. Available options are based on current profile selection.

Delete a Profile

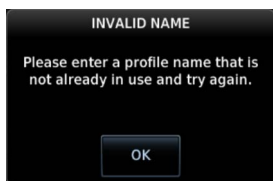
The delete function removes a selected profile from all configured units.

1. Select a profile from the list.
2. Tap **Delete**.

Add a Crew Profile to a Full Profile List

1. Delete an existing entry from the profile list.
2. Create or import the new profile.

Rename a Profile



1. Select a profile from the list.
2. Tap **Rename**.
3. Enter a new profile name using the keypad.

If the name is already in use, acknowledge the message and enter a different profile name.

Reset Profile Settings

The reset function returns all settings for a crew to their factory default values.

1. Select a profile from the list.
2. Tap **Reset Profile**.
3. Confirm the request.

Activate a Profile

You may activate a crew profile from the Manage Crew Profiles page or from a pop-up list on the database start-up page (if multiple profiles are available).

If only one profile exists, the profile is activated automatically upon unit power up.



Once activated, the profile name turns green (active). All inactive profiles display in white text.

ACTIVATE FROM THE MANAGE CREW PROFILES PAGE

From the MFD Home page:

1. Tap **System** > **Crew Profile**.
2. Select a profile from the list.
3. Tap **Activate**.

ACTIVATE FROM THE DATABASE START-UP PAGE

Select Crew Profile Pop-up, GDU 700P MFD



The Default profile is selected automatically.

If multiple profiles exist, a pop-up list allows you to select a profile during power up. Tap **Crew Profile** and select from the available options. Tapping **Continue** activates the selection and closes the pop-up.

The **Crew Profile** key does not appear if only one profile exists.

Installations with a GTN Xi Series Navigator:

Crew profile options display when multiple profiles exist. Choose a profile from the list and then tap **Continue**.

Profile options do not display if only one profile exists and database confirmation is performed remotely via GTN Xi. The MFD automatically advances to the next page (e.g., EIS start-up page, MFD Home page) once it is determined that no database issues exist.¹

¹ Requires GDU TXi software v3.30 or later with GTN Xi v20.20 or later.

System Indications

Alerts Types

The unit generates messages in response to various conditions that may occur. These abbreviated messages include:

- Warnings
- Cautions
- Function and mode advisories

Warnings & Cautions

Warnings require immediate attention. A caution indicates the presence of an abnormal condition that may require pilot action. A warning may follow a caution if no attempt is made to correct the condition (e.g., altering the aircraft's path to avoid the alerted terrain or obstacle).

System & Function Advisories



Advisories provide status and operating information.

System advisories. These display on a dedicated page or slide over window. Depending on the number of advisories, this list may be scrollable.

Function or mode specific advisories. These appear as unobstructed annunciations on the associated display.

System Advisory Window

Mode Advisory Annunciation

For a complete list of all system-related advisories, refer to section 11.

Alert Annunciations

Alert annunciations are abbreviated messages that indicate an alerted function or mode. The color of the annunciation depends on the alert type.

ALERT COLORS
WARNING
CAUTION
ADVISORY

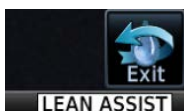
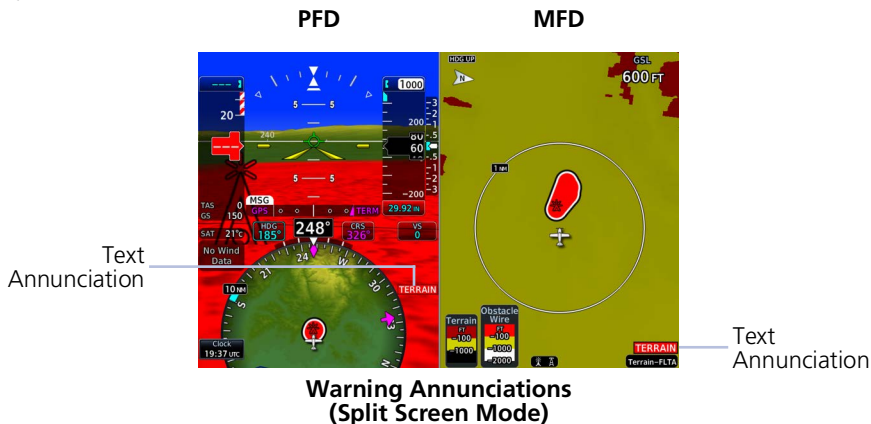
- Warnings display in white text on red background
- Cautions display in black text on amber background
- Function or mode specific advisories display in black text on white background

When an alert is triggered, the annunciation flashes by alternating text and background colors. It turns solid after five seconds. All annunciations remain active (solid) until the condition is resolved or no longer a threat.

Engine caution and warning annunciations flash until acknowledged or the condition is resolved.

ANNUNCIATION LOCATION

The location of the annunciation varies according to display and function. On the PFD, alerts annunciate to the right of the HSI. On the MFD, they annunciate in the lower right corner of the page. PFD alerts show on all configured PFD units in the TXi system.



Informational advisories pertaining to EIS modes and functions appear in the annunciator bar along the bottom of the screen.

The EIS can be configured with the aircraft's caution and warning system to provide engine alert annunciations. Read more about engine alerts in section 9.

Pop-up Alerts

Pop-up Alert Suppression

Pop-ups are suppressed if the alerted function's associated page is active on another TXi display or on the configured GTN Xi series navigator if present.¹

A pop-up window may display over the MFD if a warning or caution relating to terrain, traffic, or the backup battery occurs.

Depending on configuration, pop-up alerts may appear on one pilot display and one copilot display (if present).² Display priority is based on the available Garmin LRUs in the cockpit.

1. Pilot/copilot TXi MFD
2. Center TXi MFD
3. Pilot/copilot GTN Xi³

Closing a pop-up window on one display closes it on all configured LRUs.²



¹ TXi software earlier than v3.61: Display of the function's associated MFD page is limited to the unit only.

² Available only with TXi software v3.61. ³ Available only with GTN Xi software v20.40.

GDU 1060/1210: If the PFD is in full screen mode during an alert, a flashing page access key (**Terrain** or **Traffic**) replaces the **MFD** key.

Terrain
Key



PFD Full Screen
Mode

Terrain
Warning
Annunciation

**Warning Indications
(Full Screen Mode)**

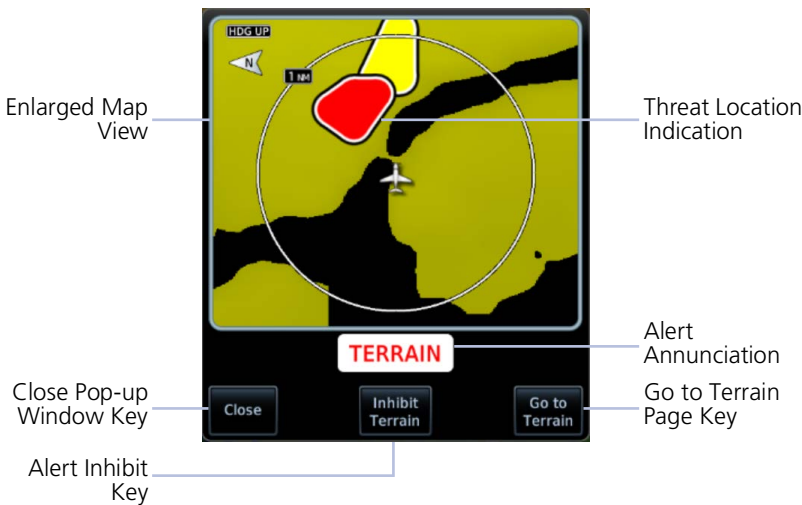
Pop-up Alert Priority

In the event of simultaneous alerts, pop-up windows display in the following order:

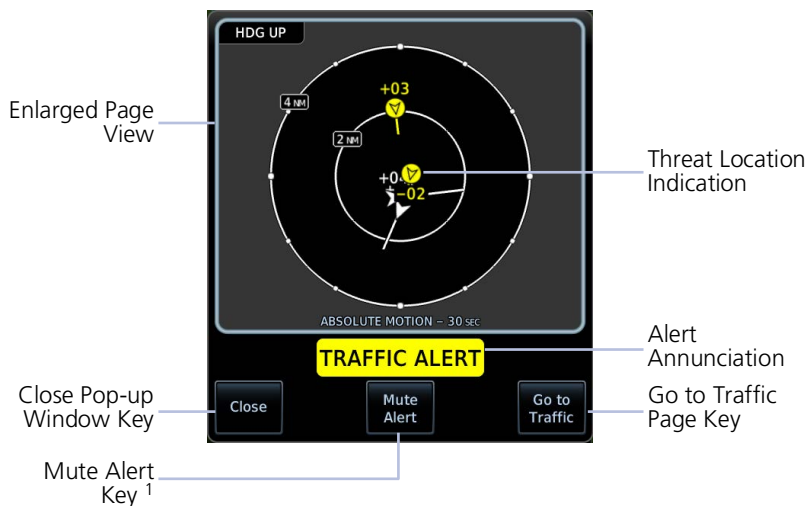
1. Terrain alerts
2. Traffic alerts
3. Battery alerts

Each pop-up alert contains:

- Enlarged view of the alerted threat location
- Alert annunciation
- Alert inhibit information, if applicable
- Control key for closing the pop-up window
- Direct access key to associated MFD page



Terrain Pop-up Alert Layout



Traffic Pop-up Alert Layout

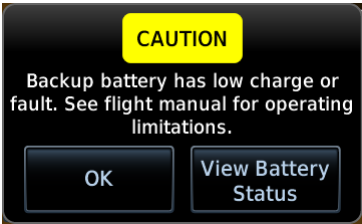
To open the indicated MFD page, tap **Go to <Page>**.

To acknowledge the alert and return to previous page view, tap **Close**.

For information about the terrain alert inhibit function, read *Alert Inhibit* in section 8. For information about the traffic alert mute function, read *Aural Alerts* in this section.

¹ ADS-B traffic only.

BATTERY POP-UP ALERTS



GDU 700P/1060/1210

These pop-ups alert to backup battery status. Available options are dependent on unit and alert type.

Tapping **OK** acknowledges the alert and closes the pop-up.

GDU 700P/1060/1210: The **View Battery Status** key provides direct access to the External LRU page.

System – External LRUs			
LRU		Status	
Battery	Charge:	75%	
	Temperature:	20°C	

Battery charge and temperature status display at the top of the LRU list.

GDU 700L: On PFD, battery pop-up alerts display in one of two locations depending on the state of the battery and the aircraft.



Vertical Pop-up Alert

Cautions display vertically along the left side of the screen.

These alerts may occur in flight. They do not display when the PFD menu is open.



Horizontal Pop-up Alert

Loss of aircraft power alerts display horizontally in front of the HSI.

This alert occurs only on ground. It displays whether the menu is open.

Aural Alerts

Some alerts are accompanied by an aural voice message. Voice gender is configured during installation. To determine which alerts provide aural indications, refer to the applicable alerts table.

MUTE ALERT

FEATURE LIMITATIONS

- Available only with GTX 345 and ADS-B software v3.20 or later



Tapping **Mute Alert** silences the active traffic alert. This key is available on the traffic alert pop-up.

The mute alert function is applicable only to the active aural alert. It does not mute future alerts.

LRU Failure Annunciations

LRU fail annunciations are designed to be immediately recognizable. If an LRU failure occurs, a red or amber “X” appears over the associated flight instrument(s) or data field(s). For additional information regarding pilot responses to LRU failures, consult the AFMS.

The annunciation color is determined by instrument type.

- Red “X” displays over primary flight instruments and data fields
- Amber “X” displays over non-primary flight instruments

PFD Failure Annunciations

During power-up, certain flight instruments are invalid until the associated equipment completes initialization. Data fields (e.g., TAS, GS, SAT) display only dashes when the input source is not available.



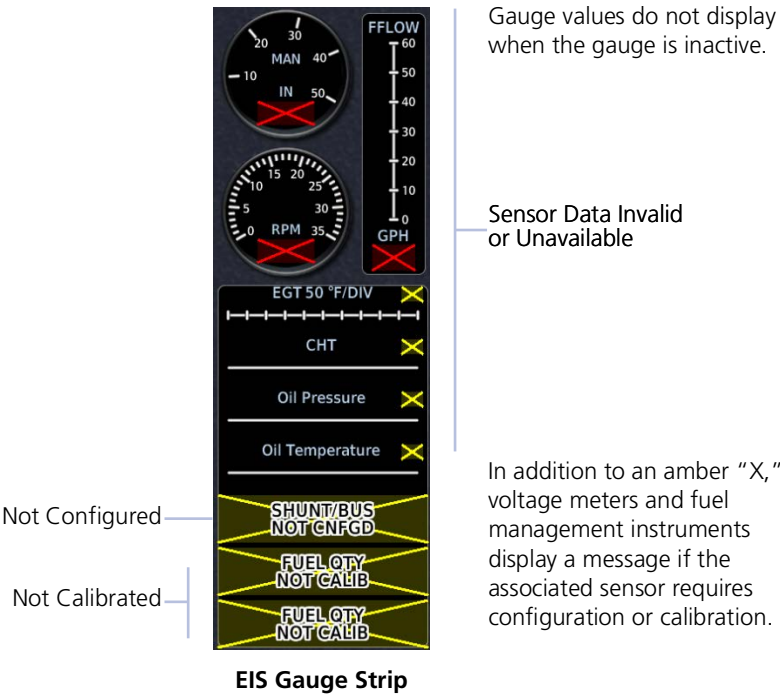
If an instrument remains flagged after one minute, check the status of the associated LRU, then contact a Garmin dealer for additional support. For Garmin LRUs, go to the External LRUs page and note any failed LRUs.

INDICATOR	ANNUNCIATION AND CONDITION
Airspeed	Annunciation: Red "X" Condition: Display is not receiving airspeed data from ADC.
Altitude	Annunciation: Red "X" Condition: Display is not receiving altitude data from ADC.
Attitude	Annunciation: Red "X" Condition: Display is not receiving attitude and heading information from the AHRS.
G-meter¹	Annunciation: Amber "X" Condition: Display is not receiving acceleration data from the AHRS.
Vertical Speed	Annunciation: Amber "X" Condition: Display is not receiving vertical speed data from ADC.
HDG FAIL	Annunciation: Red rectangle with white text "HDG FAIL" Condition: AHRS indicates that the magnetic heading value is invalid.

¹ TXi software v3.20 and later. Applicable only to GDU 700L/1060/1210 displaying G-meter in the Gauge format.

EIS Failure Annunciations

An LRU failure annunciates on the EIS when instrument sensory data become invalid or unavailable.



GDU Failure Annunciations

Fan Failure Annunciation, GDU 700()

ANNUNCIATION	CONDITION
FAN FAIL	GDU internal cooling fan failure for >10 seconds.

ADS-B Status

FEATURE REQUIREMENTS

- *Universal Access Transceiver (GDL 88, GNX 375, GTX 345)*

View last uplink time and GPS source information from the ADS-B Status page. You may access this page one of two ways.

From the Home page:

1. Tap **System** > **External LRUs**.
2. Scroll to the UAT device.
3. Select **More Info**.

From the Weather page:

Tap **FIS-B Weather** > **Menu** > **Datalink Status**.

UPLINK TIME

TEXT COLOR	MINUTES SINCE LAST UPLINK
Green	< 5
Yellow	5 to 15
	> 15

This field displays the number of minutes since last uplink. Digital values may change color depending on duration.

"> 15" displays when the time exceeds 15 minutes.

Dashes indicate when valid uplink data is unavailable (e.g., the device is offline).

FIS-B WEATHER

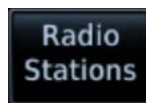
View the status of FIS-B weather products. For additional details, tap the appropriate key.

Raw Text Reports



View raw textual data for a FIS-B product.

Radio Stations



Monitor FIS-B ground station transmission status and reception completeness for certain FIS-B products.

TRAFFIC APPLICATION STATUS

View the status of all three traffic applications: • AIRB • SURF • ATAS (airborne alerts)

ANNUNCIATION	DESCRIPTION
On	Application is running. Required ownship data is available and meets the performance criteria.
Available to Run	Application is configured. Required input data is available and meets the performance criteria.
Unavailable to Run	Required input data is not available due to a failure (e.g., aircraft surveillance application process failed).
Unavailable - Fault	Required input data is available, but does not meet the performance criteria or is not available due to non-computed data conditions.

Logs



The data logging function is capable of storing approximately 100 hours of flight data in the unit's internal memory. This information is available for export to an SD card for later analysis.

FEATURE REQUIREMENTS

For external data logging:

- *SD card*

For data streaming:

- *A flyGarmin user account*
- *Garmin Pilot on a portable electronic device*
- *Flight Stream 510 wireless transceiver for flight data streaming*
- *Flight Stream 510 and GDU TXi EIS for engine data streaming*

FEATURE LIMITATIONS

- *Engine exceedance logs are available only for turbine aircraft*
- *External data logging stops when the SD card is full*

Data Logging Function

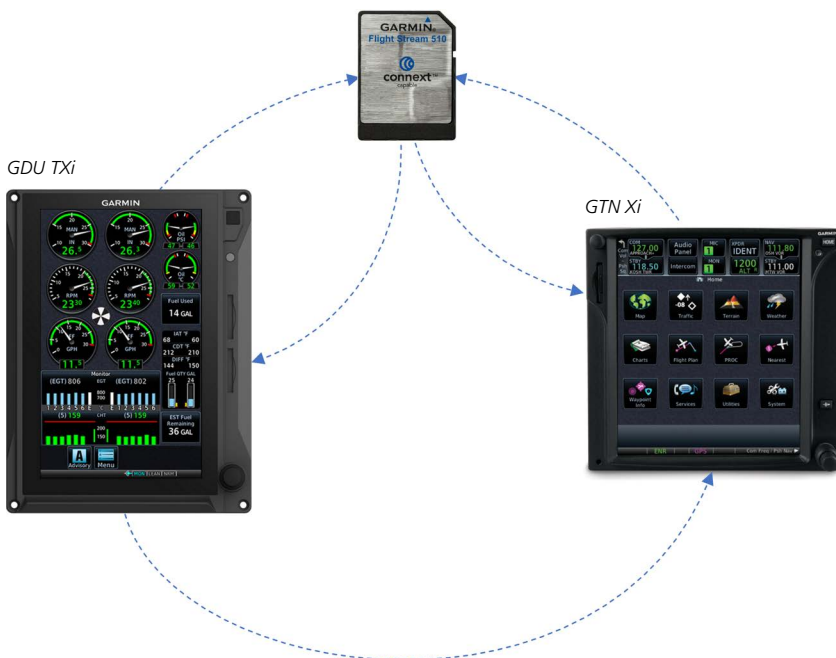
- Generates log files automatically upon unit power-up
- Records various parameters related to aircraft flight instruments, engine indications, and configuration
- Streams logged flight and engine data to Garmin Pilot
- Writes logged data to an SD card if one is present in the top/left slot
- Overwrites oldest files when the internal log reaches capacity
- Saves files in the .csv format and stores them in the "data_log" folder

Data Logging at a Glance

Upon power up, GDU begins logging flight and engine data automatically, storing the data in its internal memory.

DATA LOGGING WITH FLIGHT STREAM 510

To automatically upload data to flyGarmin.com, install Flight Stream into GDU or GTN.



If Flight Stream is present in GDU:

- GDU streams data to Garmin Pilot via Flight Stream¹

If Flight Stream is present in GTN:

- GDU transfers logs to GTN, which then streams the data to Garmin Pilot via Flight Stream¹
- GTN is the preferred location for Flight Stream installation

¹ Pilot setup required.

EXPORTING DATA TO AN SD CARD

To export data logs to an SD card, insert the card into the top/left slot of GDU.

GDU 700P



If an SD card is present in the top/left slot:

GDU writes the data to an SD card. No action is required.

If an SD card is not present:

Internal data logging still occurs. You may insert an SD card and use the export function to write GDU data to the card after landing.

What happens if there's a power interruption?

Data logging stops if power is lost. All data recorded up to that point remains stored in the internal memory. Data is not recorded for the duration of the outage. When GDU reboots, logging automatically resumes with a new log file.

In a multi-GDU system, we recommend placing an SD card in each GDU to provide continuous data logging in the event of a GDU outage.

Read more about this method in *Exporting Data Logs to an SD Card*.

File Naming Conventions

Log file names provide a useful reference to aid data analysis. They include:

- Log start date and time
- Identifier of the nearest waypoint at the indicated start time



Files with the name "log_000101_0000XX____.csv" do not contain a valid date and time stamp. These logs may include data recorded during installation.

Exporting Data Logs to an SD Card



NOTE

*Do not eject the SD card while the export function is in progress. Wait until the **Export Data Log** key is available before ejecting the card.*

Home



Exportable parameters are dependent on aircraft type.

1. Insert an SD card into top/left slot.
2. Power on GDU.
3. Go to the Logs page.
4. Tap **Export Data Log**.

Exporting Logs: 89%

Export in Progress

During export, the **Export Data Log** key is unavailable (gray), and displays the export progress percentage. It becomes available once export is complete.

What if I forget to insert my SD card before flight?

Internal flight and engine data logging occurs regardless of whether an SD card is present. After your flight, you may insert a card and download log files using the **Export Data Log** command.

With the aircraft safely on ground:

1. Power off unit.
2. Insert card.
3. Power on unit.
4. Tap **System** > **Logs** > **Export Data Log**.

Wait for the export to complete before ejecting the SD card.

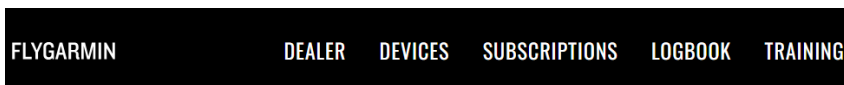
Once the files are exported onto an SD card, you may transfer the log files to your personal logbook on flyGarmin.com.

Duration varies depending on the number of files present. For lengthy transfers, it may be necessary to run the engine or operate on ground power until the export is complete.

TRANSFER SD CARD DATA TO YOUR ONLINE LOGBOOK

The **Logbook** tab on flyGarmin.com provides functions for creating and viewing multiple logbook entries.

flyGarmin Navigation Tabs



To create a logbook and upload data:

1. Sign in to your flyGarmin.com account.
2. Select **Logbook > Setup**.
3. Select **Add Aircraft Type**.
4. Provide all necessary aircraft type details, then select **Save Aircraft Type**.
5. Provide all necessary settings and pilot information, then select **Save Settings**.

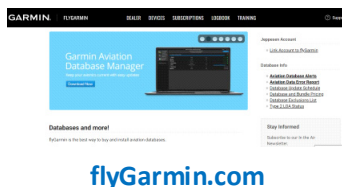
To upload flight logs to an existing logbook entry:

1. Select **Logbook > Entries > Upload Flight Logs**.
2. Select and upload the log files according to the on-screen instructions.

Data Logging with Flight Stream 510



The unit automatically streams logged flight data to the Garmin Pilot app when Flight Stream 510 is present and paired to a supported tablet or phone.



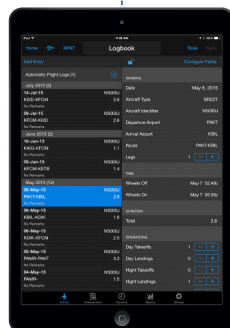
The data uploads to flyGarmin.com upon connecting to Wi-Fi with Internet.



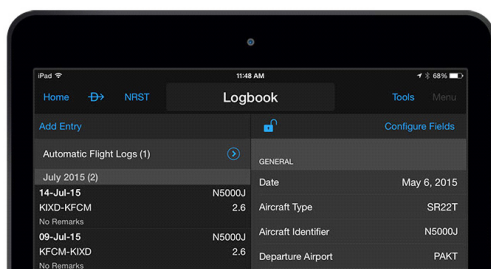
Logged flight and engine data streams to Garmin Pilot.



Garmin Pilot App



Your portable device downloads the data upon connecting to Flight Stream 510.



This includes any previous flights not already downloaded.

What happens if I forget to bring my tablet on the flight?

No need to worry. During flight, GDU records log data to the internal storage and SD card (if present). The next time you fly with your tablet, the data will stream to Garmin Pilot. This includes all previously recorded flights. The device downloads the data upon connecting to Flight Stream 510.

Exceedance Logging



Unlike flight data logging, which records parameters at a specific rate, exceedance logs record information about a specific gauge when its value exceeds a threshold.

FEATURE REQUIREMENTS

- *Turbine EIS enablement*

OR

- *Fixed-wing aircraft with variable V_{NE}*

FEATURE LIMITATIONS

- *Engine exceedances apply to turbine aircraft only*

Not all gauges have exceedance logging. Availability varies depending on airframe and engine.

Exceedance Recording Function

- Generates an exceedance file after a gauge parameter exceeds its maximum allowable time at an elevated level
- Engine exceedances record in-depth data about the exceeded gauge, and contextual details about other parameters during that time
- Advisories notify 30 seconds after the exceedance ends (during flight), and once the aircraft is on ground
- Entries are available for viewing on the dedicated summary page

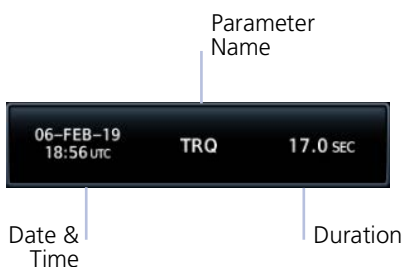
Exceedances are installer configured to match the aircraft limitations in the POH. Pilots are responsible for knowing and respecting all aircraft and engine limitations.

Exceedance Gauges

- Torque
- Prop RPM
- NG/N1
- N2
- Turbine Engine Temperature
- Oil Temperature
- Oil Pressure
- Fuel Pressure

Many engine gauges have a maximum allowable time limit that the parameter may operate at an elevated level. This “grace period” is indicated on the gauge’s associated countdown timer. Log entries do not generate until after this timer expires.

The duration of the exceedance is the amount of time that the indicated parameter exceeds the limit, which includes the grace period time.



Each log entry includes the parameter name, and the date, time, and duration of the exceedance.

Date	Gauge	Exceed	Advisory
06-FEB-19 20:28 UTC	ITT	Unknown	ACK Alert
06-FEB-19 18:57 UTC	ITT	Unknown	ACK Alert
06-FEB-19 18:56 UTC	TRQ	17.0 SEC	UNACK Alert
06-FEB-19 18:56 UTC	NP	13.0 SEC	UNACK Alert
06-FEB-19 18:56 UTC	NG	11.0 SEC	ACK Alert
06-FEB-19 18:56 UTC	TRQ	15.0 SEC	ACK Alert
06-FEB-19 18:55 UTC	NG	27.0 SEC	ACK Alert
06-FEB-19 18:55 UTC	TRQ	49.0 SEC	ACK Alert

Exceedances Summary Page

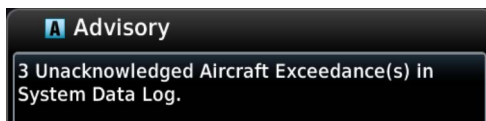
INCOMPLETE ENTRIES



If GDU loses power while recording an exceedance, the system marks the duration as “Unknown.”

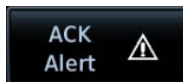
Incomplete log files contain data recorded up until the time of the interruption. Some fields are not available due to the GDU not having knowledge of parameters at the end of the exceedance.

Exceedance Alerts



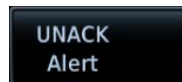
An advisory informs that an exceedance alert is available for viewing in the data log. These alerts occur 30 seconds after the exceedance ends and upon landing. Notifications persist until all exceedances are acknowledged.

ACK Alert



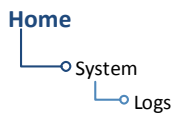
Tapping this key acknowledges the corresponding alert. An alert symbol indicates the advisory condition requires attention.

UNACK Alert



Tapping this key unacknowledges the corresponding alert. Unacknowledged entries rise to the top of the summary page.

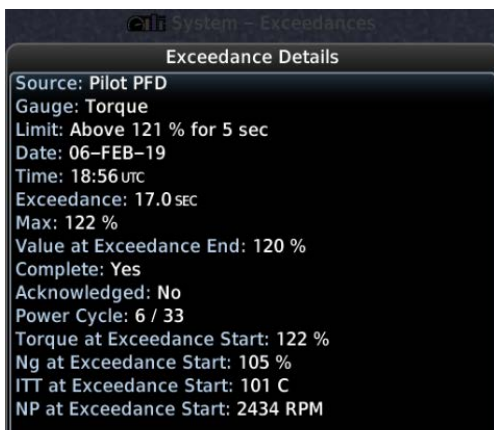
ACKNOWLEDGE AN EXCEEDANCE ALERT



To view and acknowledge the exceedance advisory:

1. Go to the Logs page.
2. Tap **Exceedances**.
3. Review the list of alert entries.
4. Tap **ACK** for each unacknowledged entry.

Exceedance Details



Select an exceedance file to view gauge details.

Duration and highest value are useful for determining whether maintenance actions are required.

The provided values of other engine parameters may be useful when determining if an exceedance resulted from sensor testing or failure.

Exceedance Log Data

- Gauge and exceeded parameter
- Date & time of occurrence
- Duration (total amount of time the parameter exceeded the limit, which includes the grace period time)
- Highest gauge value
- Power cycle
- Max parameter value during exceedance period
- Parameter value at end of exceedance
- Value of other engine parameters at beginning of exceedance
- Any interruptions (e.g., power loss)
- Completeness of data
- Alert state (i.e., acknowledged or unacknowledged)

The GDU uses an internal circular log to store exceedances entries for 100 power cycles that have had an exceedance. The user should periodically export the exceedance log to SD card.

If date and time parameters are not available (e.g., during engine start when the navigator has yet to acquire a GPS signal), use the power cycle parameter value to narrow down the time of occurrence.

Exporting to SD Card

Exportable parameters are dependent on aircraft type.

1. Insert an SD card into top/left slot.
2. Power on GDU.
3. Go to the Logs page.
4. Tap **Export Data Log**.

INTENTIONALLY LEFT BLANK

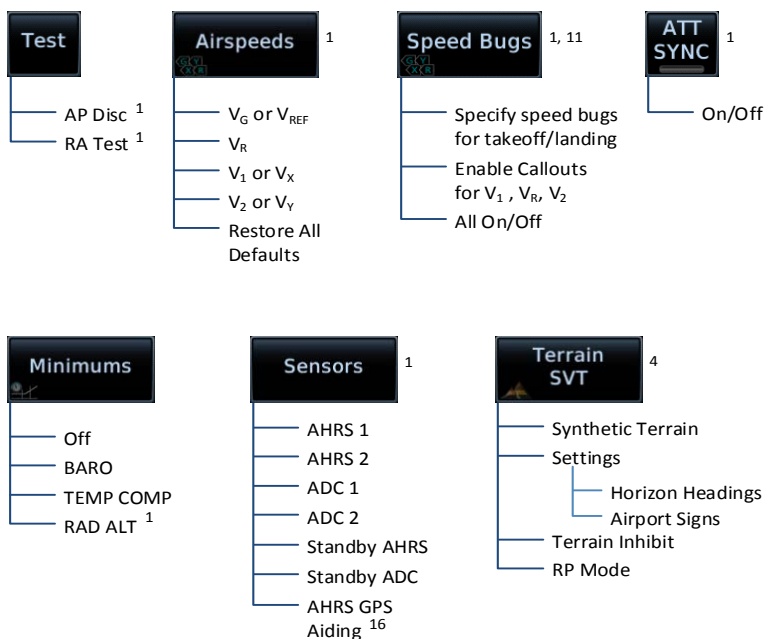
3 Primary Flight Display

PFD SETUP	3-4
FLIGHT INSTRUMENTS	3-15
SUPPLEMENTAL FLIGHT DATA.....	3-72

PRIMARY FLIGHT DISPLAY MENU FUNCTIONS



The PFD menu provides access to various controls, sub-menus, and setup options.



¹ Dependent upon unit configuration. ² Not available on GDU 700L PFD.

³ GDU 1060/1210: Units, audio, backlight, status, and database options are located in the System page of the MFD.

⁴ Menu functions pertain to the specified terrain function. ⁵ GDU 700L only.

⁶ In multi-PFD systems, a SYNC Control page access key replaces Database SYNC. Options are dependent upon display type. ⁷ Feature requires Flight Stream 510 wireless transceiver.

⁸ Available with TXi software v3.21 and later. ⁹ Available with TXi software v3.40 and later.

¹⁰ Available with TXi software v3.50 and later. ¹¹ Available with TXi software v3.61 and later.

¹² WX-500 only. ¹³ Installations with an autopilot: Flight Director key replaces Aircraft Symbol key when configured for flight director. ¹⁴ Available with TXi software v3.70 and later.

¹⁵ Available with TXi software v3.80 and later. ¹⁶ Available with TXi software v3.85 and later.

HSI Setup ²

- HSI Map
- Overlays¹
 - TOPO
 - Terrain
 - Obstacles & Wires
 - Traffic
 - NEXRAD/PRECIP
 - Lightning/Stormscope¹
 - Select mode¹²
- Bearing Pointers
 - Select source
- Wx Datalink
 - Select source

Bearing Pointers ⁵

- Select source

PFD Setup ³ **GDU 1060³** **GDU 1210**

- Clock/Timer
 - Display Clock/Timer
 - Select time format
 - Set local offset
- TEMP/DALT
 - Outside Air TEMP
 - Density Altitude
 - Select Air TEMP REF
 - Select TEMP units
- Menu Timeout
 - Set timeout
- LOC CDI Prompting¹
- Stabilized Approach^{1, 11}
 - Select alert options
- Wind Field
- AGL Field
- GPS Roll Indicator
- SYNC Control
- CDI/VDI Preview¹
- Altitude Setup^{1, 11}
 - Enable BARO Alert⁸
 - Set transition altitude
 - Select altitude knob function
- G-Meter⁸
 - Display G-Meter
 - Select style⁹
 - Reset Min/Max^{10, 13}
- Aircraft Symbol^{10, 13}
 - Select symbol type
- HDG Mode¹⁴
- DME Inset¹⁴
- Extreme Attitude^{1, 15}
- Declutter
- Auto EDM¹

System ⁴ **GDU 700()**

- PFD Setup
 - Clock/Timer
 - TEMP/DALT
 - Menu Timeout
 - CDI/VDI Preview¹
 - Auto EDM¹
 - GPS Roll Indicator
 - Wind Field
 - Stabilized Approach¹¹
 - Database SYNC⁶
 - Altitude Setup^{1, 11}
 - G-Meter⁸
 - Aircraft Symbol^{10, 13}
 - HDG Mode¹⁴
 - DME Inset¹⁴
 - Extreme Attitude
 - Declutter^{1, 15}
 - LOC CDI Prompting¹
- Units
 - Select units
- Audio
- Backlight
 - Set intensity offset
- Status
 - External LRUs
- Databases
- Flight Stream⁷
- Logs
 - Export Data Log
 - Aircraft Log¹
 - Exceedances¹
- Inhibit CO Alerts¹⁵

PFD Setup

GDU 700() PFD

Setup selections are accessible via the **System** menu key (**Menu > System > PFD Setup**).

PFD SYSTEM SELECTIONS, GDU 700()			
PFD Setup	• Access CDI/VDI Preview, Auto EDM, LOC CDI Prompting, and Wind Field functions • Access synchronization options: BARO, CDI, Database SYNC • Control clock/timer • Control Outside Air Temp/Density Altitude display • Toggle Wind Field on or off • Set menu display timeout • Access Stabilized Approach alert inhibit options⁵ • Toggle AGL Field on or off • Toggle GPS Roll Indicator on or off • Enable altimeter setting visual prompt (BARO Alert) for a specified transition altitude¹ • Set altitude knob function: Push SYNC or Coarse/Fine adjust⁵ • Access G-Meter Setup menu¹ options: Display G-Meter, Style ², Reset Min/Max • Toggle Aircraft Symbol to Chevron or Standard^{3, 4} • Access heading mode and slewing control options⁶ • Display DME information window⁶ • Toggle extreme attitude declutter mode on or off⁷		
Units	Except where noted, unit selections synchronize across all configured GDUs. Certain LRUs may contain unit settings independent of the TXi system. <table><tr><td> • NAV Angle • Temperature • Wind Speed </td><td> • BARO Pressure • Distance </td></tr></table> Altitude unit settings do not affect the altitude tape.	• NAV Angle • Temperature • Wind Speed	• BARO Pressure • Distance
• NAV Angle • Temperature • Wind Speed	• BARO Pressure • Distance		

PFD SYSTEM SELECTIONS, GDU 700()

Audio	• Set click volume
Backlight	• Adjust display brightness
Status	• View unit and software information • Check status of all configured LRUs
Databases	• View information about active and standby databases • Perform a manual database update
Flight Stream	• Access Bluetooth Setup and Wi-Fi Setup menus
Logs	• Access data, aircraft, and exceedance logs
Inhibit CO Alerts	• Toggle CO caution alerts on or off⁷

¹ Available with TXi software v3.21 and later.

² Style setup option available with TXi software v3.40 and later. Not available for GDU 700P.

³ Available with TXi software v3.50 and later. ⁴ Installations with an autopilot: Flight Director key replaces Aircraft Symbol key when configured for flight director. Availability dependent upon configuration.

⁵ Available with TXi software v3.61 and later. ⁶ Available with TXi software v3.70 and later.

⁷ Available with TXi software v3.80 and later.

GDU 1060/1210 PFD

Setup selections are accessible via the **PFD Setup** key in the main PFD menu (**Menu > PFD Setup**).

PFD SETUP SELECTIONS, GDU 1060/1210	
Clock/Timer	Control clock/timer
TEMP/DALT	- Control Outside Air Temp/Density Altitude display - Specify air temperature units and reference type
Menu Timeout	Set menu display timeout
LOC CDI Prompting	Allow prompts for switching the CDI source from GPS to LOC^{2, 3}
Wind Field	Control wind field function
CDI/VDI Preview	Enable preview indicators for VDI Glidepath/Glideslope deviation, and VOR/LOC course and deviation¹
Auto EDM	Enable automatic EDM⁴
SYNC Control	Access synchronization options: BARO or CDI⁵
Stabilized Approach ¹³	- Access Stabilized Approach alert inhibit options: - Approach Speed • BARO/GPS Mismatch • Course - Crosswind • Flaps • Gear • Glidepath/Glideslope • Tailwind
Database SYNC	- View information about active and standby databases - Perform a manual database update

PFD SETUP SELECTIONS, GDU 1060/1210

AGL Field	• Display GPS height above terrain (AGL)
GPS Roll Indicator	• Display GPS navigator's roll steering command⁷
Altitude Setup	• Enable altimeter setting visual prompt (BARO Alert) for a specified transition altitude⁸ • Set altitude knob function: Push SYNC or Coarse/Fine adjust¹²
G-Meter	• Access G-Meter Setup menu⁸ options: • Display G-Meter • Style⁹ • Reset Min/Max
Aircraft Symbol	• Toggle symbol type to Chevron or Standard^{10, 11}
HDG Mode	• Access heading mode and slewing control options¹⁴
DME Inset	• Display DME information window¹⁴
Extreme Attitude Declutter	• Toggle extreme attitude declutter mode on or off¹⁵

¹ Available only when a configured GTN is present.

² Available only when a configured GPS/NAV navigator (GTN 650/750 or GNS 430/530) is present.

³ For installations with a GFC 600 or GFC 500 autopilot: LOC CDI prompting is suppressed when the autopilot is armed to capture the localizer. The TXi CDI source automatically switches from GPS to LOC once the autopilot couples to the localizer.

⁴ Available only with GFC 600 installation and cabin altitude threshold configured.

⁵ Available for systems with multiple sources. ⁶ Available when M_{MO} value is configured.

⁷ Indicator is hidden when flight director is active. ⁸ Available with TXi software v3.21 and later.

⁹ Style setup option available with TXi software v3.40 and later.

¹⁰ Available with TXi software v3.50 and later. ¹¹ Installations with an autopilot: Flight Director key replaces Aircraft Symbol key when configured for flight director. Availability dependent upon configuration.

¹² Altitude Setup menu and knob function setting available with TXi software v3.61 and later.

¹³ Function availability and alert options dependent upon configuration. Requires TXi software v3.61 or later.

¹⁴ Available with TXi software v3.70 and later. ¹⁵ Available with TXi software v3.80 and later.

Synchronization Options

FEATURE LIMITATIONS

- *Multi-PFD systems only*
- *SYNC Control availability dependent upon configuration*



Selecting **SYNC Control** opens a pop-up menu. Selected functions synchronize across all TXi PFDs.

GDU 1060/1210: The database synchronization option resides on the MFD System Status page.

BARO	• Synchronizes the current barometric pressure value¹
CDI SRC	• Synchronizes the selected CDI source²
DB	• Initiates the database synchronization function • GDU 700() only

¹ Option not available if BARO synchronization function is configured as always on.

² Option not available if CDI synchronization function is configured as always on.

Altitude Knob Control Options

FEATURE REQUIREMENTS

- TXi software v3.61 or later

FEATURE LIMITATIONS

- Feature availability dependent upon installer configuration
- Installations with GCU 485 model -06 or -16: Coarse/fine adjustment is the altitude knob's default functionality; setup menu option not available

Customize altitude knob functionality. Select **Altitude Knob** to choose between the available control options.



When **Coarse/Fine** is selected:

- Pushing and turning the inner knob adjusts selected altitude using fine adjustment increments (100 ft)
- Turning the inner knob without pushing adjusts selected altitude using coarse increments (1,000 ft for altitudes above the transition altitude; 500 ft otherwise)



When **Push SYNC** is selected:

- Pushing the inner knob synchronizes selected altitude to the current aircraft altitude
- Rotating the inner knob adjusts the selected altitude using 100 ft adjustment increments

PFD Selection

FEATURE REQUIREMENTS

- *TXi software v3.61 or later*
- *Copilot PFD*
- *GCU 485 model -06 or -16*

FEATURE LIMITATIONS

- *Pilot transfer of Selected Side not available while EDM is active*

OPTIONAL COMPONENTS

Feature availability is dependent upon software version.

- *GTN Xi series navigator (with GTN Xi software v20.40 or later)*
- *GTX 345 (with ADS-B board v3.51 or later)*

For dual PFD installations with pilot and copilot PFDs, the Selected Side function allows you to choose which PFD will control system functions.



Pushing the **XFR** key on the PFD controller toggles the active side selection between pilot and copilot. Pilot PFD is the default selection.

When you enable the Selected Side function:

- GFC 600 uses only the selected PFD's reference bugs and active Nav and ADAHRS source selections
- GTN Xi uses only the selected PFD's reference bugs and ADAHRS sensor data for navigation (including barometric altitude for VNAV path calculation)
- GTN Xi sends ADS-B and enhanced surveillance data parameters relevant to the selected PFD to the interfacing transponder
- GTX 345 transmits ADS-B and EHS data parameters relevant to the selected PFD
- Alert audio plays only from the selected PFD

If the selected PFD is offline, the active selection automatically switches to the other PFD.

You can toggle the active side selection while the autopilot is offline, such as during power up and preflight procedures.

For information about Selected Side functionality for the GTN Xi series navigator, consult *GTN Xi Series Pilot's Guide*.

SELECTED SIDE INDICATION

A green arrow at the top of the display indicates the active side selection: pilot PFD or copilot PFD. Arrow direction is based on aircraft type.

Selected Side
Indication



	PILOT PFD	COPILOT PFD
Fixed Wing		
Rotorcraft		

Reference Bugs and Controls

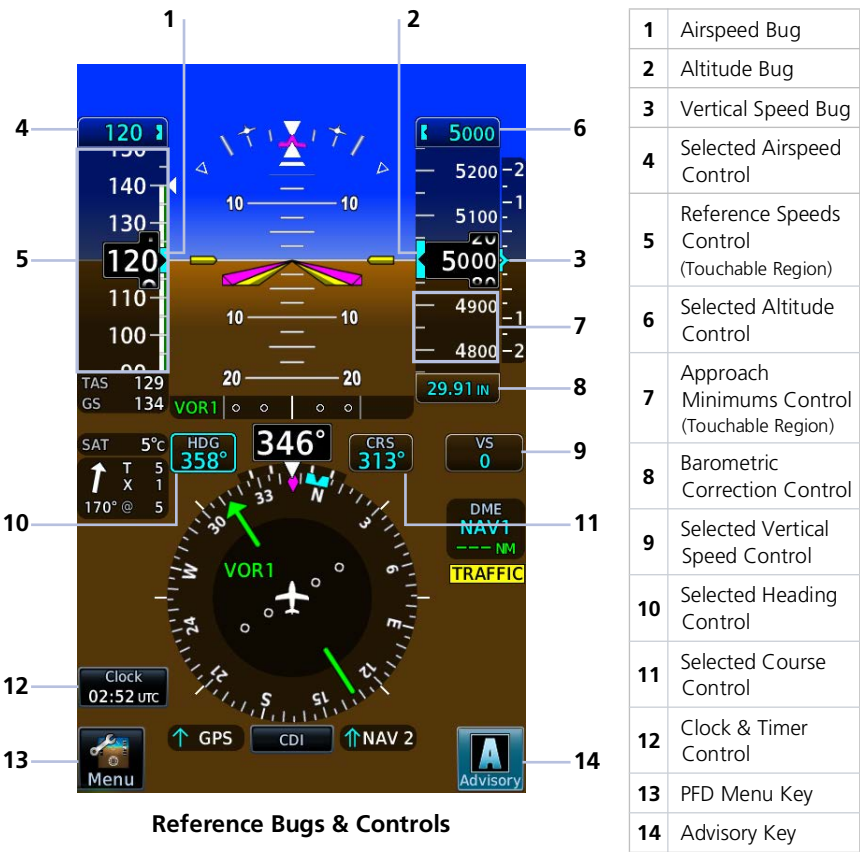
PFD Controls

- Airspeed
- Altitude
- Barometric Correction
- Heading
- Vertical Speed
- Selected Course

The PFD relies on touchscreen and control knob interactions for instrument and course adjustments.

Only one control is active at a time. Control automatically reverts to heading 10 seconds after the last entry is made.

Additional information shown is determined in menu and setup, including synthetic vision depictions and HSI Map overlays.



SETTING A REFERENCE BUG

Selecting any control opens a dedicated menu. Available options are dependent upon system configuration.



GDU 700L

On GDU 700L: Menu slides out from the left side of the screen.

On GDU 700P/1060/1210: Menu slides up from the bottom of the screen.



GDU 700P/1060/1210

There are three methods for operating PFD controls:

1. Touch and Turn

Tap the control to move knob focus to that control, then:

- turn inner knob to change the selected value

OR

- push knob to synchronize with the current value¹

2. Turn and Turn

Turn outer knob to move knob focus to the desired control, then:

- turn inner knob to change the selected value

OR

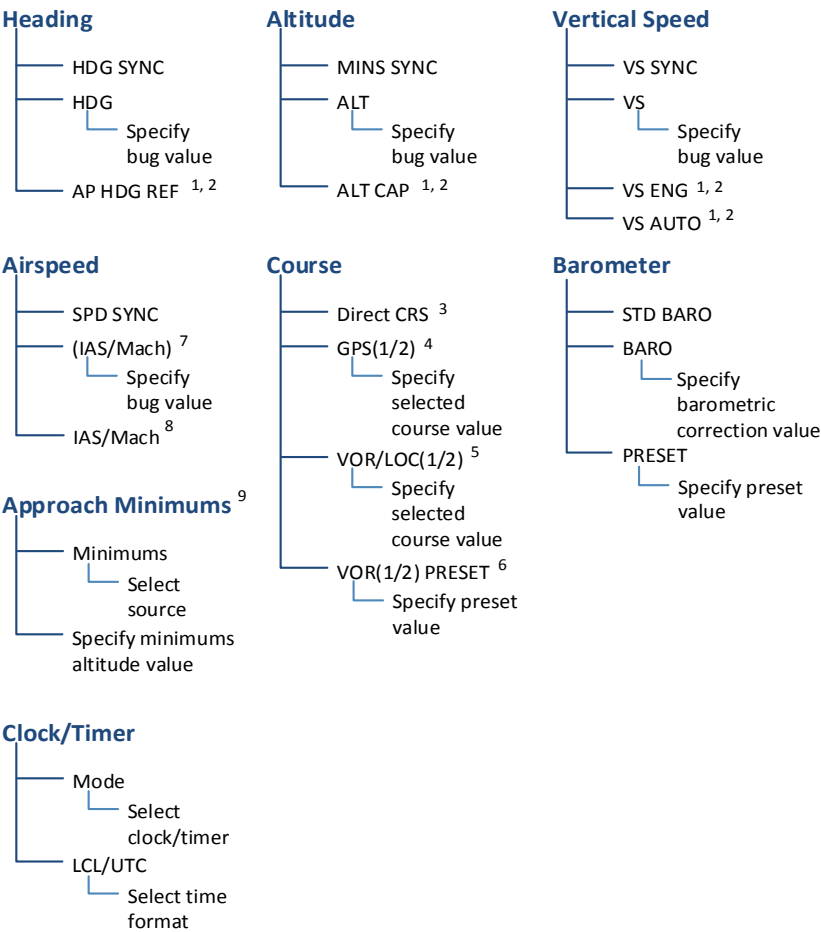
- push knob to synchronize with the current value¹

3. Touch and Touch (GDU 700P/1060/1210)

Tap the control and use the keys provided in the associated context menu.

- Center key opens a keypad for numeric entries
- Additional controls reside on the left and right

¹ TXi software v3.61 and later: Push-to-synchronize option not applicable to selected altitude if Altitude Knob control is set to Coarse/Fine.



Reference Controls & Menu Options

¹ Requires an autopilot. ² Not applicable to all autopilots. ³ Requires an active GPS source.
⁴ Available during OBS mode only. ⁵ Available only when VOR/LOC is the active CDI source.
⁶ Preset course functionality not available when GDU 620 is present (multi-PFD systems only).
⁷ IAS and Mach reference controls are mutually exclusive. **IAS** key label changes to **Mach** when you enable the Mach Selection control option.
⁸ If configured for Match Selection and interfaced with a GFC 600, tapping **IAS/Mach** toggles the Mach Selection function on or off. This feature is available only with TXi software v3.61 and later.
⁹ Approach Minimums control shortcut available only with TXi software v3.61 and later.

Flight Instruments

Supplemental Flight Data

- GPS Navigation Status
- DME
- Radar Altitude
- Wind
- Clock/Timer

The top portion of the display provides attitude and air data.

The lower portion provides a horizontal situation indicator and supplemental flight data fields.

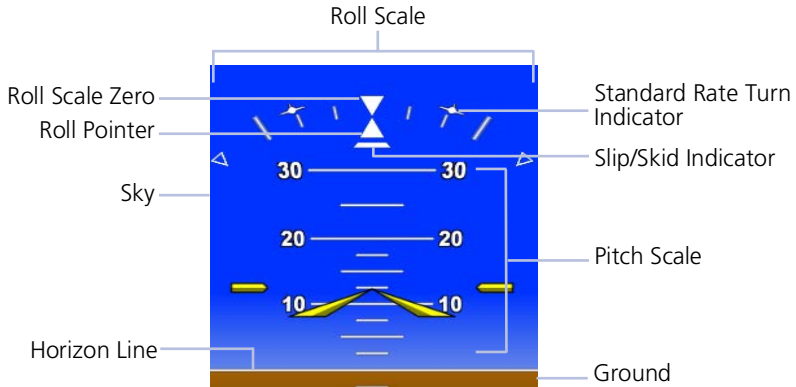




PFD Landscape Layout, Rotorcraft

Attitude Indicator

The attitude indicator provides a virtual representation of the ground and sky.



ROLL POINTER



Fixed Pointer



Sky Pointer

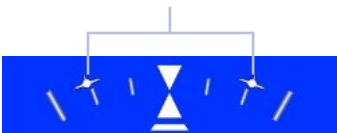
The roll pointer indicates angle of bank at varying degrees on the roll scale.

Pointer type is configured at installation as either fixed or sky.

- Major tick marks: 30° and 60°
- Minor tick marks: 10°, 20°, and 45°

STANDARD RATE TURN INDICATORS

Standard Rate Turn Indicators



- Show the required bank angle needed to maintain a standard rate turn
- Dynamically change position based on True Airspeed
- Remain at 30° when TAS is at or above 210 kts

Standard rate turn indicators are removed when:

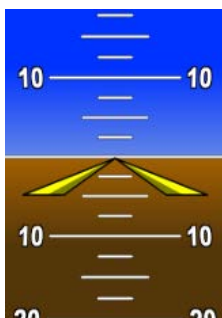
- TAS is invalid
- TAS is less than 60 kts
- Aircraft is on the ground

SLIP SKID INDICATOR



- Moves laterally away from roll pointer center (max 7° side-slip indication)
- Alignment with roll pointer denotes coordinated flight
- If attitude data becomes invalid, slip/skid indicator is removed

PITCH SCALE



- Markings range between -90° and 90° angles
- Major pitch graduations occur every 5° with SVT and 10° without SVT
- Each minor pitch mark is 5° from 25° nose down to 45° nose up
- Precision pitch marks are at each 2.5° from 20° nose up to nose down attitude

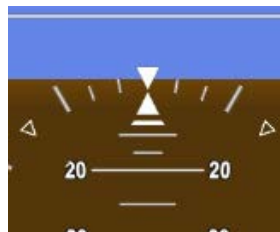
AWARENESS BANDS

Ground Awareness Band



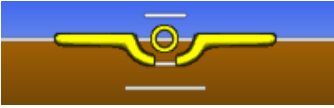
Appears when current pitch attitude causes the sky presentation to completely fill the display.

Sky Awareness Band



Appears when current pitch attitude causes the ground presentation to completely fill the display.

Aircraft Symbol



Rotorcraft

Depending on aircraft type, the aircraft symbol is either fixed wing or rotorcraft. For requirements, consult the POH.

AIRCRAFT SYMBOL KEY

FIXED-WING AIRCRAFT ONLY

FEATURE REQUIREMENTS

- TXi software v3.50 or later
- Aircraft Symbol/Flight Director option configured as "Pilot Control"



Tapping **Aircraft Symbol** toggles the symbol type between chevron and standard. This option resides in the PFD Setup menu.



Chevron



Standard

Aircraft symbol type selection automatically synchronizes with GI 275 (if present).¹

¹ Applicable only to TXi software v3.50 and later & GI 275 software v2.50 and later.

Attitude Sync



ROTORCRAFT ONLY

FEATURE LIMITATIONS

- Availability dependent upon configuration

Function disables when:

- Current pitch and horizon line differ by more than $\pm 8^\circ$
- Attitude data is invalid

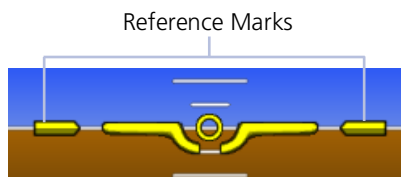
ATT SYNC KEY



- Synchronizes aircraft symbol to the horizon line for all configured GDUs
- Reference marks provide an absolute pitch reference
- Useful during pitch up or pitch down maneuvers



Attitude Sync Off



Attitude Sync On

Extreme Attitude Indications

FEATURE REQUIREMENTS

For extreme attitude declutter mode option:

- TXi software 3.80 or later

FEATURE LIMITATIONS

- Availability of selectable mode option dependent upon configuration



Extreme Pitch

To aid in recovery from extreme pitch attitude, red chevrons are shown between major pitch marks from 80° nose down to 30° nose down, and from 80° nose up to 50° nose up. Red chevrons always point toward 0° pitch.

If enabled, the PFD employs an extreme attitude declutter mode to improve instrument scan and facilitate aircraft recovery from extreme attitudes.



Extreme Roll

The PFD declutters if pitch exceeds 20° nose down or 30° nose up, or if bank angle exceeds 65°. When decluttered, HSI Map reverts to a standard HSI depiction until the PFD exits extreme attitude declutter mode.

The HSI Map option is accessible from the PFD menu. For information about this advanced feature, read *HSI Map* in section 4.

Data removed during extreme attitudes

- Open menus and keyboards
- GS
- TAS
- Air temperature
- Flight director command bars
- Marker beacon annunciation
- Vertical deviation indicator
- Clock/timer
- Selected Altitude control
- Vertical Speed control
- Selected IAS
- Bearing pointer window(s)
- GPS navigation status
- Fast/Slow indicator
- **CDI** source selection key
- PFD **Menu** key
- PFD **Full** key (GDU 1060/1210 only)
- Standard rate turn indicators
- GPS height above terrain (AGL)
- Wind field
- Selected heading
- Selected course
- Barometric pressure setting
- DME
- **Advisory** key

EXTREME ATTITUDE DECLUTTER MODE KEY

You may toggle extreme attitude declutter mode on or off at any time.¹ This mode is active by default.

Open the PFD Setup menu and tap **Extreme Attitude Declutter**.

AHRS(1/2) KEY



- Allows pilot to change the current AHRS source while extreme attitude decluttering is in effect
- Replaces **PFD Menu** key during extreme attitudes

¹ TXi software earlier than v3.80: Selectable mode option not available. PFD declutters automatically when conditions for extreme attitude declutter mode are met.

VNAV Guidance Indications



WARNING

Do not rely solely on VNAV guidance when navigating horizontally and vertically around user-defined airports. It is the pilot's responsibility to ensure separation from terrain and obstacles during an approach to a user-defined airport.

FEATURE REQUIREMENTS

- Configured GTN (v6.50 or later)
- Valid VNAV data
- GFC 500/600 for autopilot VNAV mode status annunciation

The vertical navigation (VNAV) feature provides vertical profile guidance during the descent phase of flight. Guidance is based on altitude constraints associated with lateral waypoints in the active flight plan.

VNAV Guidance Function

- Presents vertical path guidance along the descent as either a line joining two waypoints with specified altitudes or a desired linear vertical path (i.e., the vertical angle from the specified waypoint or altitude)
- Integrates vertical waypoints into the active flight plan
- Supports both manual and autopilot coupling

VNAV PFD INDICATIONS

VNAV data received from the navigator display as magenta indications on the PFD.



INDICATOR	VNAV DATA
Barometric Altimeter	- Target altitude reference field - Modified selected altitude knob increments
VSI	Required vertical speed indication
VDI	VNAV source and vertical deviation indications¹
AFCs Status Box	VNAV mode annunciation (requires Garmin autopilot)

¹ Systems with BARO synchronization disabled during normal mode: If the Selected Side function is enabled, VDI is based on the barometric altitude displayed on the selected PFD. Applicable to TXi software v3.61 and later.

Airspeed Indicator



The airspeed indicator is configured at installation to meet the requirements of the AFM/POH. It supports ten custom airspeed tape markings with installer configurable labels.

Available units: • knots (default) • kilometers per hour
• statute miles per hour

FEATURE REQUIREMENTS

For Mach Selection capability:

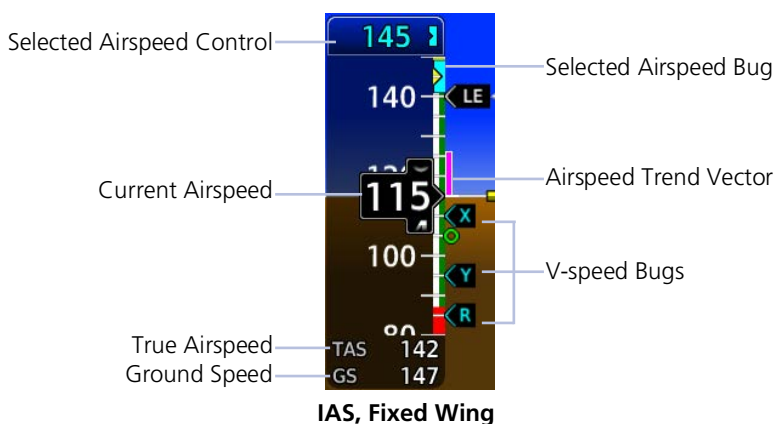
- TXi software v3.61 or later
- Airspeed indicator configured for M_{MO} value
- Mach Selection control option enabled on GDU

For jet aircraft reference speeds, markings, and aural callouts:

- TXi software v3.61 or later

FEATURE LIMITATIONS

- Automatic transition between IAS and Mach airspeed reference units dependent upon installer configuration
- Automatic hiding of the low-speed awareness band during takeoffs available with TXi software v3.50 and later
- Approach speed cue available only for turbine aircraft with GEA 71 EIS and AOA interface (with TXi software v3.61 or later)
- Aural callout availability for jet takeoff speeds dependent upon configuration



Airspeed Indicator Function

- Provides indicated airspeed, true airspeed, and ground speed
- A fixed pointer, with rolling number gauge, indicates current airspeed on a moving tape
- Actual airspeed tape colors and markings vary by installation
- Data fields display true airspeed, ground speed, and, for some configurations, Mach
- As airspeed increases, tape scrolls down

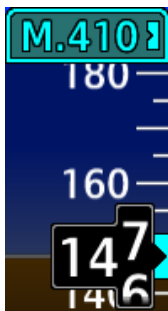
AIRSPEED CONTROLS & INDICATIONS

Selected Airspeed	• Sets airspeed bug on tape if configured • Both the bug and its digital value display in cyan • Automatically synchronizes across all connected TXi PFDs • Manual toggle between IAS and Mach available if configured for Mach selection • Not selectable during EDM selected data lockout (both field and bug turn gray)
Airspeed Trend Vector	• Magenta trend vector at the right of the airspeed tape • 6 second prediction of airspeed based on current acceleration • Absent if airspeed remains constant or if any data necessary to calculate airspeed is not available due to a system failure
Speed Range Strip	• Multi-colored strip at the right of the moving tape • Actual colors and patterns vary according to aircraft type
Selected Airspeed Bug	• Set using the selected airspeed control • Present on the right, inboard side of the tape if configured • Parks at the top or bottom of the tape when selected airspeed is outside the visible range • Parks at the equivalent IAS value when Selected Airspeed is in reference to Mach • Changes from cyan to gray during EDM selected data lockout
Approach Speed Cue¹	• Provides a target for maintaining the correct approach speed regardless of weight or changes in atmospheric conditions • Hollow green circle on the airspeed tape • Indicates approximate airspeed 1.3 times the stall speed for the current weight, G loading, and aircraft configuration • May serve as an approximate indication of V_{REF} or V_{APP} • Movements based on aircraft angle of attack

For required airspeed limitations and markings, consult the aircraft AFM/POH.

¹ Available only for select airframes with TXi software v3.61 or later.

Mach Selection



Selected Mach value displays when the Mach Selection function is active. Airspeed bug indicates the equivalent IAS value on the tape.

Mach Selection and bug setting controls are accessible via the **Selected Airspeed** control.

Toggle Mach Selection on or off by tapping **IAS/Mach** or by pushing the knob. Once active, tap **Mach** to specify an airspeed bug value.

OPTIONAL AUTOMATIC TRANSITION BETWEEN AIRSPEED REFERENCE UNITS

During climb, airspeed reference units automatically change from IAS to Mach once the aircraft attains the configured transition altitude or airspeed. Units change back to IAS once the aircraft reaches the transition altitude or airspeed while on descent. The system determines aircraft climb or descent by the relationship between current and selected altitudes.

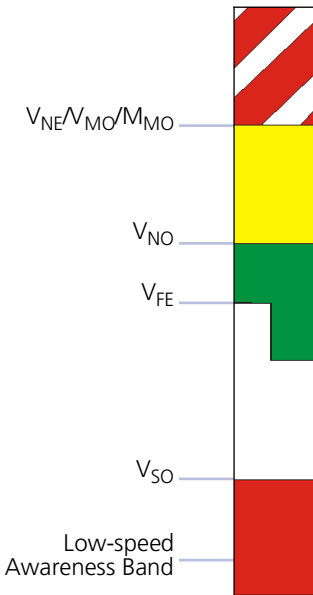
Transition altitudes and airspeeds are configurable. Consult the AFMS to determine feature availability and the configured transition altitudes and airspeeds for your aircraft.

Reference Markings

FIXED WING

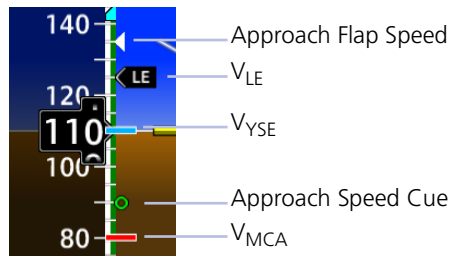
The V_{NE}/V_{MO} barber pole and V_{NO} yellow range are configurable as fixed values or variable values that change with altitude. M_{MO} may be configured when a variable V_{NE}/V_{MO} is configured.

For configurations with multiple variable $V_{NE}/V_{MO}/M_{MO}$ or V_{NO} values, the bottom of the indicating range and alerting parameters extend to the lowest airspeed based on calculations. For fixed-wing installations, exceeding a variable $V_{NE}/V_{MO}/M_{MO}$ value may be configured to create an airspeed exceedance entry.



A white triangle reference marking is configured for airframes that are placarded with approach flap speed limitations.

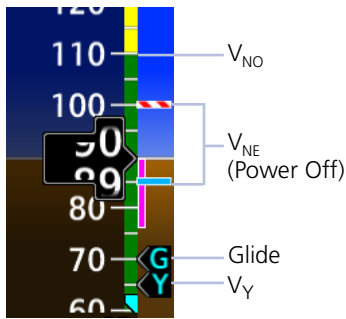
During takeoff, the low-speed awareness band and approach speed cue are hidden until the system determines that aircraft takeoff is complete.¹



For TXi installations supporting AOA, the low-speed awareness band incorporates angle of attack information directly onto the airspeed tape to provide a real-time reference for stall speed. For all other installations, this indication is static.

¹ Automatic hiding of the low-speed awareness band during takeoffs available with TXi software v3.50 and later. Approach speed cue available only for select airframes with TXi software v3.61 or later.

ROTORCRAFT



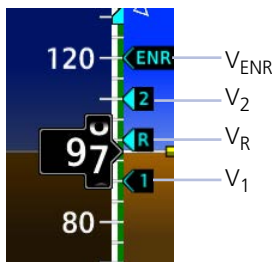
Two types of markings are available for denoting maximum auto rotation speed (V_{NE} power off). This option is configured during installation.

Variable V_{NE} is based on values in the airspeed limitations placard.

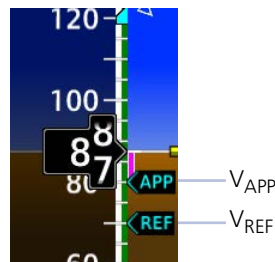
TURBINE AIRCRAFT

Up to six markings are available for denoting takeoff and landing speeds. Takeoff speed markings are automatically disabled when airspeed exceeds the installer configured declutter speed.

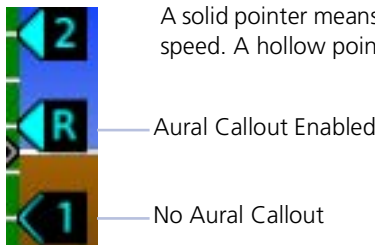
Takeoff



Landing



Takeoff Speeds with Aural Callouts



A solid pointer means that an aural callout is enabled for that takeoff speed. A hollow pointer means the speed has no aural callout.

Aural Callout Enabled

No Aural Callout

OVERSPEED INDICATIONS

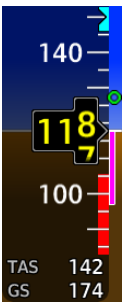


Current airspeed and Mach values turn yellow when the trend vector enters the overspeed range.



Indicator backgrounds turn red when current airspeed exceeds $V_{NE}/V_{MO}/M_{MO}$.

UNDERSPEED INDICATIONS



Current airspeed turns yellow when the trend vector enters the low speed awareness range.



Indicator background turns red when current airspeed remains within the low speed awareness range.

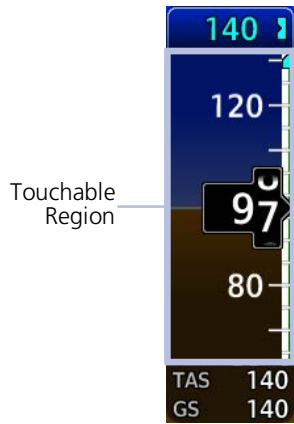
Angle of Attack Data Lost Indication



A red flag indicates when angle of attack data are lost and the system cannot calculate the low speed awareness range.

If configured, the indicator defaults to yellow and/or red bands instead of the low speed awareness band.

Reference Speeds



As a shortcut, the entire airspeed tape is selectable. Tapping anywhere in this region provides direct access to V-speed settings. Tapping this area again, or tapping the underlying display, closes the menu. These controls are also accessible via the **Menu** key.



FIXED WING & ROTORCRAFT

On/off controls reside in the Airspeeds page of the PFD menu.

- Tap **Menu** > **Airspeeds**, or
- Tap anywhere on the airspeed tape

Enable V-speeds by toggling the appropriate on/off keys.

Set or change V-speed reference values by tapping the associated data entry key and entering the value on the provided keypad.

Fixed-wing Reference Speeds:

Glide or V_{REF} V_Y or V_2 V_X or V_1 V_R

Rotorcraft Reference Speeds:

Glide V_Y V_R

V_X , V_Y , V_R and Glide default values are set during the installation process. V-speeds can be set for an individual flight, but will default back to install presets following a power cycle. When active (on), the V-speeds are displayed at their appropriate locations on the right of the airspeed tape.



A positive or negative value indicates the difference between the currently selected V-speed and its default value. Tapping **Restore All Defaults** resets all V-speeds to their default values.

TURBINE AIRCRAFT

On/off controls reside in the Speed Bugs page of the PFD menu.

- Tap **Menu > Speed Bugs**, or
- Tap anywhere on the airspeed tape

For convenience, the Speed Bugs page automatically opens during power up to present V-speed settings for takeoff.

Opening the page while the aircraft is in-air provides V-speed settings for landing.

Enable/disable individual V-speeds by:

- Toggling the appropriate on/off keys, or
- Tapping **All On** to enable all displayed speed settings

Tapping **All Off** disables all displayed V-speed settings.



Set or change V-speed reference values by tapping the associated data entry key and entering the value on the provided keypad.

Takeoff Reference Speeds:

 V_1
 V_R
 V_2
 V_{ENR}^1

Landing Reference Speeds:

 V_{AP}
 V_{ENR}^1
 V_{REF}

Enable Aural Callouts for Takeoff Speeds



Allow audio callouts to alert you when the aircraft exceeds a configured takeoff speed. Tapping the **Aurals** icon enables callouts for all takeoff V-speed(s): V_1 , V_R , V_2

Callouts for individual takeoff speeds may be disabled by your installer.

¹ Depending on airframe, the V_{ENR} takeoff reference speed may be configured with a custom label.

Fast/Slow Indicator

PC-12 AIRCRAFT ONLY

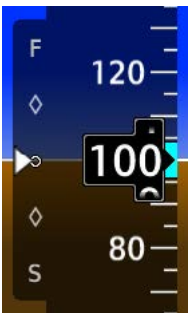
FEATURE LIMITATIONS

Fast/Slow indicator and G-meter vertical tape are mutually exclusive. Only the G-meter digital readout format is available when the Fast/Slow indicator is active.

The Fast/Slow indicator shows when indicated airspeeds are less than 150 KIAS. For details on operation, consult the AFM/POH.



On narrow layouts, the indicator appears along the inside edge of the airspeed tape.



On expanded layouts, it appears to the left of the airspeed tape.

Speed Constraint Indications

FEATURE REQUIREMENTS

- GTN Xi series navigator with software v21.02 or later
- Speed Constraint setup option enabled by pilot on the configured GTN Xi
- VNAV function enabled by installer

Constraint values display in a dedicated field. The location of this field varies depending on PFD layout.



**GDU 700L/1060/1210
(Full Screen View)**



**GDU 700P
GDU 700L (Menu Open)
GDU 1060/1210 (MFD/PFD/EIS)**

White bars indicate constraint type. The position of the airspeed value (above or below a single bar, or between two bars) denotes the required airspeed relative to that constraint.



**At or above
target airspeed**



**At target
airspeed**

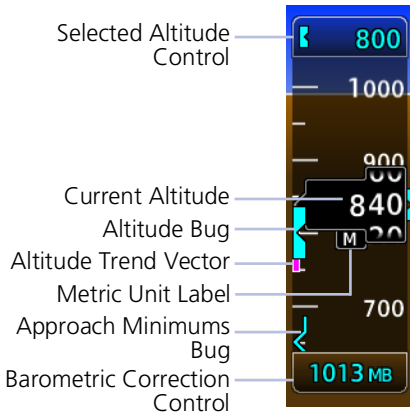


**At or below
target airspeed**

Barometric Altimeter

FEATURE LIMITATIONS

- 20 ft (20 m) digital resolution
- Range: -1,000 to 99,000 ft (-305 to 30,175 m)



Barometric altitude is displayed on a moving tape with a fixed pointer and digital readout.

Units are either feet or meters depending on configuration. A metric unit label displays for metric altitudes.

Altimeter tape limits are configured to match aircraft performance. For more information, consult the AFM/POH.

ALTIMETER CONTROLS & INDICATIONS

Selected Altitude Control

- Sets altitude bug on altimeter when configured
- Bug automatically synchronizes across PFDs
- If barometric minimums is set, selected altitude may be set to same value as barometric minimums
- Determines visual and aural alerts for approaching and deviating from selected altitude.
- Not selectable during EDM selected data lockout (both field and bug turn gray)
- Provides source data for the range to altitude arc display on MFD map (GDU 1060/1210 only)
- Controls altitude preselect mode for various autopilots
- Bug is removed upon system startup or by rotating control knob to a value <-1,000 ft

Altitude Trend Vector

- Magenta trend vector at the left of the altitude tape
- 6 second¹ prediction of altitude based on current vertical speed
- Absent if aircraft altitude remains constant or if data needed for rate calculation is not available due to a system failure

ALTIMETER CONTROLS & INDICATIONS

Barometric Correction Control

- Displays digits in cyan at the bottom of altimeter tape
- Displays digits in yellow when pilot and copilot TXi PFD settings differ by more than 0.03 in Hg
- Flashes momentarily when changing the setting via an external source
- Select units to display in inches of mercury (in Hg), Hectopascals (hPa), or Millibars (Mb)
- Selecting **STD BARO** sets the barometric pressure setting to 29.92 in/1013 Mb

Approach Minimums Bug

- Displays as a colored bracket for selected MDA/DH minimums set on the altimeter tape
- Bug is parked to the top or bottom of altimeter tape when value is outside current visible range
- Cyan bracket denotes BARO or RAD ALT minimums altitude
- Magenta bracket denotes TEMP COMP minimums altitude
- White bracket denotes close proximity above minimums altitude
- Yellow bracket denotes below minimums
- Triggers “Minimums, Minimums” aural alert at MDA/DHs
- Set by tapping either the lower half of the altitude tape or **Menu > Minimums**

Altitude Bug

- Set using the Selected Altitude control
- A portion of the bug displays at the top or bottom of the altitude tape if selected Altitude Bug is out of visible range
- Bug width is 100 units wide, depicting (+/-) 50 unit increments
- Turns gray during EDM selected data lockout

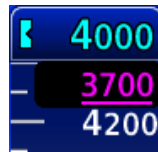
¹ Some airframes are configured for 10 second prediction. Consult the AFMS for aircraft specifics.

TARGET ALTITUDE REFERENCE FIELD

The location of this field varies depending on PFD layout.



**GDU 700L/1060/1210
(Full Screen View)**



**GDU 700P
GDU 700L (Menu Open)
GDU 1060/1210 (MFD/PFD/EIS)**

Magenta bars indicate target altitude type. The position of the target altitude value (above or below a single bar, or between two bars) denotes where the aircraft should be relative to the displayed altitude.



**At or above
target altitude**



**At target
altitude**



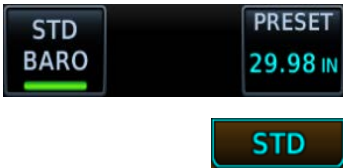
**At or below
target altitude**

Adjusting Barometric Pressure



Controls for adjusting barometric pressure reside in the Barometer control menu.

STD BARO	• Toggles between standard barometric pressure and the pilot specified value
BARO	• Allows entry of selected barometric pressure value • Available only when STD BARO is inactive
PRESET	• Allows entry of destination barometric pressure while the aircraft is above transition level (FL 180) • Preset does not affect altimeter readout • Transitioning from standard to preset sets the altimeter to the preset value • Available only when STD BARO is active



When **STD BARO** is active:

- Barometric Correction control below the altimeter annunciates **STD**
- The barometer preset value may be changed
- Changes to the preset value are displayed in a pop-up field above the Barometric Correction control

SET BAROMETRIC PRESSURE

1. Tap the Barometric Correction control or turn the PFD control knob to select it.
2. Push the inner control knob or tap the **STD BARO** key to select standard barometric pressure or pilot-selected barometric pressure.
3. If **STD BARO** is active, turn the inner knob or tap the **PRESET** key to set the barometer preset value.
4. If **STD BARO** is not active, turn the inner knob or tap the **BARO** key to set the barometer current value.

Adjusting Selected Altitude

FEATURE LIMITATIONS

The MINS SYNC function is available only when the following conditions are met.

- Minimums source is configured as barometric altimeter or temperature compensated barometric altimeter
- **BARO** source key is active
- A valid minimums altitude setting



Adjustments to the selected altitude value cause the alert function to reset.

MINS SYNC

- Synchronizes altimeter bug to the MDA/DH setting

ALT

- Allows entry of a selected altitude value

Adjusting Selected Altitude via Inner Knob



Inner knob functionality is based on the Altitude Knob pilot setting in the setup menu.¹

Coarse/Fine:

- Turning adjusts in increments based on altitude:
 - Below transition altitude: 500 ft increments
 - Above transition altitude: 1,000 ft increments
- Pushing and turning adjusts in 100 ft increments.

Push SYNC:

- Turning adjusts in 100 ft increments.
- Pushing synchronizes to the current aircraft altitude.

Read more about this feature in *Altitude Knob Control Options*.

TXi Installations with GCU 485 Model -06 or -16 (CDI/XFR/BANK)

You may also use the ALT knob on GCU 485 to make incremental adjustments to selected altitude. This knob functions the same as the inner knob on GDU.

¹ Availability dependent upon installer configuration.

SELECTED ALTITUDE ALERTING

Visual and aural indications alert when the aircraft is approaching the selected altitude.

DISTANCE FROM SELECTED ALTITUDE	INDICATION ¹
1,000 ft	• Black text on cyan background • Optional aural tone ²
200 ft	• Flashing cyan text on transparent background • Optional aural tone ²
0 ft Altitude deviates more than ± 200 ft from selected altitude	• Flashing yellow text on transparent background • Aural tone

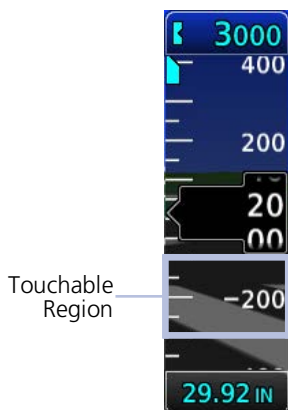
¹ Alert duration: five seconds. ² Aural tone dependent upon configuration.

DISABLE ALTITUDE BUG



Turn inner knob counterclockwise until the selected altitude value is $< -1,000$ ft. Values less than $-1,000$ display as a series of dashes.

Approach Minimums



As a shortcut, the lower half of the altitude tape is selectable. Tapping anywhere in this region provides direct access to the MINS keypad. Tapping this area again, or tapping the underlying display, closes the keypad.

These controls are also accessible via the **Minimums** key in the PFD menu (**Menu > Minimums**).



Set Approach Minimums



1. Open the MINS keypad.
2. Tap **Minimums** and select one of the following source options:
 - Baro • Temp Comp • Rad Alt¹
 - Off
3. Specify the minimums altitude value for the approach.

¹ Applicable only to installations equipped with a radar altimeter.

TEMPERATURE COMPENSATION MINIMUMS



GTNs and TXi displays use only one destination airport temperature for calculating compensated altitudes. Changing the temperature on one of these units automatically recalculates the value across all connected GTNs and GDUs.



WARNING

Always verify minimums (including calculated temperature compensated minimums) with their published values.

FEATURE LIMITATIONS

- The TEMP COMP minimums function is available only when a destination airport is present on the external navigator
- Cycling power to the GDU, or changing the approach on the navigator, clears all minimums data
- Altimeter tape touch functionality available with TXi software v3.61

TEMP COMP Function

- Calculates compensated altitude for the destination airport
- Indicates temperature compensated minimums on altitude tape
- Synchronizes destination temperature between GDU and GTN



Tap **Menu > Minimums > Minimums**, and select **TEMP COMP**.

Selecting this source option when a destination airport temperature is not available automatically opens the TEMP at DEST keypad.

Specify a temperature value to enable the function.

Minimums Bug Indication



Compensating the approach minimums bug determines the location of the minimums reference on the altimeter. Temperature compensation increases the altitude of the minimums bug to correct for cold temperatures. It does not adjust barometric altitude.

When temperature compensation is active, the minimums bug and numeric value turn magenta.

Set Temperature Compensated Minimums



1. Tap **Altitude**. Specify the minimums altitude value.
2. Specify the destination airport's current temperature in the Temp at DEST window that appears.



GDU automatically calculates the compensated value. By default, this field displays the compensated minimums altitude.

Always verify minimums (including calculated temperature compensated minimums) with their published values.

Dest Temp
Key

Compensated
Altitude Value

Loading a different destination airport into the flight plan clears the minimums bug and its associated parameters (whether compensated or not).

To re-enable temperature compensation, reselect TEMP COMP and define all parameters for the new destination as described above.

Resume Temperature Compensation after Airport Data is Lost

If the destination airport is lost (e.g., the pilot clears the flight plan on the navigator):

- Temperature compensation is unavailable
- Minimums altitude source type switches from TEMP COMP to BARO

To resume temperature compensation:

1. Reselect **TEMP COMP** after loading a new destination airport into the navigator.
2. Enter a new minimums altitude and temperature at destination.



Both TEMP COMP and BARO share the minimums altitude function. Switching between these two source types does not alter the specified altitude value.

MDA/DH Alerting

MDA and DH alerts are based on barometric altitude or optional radar altitude. Alerting is inhibited while the aircraft is on the ground. The radar altimeter option is only available in aircraft with a compatible radar altimeter interfaced to the TXi system.

Both radar altitude and barometric minimums settings are lost following a unit power cycle. Controls for setting MDA and DH alerting behavior reside in the Minimums menu.

CONDITION	ALERT INDICATION
Initial MDA/DH	When the MDA/DH is initially set: • Cyan altitude value with bracket pointer displays at the bottom of the altimeter tape • BARO or RAD ALT minimums remain boxed until the MDA/DH altitude is visible on the altimeter tape • Minimums display “RA” when using RAD ALT settings Once MDA/DH altitude is within altitude tape display range, the boxed value is removed and the cyan minimums bug begins tracking with the altimeter tape.
Within 100 ft of MDA/DH	• Bug turns white
Reaching MDA/DH	• Bug turns yellow • RA value turns yellow. • Aural alert: “Minimums, minimums”

As the aircraft ascends, the bug changes color to reflect the current altitude above minimums. Alerting is armed once the aircraft is >150 ft above min altitude.

- White at 50 ft above min
- Cyan at 150 ft above min

RA value turns white once the aircraft is on ground (i.e., system detects weight on wheels; TAS is < 50 kt).

BARO Alerting

FEATURE REQUIREMENTS

- TXi software v3.21 or later

Enable this function to receive altimeter setting prompts when reaching a specified transition altitude, or when inadvertently setting the incorrect selected baro mode (STD BARO vs. non-STD BARO).



When an alert is triggered, the altimeter setting (selected baro) value flashes.



Controls for setting BARO alerting behavior reside in the Altitude Setup menu.

BARO Alert	Toggles the alerting function on or off.
Transition Altitude	Allows entry of a numeric transition altitude.

From the GDU 700() PFD:

Tap **Menu > System > PFD Setup > Altitude Setup**.

From the GDU 1060/1210 PFD:

Tap **Menu > PFD Setup > Altitude Setup**.

BARO alerts are a useful reminder to set the standard baro value, or to input the local baro value.

ENABLE BARO ALERTING

1. Navigate to the Altitude Setup menu.
 - *GDU 700()*: Tap **Menu** > **System** > **PFD Setup** > **Altitude Setup**.
 - *GDU 1060/1210*: Tap **Menu** > **PFD Setup** > **Altitude Setup**.
2. Toggle **BARO Alert** on.
3. Select **Transition Altitude** and enter a transition altitude.

Once enabled, altimeter setting prompts trigger when the barometric altitude is valid and any of the following conditions are true.

Condition 1	Condition 2
• Aircraft climbs through the transition altitude plus 280 ft • STD BARO is inactive	• Aircraft descends through the transition altitude minus 280 ft • STD BARO is active
Condition 3	Condition 4
• Aircraft is 280 ft or lower below the transition altitude • Baro mode is STD BARO	• Aircraft is 280 ft or higher above the transition altitude • Baro mode is other than STD BARO

Alerting is stopped when a change is made to the BARO altimeter setting (i.e., STD BARO is toggled on/off) and either of the following conditions are true.

- Aircraft descends through an altitude 200 ft above the transition altitude
- STD BARO is inactive

Or

- Aircraft ascends through an altitude 200 ft below the transition altitude
- STD BARO is active

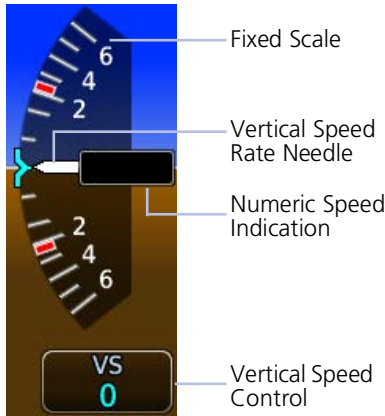
DISABLE BARO ALERTING

Open the Altitude Setup menu and toggle **BARO Alert** off.

VSI

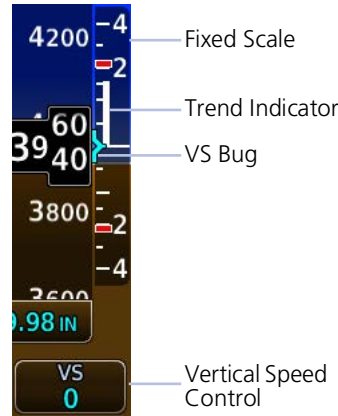
Vertical speed data displays on either a standard or arc style indicator. Format is dependent on unit type (portrait or landscape) and installer configuration.

Arc VSI



Moving needle indicates current vertical speed on a fixed scale.

Standard VSI



Trend indicator replaces needle, pointer, or indications.

AUTOMATIC FORMAT CHANGES

The arc format changes to standard when alternating between full and split views (arc during full PFD, standard during split). This occurs in the following display configurations.

GDU 1060/1210 MFD/PFD/EIS or MFD/PFD (60%/40%)

GDU 700L PFD/menu open

TCAS II RAs

FEATURE REQUIREMENTS

- Installed GTS 8000 or compatible ARINC 429 TCAS II unit
- GDU 1060/1210 with 60% PFD or 700L PFD

The TCAS II uses transponder replies to determine relative altitude, range, and bearing of any air traffic control radar beacon system (ATCRBS) or Mode S equipped aircraft with altitude reporting. Based on the information TCAS II determines the level of advisory. The TCAS II will not issue RAs for ATCRBS aircraft that reply with only Mode A information.



When the TCAS II LRU issues an RA, the VSI displays red no-fly bands to indicate areas to avoid.

A green band indicates the desired climb/descent rate.

NUMERIC SPEED INDICATION

GDU 700L/1060/1210: A numeric field displaying current vertical speed is available on both standard and arc VSIs. The type of field depends on indicator format and, in the case of arc VSI, aircraft type.



Standard VSI: This field replaces the trend indicator and moves up or down as the vertical speed changes. It displays only in expanded layouts.

- GDU 1060/1210: full screen PFD, split MFD/PFD, and split PFD/EIS
- GDU 700L: full screen PFD

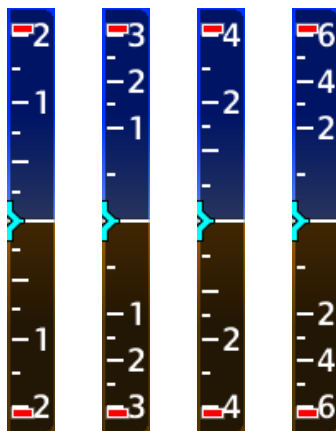


Arc VSI: Field behavior is dependent upon aircraft type:

For rotorcraft, a stationary field appears when vertical speed exceeds +/- 300 fpm (1.5 mps).

For fixed-wing aircraft, the digital value is always present.

VERTICAL SPEED RANGE



Fixed Scale Range Options

Vertical speed range is dependent upon aircraft performance. Both range and units are configured at installation. Available range types include:

- +/- 2,000 FPM
- +/- 3,000 FPM
- +/- 4,000 FPM
- +/- 6,000 FPM
- Metric:
- +/- 9 MPS
- +/- 15 MPS
- +/- 21 MPS
- +/- 30 MPS

REQUIRED VERTICAL SPEED INDICATION



When receiving VNAV data from the navigator, a magenta chevron indicates the rate of descent needed to reach target altitude.

Selected Vertical Speed



Controls for setting the vertical speed bug reside in the VS control menu. Selected VS bug limits are determined by the visible tape range.

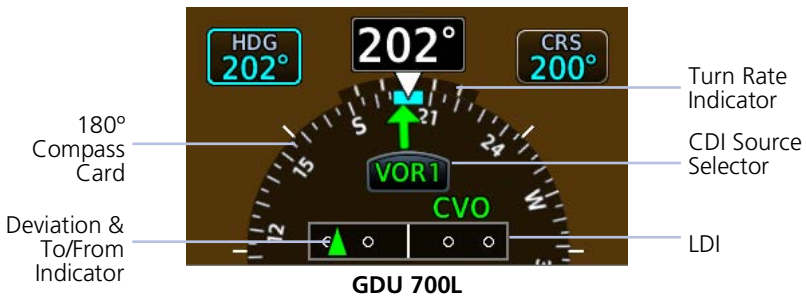
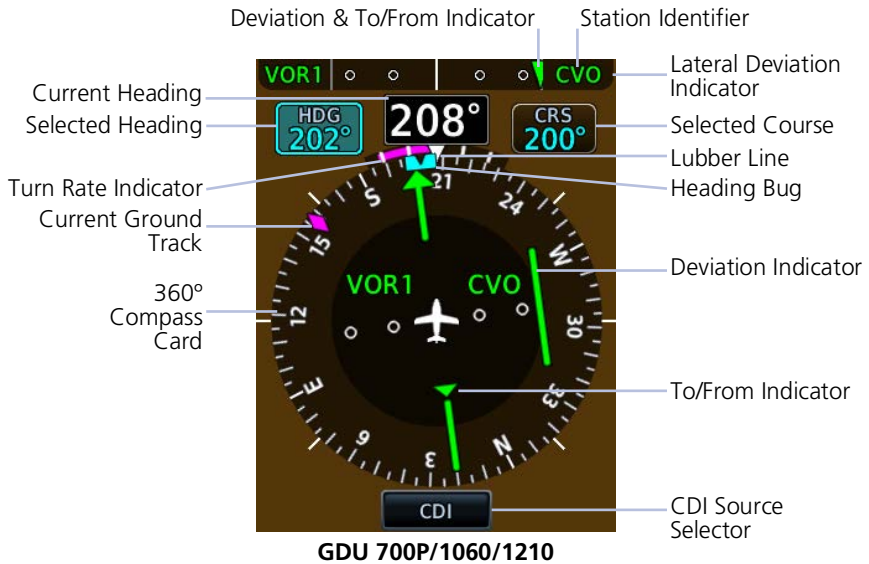
- | | |
|----------------|---|
| VS SYNC | Synchronizes selected VS bug to current vertical speed. |
| VS | Allows entry of a numeric vertical speed. |

Horizontal Situation Indicator

The HSI displays a rotating compass card in a heading-up orientation. Letters indicate the cardinal points with numeric labels every 30°. Major tick marks are placed at 10° intervals and minor tick marks are at 5° intervals. The HSI displays:

- Digital reading of the current heading
- A magenta diamond representing current track
- Turn rate, course deviation, bearing, and navigation source information

HSI layout varies according to display type.



HSI CONTROLS & INDICATIONS	
Turn Rate Indicator	• Lubber line serves as apex for the turn rate indicator, displaying aircraft centerline and direction straight ahead • Tick marks closest to the lubber line depict half-standard rate turns and tick marks furthest away denote a standard rate turn • 6 second prediction of heading based on present turn rate • If aircraft turn rates are greater than 4° per second, an arrowhead appears at the end of the vector and a heading prediction is no longer valid
Lateral Deviation Indicator	• Lateral movement of symbols indicate course deviation • GPS flight phase, navigation source, OBS status, and message annunciator display on deviation indicator borders
Current Ground Track	• Magenta diamond moves on compass card indicating aircraft current track over ground
Current Heading	• Heading displays either true north, user, or magnetic values by selecting NAV Angle units from the System Units menu
To/From Indicator	• Rotates with the course pointer on the standard HSI depicted as an upright or inverted triangle • Displays only when an active navigation source is received (GPS or VOR) • Color scheme is uniform with GPS and VOR/LOC source selections • Upward pointing symbol indicates to, downward pointing symbol indicates from
Station Identifier	• For VOR/LOC sources, the HSI displays the decoded Morse code station identifier on the right side of the CDI lateral deviation indicator • Station Identifiers display at the center of the HSI when HSI map overlays are selected off
Heading Bug	• Cyan bug on the compass card corresponds to the selected heading • Selected heading is adjustable using the PFD knobs or touchscreen controls

Setting the Heading Bug



Controls for setting the heading bug reside in the Selected Heading control menu.

GDU 700L: Set the heading bug using the inner control knob.

HDG SYNC	Synchronizes heading bug to current heading.
HDG¹	Allows entry of a numeric heading.

SELECTED HEADING REFERENCE

FEATURE REQUIREMENTS

- *TXi software v3.21 or later*



A dashed line extends from the selected heading bug to the opposite end of the compass. A pointer indicates direction.

The selected heading reference remains visible for 5 seconds following the last change in selected heading.

Selected Heading Reference

The selected heading reference (or *tail*) provides situational awareness during course intercepts.

¹ Touch key not available on GDU 700L.

OFF SCALE INDICATIONS

GDU 700L ONLY



Heading Bug
Off Scale

It is possible for the heading bug position to be out of view on the 180° compass card. When this occurs:

- Heading bug parks on the side nearest to its actual position (a partial bug is visible at the bottom of the screen)
- Down arrow indicates direction

Changing Heading Modes

FEATURE REQUIREMENTS

- TXi software v3.70 or later
- GRS 7800 for DG free mode
- GSU 75 for HPM mode

FEATURE LIMITATIONS

- Heading mode option available when the AHRS is configured for HPM or DG free mode
- SYNC to GPS Track function available only with valid GPS

Heading mode selection and slewing control options reside in the HDG Mode control menu.



You can access the menu one of two ways:

- Select **HDG Mode** from the PFD Setup menu.

Or

- Tap the current heading field atop the HSI.

Enable HPM or DG free (*free gyro*) mode by toggling the associated on/off key (**DG** or **HPM**).

Controls for slewing the heading are available once HPM/DG free mode is active.

“FREE” annunciates above the current heading value when DG free mode is active. “HPM” annunciates when HPM mode is active.

HDG +/-

Slews the heading in the indicated direction when you tap and hold the key. Depending on configuration, the heading will slew at a slow rate for a predefined number of seconds and then at a faster rate until you release the key.

SYNC to HDG Bug

Slews the heading at a steady rate until reaching the value set in the heading bug.

SYNC to GPS Track¹

Slews the heading at a steady rate until reaching the current GPS Track direction.

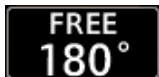
HPM/DG free mode decouples the AHRS indicated heading from the magnetometer, preventing continuous updates of the actual magnetic heading value. In such cases, manual correction of the AHRS heading is required. This option is helpful when no magnetometer information is available (e.g., at extreme latitudes).

¹ Available only with valid GPS.



Current heading value and heading mode annunciation ("HPM" or "FREE") turn yellow when:

- The system recommends enabling HPM/DG free mode, such as when a loss of magnetometer information occurs during normal operation.
- The system recommends disabling HPM/DG free mode because the magnetometer is providing valid information.



These indications are white when no change in mode setting is necessary.

AUTOMATIC HEADING MODE SYNCHRONIZATION



HDG Mode menu options are grayed out while synchronization is active.

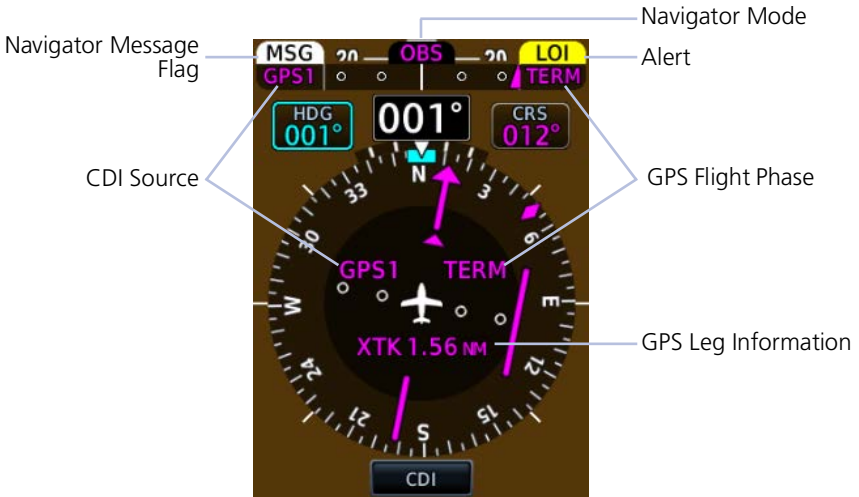
If configured, heading mode settings automatically synchronize between AHRS 1 and AHRS 2. When this occurs, the menu of the AHRS not receiving commands is unavailable until synchronization is complete.

If multiple displays are controlling the same AHRS, the second menu to be opened is unavailable until the original menu closes. This prevents the possibility of multiple simultaneous control inputs.

HSI Annunciations

HSI annunciations display on the HSI compass card and around the lateral deviation indicator. They include:

- GPS flight phase
- CDI source
- VOR/LOC station Morse Code identifier (when VOR is active)
- Navigator modes and messages
- Cross track error (if CDI deflects beyond full scale)



NAVIGATOR ANNUNCIATIONS

WPT

Waypoint Alert

- Blinking text depicts arrival at a waypoint
- Appears within the Lateral Deviation Indicator

SUSP

GPS Waypoint Sequencing Suspended

- Appears above the Lateral Deviation Indicator

LOI

Loss of GPS Integrity Alert

- GPS integrity is insufficient for the current phase of flight
- GPS information is not present or is invalid for navigation use
- Appears above the CDI Lateral Deviation Indicator

MSG

Message Flag

- Blinking text signals a message is queued in the navigator
- Appears above the CDI Lateral Deviation Indicator

OBS

Omnibearing Selector Active

- Appears above the CDI Lateral Deviation Indicator

GPS1

CDI Source

- Displays current CDI source information
- Appears within the CDI Lateral Deviation Indicator/Center of HSI

VOR1

LOC1

GPS FLIGHT PHASE ANNUNCIATIONS

GPS flight phase annunciations appear in the following locations.

- CDI lateral deviation indicator
- Center of the HSI

Phase of flight annunciations are a direct indication of the current CDI behavior for the selected navigation source. Not all annunciations are available for every navigator.

These annunciations typically display in magenta. The color changes to yellow when cautionary conditions exist.

0.3NM	0.3 NM CDI scale
1NM	1.0 nm CDI scale
DPRT	Terminal level with departure procedure as the active navigation
DR	Dead reckoning (CDI not available)
ENR	En route
HDG LEG	Navigator provides heading information only (CDI not available)
LNAV	Lateral Navigation Approach
LNAV+V	Lateral Navigation Approach with advisory vertical guidance provided. LNAV+V uses published LNAV minimums.
L/VNAV	Lateral and Vertical Navigation Approach
LP	Localizer Performance Approach
LP +V	Localizer Performance Approach with advisory vertical guidance provided. LP +V uses published LP minimums.
LPV	Localizer Performance with Vertical guidance approach.
MAPR	Missed Approach
OCN	Oceanic
TERM	Terminal
VISUAL	Visual Approach

Course Pointer Types

Course pointer type varies between single or double line arrows depending on the active CDI source. Arrow tips point in the direction of the selected course or GPS desired track.

ICON	SOURCE
	LOC1, VOR1
	GPS1

ICON	SOURCE
	LOC2, VOR2
	GPS2

COURSE POINTER PREVIEW



A dashed gray arrow indicates the VOR/LOC preview selected course. This pointer displays when the CDI/VDI Preview function is active.

For more about CDI/VDI preview indications, read *CDI/VDI Preview* in this section.

Cross Track Error Indication



The GPS navigator determines CDI scaling. A cross track error displays if the CDI exceeds the full scale deviation value during GPS navigation.

Cross Track Error Annunciation

CDI Source Selection



The **CDI** key toggles between two sources or opens a source selection menu to select between more than two sources.



On GDU 700L: Source selections display on the **CDI** key.

For systems with a GTN that has a GTN CDI key, the CDI source may be switched between GPS and VOR from a PFD or with the GTN CDI key.

CDI SOURCE SYNC

When CDI SRC sync is on or the system is configured to always sync, the CDI source selected on any PFD is automatically selected on all PFDs in the system.

In a system with two or more navigation units available as CDI sources (GPS/VOR 1 and GPS/VOR 2) and CDI SRC sync off, the CDI source can be selected on each PFD independently.

SOURCE SELECTION ALERT

The **CDI** key flashes to alert the pilot when a change in source selection is necessary. Conditions include:

- Localizer or ILS approach is active on the GPS navigator
- Active waypoint is the final approach fix
- Localizer deviation is less than full scale deflection
- Localizer approach course is within 20° of GPS track

Selected Course with VOR/LOC



Selecting a VOR/LOC source activates the Selected Course control. All course indications and annunciations turn green.

Selecting **Direct CRS** centers the VOR CDI needle and sets a direct-to-fix course.

Selected Course with GPS

FEATURE LIMITATIONS

- In multi-PFD systems, preset course functionality is not available when GDU 620 is present

When a GPS source is selected, the Selected Course field displays course values in magenta. This function provides the preset course for VLOC guidance upon switching from GPS to VOR or LOC.



Selecting **VOR(1/2) PRESET** allows the pilot to preset VOR course selected course entries. Access this function by tapping the preset key and entering the value on the numeric keypad or by turning the inner control knob on the PFD.

GPS AUTO displays the automatic course computed by the navigator. There is no touch key available. Selected GPS course cannot be changed during non-OBS GPS operation.

Automatic Source Selection

FEATURE REQUIREMENTS

- Integrated GPS and NAV radio source (GTN 650/750 or GNS 430/530) for VOR/LOC to GPS auto-switch functionality during missed approaches
- GTN 650/750 for GPS to LOC auto-switch functionality during localizer approach

FEATURE LIMITATIONS

- In multi-PFD systems, the missed approach auto-switch function disables when GDU 620 is present

The system automatically changes the navigation source during the following conditions.

CONDITION	CDI RESPONSE
Missed approach procedure activates	Navigation source automatically switches from VOR/LOC to GPS.
Autopilot captures the localizer	Navigation source automatically switches from GPS to LOC. ¹
Localizer approach without autopilot	Navigation source automatically switches from GPS to LOC. ²

Localizer auto-switching is available without an autopilot when:

- the CDI synchronization installer setting on GDU is configured as always on
- the CDI key installer setting on GTN is configured as enabled

In this configuration, the CDI SRC option is absent from the PFD Setup menu's SYNC Control options.

Auto-switch functionality requires the appropriate configuration of GDU and GTN. If the CDI source option is not available on GTN, or if the option is absent from both GTN and GDU, it means that CDI sources can desynchronize. Contact the installer to configure your system to allow localizer auto-switching.

¹ GFC 600 autopilot only.

² GPS to localizer auto-switching is available without autopilot when CDI SRC synchronization function is configured as always on and the CDI key on GTN is enabled. In this configuration, the pilot-selectable CDI SRC option is absent from the SYNC Control pop-up menu.

Auto-Slewing

FEATURE LIMITATIONS

- In multi-PFD systems, auto-slew functionality is not available when GDU 620 is present

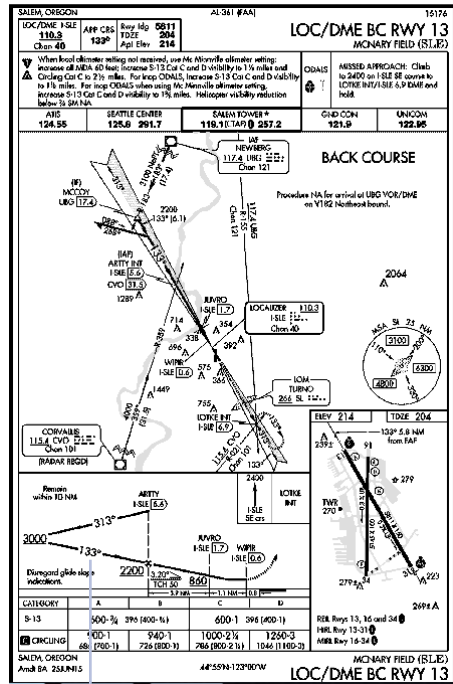
When changing the CDI source from GPS to LOC, the system auto-slews the selected course to the localizer course if:

- A localizer-based approach exists in current flight plan of GPS navigator
- A localizer frequency is active in navigator (LOC, BC, SDF, ILS, LDA)

When a LOC BC approach is activated, the course pointer slews to the LOC front course to provide appropriate back course HSI needle indications.



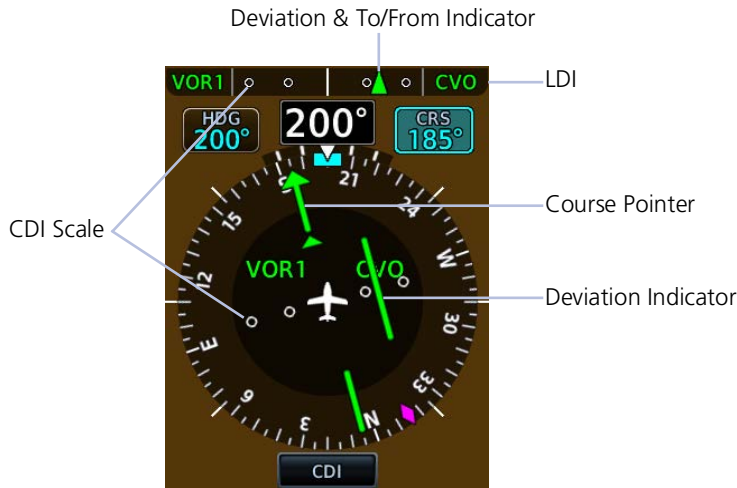
Course Selection and Pointer Auto-Slewed to 313° (front course of the back course approach)



Final Inbound Approach Course 133°

LDI

Deviation indicators move left or right along the CDI scale to portray aircraft position relative to the selected course.



Reverse Sensing Correction

The lateral deviation indicator incorporates an automatic reverse sensing correction. When the aircraft heading is more than 107° left or right from the selected course, the lateral deviation indicator will reverse CDI indications and the to/from pointer.



VDI

The VDI displays to the left of the altitude tape when:

- The aircraft is in an ILS or GPS approach with vertical guidance
- The GDU is receiving VNAV vertical guidance from the GTN

Status annunciations replace VDI indications when data is invalid or not available.



ILS Glideslope

For ILS glideslope indications, the VDI is a green diamond. An ILS glideslope indication appears if the current heading is within 107° of the selected course. This prevents the glideslope from displaying during localizer backcourse approaches.



No GS

Annunciates if a localizer frequency is tuned and there is no glideslope signal.



GPS Glidepath

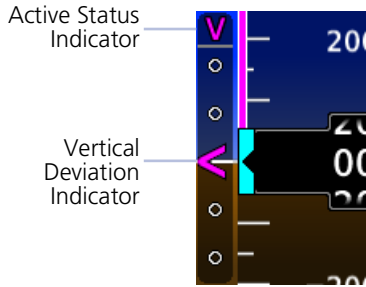
GPS glidepath vertical guidance is similar to glideslope indications. A magenta diamond glidepath indication displays for GPS approaches supporting vertical guidance (LNAV+V, LRVNAV, LPV, LP +V, VISUAL).



No GP

Annunciates if GPS glidepath data becomes invalid while the VDI is displayed.

VNAV Deviation Indications

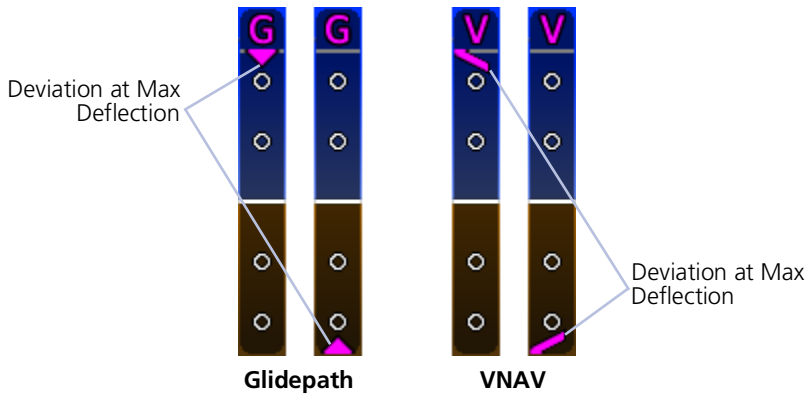


VNAV deviation guidance displays when:

- CDI source is GPS and the navigator is providing valid VNAV deviation data
- CDI source is VOR/LOC, the navigator is providing valid VNAV deviation data, and a VNAV mode is armed or active on the autopilot

These indications are absent when deviation data is invalid.

MAXIMUM DEFLECTION INDICATIONS



CDI/VDI Preview

FEATURE REQUIREMENTS

- A configured GTN



A preview option provides advance indications of approach lateral and/or vertical deviations, allowing the pilot to anticipate changes and make smooth transitions during approach. By default, this function is active.

All preview indications and annunciations are gray.

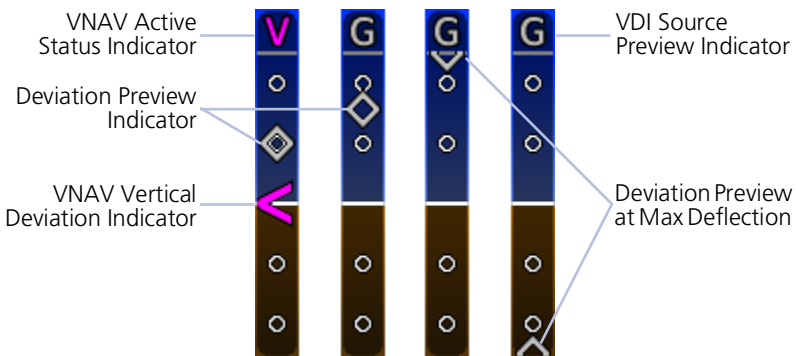
Approach Preview Data

- CDI Source ¹
- CDI VOR/LOC Deviation ²
- HSI Map VOR/LOC Course
- VDI Glidepath/Glideslope
- VOR/LOC Course
- VOR/LOC Morse Code ID

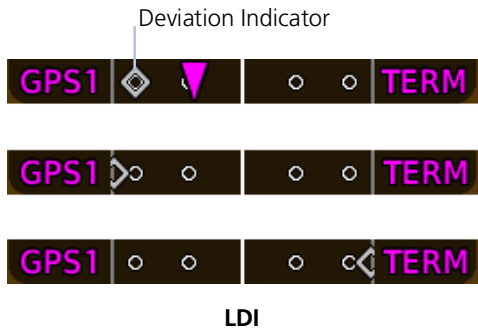
¹ Only if Morse code is not available. ² Includes LOC backcourse deviation.

VDI PREVIEW

For ILS and GPS approaches, the VDI provides a preview of glideslope or glidepath deviation.

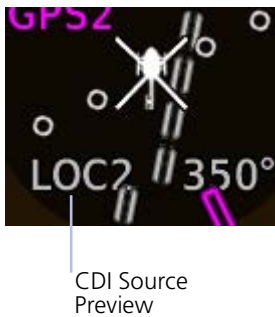
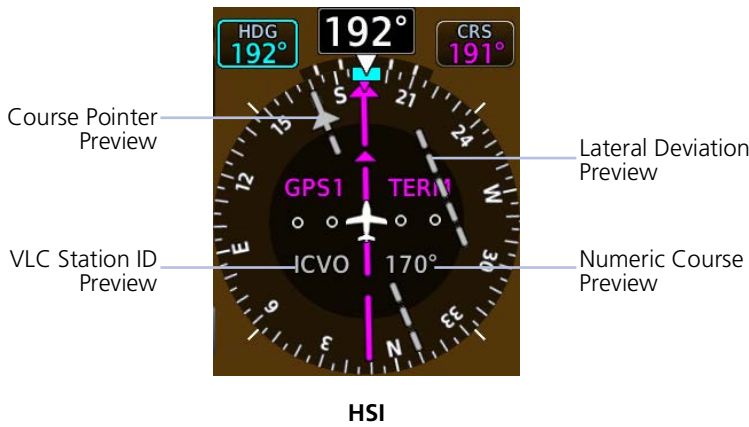


CDI PREVIEW



For VOR or ILS approaches, the CDI provides a preview of VOR or localizer deviation.

GDU 700P/1060/1210:



If a station ID is not available, the CDI source annunciates in its place (e.g., VOR, LOC).

To change the course preview:

1. Open the Course control menu.
2. Select **VOR/LOC PRESET**.
3. Specify a new preset value.

GDU 700L:

Preview data include a course pointer and lateral deviation indicator.



Supplemental Flight Data

Bearing Pointers

FEATURE REQUIREMENTS

To receive TACAN navigation source data, GDU requires the following components.

- TXi software v3.40 or later
- Purchased TACAN feature enabled on GDU
- Supported TACAN receiver

Aircraft equipped with two ADF receivers:

- TXi software v3.61 or later for dual ADF data display

FEATURE LIMITATIONS

Bearing pointers are mutually exclusive with the HSI Map. They do not display when HSI Map is active.

ICON	BEARING
	1
	2

Two pilot-enabled bearing pointers are available for the display of bearing information from a VOR, GPS, ADF, or TACAN source.

Each bearing pointer arrow tip points to the signal source in relation to the current aircraft heading.

Bearing pointers do not override CDI features.

Bearing pointer windows indicate bearing type and navigation source. They display to the left and right of the compass card.

Dual Bearing Pointers, GDU 700P/1060/1210



Single Bearing Pointer, GDU 700L



Bearing Pointer Setup

Bearing Pointer Source Options

- GPS 1/2
- NAV 1/2
- ADF 1/2
- TACAN

Bearing pointer controls reside in the PFD menu. Available source options are dependent upon configuration.



Bearing
Pointer 1
On/Off



Bearing
Pointer 2
On/Off

Toggle a bearing pointer on or off by tapping the corresponding icon.



GDU 700L

Navigation source options are provided for each pointer. GDU retains all pointer selections during a unit power cycle.

GDU 700P/1060/1210:

Controls for each bearing pointer may reside on a dedicated menu pane depending on the number of configured navigation sources.



GDU 700P/1060/1210

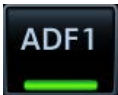
SET BEARING POINTERS

1. Turn off HSI Map if active.
2. Navigate to the Bearing Pointers sub-menu.
 - *GDU 700P/1060/1210*: Tap **Menu** > **HSI Setup** > **Bearing Pointers**.
 - *GDU 700L*: Tap **Menu** > **Bearing Pointers**.
3. Toggle bearing pointer(s) on.
4. Select a bearing navigation source for the active pointer(s).

Selecting a bearing pointer does not necessarily make it visible. Bearing pointers are absent when:

- GDU is not receiving valid data from the selected navigation source
- NAV radio is not receiving tuned VOR station
- NAV radio is tuned to a Localizer frequency
- GPS is the navigation source and active waypoint is not selected
- ADF is selected and the signal is not received
- TACAN is selected and the signal is not received
- HSI Map function is on

ADF Display



Selecting **ADF** sets the bearing navigation source for the associated pointer to the externally tuned ADF.

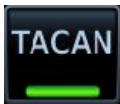
Dual ADF Configuration



Bearing pointers point to the indicated station when active.

Bearing pointer windows show the ADF source(s) and direction to station.

TACAN Display

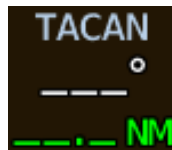


Selecting **TACAN** sets the bearing navigation source for the associated pointer to the tactical air navigation (TACAN) system.

An information window provides the bearing and distance to the station. The location of this window varies depending on unit type. On GDU 700P/1060/1210, it displays below the Vertical Speed control. On GDU 700L, it displays to the left of the airspeed tape.



This display overrides the DME information window if active.



Dashes indicate when a TACAN signal is invalid.

GPS NAV Status Field

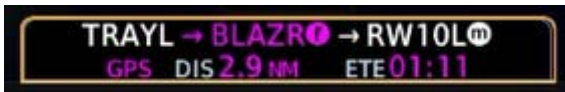
FEATURE LIMITATIONS

From and next waypoints may be removed when the number of characters in a full route exceeds the allowable space.

A GPS NAV status field displays at the bottom of the PFD when the GPS navigator associated with the selected CDI source has an active valid leg. Information is based on calculations by the GPS navigator. Annunciations and window width vary according to GDU type.

GDU 700P/1060/1210:

- From, to, and next waypoints
- Leg and waypoint types
- Identifier, distance, and ETE of active waypoint



GDU 700L:

- Active waypoint identifier and distance
- GPS message status



ACTIVE LEG GPS NAVIGATION STATUS FIELD SYMBOLS

	IAF		Arc Left
	FAF		Holding Pattern (Right Turns)
	MAP		Holding Pattern (Left Turns)
	MAHP		Right Arrow
	Parallel Track		Direct To
	Arc Right		

Relative Wind Data

Wind Data

- Wind direction and speed
- Headwind (H)
- Tailwind (T)
- Crosswind (X)

The system computes current wind conditions during flight and displays the data in a dedicated field.

Status annunciations replace wind data when conditions are calm or the configured data source is invalid or not available. Wind data displays when ground speed is 30 knots and higher.

A white arrow indicates wind direction relative to the aircraft's heading. The numeric display of the wind direction is dependent on the NAV Angle system unit setting.



WIND STATUS ANNUNCIATIONS

**Wind
Calm**

Calculated winds are less than or equal to 2.5 kts.

**No Wind
Data**

GPS track or TAS data are invalid or unavailable.

User Mag Var Display

A dedicated field displays the pilot-specified magnetic variation (*mag var*) value when the NAV Angle unit setting is User.



Magnetic variation displays as a numeric value in degrees E or W.



The field is highlighted in cyan when the value is updated, either by way of the pilot setting on GDU or crossfilling from GTN Xi.¹

The location of this field varies depending on unit type. On GDU 700P/1060/1210, it displays below the relative wind data window. On GDU 700L, it displays to the right of the window.



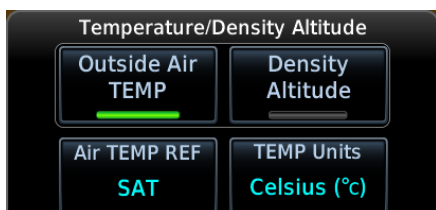
GDU 700L



GDU 700P/1060/1210

¹ TXi software v3.70 and later: NAV Angle unit settings synchronize with GTN Xi v20.43 and later.

Temp/DALT Display



A dedicated field displays outside air temperature or density altitude based on pilot selection. These reference values are mutually exclusive. Selecting one automatically turns off the other.

Outside Air TEMP



Displays the pilot selected air temperature reference.

Units are available in °C or °F.

Options include:

- Static Air Temperature
- Total Air Temperature
- Temperature difference from ISA

Density Altitude



Displays pressure altitude corrected for nonstandard temperature.

Units display in increments of ten, and are in feet or meters depending on altitude tape configuration.

DME Display

FEATURE REQUIREMENTS

- A supported DME transceiver

For Dual DME source tuning:

- Second DME transceiver and TXi software v3.61 or later

OR

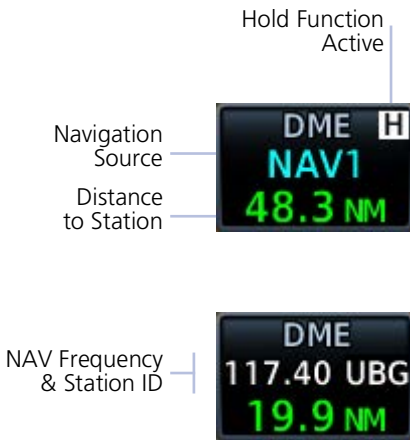
- GDM 450R/4500 and TXi software v3.85 or later

FEATURE LIMITATIONS

Enabling the TACAN bearing pointer source option replaces the DME information window with a TACAN display, rendering the DME control menu inaccessible. To view DME data and access the control menu, disable the TACAN source option(s) in the Bearing Pointers menu.

Hold capability and display of tuned NAV frequency and station ID dependent upon the configured DME.

When interfaced with a DME receiver, a selectable DME information window is available for display on the PFD.



Information includes:

- DME source (NAV1, NAV2, EXT)
- HOLD function active status ("H")
- Tuned NAV frequency and station ID
- Distance to station (GPS or DME)

Depending on the configured DME, the tuned NAV frequency and station ID may display in place of the DME source.

DUAL DME INSTALLATIONS



For aircraft equipped with two DME units, TXi provides a selectable DME information display and dual bearing pointer windows, allowing you to view the distance to two NAVAIDs simultaneously. TXi accomplishes this by displaying one NAV source in the DME information window and the other in the bearing pointer window.



Bearing Pointer Window

DME Control Status Indications

COLOR DEFINITIONS ¹	
Green	Information provided directly from the DME.
White	Information provided by database or non-DME source.
Cyan	Information not available. Control source is external.

¹ TXi software earlier than v3.85: DME distance, frequency, and station identifier appear in green regardless of the control source.

The color of information indicates the DME control source.

DME INFORMATION WINDOW



DME/DME1/DME2

In dual-DME installations, the number indicates which channel (DME 1 or DME 2) is displaying information.

- “DME1” is associated with NAV1
- “DME2” is associated with NAV2



Frequency

- White means the frequency is repeated from the NAV radio and may not reflect the actual DME frequency.
- Green means the frequency is received/available via the DME.¹



Station Identifier

For GDM 450R/4500 and Collins DME units:

- Green means the station identifier has been confirmed by the DME.
- Dashes indicate when the station identifier is unconfirmed or unavailable.



For all other DME units:

The station identifier received from the NAV radio is repeated (in white) regardless of whether it is looked up in the database or confirmed by radio.

Always verify DME-displayed identifiers independently to ensure accuracy.



Hold Status

For DME units that support tuning hold controls via the GDU, the HOLD function active status indication ("H") flashes white when the pilot activates the hold function for the selected DME source.

For non-GDM installations:

- Frequency stops updating and turns white. The system assumes that the NAV frequency at the time the hold function was selected is still the frequency on the DME.
- Identifier flashes white indicating that it may no longer match.



EXT/NAV1/NAV2

"EXT" (external) replaces frequency and identifier when:

- TXi is no longer the control source (i.e., DME control is via the external panel-mount DME unit).

Or

- The NAV radio source is unavailable.



"NAV1" or "NAV2" may annunciate depending on the selected tuning source/DME channel. In these cases, the information window is display only.

Distance to Station

- Valid distance-to-station values appear in green. This information is always provided directly from the DME unit.
- Dashes indicate when the DME signal is invalid.



DME Fault

A yellow "X" displays over the information window if one of the following occurs.

- Communication with the DME unit is lost
- The GDM unit reports a fault



¹ Requires GDM 450R/4500 and TXi software v3.85 or later.

BEARING POINTER WINDOWS



These windows display DME information according to the data source.

Station ID

Distance to Station

DME DISPLAY ELEMENTS

Station Identifier	- Green means the NAV radio source is valid - White means the data sourcing is via database
Distance to Station	Green means the distance-to-station value is valid. For this to be true, the following conditions must be met. - DME is not in hold - DME identifier matches the NAV identifier (if available) - At least one DME is co-located with the NAVAID. If not, the system defaults to GPS Magenta means the data sourcing is via GPS.

Read more about bearing pointer window display elements in the following sample scenarios.

Dual DME Configuration



Green indicates that the distance-to-station value is received from the DME tuning source.



Magenta indicates that the distance-to-station value is derived from the GPS database.



Bearing pointer window is empty when NAV information is absent.



DME Signal Valid

- Station ID from NAV source matches ID in database
- NAV1 and NAV 2 distance-to-station is computed from DME tuning source

Dual Control Sources

- Station ID from NAV source matches ID in database
- NAV1 distance-to-station is from DME tuning source
- NAV2 distance-to-station is computed from GPS and the navigation database

NAV1 Distance Invalid

- NAV1 information is absent
- NAV2 station ID does not match database; distance-to-station is computed from GPS and the navigation database

DME Unavailable

- Station ID from NAV source matches ID in database
- NAV1 and NAV 2 distance-to-station is computed from GPS and the navigation database

DME Setup



Selecting **DME Inset** displays the DME information window on the PFD. This option resides in the PFD Setup menu.

GDU 700(): **Menu > System > PFD Setup > DME Inset**

GDU 1060/1210: **Menu > PFD Setup > DME Inset**

The information window is selectable. Tapping it opens a menu of available tuning options.



Selectable DME Information Window

DME Control Menu
(NAV1 & Hold Functions Active)

DME TUNING OPTIONS

AUTO

- Enables automatic tuning using the selected CDI VHF source as the DME tuning source¹
- Allows independent DME selections on each configured PFD

NAV(1/2)

- Allows manual selection of the DME tuning source
- Options: NAV1 or NAV2

Hold

- Activates/deactivates the tuning hold function for the selected DME source (NAV1 or NAV2)
- Switching between NAV1 and NAV2 automatically cancels the function
- Not all installations have the DME HOLD function

¹ Available only with TXi software v3.61 and later.

Single GDM Installations Configured for Single-channel Audio¹

The DME automatic tuning option is available when the following conditions are met.

- CDI synchronization is enabled
- DME selection is synced between all configured TXi PFDs
- The DME channel is selected for audio output on the GDM unit

¹ Available only with TXi software v3.85 and later.

Marker Beacon Symbols



Inner Marker
Symbol

Marker beacon symbols display to the left of the altitude tape when interfaced to a marker beacon receiver.

All marker beacon light and audio tests are performed through controls on the associated audio panel.

SYMBOL	MARKER TYPE
I	Inner
M	Middle
O	Outer

Radar Altitude

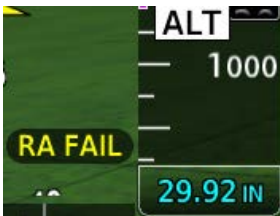
When interfaced with a radar altimeter, GDU indicates radar height to the left of the altitude tape. An altitude field automatically appears when the aircraft is at or below 2,500 ft AGL or within operating range of the installed RA unit (generally 2,500 ft AGL).

Brown and white hash marks depict a ground awareness band on the altitude tape. When the radar altimeter readout is 0 ft, the ground band will level with the altitude pointer.



As the aircraft climbs, the ground band moves down the tape in equal amounts to the indicated RAD ALT value.

Ground Band



If the radar altimeter fails, the message "RA FAIL" annunciates in place of the current radar value.

Audio Alert (Rotorcraft Only)

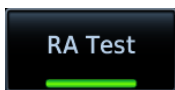


Depending on configuration, an audio alert is active when the radar altimeter indicates the aircraft is descending below 100 ft.

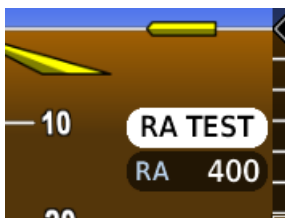
RA Test

FEATURE LIMITATIONS

- Self-test functionality is not available on all radar altimeters
- Sequence duration is approximately 12-20 seconds
- For test values and unit limitations, consult the appropriate third-party radar altimeter documentation



A test function verifies communication between the GDU and the radar altimeter.



During the test sequence:

- Ground awareness band displays on the altimeter tape
- RA readout field displays a test altitude value (e.g., 50 ft AGL)
- "RA TEST" annunciates to the right of the compass card

To suspend the test, deselect **RA Test** or exit the Test menu.

AGL Display

FEATURE LIMITATIONS

- GPS height above terrain value not available for display above 2,500 ft AGL

For systems without a radar altimeter, GDU allows you to display GPS height above terrain to the left of the altitude tape. Units are either feet or meters depending on configuration.

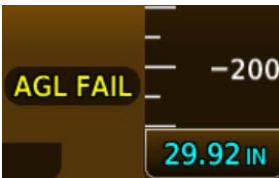


Altitude Field

To toggle the altitude field on or off, open the PFD Setup menu and tap **AGL Field**.



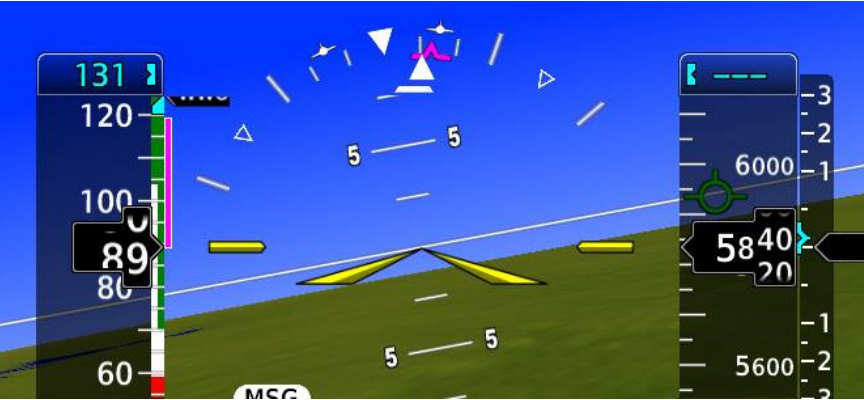
The altitude value is absent when the aircraft exceeds 2,500 ft AGL. It returns once the aircraft is at or below 2,500 ft AGL.



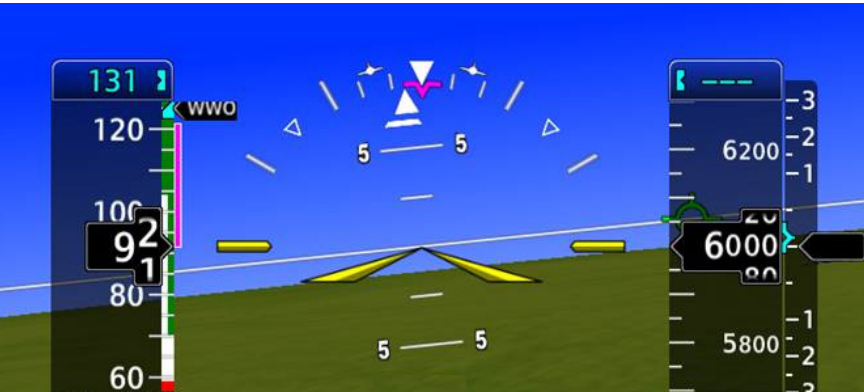
If GPS fails, or if terrain elevation is unavailable or invalid, the message "AGL FAIL" annunciates in place of the current GPS altitude value.

GPS Roll Indicator

A magenta GPS Roll Indicator adjacent to the roll pointer provides roll steering guidance during GPS navigation. The GPS navigator's roll steering command is used to drive the indicator whenever valid AHRS roll, GPS deviation, and GPS roll steering values are available and the flight director is inactive.



GPS Roll Indicator, Ground Pointer



GPS Roll Indicator, Sky Pointer

G-meter

FEATURE REQUIREMENTS

- TXi software v3.21 or later

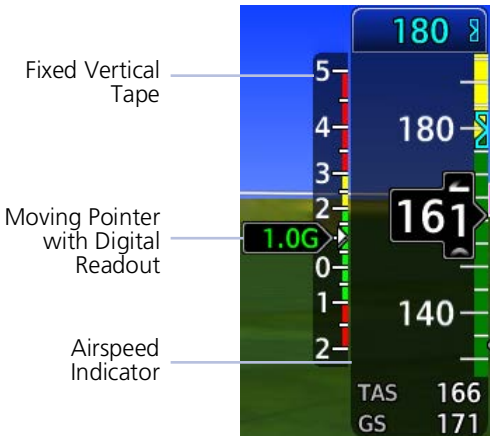
FEATURE LIMITATIONS

Selectable instrument format options available with TXi software v3.40 and later.

PC-12 aircraft: G-meter vertical tape and Fast/Slow indicator are mutually exclusive. Only the G-meter digital readout format is available when the Fast/Slow indicator is active.

An optional G-meter indicates acceleration (G-force) along the aircraft's vertical axis. This information can be valuable during turbulence or aerobatic maneuvers.

Instrument format options are dependent upon screen layout. In expanded layouts, the G-meter may be displayed as a fixed vertical tape to the left of the airspeed indicator.



Graduations, markings, and white markers show the attained extremes for the current flight.

A moving pointer and digital readout display the two digit g-force value in the color of the currently active marking.

Markings are installer configurable.



In spatially constrained layouts, the digital readout is below the airspeed tape.

Minimum 0.9G
Maximum 1.1G

Attained minimum and maximum acceleration values are viewable in the G-Meter Setup menu.

From the GDU 700() PFD:

Tap **Menu** > **System** > **PFD Setup** > **G-Meter**.

From the GDU 1060/1210 PFD:

Tap **Menu** > **PFD Setup** > **G-Meter**.

G-METER SETUP OPTIONS

G-Meter Setup Menu

- Display G-Meter
- Style
 - Select instrument format^{1, 2}
- Reset Min/Max

From the setup menu you may:

- *Display G-meter:* Tap **Display G-Meter**.
- *Change instrument format:* Tap **Style**. Options include Gauge and Digital.^{1, 2}
- *Reset attained minimum and maximum acceleration values:* Tap **Reset Min/Max**.

During a power cycle, GDU retains the display setting, but not the indicated minimum and maximum acceleration values.

¹ Available with TXi software v3.40 and later. ² Instrument format options dependent upon screen layout.

Clock/Timer



A generic timer counts up in an H:MM:SS format. There are no aural alerts or pilot advisories associated with this feature. Turning the clock/timer function off removes the control key from the instrument display.

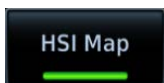
Mode	Toggles between Clock and Timer.
Start/Stop	Pauses timer.
Reset	Resets timer to zero. Count resumes when the timer is reset during operation.

INTENTIONALLY LEFT BLANK

4 Advanced Features

HSI MAP	4-2
MAP DISPLAY, GDU 700L	4-7
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AUTOMATIC FLIGHT CONTROL SYSTEM	4-22
AEROBATIC FLIGHT	4-41
CARBON MONOXIDE ALERTS	4-42
STABILIZED APPROACH ALERTS	4-45

HSI Map



Tapping **HSI Map** places a map within the HSI. This key resides in the PFD menu.

FEATURE REQUIREMENTS

- Valid GPS position and magnetic heading
- FIS-B, Iridium transceiver, or SiriusXM receiver (weather overlays only)
- Active weather subscription (Connex and SiriusXM only)
- WX-500 or WX-1000E receiver (Stormscope overlay only)
- Valid VNAV data (ATK waypoints)

FEATURE LIMITATIONS

- GDU 700P/1060/1210 only

HSI MAP INTERACTIONS

- Map interactions are zoom only. Pan functionality is not available
- Overlays declutter at a predefined map range for specified data depictions

WHEN HSI MAP IS ACTIVE

- Standard HSI view expands to incorporate a basemap and selected overlay
- Map overlays replace standard HSI indications
- HSI display enlarges to a 270° compass card view to accommodate the map
- CDI and annunciations are removed
- CDI indications appear only on the LDI
- Status icons indicate active map overlays
- Bearing pointers do not display
- HSI reverts to a standard depiction in the absence of a valid magnetic heading or GPS source

Data depicted when HSI Map is active

- | | |
|-------------------------|-------------------------------------|
| • GPS flight plan | • Airspace boundaries and altitudes |
| • TFRs | • Runway and SafeTaxi detail |
| • Range to altitude arc | • Terrain alerts, if configured |



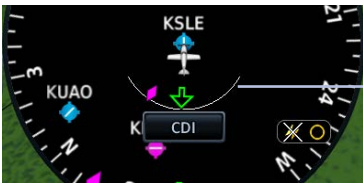
Standard HSI



HSI Map

HSI ARC

When a heading bug, track indicator, course preview pointer, or course pointer moves to a position on the HSI circle behind or below other PFD controls, indicators depict on a miniature arc, representative of the hidden portion of the HSI circle.



HSI Arc

CDI



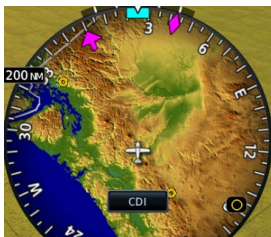
GPS NAV Status Field

LDI indications include:

- CDI source
- Course deviation
- GPS phase of flight
- Course deviation preview
- Navigator message annunciations
- To/from
- VLOC station identifier
- Waypoint arrival annunciation

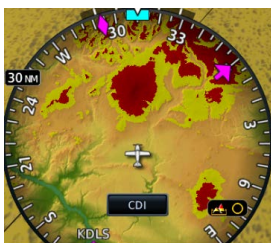
HSI Map Overlays

TOPO



- Topographic map depicting terrain elevation using a color scale similar to Sectional Charts

TERRAIN



- Terrain map depicting terrain elevation relative to aircraft altitude
- Uses the same color scale as the dedicated Terrain page
- NEXRAD weather overlay is disabled to prevent confusion of depicting radar and terrain data with similar colors

OBSTACLES AND WIRES



- Depicts obstacle and wire elevations relative to aircraft altitude
- Uses the same color scale as the dedicated Terrain page
- Data removed when viewing range is >5 nm

TRAFFIC



- Displays traffic using a NORMAL altitude filter
- Data removed when viewing range is >20 nm

NEXRAD/PRECIP & TFRS



- Overlays NEXRAD/PRECIP datalink weather from ADS-B⁵, SiriusXM, or Garmin Connex^{1, 2} as selected in the HSI Setup menu
- Precipitation intensity displays in accordance with NEXRAD/PRECIP legend viewable on the dedicated weather page
- Weather and TFR product age display on HSI Setup Menu

STORMSCOPE



- Overlays Stormscope lightning data on map³
- Cell or strike mode selections available with WX-500

LIGHTNING



- Depicts SiriusXM, FIS-B, or Connex lightning strike information on the basemap^{1, 4}

¹ Requires aircraft to be equipped with appropriate datalink receiver and an active datalink weather subscription service, if applicable. FIS-B is not a subscription based service.

² NEXRAD/PRECIP and terrain overlay functions are mutually exclusive. Enabling one automatically disables the other.

³ Requires aircraft to be equipped with compatible WX-series Stormscope system.

⁴ Stormscope and SXM lightning overlay functions are mutually exclusive. When one is installed and selected in the PFD menu, the other is not available.

⁵ When displaying FIS-B NEXRAD data, the overlay is set to display the Regional product.

Overlay Status Icons

Icons represent the presence of overlay data. For real time products (e.g., TFRs, NEXRAD/PRECIP), these icons indicate the recency of data.

The absence of an overlay icon means one of two possible conditions:

1. Overlay is not present at the current zoom level.
2. Overlay control is off.



Obstacles overlay data is active on the display.



Power line overlay data is active on the display.



TFR overlay is active.



Terrain overlay is active on the display.



Traffic system is active and displaying information.



NEXRAD or PRECIP overlay is active and displaying received information.



Lightning overlay is active and displaying SiriusXM, FIS-B, or Connex lightning information.



Stormscope overlay is active and displaying lightning strikes.

FAULT INDICATION



A fault icon means overlay data is not available, not received, or has failed (where relevant).

OLD DATA



An old data icon means overlay data is not current but remains displayed.

Map Display, GDU 700L



ROTORCRAFT ONLY



Tapping **Map** overlays a full screen map on the PFD.

FEATURE LIMITATIONS

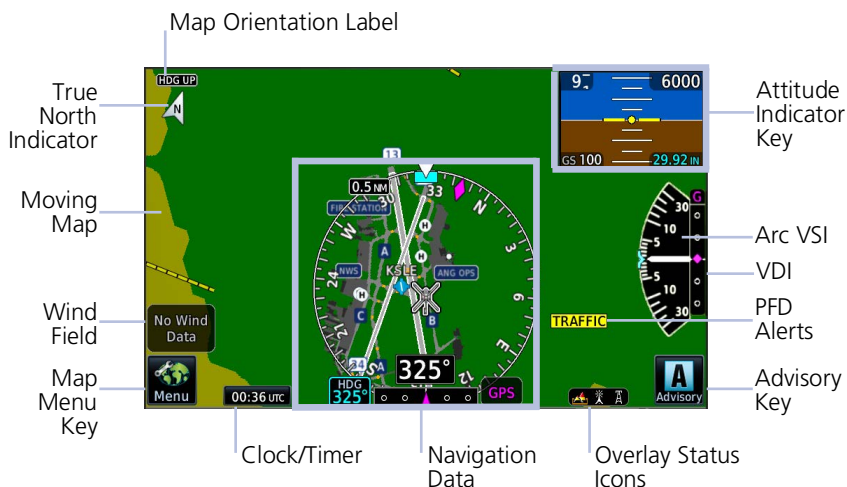
- *GDU 700L only*
- *Map display feature must be configured as available*

MAP INTERACTIONS

- Map interactions are single-finger zoom controlled
- Pan functionality is not available
- Overlays declutter at a predefined map range

WHEN MAP IS ACTIVE

- VFR style moving map with 360° compass card replaces standard PFD layout
- Road and city information display in support of VFR operations
- Airspeed, altitude, BARO correction, and ground speed values display within a selectable attitude indicator key
- Display automatically returns to full screen PFD in the event of extreme attitudes
- PFD Menu key changes to accommodate only the map display



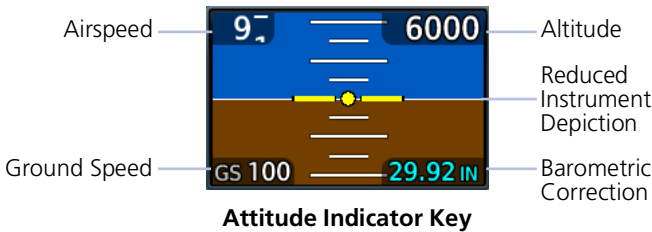
GDU 700L PFD Map Features

Data depicted when PFD Map is active

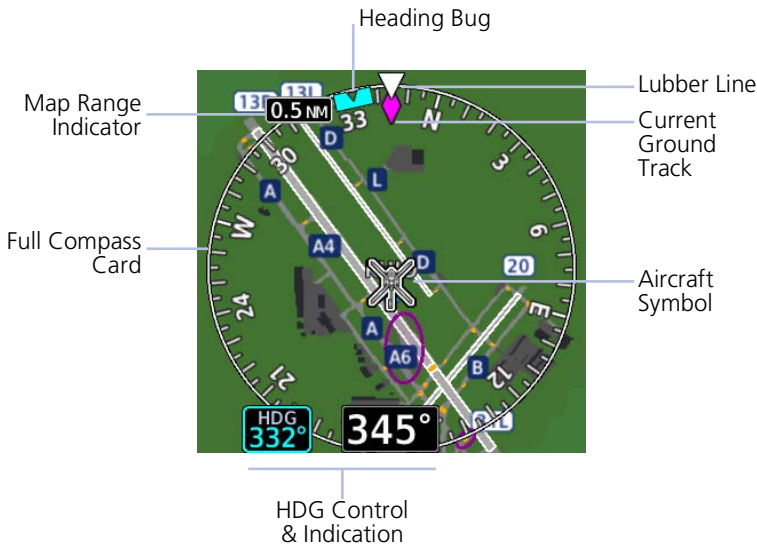
- Full compass card view
- Lubber line and heading bug
- Current ground track
- Selected Heading control and current heading field
- Selected BARO control
- Reduced attitude display
- Arc VSI (if configured)
- Wind field
- Clock/timer
- Autopilot annunciations
- Active fix and distance to active fix (if available from the navigator)

ATTITUDE INDICATOR KEY

To return to full screen PFD, tap the attitude indicator key. This key displays a reduced attitude depiction and instrument data fields. It resides in the upper right corner of the display when full screen map is active. Data fields are display only.



360° COMPASS CARD



GDU 700L Map Overlays



Tapping **Menu** while the map is active opens an overlay menu. Overlay selections display in a dedicated menu.

Overlay Selections

- TOPO
- Terrain
- Obstacles
- Wires
- Traffic
- Weather Datalink
 - SiriusXM
 - FIS-B
 - Connex
 - NEXRAD/PRECIP
 - Datalink Lightning
- Stormscope
 - Off
 - Strike Mode
 - Cell Mode

Overlay Controls



Control keys enable the specified overlay function only and do not activate interfaced equipment. Overlay keys remain active even in the absence of required data.

GDU 700L MAP OVERLAYS	
TOPO	• Overlays topographical data and ground elevation scale • Depictions are similar to a VFR sectional
Terrain	• Overlays terrain map data • Color shading depicts terrain elevation relative to the aircraft's altitude
Obstacles & Wires	• Depicts obstacle and wire elevations relative to aircraft altitude • Uses the same color scale as the dedicated terrain page • Data removed when viewing range is >5 nm
Traffic	• Overlays traffic information • Filter selection on Traffic page determines altitude range • Optional
NEXRAD/PRECIP & TFRs	• Overlays NEXRAD/PRECIP datalink weather from ADS-B, SiriusXM, or Garmin Connex as selected in the Overlay menu • Precipitation intensity displays in accordance with NEXRAD/PRECIP legend viewable on the dedicated weather page • Weather and TFR product age display
Stormscope	• Overlays Stormscope strike or cell data • Data is removed after 4 minutes

Overlay Status Icons

Icons indicate which overlays are present at the current map range. A crossed out icon means the overlay is active, but data is unavailable due to a failure, test, or standby condition (where relevant).

The absence of an overlay icon means one of two possible conditions:

- 1. Overlay not present at the current detail level or zoom setting.
- 2. Overlay control is off.

	Obstacle		Power Line
	Stormscope		Traffic
	Terrain		Data not available

SVT



NOTE

Synthetic terrain aids pilot awareness of terrain and obstacles in front of the aircraft. It does not replace primary flight instruments.

FEATURE REQUIREMENTS

- Valid attitude and magnetic heading
- Valid 3D GPS position
- Valid terrain and obstacle databases
- Feature enablement to continue service beyond the free trial period

FEATURE LIMITATIONS

- Loss of any required data, including temporary loss of GPS, causes SVT to be removed until data is restored
- SVT terrain alerting functionality is not compliant with TSO-C151c
- Aural and visual alerting functions are in accordance with appendix H.2 of AC 20-167A

SVT depicts only terrain contours and obstacles from the associated databases. Optional depictions (e.g., airport signs, aircraft heading) are dependent upon pilot selection. Power line depictions are available for rotorcraft only.

SVT Setup



SVT setup selections are accessible via the **Terrain/SVT** key.

Synthetic Terrain	Activate SVT function.
Settings	Activate horizon heading markings and airport signs.
Terrain Inhibit	Deactivate terrain aural and visual alerts.

SVT Features



NOTE

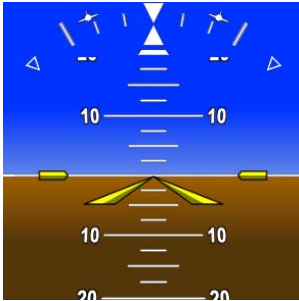
GDUs with PFD functionality have a 10-hour free trial of SVT. A timer in the Terrain/SVT menu indicates current trial status. For details on how to enable SVT once the trial expires, read G500(H) TXi (H)SVT Enablement in this section.

SVT presents a forward looking view of the terrain, obstacles, and traffic surrounding the aircraft. This optional feature offers a visual enhancement to the PFD.

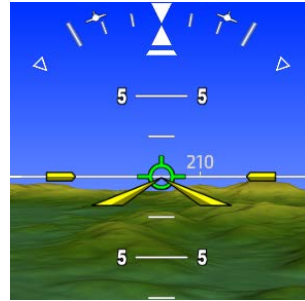
SVT Features	
PFD	MFD
• Terrain Alerting • Obstacle Alerting • SVT Alert Inhibit • Flight Path Marker • Zero Pitch Line • Arc-second Grid Lines • Horizon Heading Marks • Traffic Display • Airport Signs • Runway Display • 3D Alert Symbols	• Terrain Alerting • Obstacle Alerting • SVT Field of View • SVT Alert Inhibit Controls

Pitch Scale Expansion

When SVT is active, the pitch scale expands to display fewer degrees of pitch. This change in scale is necessary to maintain an accurate representation of terrain.



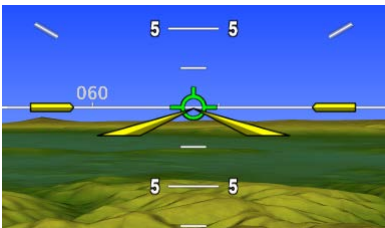
Default Pitch Scale



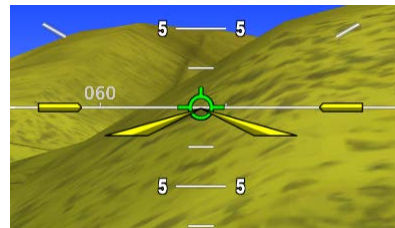
Pitch Scale with SVT

Zero-Pitch Line

This horizontal reference line spans the width of the PFD. It is useful when quickly determining whether the aircraft's current altitude is above or below the approaching terrain.



Zero-Pitch Line Above Mountains



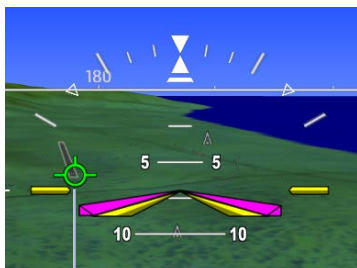
Zero-Pitch Line Below Mountains

Flight Path Marker



WARNING

Do not use the Flight Path Marker as a flight director.



Flight Path Marker

The flight path marker displays the aircraft's projected path relative to runways, traffic, terrain, and obstacles.

It indicates the aircraft's current lateral and vertical path, accounting for wind speed and direction relative to the 3-dimensional terrain depiction.

The flight path marker displays when SVT is enabled and active, and ground speed is greater than 30 knots. This is useful when assessing potential conflicts with terrain, obstacles, and traffic.

Displayed terrain or obstacles in the aircraft's flight path depicted extending above the FPM indicate a possible conflict.

SVT Terrain and Obstacles

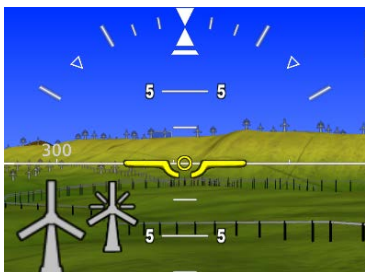


SVT Features



SVT depicts terrain in front of the aircraft. The color palette indicates terrain and obstacles at varying elevations.

This imagery derives from the aircraft's attitude, heading, GPS 3D position, and terrain and obstacle databases.

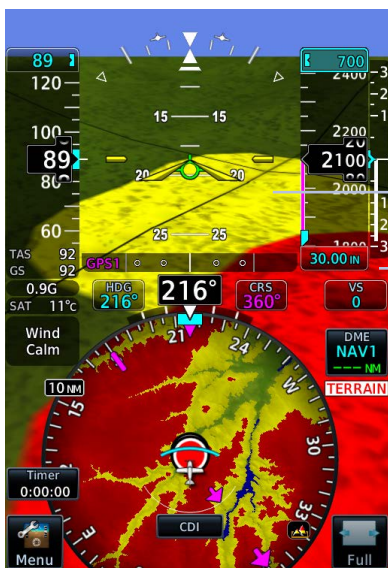


Two-dimensional symbols represent an obstacle's relative height above terrain and distance from the aircraft.

- Includes towers, power lines, and wind turbines
- Obstacles never obstruct instrumentation
- Size varies according to aircraft proximity

TERRAIN AND OBSTACLE ALERTS

When interfaced to a GTN with TAWS-A, TAWS-B or HTAWS enabled, or when TXi Terrain FLTA or TAWS-B is enabled, terrain and obstacle alerts include alert depictions in SVT.

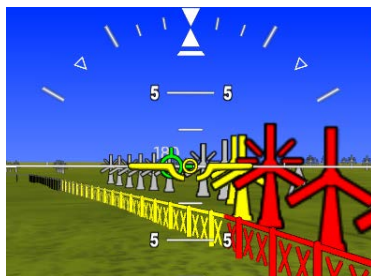


Alerted Terrain

In the event of a terrain or obstacle alert, SVT shading changes to red or yellow, reflecting alert status.

SVT Caution Shade

SVT Warning Shade



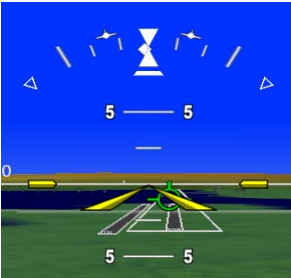
Alerted Obstacles

SVT Runways



Runway depictions provide improved awareness of runway location with respect to surrounding terrain. Runway thresholds correspond to their correct elevation with respect to terrain.

Runways oriented within 45° of aircraft heading are represented in white, others in gray.



The runway is brighter with a white box outline when a runway approach is active. This improves recognition of the intended runway.

Detail is added as the aircraft becomes closer to the runway. This includes runway numbers and centerlines. Taxiways are not shown in SVT.

SVT Traffic

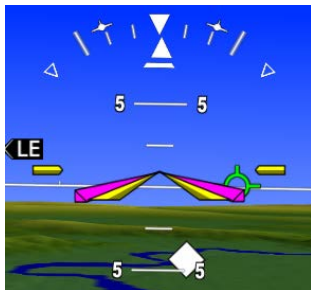
Detected intruders display in their relative location as determined by the aircraft traffic systems. Traffic in 3D appears larger when closer, and smaller when further away. Intruders above the zero pitch line are above the aircraft's altitude and intruders below are lower.



SVT Traffic Display

Advanced Features

A white diamond indicates proximate traffic. A yellow circle indicates alerted traffic.



Proximity Advisory



Traffic Advisory

Selectable Display Features

Pilot selectable display features provide additional indications in synthetic vision.

HORIZON HEADINGS



Horizon heading marks synchronize with the HSI. Tick marks and compass heading display in 30° increments on the zero-pitch line. These markings are not visible behind the altimeter and airspeed indicator.

AIRPORT SIGNS



Signs denote airport location and ID. These indications are useful when flying into an unfamiliar airport.

The type of indication is dependent on the aircraft's approach distance.

To reduce clutter, signs for private airports display only if the airport is in the active flight plan.

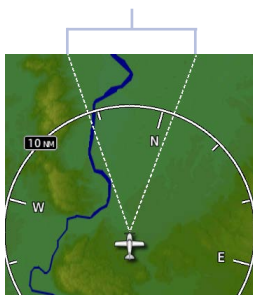
RANGE FROM AIRPORT	SVT AIRPORT INDICATION
15 nm	Signpost and frame only.
8 nm	Identifier annunciates within frame.
4.5 nm	No indication (runway visible).

SVT FIELD OF VIEW (MFD)

FEATURE LIMITATIONS

- Available only on GDU 1060/1210
- Displays only when the Synthetic Terrain function is active

SVT Field of View



On the MFD:

The Map page offers a depiction of the PFD's lateral view.

Terrain and traffic within this V-shaped indicator are visible in synthetic vision.

G500(H) TXi (H)SVT Enablement

For convenience, (H)SVT enablement is available for download at flyGarmin.com. This process is similar to downloading databases and does not require a visit to the dealer.

You must be the aircraft account holder to download this feature. Contact your Garmin dealer for assistance.

To add (H)SVT to your TXi system:

1. Sign in to your flyGarmin.com account.
2. Select the **Devices** tab, then locate your aircraft and G500(H) TXi unit.
3. Scroll to the "Feature Enablements" section. Select **Buy Enablement** and then complete the checkout process.
4. Insert a blank SD card into the computer/card adapter. Complete the on-screen prompts to create an enablement card.
5. Place the enablement card in the top/left slot of the TXi PFD.
6. Power up all configured GDUs.
7. Activate (H)SVT if necessary.

Enablement occurs upon system startup. While this process is quick, use of a ground power unit or running the engine may be advised depending on the aircraft.

Automatic Flight Control System

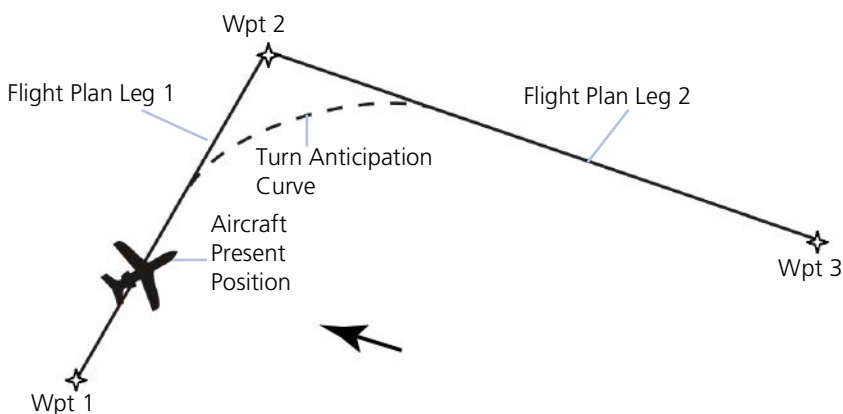
AFCS Interface Functions

- | | |
|----------------------|-------------------------------|
| • GPSS | • Altitude capture |
| • Flight director | • Lateral/vertical deviations |
| • Mode annunciations | • VS preselect |
| • Half-bank mode | • Servo heading reference |

GPSS

GPSS provides roll command signals from the navigator to the autopilot in order to:

- Make smooth transitions when passing waypoints
- Fly curved leg types (e.g., procedure turns and holding patterns)



GPSS Turn Anticipation

If the autopilot is capable of receiving GPSS Roll Steering information and when navigating using GPS guidance, data is transmitted through a digital communications bus from the GDU to the autopilot. In dual GPS installations, the PFD sends Roll Steering information for the currently-selected GPS source. For these autopilots there is no pilot action required to use GPSS.

For autopilots that are not GPSS Roll Steering capable, the GDU converts GPSS turn commands into a heading error signal for the autopilot. When autopilot is operated in HDG mode and GPSS is selected on the PFD, the autopilot flies the turn commands from the GPS navigator selected on PFD 1. An additional touchscreen button is provided on the selected heading popup window for toggling the autopilot heading reference between GPSS and selected heading. After setting the heading bug, "GPSS" displays in selected heading field.



The **AP HDG REF** key is present if an autopilot is interfaced with the GDU and is configured to use GPSS emulation. This key changes the autopilot's heading reference between HDG bug and GPSS.

When GPSS is selected:

- Heading bug changes from solid to a cyan outline
- Selection key annunciates a GPSS label

The heading bug may still be adjusted by the pilot as a visual reference without affecting GPSS to the autopilot.

GPSS Mode Icons



Roll steering function is active, but no commands are sent.



GDU is sending roll steering commands to autopilot (pilot side only).



GDU is sending roll steering commands to autopilot (copilot side or second PFD only).



GDU is sending selected HDG commands without GPSS.



Heading bug on HSI is solid.

Flight Director

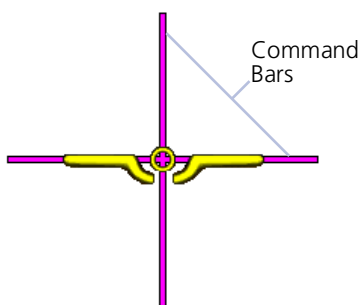
FEATURE REQUIREMENTS

- Autopilot flight director commands interfaced to the TXi system

FEATURE LIMITATIONS

The TXi system limits the distance the flight director commands may deviate from the aircraft attitude icon. If the pitch command provided by the flight director is greater than that allowed by the display, then the command bars will display at the maximum distance allowed by the PFD. As the aircraft pitch changes to satisfy the command bars, the bars will continue display at the maximum distance from the aircraft attitude symbol until the aircraft pitch deviation is within the command display limit.

Command Cues



Dual Cue, Rotorcraft

Flight director displays commands from the external flight director on the PFD. Available display options are dependent upon aircraft type and configuration.

- *Fixed wing:* The PFD allows you to display commands as either a single or dual cue.
- *Rotorcraft:* Commands always display as a dual cue.

FLIGHT DIRECTOR KEY

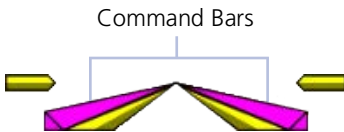
FIXED-WING AIRCRAFT ONLY

FEATURE REQUIREMENTS

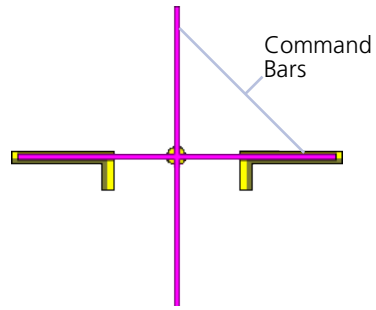
- TXi software v3.50 or later
- Aircraft Symbol/Flight Director option configured as "Pilot Control"



Tapping **Flight Director** toggles the command type between single cue and dual cue. This option resides in the PFD Setup menu.



Single Cue

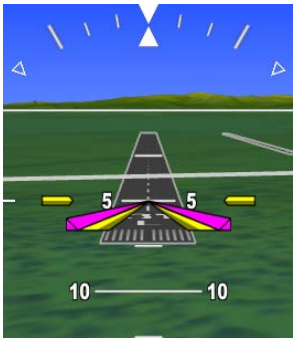


Dual Cue

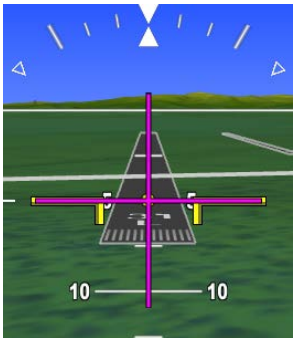
COMMAND INDICATIONS

Command bars represent pitch and roll guidance.

Aircraft at Commanded Attitude

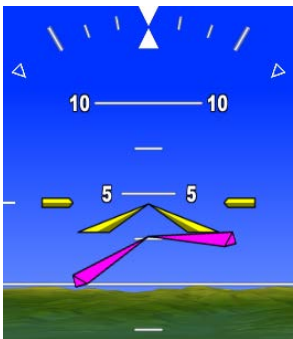


Single Cue

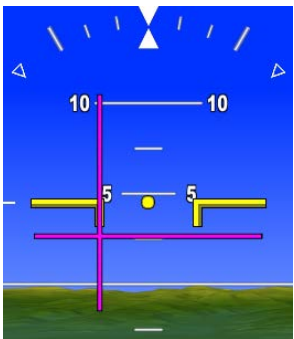


Dual Cue

Descend Left Command



Single Cue



Dual Cue

CONDITION	INDICATION
Commanded pitch is greater than GDU maximum allowable limit.	Command bars display at maximum allowable limit.
Aircraft pitch exceeds $-20^{\circ}/+30^{\circ}$.	Command bars and annunciations declutter.
Aircraft roll exceeds $\pm 65^{\circ}$.	
Attitude information becomes invalid or unavailable.	Command bars do not display.

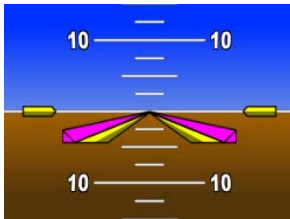
AFCS ENGAGED INDICATION

FEATURE LIMITATIONS

"AFCS not engaged" indications (hollow command bars) available only for single cue depictions.

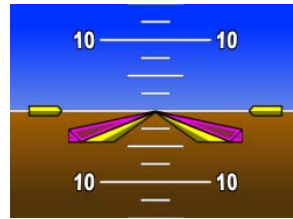
For some autopilots, single cue command bars alternate between hollow and solid to show autopilot status.

AFCS Engaged



Command bars are solid.

AFCS Not Engaged



Command bars are hollow.

IAS/FLC & VS Bug Indications

FIXED-WING AIRCRAFT ONLY

FEATURE REQUIREMENTS

- *TXi software v3.21 or later*
- *TXi software v3.61 or later for FLC bug indications*

FEATURE LIMITATIONS

- *Applicable to GFC 500/600 and DFC90 autopilots only*
- *FLC mode applicable to GFC 600 only*

Airspeed and vertical speed bugs may be hollow or solid depending on the state of the corresponding flight director mode.

AIRSPEED BUG INDICATIONS

For GFC 600 systems that support FLC mode instead of IAS vertical mode, you may use a Mach reference for the airspeed bug.

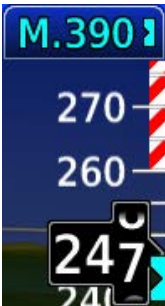
When interfaced with GFC 600 configured for FLC mode, changing the airspeed reference between Mach and IAS units will result in a corresponding change in the FLC reference displayed in the AFCS status box. Read more about Mach reference values in *Mach Selection*.



The airspeed bug turns solid and automatically changes to the current airspeed value when IAS flight director mode is activated.

Flight director mode is IAS.

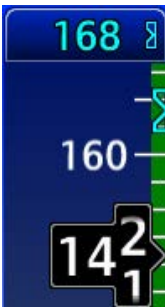
Flight Director IAS Mode Active



The bug turns solid and automatically changes to the current Mach value when Mach Selection is active and FLC flight director mode is activated.

Flight director mode is FLC.

Flight Director FLC Mode Active



The airspeed bug is hollow when IAS/FLC flight director mode is inactive.

Flight director mode is a vertical autopilot mode (e.g., ALT, VS).

Flight Director IAS/FLC Mode Inactive

VERTICAL SPEED BUG INDICATIONS



The vertical airspeed bug is hollow when VS flight director mode is inactive.



ALT 12000

Flight director mode is ALT.

Flight Director VS Mode Inactive



The bug turns solid and automatically changes to the current vertical speed when VS flight director mode is activated.



VS ↑1000
ALT

Flight director mode is VS.

Flight Director VS Mode Active

Altitude Preselect Functions

FEATURE LIMITATIONS

- Mode availability is dependent upon autopilot type

Arming Altitude Capture

Depending on the autopilot and flight director interface, GDU provides altitude preselect capabilities. This allows autopilot to capture a preselected altitude.

If arming is required, **ALT CAP** arms the altitude capture function and sets the altitude bug.



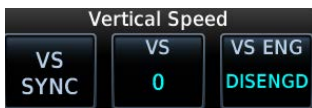
1. Specify altitude value.
2. Set **ALT CAP** to "ARMED."

An optional **ARM** key is available on the GCU 485 control panel.

If the autopilot does not support altitude preselect functionality, then the selected altitude field acts as an altitude alerter. For more about the selected altitude and altitude alerting functions, read *Barometric Altimeter* in section 3.

Enabling Vertical Speed Control

Depending on the autopilot and flight director interface, GDU provides vertical speed commands to autopilots.



1. Specify vertical speed value.
2. Set **VS ENG** to ENGAGED.



VS key label changes to reflect mode status. "AUTO" indicates the autopilot is actively scheduling vertical speed.

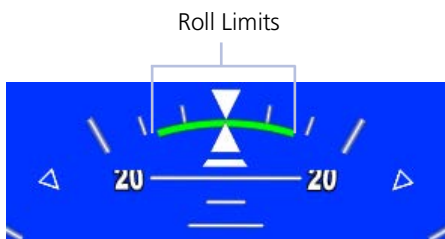
Low Bank Mode

FEATURE REQUIREMENTS

- PFD
- GFC 600 with low bank mode support

This feature is intended for aircraft that operate at high altitude or that would otherwise require low bank autopilot support (e.g., multi-engine aircraft experiencing engine failure during takeoff).

When crossing the altitude threshold, low bank mode will automatically activate in aircraft configured with low bank support. You may also activate or deactivate low bank mode manually using the installed external switch.



A green arc appears on the roll scale to indicate the roll limits.

Actual roll limits are determined by the autopilot and based on airframe type.

Selected Side/Sensor

When the Selected Side function is enabled, the GFC 600 flight director uses sensor and bug data from the selected PFD (pilot or copilot) only. These data include:

- Selected CDI NAV source
- Selected AHRS sensor
- Selected ADC sensor
- Selected bugs
- Baro setting

For more about the Selected Side function, read *PFD Selection* in section 3.

Servo Heading Reference



ROTORCRAFT ONLY

FEATURE REQUIREMENTS

- PFD
- GFC 600H

For rotorcraft, the option to beep heading using the yaw beep switch is available while hovering or in ground speed mode. This displays as a servo heading reference bug on the HSI and as a numeric value in a field adjacent to the HSI.

The servo heading reference is adjustable only through the HFCS interface. Indications on the TXi PFD are display only.

Servo Heading
Reference Bug

Servo Heading
Reference Value



The reference bug represents the heading that the yaw servo is attempting to achieve while in hover.

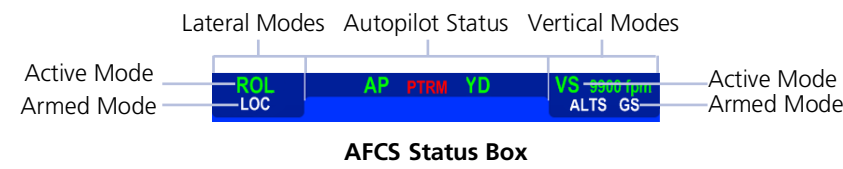
AFCS Mode Annunciations

FEATURE LIMITATIONS

- Applicable to GFC 500/600, KFC 225/275/325, DFC90
- Custom textual alert annunciations available only for GFC 600 and based on configuration

Autopilot mode status annunciates in a box along the top edge of the PFD.

- Green indicates active status
- White indicates armed mode



For details about DFC90 autopilot operation and applicable mode annunciations, consult *Avidyne DFC90 Digital Autopilot Pilot Guide*.

AFCS Basic Mode Annunciations

ROTORCRAFT ONLY		FIXED WING & ROTORCRAFT	
CPLD	Flight Director Coupled	AP	Autopilot
FTR	Force Trim Release	YD	Yaw Damper
PRY	Pitch, Roll, and Yaw	CWS	Control Wheel Steering

AFCS Lateral & Vertical Mode Annunciations

VERTICAL MODE ANNUNCIATIONS		LATERAL MODE ANNUNCIATIONS	
ALT	Altitude Hold	BC	Backcourse Navigation
ALTS	Selected Altitude Capture	GPS	GPS Approach
APR	Approach		GPS Navigation
GA	Go Around	HDG	Heading
GP	Glidepath	LVL	Level Hold
GS	Glideslope	LOC	Localizer Approach
IAS	Indicated Airspeed		Localizer Navigation
LVL	Level Hold	NAV	Navigation
OSP	Overspeed Protection ¹	ROL	Roll Hold
PIT	Pitch Hold	ROL	Roll Hold
USP	Underspeed Protection ¹	VAPP	VOR Approach ¹
VNAV	Vertical Navigation	VOR	VOR Navigation
V ¹			
VPTH	Vertical Path		
VS	Vertical Speed		
FLC	Flight Level Change ¹		

¹ GFC 600 only.

AFCS Alert Annunciations

GFC 500/600 ALERTS, FIXED WING	
PFT	Preflight test in progress.
PFT	Preflight test failure.
AFCS	Autopilot failure.
AP	Autopilot is automatically disengaged.
AP	Autopilot is manually disengaged. ¹
PTRM	Pitch trim failure. Autopilot may remain engaged. If autopilot disengages, re-engagement cannot occur until after the problem is resolved. May be red or yellow.
↑ELE	Mistrim condition requiring nose up elevator trim.
↓ELE	Mistrim condition requiring nose down elevator trim.
←AIL	Mistrim condition requiring roll trim to the left.
AIL→	Mistrim condition requiring roll trim to the right.
←RUD	Mistrim condition requiring rudder trim to the left.
RUD→	Mistrim condition requiring rudder trim to the right.
MAXSPD	Overspeed protection is active.
MINSPD	Underspeed protection is active.
YAW	Yaw damper failure. May be red or yellow.
RB PFT	Rudder bias preflight test in progress. ^{3, 6}
RB OFF	Rudder bias is manually turned off. ²
RB FAIL	Rudder bias system is inoperative. ²
R ENG	Low power detected on right engine. Rudder bias system is active. ²
L ENG	Low power detected on left engine. Rudder bias system is active. ²
V	VNAV is automatically disabled during normal operation. ^{3, 6}
ICE SPD	Airspeed is below the airframe's minimum speed for operating in icing conditions. Anti-ice function is active. ^{3, 6, 7}

GFC 500/600 ALERTS, FIXED WING

GLIDE

Smart Glide is active. Appears during IAS vertical mode.⁵

EDM

Emergency Descent Mode is active.

EDM

Emergency Descent Mode is inhibited or in override mode.

ESP

Electronic Stability and Protection is active.^{3, 4}

¹ Annunciation flashes for 5 seconds, then turns off. ² GFC 600 twin-engine aircraft only. ³ GFC 600 only.

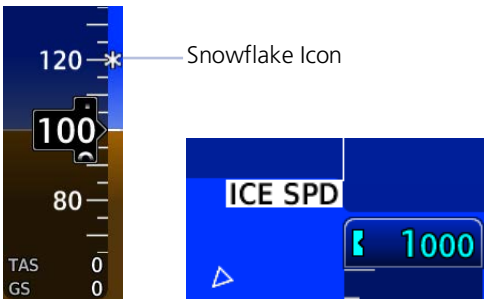
⁴ Available with TXi software v3.21 and later. ⁵ Available with TXi software v3.30 and later.

⁶ Available with TXi software v3.80 and later.

⁷ Annunciation overridden when underspeed protection (“MINSPD”) is active.

On aircraft equipped with GFC 600, a custom five-character textual alert may be configured to annunciate in place of one or more AFCS alerts. Consult the AFMS to determine the available message types for your aircraft.

FOR FIXED-WING AIRCRAFT: A snowflake icon appears on the airspeed tape when TXi EIS or GMC 605 detects that the anti-ice function is active.



Anti-ice Active Indications

“ICE SPD” annunciates once the airspeed falls below the minimum speed required for flight into known icing conditions.

ACKNOWLEDGE GFC 500 AFCS ALERTS

FEATURE REQUIREMENTS

- *TXi software v3.30 or later*

FEATURE LIMITATIONS

- *Applicable only to GFC 500 with GI 275*

A cyan border indicates when GFC 500 AFCS alert annunciations require pilot acknowledgment.

Examples:



Acknowledgeable AFCS Alert Annunciations

You can acknowledge AFCS alert annunciations one of two ways:

- Tap the indicated alert annunciation(s) in the AFCS status box.

OR

- Push the control knob.

GFC 600H ALERTS, ROTORCRAFT

PFT

Preflight test failure.

FCS

HFCS failure. May be flashing or solid.

P R Y

Servo failure.

P R Y

Out of detent.

P R Y

Out of detent > 30 seconds.

P R Y

Servo is disengaging.¹

CPLD

HFCS Flight Director is disengaging.¹

LOWALT

HFCS low altitude protection feature is active.

MAXSPD

Overspeed protection is active.

LOWSPD

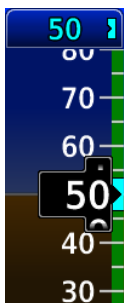
Underspeed protection is active.

YAW

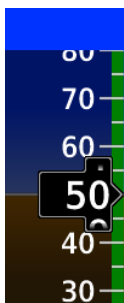
Yaw axis failure. May be red or yellow.

¹ Annunciation flashes for 10 seconds, then turns off.

FOR ROTORCRAFT: The selected airspeed bug and control field are not present during overspeed and underspeed protection modes.



Normal IAS
Operating Mode



Underspeed
Protection Active

IAS 35

Autopilot maintains airspeed at the current bug value.
This value annunciates with the active vertical mode.

Autopilot Preflight Test



WARNING

Do not use a third-party autopilot if it does not disengage normally during the preflight test.

FEATURE LIMITATIONS

- *AP DISC availability is dependent upon unit configuration*

If the system detects a malfunction in the AHRS, “ATTITUDE” annunciates on the PFD and the autopilot disconnects. For more information, consult the autopilot operating instructions.

The GAD 43e uses AHRS attitude output to generate analog signal information for an installed ADI-gyro based autopilot system. Should an error occur within the GAD 43e or AHRS output, the autopilot will disengage. Test the autopilot disconnect before each flight.

TEST THE AUTOPILOT DISCONNECT

1. Power up all avionics. Allow time for self-test completion.
2. Engage autopilot.
3. Select **Test** from the PFD Menu.
4. Tap **AP DISC**.
5. Verify autopilot disconnects.

Aerobatic Flight

Aerobatic Enablement

FEATURE REQUIREMENTS

- *GSU 75*
- *TXi software v3.21 or later*
- *Purchased Aerobatic feature enabled on GDU*

With the aerobatic feature enabled, the PFD continues to display attitude during high acceleration and roll rates.

Enabling the aerobatic feature does not alter extreme attitude indications. Exceeding the limits of the AHRS still results in a red “X” failure annunciation on the PFD. If this should occur, fly the aircraft straight and level and allow the AHRS time to recover.

For more about extreme attitude indications, read *Extreme Attitude Indications* in section 3.

Carbon Monoxide Alerts



NOTE

Always take appropriate emergency measures as described in the AFM/POH.

FEATURE REQUIREMENTS

- GCO 14
- GI 275 or GTN Xi series navigator (host)
- TXi software v3.80 or later

OPTIONAL COMPONENTS

- Additional alerting features, including aural caution alerts and custom advisories, are available only on capable GTN Xi¹ and GI 275² units in the system

CO concentration alerts and inhibit control are available for installations equipped with a Garmin carbon monoxide detector.

GCO 14 Features

- Issues visual caution alerts when CO concentration meets the activation criteria
- Includes alert inhibit option

¹ Available with GTN Xi software v 21.01 and later. ² Available with GI 275 software v3.40 and later.

Alert Inhibit



The **Inhibit CO Alerts** control is accessible via the System page. Toggle the function on or off when appropriate.

From the MFD Home page:

Tap **System** > **Setup** > **Inhibit CO Alerts**.

From the GDU 700() PFD:

Tap **Menu** > **System** > **Inhibit CO Alerts**.

From the GDU 700() EIS:

Tap **Menu** > **System** > **Inhibit CO Alerts**.

Always use discretion when inhibiting CO caution alerts. Re-activate the alert function when appropriate.

CO Alert Indications

When a CO alert occurs:

- A textual caution annunciation appears at the bottom of the screen

A caution alerts you when the concentration of carbon monoxide in the cabin meets the activation criteria. CO alerts have a lower priority than terrain and traffic alerts.

PFD Annunciation



EIS/MFD Annunciator Bar



CO Level Alert Annunciation

ANNUNCIATION	CONDITION POP-UP AURAL MESSAGE
CO LEVEL	Condition: CO concentration meets the activation criteria Pop-up Alert: No Aural Message: No

CO caution alerts may be accompanied by an aural message on the host LRU.

Stabilized Approach Alerts

FEATURE LIMITATIONS

- *Annunciation alerting levels, alert behavior, and aural voice message availability are dependent upon configuration and may vary for each alerting altitude zone*
- *Not all stabilized approach alerts may be available for all configurations*

Stabilized Approach is a GPS-based monitoring system that checks certain parameters to ensure the aircraft operates within the stabilized approach envelope. Should the flight path deviate outside the configured parameters, aural alerts and/or annunciations will notify the crew of the condition that must be addressed to maintain a stable approach.



Alert Annunciation

Alert annunciations appear adjacent to the airspeed tape. They can be white (advisory) or yellow (caution) depending on the configured alerting level.

Aural alert indications notify the flight crew of unstable conditions during the approach.

Depending on configuration, each alert may consist of a visual annunciation and/or an aural voice message. The number of times a voice message repeats is configured by the installer.

Stabilized Approach Function

- Automates parameter monitoring typically performed by the copilot and alerts the pilot when stabilized approach criteria are not met
- Provides real-time guidance to crew members when corrections are needed to bring the aircraft back within the stabilized approach envelope
- Uses course deviations, IAS, and GS inputs for GPS-based monitoring
- Calculates wind data in real time while the aircraft is airborne (wind calculations are not based on reported winds at the ground station)

Stabilized approach alerts occur automatically when the aircraft is on approach to a runway and within a certain vertical zone. For information regarding the stabilized approach alerts for your specific aircraft, including alerting conditions, consult the AFMS.

Alert Types

Not all of the following alerts may be available.

ALERT TYPE	CONDITION
Approach Speed	Approach speed has deviated from V_{REF} by more than the configured threshold.
Baro/GPS Mismatch	Barometric and GPS altitudes differ by more than the configured threshold value.
Crosswind	The currently calculated crosswind component exceeds the maximum demonstrated crosswind or the crosswind limitation from the AFM/POH.
Descent Speed ^{1, 2}	<i>VFR Approach</i> : Descent speed is greater than the configured descent speed threshold for the active approach type.
	<i>IFR Approach</i> : Descent speed differs from the computed target speed by more than the configured descent speed tolerance.
Flaps Not in Landing Configuration ^{1, 2, 3}	Flaps are not in position for landing.
Gear Not in Landing Configuration ^{1, 2}	Gear is not down and locked.
Lateral Deviation	The magnitude of the lateral deviation is greater than the configured threshold value.
Tailwind	The currently calculated tailwind component exceeds the maximum tailwind component from the AFM/POH.
Vertical Deviation ^{1, 4}	The magnitude of the vertical deviation is greater than the configured threshold value. The type of indication is applicable to the type of approach being flown (Glidepath or Glideslope).

To determine the configured threshold values for your specific aircraft, consult the AFMS.

¹ Alert type not available when configured for TAWS-A. ² Available with TXi software v3.80 and later.
³ For manufacturer recommended or required flap positions, consult the AFM/POH.
⁴ "Glidepath" for GPS approaches; "Glideslope" for non-GPS approaches.

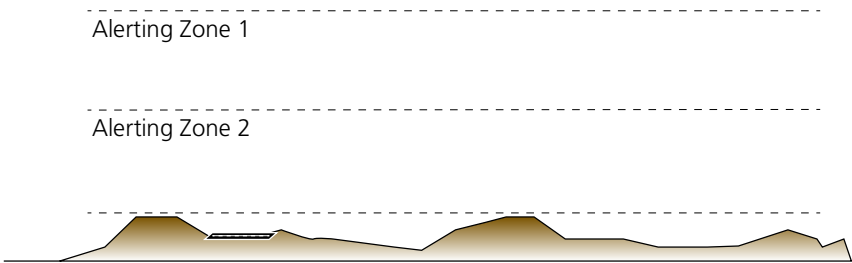
Runway Alerting Zones

Stabilized Approach caution alerts occur only while the aircraft is situated vertically within the Vertical Alerting Zone while in the Horizontal Alerting Zone.

Vertical Alerting Zone

The Vertical Alerting Zone accounts for IFR and VFR conditions with respective alert zones.

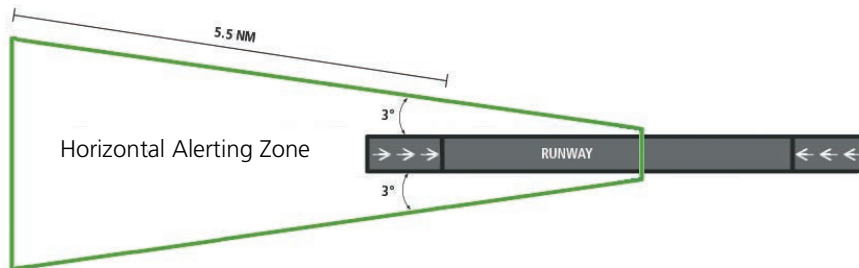
The system issues vertical IFR and VFR approach alerts between two sets of installer configurable altitudes: Alerting Zone 1 and Alerting Zone 2. Alerting levels may be configured as advisory, caution, or none.



To receive IFR stabilized approach alerts, you must load an active instrument approach procedure to the active flight plan.

Horizontal Runway Alerting Zone

The system considers the aircraft to be within a runway alerting zone when the aircraft is on approach to a runway and situated within the limits of the runway's horizontal alerting zone.



Horizontal IFR/VFR Alerting Zone

The system uses the approach loaded into the navigator to determine the alerting thresholds. Straight-in instrument approaches will trigger IFR alerting, whereas circling or visual approaches will trigger VFR alerting. If no approach is loaded, but an airport is the last waypoint in the flight plan, VFR alerting will be used.

Alert Inhibit

Alert Suppression Options

- Inhibit All
- Speed Inhibit
- BARO Inhibit
- Crosswind Inhibit
- Tailwind Inhibit
- Course Inhibit
- GP/GS Inhibit
- Flaps Inhibit
- Gear Inhibit
- Sink Rate Inhibit

Suppression controls allow you to eliminate nuisance alerts during the associated conditions. These controls are accessible from the PFD Setup menu.

GDU 700(): **Menu > System > PFD Setup > Stabilized Approach**

GDU 1060/1210: **Menu > PFD Setup > Stabilized Approach**

Alert suppression control settings default back to install presets following a power cycle.

Stabilized Approach Alert Indications

Alert Priority

Alerts occur in order of priority, from highest (1) to lowest (9):

- 1. Gear
- 2. Flaps
- 3. Speed
- 4. BARO
- 5. Glidepath/Glideslope
- 6. Course
- 7. Sink Rate
- 8. Xwind
- 9. Twind

In the event of simultaneous alerts, only the highest priority alert displays, regardless of whether the alert is configured as aural, visual, or both.

Available aural and visual indications are dependent upon configuration.

ALERT TYPE	ALERTING ZONE 1		ALERTING ZONE 2	
	AURAL	VISUAL	AURAL	VISUAL
Approach Speed	"Speed"	SPEED	"Speed, Speed"	SPEED
BARO/GPS Mismatch	"Baro"	BARO	"Baro, Baro"	BARO
Crosswind	"Crosswind"	XWIND	"Crosswind, Crosswind"	XWIND
Descent Speed	"Sink Rate"	SINK RATE	"Sink Rate, Sink Rate"	SINK RATE
Flaps Not in Landing Configuration	"Flaps"	FLAPS	"Flaps, Flaps"	FLAPS
Gear Not in Landing Configuration	"Gear"	GEAR	"Gear, Gear"	GEAR
Lateral Deviation	"Course"	COURSE	"Course, Course"	COURSE
Tailwind	"Tailwind"	TWIND	"Tailwind, Tailwind"	TWIND
Vertical Deviation (GPS Approach)	"Glidepath"	GLIDEPATH	"Glidepath, Glidepath"	GLIDEPATH
Vertical Deviation (Non-GPS Approach)	"Glideslope"	GLIDESLOPE	"Glideslope, Glideslope"	GLIDESLOPE

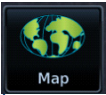
5 Multi-Function Display

- MFD SETUP 5-4
- MAP..... 5-6
- CHARTS 5-34
- ACTIVE FLIGHT PLAN 5-39
- WAYPOINTS..... 5-52
- SIRIUSXM AUDIO ENTERTAINMENT..... 5-58
- EXTERNAL VIDEO 5-61



MULTI-FUNCTION DISPLAY APPS & FUNCTIONS

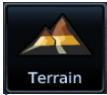
Menu selections vary based on features and optional equipment installed with Garmin avionics.



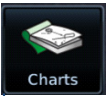
- Overlays
- Map View
- Map Setup
- Map Detail



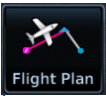
- Mode
- Test
- Motion Vector ¹
- Vector Duration ¹
- Altitude Filter
- TCAS II ⁵



- View
- Layers
- (H)Terrain-FLTA
- TAWS ⁵



- Selection
- Color
- Streaming
- Type



- Edit Data Fields
- Collapse All Airways



- SiriusXM ¹
- FIS-B WX ¹
- Connex WX ¹
- Stormscope ¹
- Radar ¹



- Airport
- Intersection
- VOR
- VRP
- NDB



- Fuel Computer
- Engine & Flight Hours
- Advisories
- Lean



- Smart Glide
- Emergency Descent

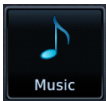
¹ Dependent upon unit configuration. ² Available status screens dependent upon configured LRUs.
³ SD/HD designations determined at configuration. ⁴ Requires Flight Stream 510 wireless transceiver.
⁵ Optional.



1

Video

Source ³
(1 – 4)



Music

Status



System

- System Status
 - Database SYNC
 - Chart Streaming
 - Update
- External LRUs ²
- Setup
 - Date/Time
 - Nearest Airport Criteria
 - Startup Page Selection
 - WX Shortcut
 - Inhibit CO Alerts
 - Click Volume
- Units
- Backlight
- Crew Profile
- Flight Stream ⁴
- Logs
 - Aircraft Log
 - Exceedances ¹
 - Export Data Log

MFD Setup

MFD SYSTEM SELECTIONS	
System Status	• View unit, software, and database information • Check standby database availability • Access Database SYNC, Chart Streaming, and manual database update functions
External LRUs	• Check status of all configured LRUs
Setup	• Select the MFD startup page and visibility • Set nearest airport criteria • Set the clock and click volume • Create Weather page shortcut • Toggle CO caution alerts on or off
Units	Specify units of measure for displayed data. Selections synchronize across all configured GDUs. Certain LRUs may contain unit settings independent of the TXi system. • Distance • Altitude • Temperature • NAV Angle
Backlight	• Adjust display brightness
Crew Profile	• Access crew profile management function
Flight Stream	• Access Bluetooth Setup and Wi-Fi Setup menus
Aircraft Log	• View engine and airframe cycle counters
Exceedances	• View and acknowledge exceedance advisories • Availability dependent upon unit configuration
Export Data Log	• Save logged data to SD card

Nearest Airport Criteria

Filtering criteria includes settings for minimum runway length and hard or soft runway surface options. Selections determine which airports are available when using the nearest airport search feature.

RUNWAY SURFACE	MINIMUM RUNWAY LENGTH
Aircraft runway surface selection: - Any - Hard/Soft - Hard Only - Water	Specify the shortest distance required for takeoff and landing. In the AFM/POH, this is typically the distance given for sea level and the coldest air temperature.

MFD Display Size Options

MFD 40/60% Resize

FEATURE LIMITATIONS

- GDU 1060/1210 PFD/MFD and MFD/MFD units only
- Availability dependent upon configuration
- Accessible from the MFD Home page only



The **Resize** key allows you to change the size of the MFD. This option resides on the MFD Home page.

Tapping the key once expands the MFD from its default 40% screen area to 60%. Tapping it again returns the MFD to 40% area.

MFD Full/Split Screen

FEATURE LIMITATIONS

- GDU 1060/1210 MFD/MFD and MFD/MFD/EIS units only



Tapping **Full** changes the size of one MFD to fill both MFD display areas.



Tapping **Split** restores the dual MFD view.

Map



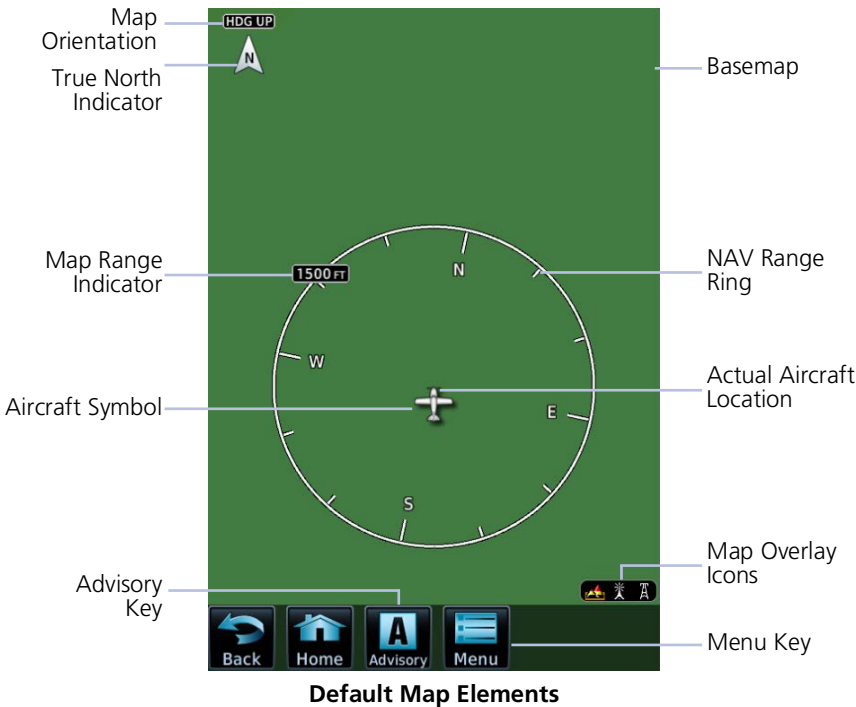
To increase situational awareness, the Map page depicts the aircraft's current position relative to land, aeronautical, weather, and traffic information.

FEATURE REQUIREMENTS

- Active GPS source (aircraft position symbol)
- FIS-B, Iridium transceiver, or SiriusXM receiver (weather data)
- GDL 69, GDL 88, GNX 375, GTX 345, or GSR 56, with appropriate weather service subscription (NEXRAD overlay)
- Active weather subscription (Connex and SiriusXM only)
- Valid charts database (ChartView, FliteCharts) with effective database cycle, and aircraft to be airborne
- GWX 68, GWX 70, or GWX 75 (airborne weather radar overlay)
- Compatible WX-series Stormscope receiver (lightning data)
- TIS/TAS/TCAS II/TCAS II traffic device (traffic data)

FEATURE LIMITATIONS

- Onboard weather radar not available for GDU 700() EIS/MFD configuration



MAP OBJECTS

Aircraft Symbol	Depicts current aircraft position and orientation. • Tip represents actual aircraft location • Symbol type is dependent upon configuration • Absent if GPS source is not available
Basemap	Presents a graphical depiction of land and water data. Basemap is always depicted.
Page Orientation Label	Three orientations: 1. North Up orients map to north. 2. Heading Up orients map to current aircraft heading. 3. Track Up orients map to current aircraft GPS track.
North Indicator	Indicates True north. Tap to orient the map North Up.
Map Range Indicator	Displays current map range in upper left quadrant of range ring (i.e., distance from the aircraft to range ring). Distance is associated with settings in the map setup menu.
NAV Range Ring	Displays current direction of travel on a rotating compass. Absent during map interactions (i.e., pan mode). Orientation: Magnetic north
Map Overlay Icons	Indicates status of overlays at the current map range. Includes: lightning, obstacles, power lines, precipitation, Stormscope, terrain, and traffic
Menu Key	Accesses map overlay controls and pilot selectable settings.

FEATURE LABELS

To maintain readability, map feature labels remain uniform at all zoom levels.

TRAFFIC UNITS

System Units page selections do not affect the display of traffic on the Map page.



If altitude system units are set to meters, the traffic overlay icon will change, serving as a reminder that traffic altitude values remain in feet.

LAND AND WATER DEPICTIONS

Land and water data are for general reference only. Data accuracy is not suitable for use as a primary navigation source. The information is intended to supplement and not replace official government charts and notices.

DATA DRAWING ORDER

The electronic map draws data in order of priority, from highest (1) to lowest (39), with higher priority features drawn atop those of lower priority.

LEVEL	FEATURE
1	Traffic
2	Ownship
3	Flight Plan
4	TAWS FLTA
5	Fuel Range Ring
6	Weather Radar
7	Charts
8	Stormscope
9	Point Obstacles
10	Line Obstacles
11	TFR
12	Freezing Level
13	Cell Movement
14	Lightning
15	METAR
16	Winds Aloft
17	SIGMET
18	AIRMET
19	Cyclone
20	County Warning

LEVEL	FEATURE
21	PIREP
22	AIREP
23	City Forecast
24	Surface Conditions
25	Airspace
26	Waypoints
27	Airways
28	Reference Grid
29	Turbulence
30	Icing
31	Radar Coverage
32	Echo Tops
33	NEXRAD
34	Cloud Tops
35	SafeTaxi
36	Runways
37	Terrain
38	Basemap
39	Topo

Map Interactions

Typical map interactions include zoom, pan, and object selection.

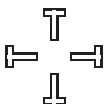
PAN & ZOOM

Panning allows movement of the map in any direction without change to the current zoom setting. Zooming adjusts the current magnification level between pilot specified range parameters.

OBJECT SELECTION

Tapping any object or location on the map displays a map pointer and an information banner.

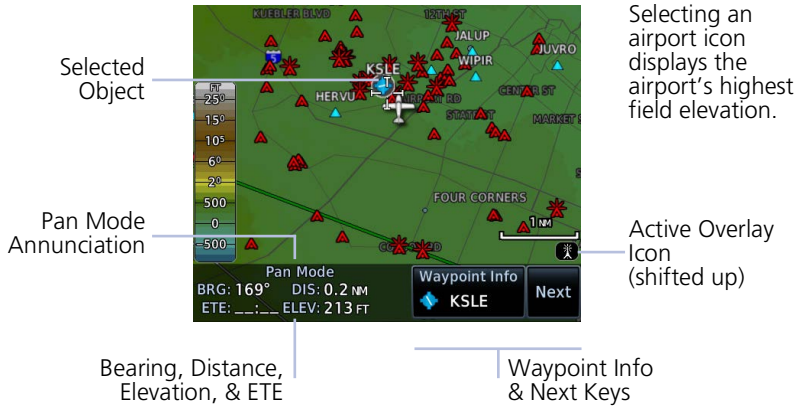
MAP POINTER



This symbol indicates point of contact on the map. A gray circle highlights any selected waypoint or obstacle.

INFO BANNER

Available information and controls are dependent upon object or location type and proximity to other objects.



Info Banner Features

- Pan mode annunciation
- Bearing, distance, and estimated time en route from current aircraft position to map pointer
- Location elevation
- **Next** key for stacked objects
- Maximum altitude AGL and MSL for obstacles
- Information page access key for the selected waypoint, airspace, airport, airport surface hot spot, or TFR

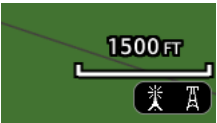
MAP SCALE

FEATURE REQUIREMENTS

- TXi software v3.21 or later



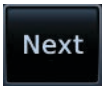
A scale replaces the NAV Range Ring during map interactions, minimizing clutter on the map. This scale depicts half of the current map range (i.e., half of the distance from the aircraft to range ring).



The ring returns once you exit pan mode. To exit pan mode:

- tap the **Back** key, or
- leave and then reopen the map application

STACKED OBJECTS



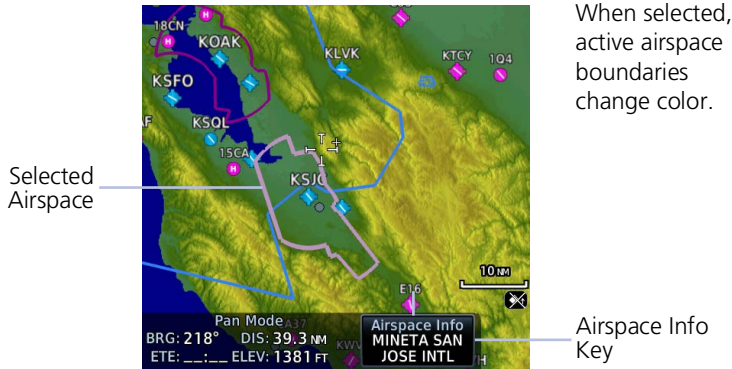
To move the selector through each object in proximity of the map pointer, select an object and tap **Next**.



Overlapping objects may be difficult to identify at a given zoom level.

AIRSPACE INFO

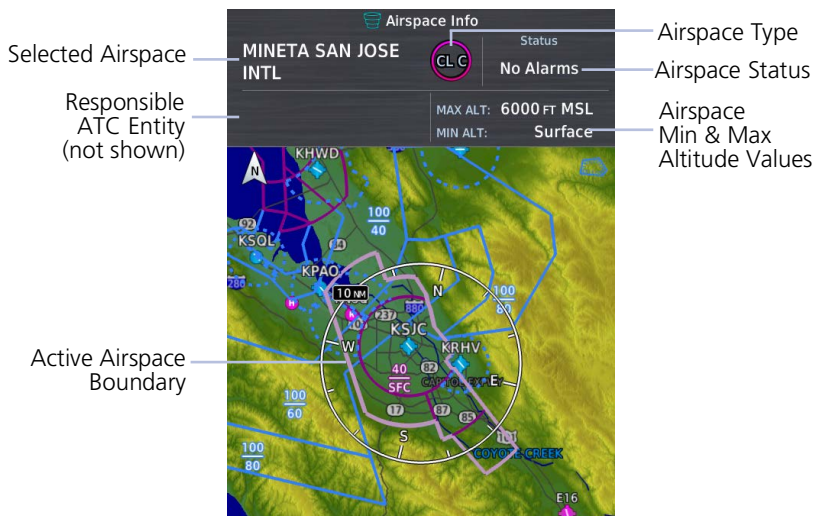
Selecting an airspace on Map shows the United States National Airspace System (NAS) altitude limits up to the selected range.



Airspace Info Page Features

Airspace info pages have the same basic layout as waypoint info pages. The upper portion contains specific data of the selected airspace. The remainder of the page displays the active airspace boundary and a map of the surrounding area. Data include:

- Airspace status and type
- ATC entity responsible for the airspace, if applicable
- Floor and ceiling altitudes



Map Views



The **Map View** key in the Map Menu may be used to select a user-configurable map view.



Three different map views may be individually configured. All settings configured in the map menu and map setup menu are saved.

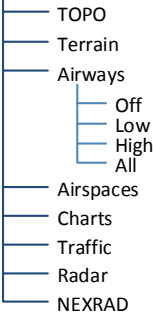
Views may be reconfigured at any time. Any changes to a view configuration are immediately shown on any MFD in the system using that view.

Selecting any view loads previously saved view settings.

Tap the **Customize** key to change a view name or reset both the name and all menu settings to their default values.

Map Overlays

Overlay Selections



Various map overlay selections may be configured. The following overlay functions are mutually exclusive. Enabling one automatically disables the other.

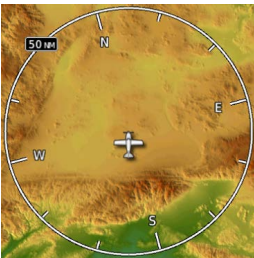
- NEXRAD, terrain, and weather radar overlays
- Stormscope and SXM lightning
- Cloud tops and echo tops
- Charts and SafeTaxi

Overlay Controls



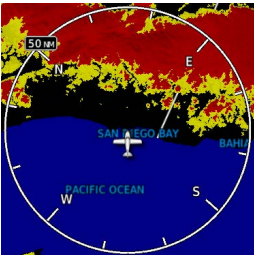
Control keys enable the specified overlay function only and do not activate interfaced equipment. Overlay keys remain active even in the absence of required data.

TOPO



- Overlays topographical data and ground elevation scale
- Depictions are similar to a VFR sectional

TERRAIN



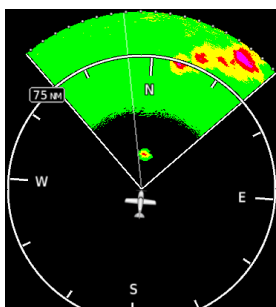
- Overlays terrain map data
- Color shading depicts terrain elevation relative to the aircraft's altitude

TRAFFIC



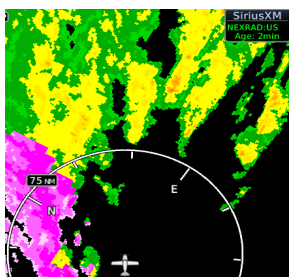
- Overlays traffic information
- Filter selection on Traffic page determines altitude range
- Optional

RADAR



- Overlays airborne weather radar information
- Depictions are identical to those on Weather Radar page
- Excludes WATCH shading
- Optional

NEXRAD



- Overlays datalink precipitation weather information, including product age
- Optional

Overlay Status Icons

Icons indicate which overlays are present at the current map range. A crossed out icon means the overlay is active, but data is unavailable due to a failure, test, or standby condition (where relevant).

The absence of an overlay icon means one of two possible conditions:

1. Overlay not present at the current detail level or zoom setting.
2. Overlay control is off.



Obstacle



Power Line



Stormscope



Traffic



Terrain



Data not available

Map Detail



Select the level of detail on the map.
Changes to the map detail level take effect immediately.

FEATURE	MAP DETAIL LEVEL			
	FULL	HIGH	MED.	LOW
Small Cities	•			
Medium Cities	•			
Large Cities	•			
Freeways	•			
Highways	•			
Roads	•			
Railroads	•			
Basemap Labels	•			
VORs	•	•		
NDBs	•	•		
Line Obstacles	•	•		
Point Obstacles	•	•		
Airspaces that are not prohibited or restricted	•	•		
Waypoints	•	•	•	
SafeTaxi	•	•	•	
Prohibited Airspaces	•	•	•	

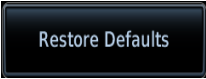
Map Setup

Map Setup

- Map
- Aviation
- Land
- Traffic
- Weather

Overlay data controls are located in the Map Setup menu. Changes to an overlay setting take effect immediately. Selections are organized into five groups.

RESTORE DEFAULTS

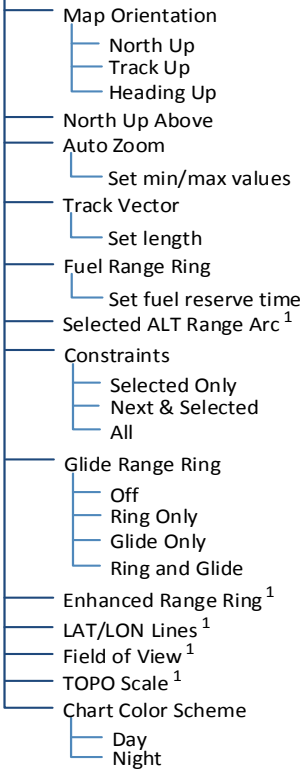


Restore Defaults

Located in the bottom of each list. This key restores all original factory settings for the active group tab.

Map Selections

Map



All Map tab selections provide on/off and range setting options unless otherwise noted.

Shared Map Settings

Changes to the following map settings also take effect on the configured datalink weather app(s). You may adjust these settings from the appropriate Map Setup tab.

- North Up Above range setting and LAT/LON Lines (Map tab)
- Runway Extensions and waypoint display range settings (Aviation tab)
- Road Detail, City Detail, State/Province Borders, and River/Lake Detail (Land tab)

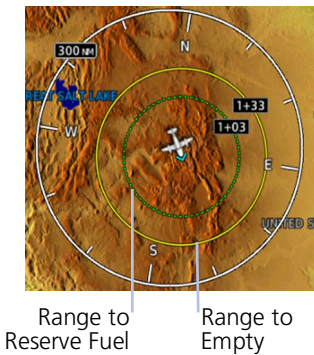
¹ On/off functionality only.

Map Orientation	• Specifies map display orientation • Label above North indicator shows current orientation
North Up Above	• Specifies range at which the map orientation changes to North Up

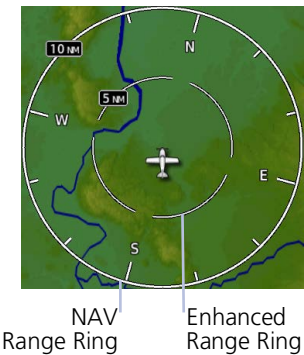
Auto Zoom	- Automatically adjusts page to display the next waypoint in flight plan at the closest possible map range - Presents SafeTaxi data while aircraft is on ground - Automatic functionality is overridden during manual zoom interactions. It resumes when: - Another waypoint is in sequence - Aircraft transitions from on ground to airborne - Auto zoom range matches manual override range - Set minimum and maximum values to best meet operational needs
Track Vector	- Indicates current ground track - End of arrow represents aircraft position at the specified time interval
Fuel Range Ring	- Estimates the remaining range at the current fuel consumption rate and ground speed¹ - Calculations are based on the pilot specified fuel quantity in the fuel computer - Dashed green circle indicates selected range to reserve fuel - Solid yellow circle indicates total endurance range
Selected ALT Range Arc	Represents location at which the aircraft is expected to reach selected altitude²
Constraints	Displays altitude and/or speed³ constraint labels within the flight plan
Glide Range Ring	- Identifies map region within estimated gliding distance and/or best glide airport - Options: Ring Only, Glide Only, Ring and Glide
Enhanced Range Ring	- Provides a more precise indication of distance between the aircraft and map objects - Fixed range value at half the distance of NAV range ring
LAT/LON Lines	Displays latitude and longitude lines
Field of View	- Depicts lateral terrain view presented in SVT⁴ - Synthetic Terrain function on PFD must be active for indication to display
TOPO Scale	Displays a topographical elevation scale
Chart Color Scheme	Changes chart overlay color for day or night view

¹ Requires EIS. ² Systems with at least one PFD. ³ Speed constraints available with GDU TXi software v3.80 and later and GTN Xi software v21.02 and later. ⁴ GDU 1060/1210 only.

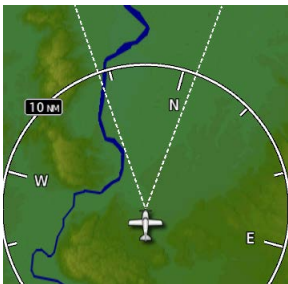
Fuel Range Ring



Enhanced Range Ring



Field of View
(GDU 1060/1210 Only)



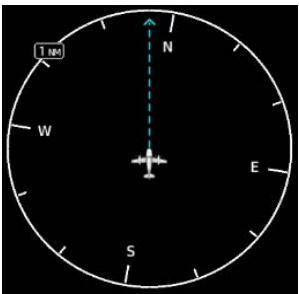
Selected ALT Range Arc



TOPO Scale



Track Vector



When choosing map scales:

- Consider which map scales or zoom levels to use to achieve the desired level of detail and map information.
- Generally, use smaller map scales or zoom levels in and around terminal areas and whenever precise navigation is required (e.g., airway or radial intercepts).
- During cruise flight, map scale settings often increase to balance navigation, situational awareness, and decision making.

Map Orientation sets the map page to either North Up, Track Up, or Heading Up. North Up is useful when zoomed out to view the entire route or a frontal system on a NEXRAD display.

Using the North Up Above feature causes the screen to switch at certain zoom levels. This is useful as a shortcut to quickly increase situational awareness.

Track vector length options display as a dashed line and arrow extending from the aircraft icon, showing current track and distance the aircraft will travel in the selected time.

If the track vector is placed over a point on the map, and no data is entered into the system, the GDU indicates a wind-corrected GPS track to that point. This is useful for intercepting airways and radials, making small but positive lateral corrections during approaches, and in setting up for arrivals in the terminal area.

GLIDE RANGE RING

FIXED-WING AIRCRAFT ONLY

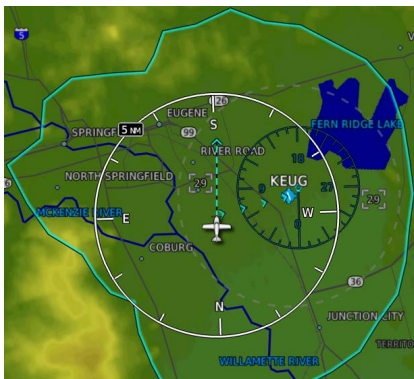
FEATURE REQUIREMENTS

- GTN Xi series navigator (host) configured for Glide Range Ring
- Datalink winds or compatible PFD (Glide Range Ring wind compensation)
- For best glide performance, the aircraft must be configured in accordance with POH guidance

FEATURE LIMITATIONS

Glide Range Ring is a function of the GTN Xi series navigator. GDU does not calculate glide range, nor does it determine best glide airports; it only displays this data when received from GTN Xi. For more about this feature, consult GTN Xi Series Pilot's Guide.

Ring and Glide Display



This overlay receives wind information from the ADAHRS if present. It does not require an ADAHRS unit for operation.

Identifies map region and features within gliding distance. A cyan border indicates where the projected glide descent reaches 50 ft above terrain. Calculations are performed by the host GTN Xi series navigator.

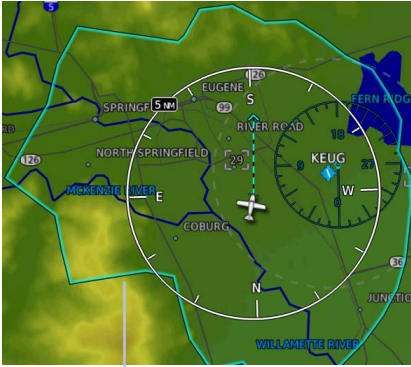


The Glide Range Ring is an estimate based on the best glide speed and glide ratio published for the aircraft. For more information, consult the AFM/POH.

Glide Range Display Options

- Off
- Ring Only
- Glide Only
- Ring and Glide

To enable, tap **Glide Range Ring** and select a display option.

Ring Only Display

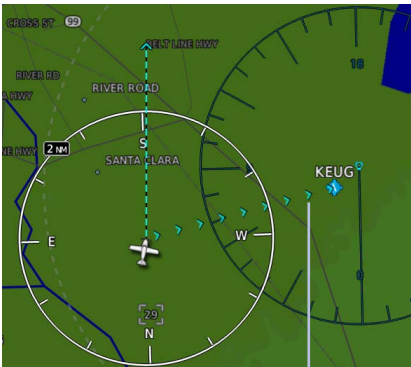
Glide Range
Ring

The Glide Range Ring depicts the estimated glide range down to 50 ft AGL. It does not show beyond 50 ft AGL where the aircraft reaches the ground.

Data updates occur approximately every 5 seconds.

Display
Ring Only

Enable Glide Range Ring to stay aware of all airports within gliding distance. This is extremely helpful should you experience engine failure.

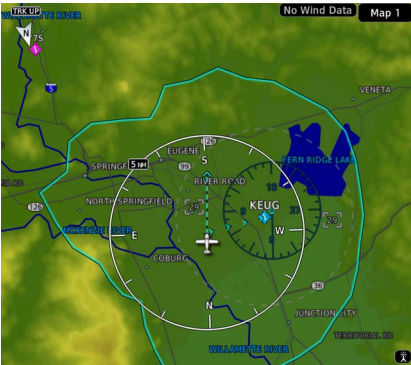
Glide Only Display

Best Glide
Airport
Indicator

Best Glide Airport Indicator

Cyan arrows point toward the best glide airport at any given time. Best glide airport selection is based on distance from current location, runway length, and weather (if available). Airports beyond the range ring are not considered for routing.

Display
Glide Only



“No Wind Data” Indicator

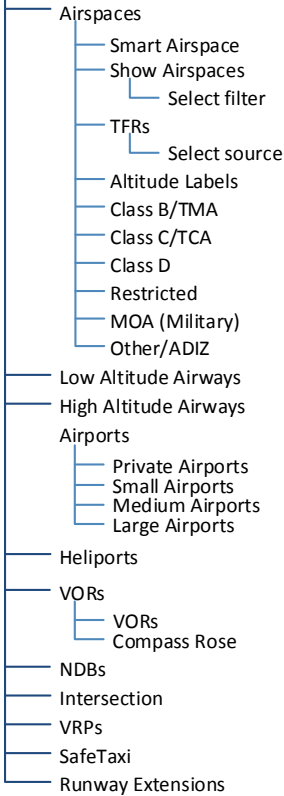
This indication appears when FMS winds and datalink winds aloft are either unavailable or invalid.

The Glide Range Ring does not adjust to account for current winds when the indicator is present.



Aviation Selections

Aviation



Setup options allow the customization of aeronautical information.

- Filter airspace data according to altitude
- Select a TFR source
- Specify airway types and range values
- Customize the display of SafeTaxi and waypoint data
- Display runway extensions for a destination airport

With the exception of TFRs, all Aviation tab selections provide on/off and range setting options.

The runway feature extends the runway's centerline out 5 NM from the runway threshold. This is useful when setting up for a visual approach, especially at airports with parallel runways or low visibility.

AVIATION DATA SYMBOLS

	Non-towered, non-serviced airport ¹		Non-towered, serviced airport ¹
	Towered, non-serviced airport ¹		Towered, serviced airport ¹
	Soft surface, non-serviced airport		Soft surface, serviced airport
	Restricted (private) airport		Unknown airport
	Heliport		ILS/DME or DME only
	Intersection		LOM
	NDB		TACAN
	VOR		VOR/DME
	VORTAC		VRP
	User airport		User waypoint
	Runway extension		ATK
	Fly-over waypoint ²		

¹ Symbol depicts orientation of longest runway.

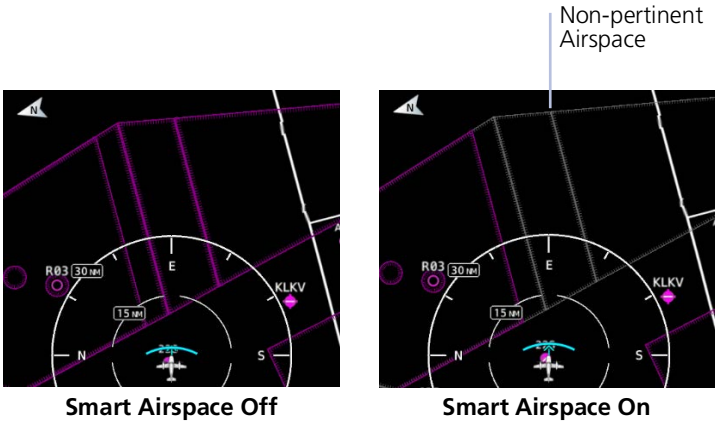
² Requires GDU TXi v3.50 or later with GTN Xi v20.30 or later.

Smart Airspace

Garmin's Smart Airspace feature automatically de-emphasizes non-pertinent airspace away from the aircraft's current altitude.

SMART AIRSPACE CRITERIA	
AIRCRAFT ALTITUDE	AIRSPACE PROXIMITY TO AIRCRAFT ¹
Sea level	>1,000 ft
>10,000 ft	2,000 ft
¹ Vertical distance above and below aircraft altitude.	

An airspace boundary turns gray when its vertical proximity to the aircraft is >1,000 ft. This range increases linearly to 2,000 ft as the aircraft ascends to 10,000 ft.



AIRSPACE DATA SYMBOLS

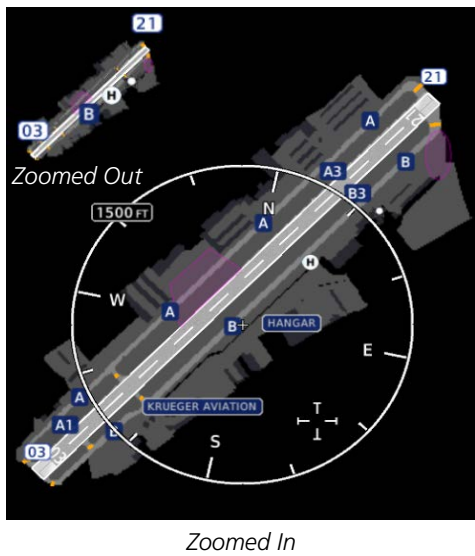
	Class B Altitude Label (ceiling/floor)		Class C/TCA
	Class C Altitude Label (ceiling/floor)		Class D
	Class D Altitude Label (ceiling only)		Restricted
	TFR		MOA
	Class B/TMA		Other/ADIZ

SafeTaxi

During flight, SafeTaxi database is mutually exclusive with chart databases. SafeTaxi and hot spot data display only when the Charts overlay is off.

SafeTaxi Features

- Airport diagram overlay that includes hot spot information
- Aircraft position relative to taxiways, runways, and airport landmarks
- Automatically overrides Charts overlay function during ground operations
- Pilot selectable range options



SafeTaxi provides greater map detail and higher image resolution at lower zoom levels.

Feature labels denote:

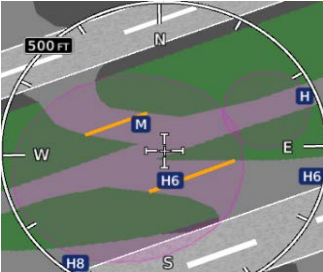
- Runways
- Taxiways
- Airport landmarks

SAFETAXI DATA SYMBOLS

	Runway		Taxiway
	Helipad		Construction Area
	Airport Beacon		Unpaved Parking

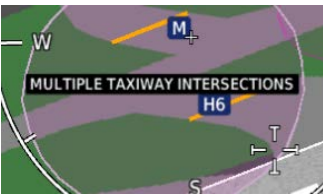
HOT SPOTS

SafeTaxi hot spots identify locations on an airport surface where positional confusion or runway incursions are likely to occur. These known problem areas require heightened attention by pilots.



The following airport features may be deemed hot spots by aviation authorities.

- Intersecting taxiways and runways
- Complex ramp areas
- Directional limitations
- Limited wing-tip clearance
- Overflight risk



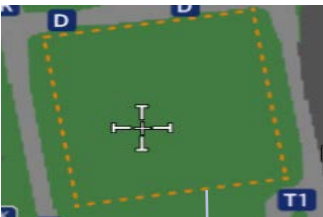
Selecting the border of a hot spot displays a brief summary of the indicated hazard and an information key.



Tap **Hot Spot Info** to view additional location details and potential hazard information. Numbering corresponds to a list on the airport diagram.



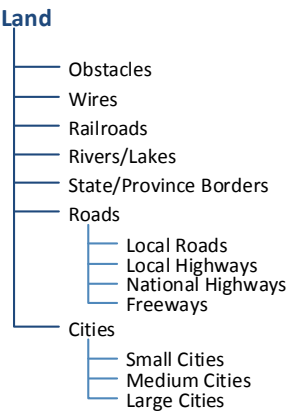
CONSTRUCTION SPOTS



Construction Area
Border

There are no expanded detail keys or notes associated with construction areas.

Land Selections



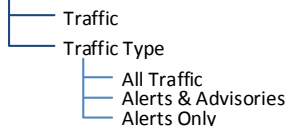
Selections include on/off and range setting options for all land features.

LAND DATA SYMBOLS

	Railroad		National Highway
	River/Lake		Freeway
	State/Province Border		Small City
	Local Road		Medium City
	Local Highway		Large City

Traffic Selections

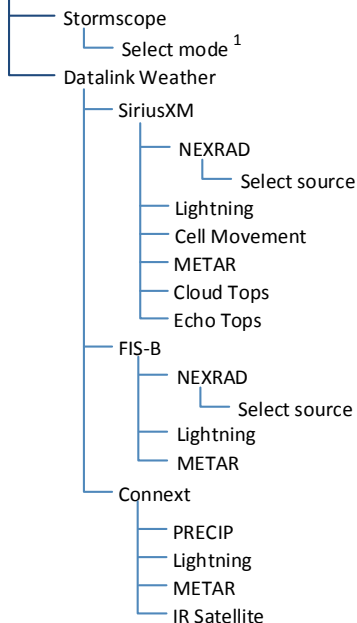
Traffic



Filter traffic data according to type. Other selections include on/off and range setting options.

Weather Selections

Weather



Setup options are available for all active weather services.

- Alternate between lightning display modes¹
- Specify a datalink weather source
- Toggle individual weather products on/off

¹ WX-500 only. Mode selection affects setting on dedicated weather page.

Charts



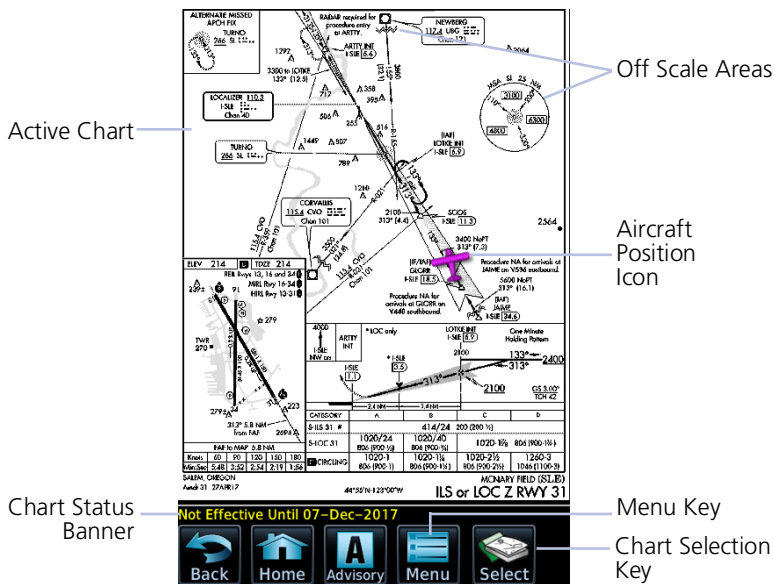
The Charts page provides terminal procedures and airport surface diagrams.

FEATURE REQUIREMENTS

- A current and valid chart database

FEATURE LIMITATIONS

- Geo-referencing is not available for some arrival and departure charts
- FliteCharts and ChartView databases are optional and mutually exclusive



Charts Page

CHART STATUS

Active chart status displays in the lower left corner of the Charts page.

Getting Newer Charts

Charts are in the process of updating.

Charts Changed

Charts are up to date.

Expired 28-Apr-2016

Database is out of date.

Not Effective Until 31-Mar-2016

Database is installed before it is current.

Chart Setup

Charts Menu

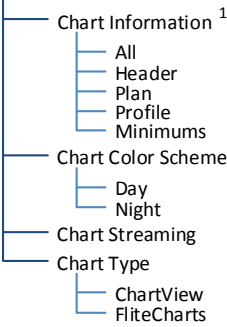


Chart setup selections reside in the Charts menu.

¹ ChartView only.

Chart Information	Displays individual sections of a chart in the ChartView database only. Options include: • All • Header • Planview • Profile • Minimums
Chart Color Scheme	Toggles chart color scheme between day and night modes. • Day mode displays black on white background • Night mode displays inverse white on black background
Chart Streaming	• Enables automatic streaming of individual charts to the GDU during database synchronization • Streams from the newest chart database • Charts from the previously loaded database remain available for display • Toggling this function off does not affect Database SYNC • No restart required
Chart Type	• Toggles chart database type between FliteCharts and ChartView • Selections must be made on the ground • Restart required for changes to take effect

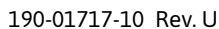
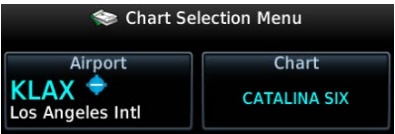


Chart Selection



To open a menu, tap **Select**.
Enter an airport identifier and choose from the available chart types.

CHART TYPES	
Information	Airport surface charts, alternate minimums, rate of climb and descent tables, inoperative components or visual aids tables, hot spots, land and hold-short operations, and other destination specific flight data.
Departure	Departure route descriptions, including takeoff minimums and obstacle notes.
Arrival	Arrival procedure instructions. Arrival charts are listed in alphabetical order.
Approach	All approach charts associated with the selected airport. Charts are listed according to level of precision and approach type. Runways are listed in numerical order.

AUTOMATIC CHART SELECTION

Automatic chart selection occurs every time the Charts page is opened. The type of chart selected is determined by aircraft status and content from the active flight plan. This function does not override manual chart selections unless a change occurs during one of the following conditions.

CONDITION	CHART DISPLAY DEFAULT
No flight plan or destination airport is present.	Nearest airport diagram.
Aircraft is on the ground.	Nearest airport diagram (regardless of flight plan).
Flight plan is present, but approach is not loaded.	Airport surface chart for previous airport in flight plan.
Approach is loaded in the selected GPS navigator.	Applicable approach chart.
No charts are available for a selected airport.	"Chart not available for <airport ID>."

Aircraft Position Icon

FEATURE REQUIREMENTS

- Aircraft position is fully within chart boundaries
- FliteCharts or ChartView is active (airborne maneuvers only)
- SafeTaxi is active (ground maneuvers only)

FEATURE LIMITATIONS

- SIDs and STARs do not support the display of aircraft position
- Not available for arrival and departure charts
- Displays only within the planview area of geo-referenced approach charts

Aircraft position displays on airport diagrams and approach charts when the current position is known.

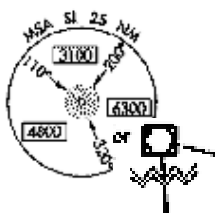
POSITION UNAVAILABLE INDICATION



A fault icon displays in the lower right-hand page corner when:

- Aircraft position data is not available
- Backup GPS is in use
- Selected chart does not contain geo-referenced data
- Selected chart layer is not All or Planview

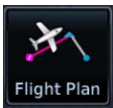
OFF SCALE AREAS



Some charts contain supplemental information within the planview. These areas are considered off scale as they are not geographically accurate.

Aircraft position depictions in off scale areas should be judged relative to the geo-referenced information within the planview only.

Active Flight Plan



Current flight plan information received from a Garmin navigator displays as a scrolling list on the Active Flight Plan page.

FEATURE REQUIREMENTS

- External GPS navigator
- Configured GTN, v6.50 or later (VNAV constraint data)
- Active flight plan

FEATURE LIMITATIONS

- Create, edit, and delete functionality not available on GDU. Any alterations to a flight plan must be performed using the external navigator
- Displays up to 100 waypoints for an active flight plan

Waypoint Identifier Column

Procedure Header

Active Leg Indicator

Current Waypoint

Waypoint Type Icon

Active VNAV Constraint Data

Waypoint Identifier Column

Active Flight Plan

ALT FPA DIS

Approach – KLAX ILS 07L

APT Info

MERMA iaf ▲

Merma

DITDA ▲

Ditda

1800 FT

0.00°

FEKIL ▲

Fekil

1800 FT

-3.00°

3.0 NM

TURKA faf ▲

Turka

1800 FT

0.00°

6.0 NM

RW07L map ▲

1800 FT

0.00°

5.0 NM

Leg Data

Selectable Data Field Columns

Airport Info Key

VS TGT -796 FPM

VS REQ -330 FPM

V DEV -550 FT

TOD 00:41

FPA -3.00°

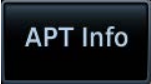
Active VNAV Constraint

FEKIL 1800 FT

Selecting a waypoint identifier opens the associated info page.

Active Flight Plan Page

AIRPORT INFO



For convenience, airport information is directly accessible from the procedure header. This includes airports specified in active approaches, arrivals, and departures.

ACTIVE VNAV CONSTRAINT WINDOW

VNAV Constraint Data

ALT - Altitude Constraints

VS TGT - Vertical Speed Target

VS REQ - Vertical Speed Required

V DEV - Vertical Deviation

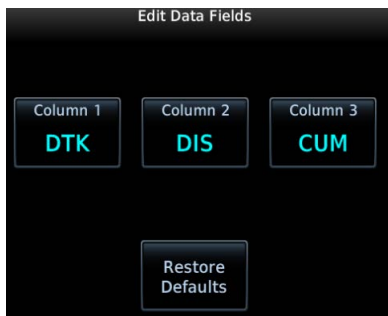
TOD/BOD - Top/Bottom of Descent

FPA - Flight Path Angle

VS TGT and FPA are pilot-specified values received from the GTN. All other constraint values are based on VNAV calculations.

The TOD field may change to display BOD once the top of descent value counts down to zero.

Edit Data Fields



To select a flight plan data column, tap **Edit Data Fields**. Columns are arranged in numerical order (1 - 3).

To restore columns to default display settings, tap **Restore Defaults**.

Data Field Selections

ALT - Altitude Constraint
 CUM - Cumulative Distance
 DIS - Distance
 DTK - Desired Track
 ETA - Est. Time of Arrival
 ETE - Est. Time En Route
 FPA - Flight Path Angle
 SPD - Speed Constraints (Advisory)

Selections are identical for each column.

By default, flight plan information fields display:

Column 1: DTK

Column 2: DIS

Column 3: CUM

Collapse All Airways

Airways automatically display as flight plan legs. A single airway may contain numerous legs. Airways without an active leg collapse for simplification. This does not affect airway legs shown on the external navigator(s).

Airway Indication

Expanded Airway Detail

Airway - V83.SAF

MASSA	177°	6.0 NM	988 NM
LYSSA	177°	8.0 NM	996 NM
ZOTOS	177°	7.6 NM	1003 NM
TAS Taos	177°	23.0 NM	1026 NM

All airways begin with an indicator field and end with an exit identifier.

Airway Collapsed Indication

Exit Waypoint

Airway - V83.SAF (collapsed)

SAF Santa Fe	---	109 NM	1091 NM
-----------------	-----	--------	---------

To hide all waypoints along an airway, but not the airway's exit waypoint, tap **Collapse All Airways**.

VNAV Guidance Indications



WARNING

Do not rely solely on VNAV guidance when navigating horizontally and vertically around user-defined airports. It is the pilot's responsibility to ensure separation from terrain and obstacles during an approach to a user-defined airport.

ALTITUDE CONSTRAINT INDICATIONS

COLOR DEFINITIONS	
White	Altitude is for reference only.
Cyan	GTN honors the constraint for vertical guidance when the VNAV function is active.
Magenta	Active constraint.

When the VNAV function is active on GTN, altitudes may be accompanied by one or two altitude restriction bars. The position of the value (above or below the bar, or between two bars) denotes the required aircraft altitude relative to that constraint.

Dual values annunciate when the aircraft needs to cross between two altitudes. Constraint values display in MSL or flight level (FL). Constraints at airports may be specified as MSL or AGL.

Examples:

Cross at or Above 5,000 ft

5000 FT

Cross at or Below 5,000 ft

5000 FT

Cross at 5,000 ft

5000 FT

Cross Between 5,000 ft and 6,000 ft

**6000 FT
5000 FT**

ALTITUDE TYPES



Database Constraint

Altitude is retrieved from the navigation database and designated for use in determining vertical guidance.



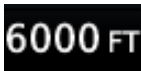
Pilot-specified Constraint

Pencil icon indicates manual designation or manual data entry.



Invalid Altitude Constraint

GTN cannot use the altitude to determine vertical guidance.



Estimated Crossing Altitude

For reference only.

System calculated estimate of aircraft altitude as it passes over the navigation point. Absence of bar(s) indicates it is not a potential constraint.



Published Reference Altitude

For reference only.

Altitude is retrieved from the navigation database, but not for use in determining vertical guidance. Bar above and/or below the value indicates constraint type.

**Altitude Constraint
Data Priority**

GDU receives constraint data from the GTN and displays it in the ALT column of the active flight plan. GTN prioritizes the data for each navigation point in the following order:

1. GTN honored constraints (includes invalid constraints)
2. Reference-only estimated crossing altitudes
3. Published reference altitudes
4. Empty (no altitude data)

INVALID ALTITUDE CONSTRAINTS

An altitude constraint is invalid if:

- Meeting the constraint requires the aircraft to climb
- Meeting the constraint requires the aircraft to exceed the maximum flight path angle (6° downward) or maximum vertical speed (-4,000 fpm)
- It results in a TOD behind the aircraft's current position
- It is within a leg type that does not support altitude constraints
- It is added to a waypoint past the FAF

Speed Constraint Indications



NOTE

The system uses jet aircraft reference speeds for procedures containing multiple speed constraints. Always verify airspeeds when loading a procedure for another type of aircraft.

FEATURE REQUIREMENTS

- GTN Xi series navigator with software v21.02 or later
- VNAV function enabled by installer

Like altitude constraints, speed constraints may be accompanied by one or two restriction bars. The position of the value (above or below the bar, or between two bars) denotes the required airspeed relative to that constraint.

Speed constraint values may display in knots, KPH, MPH, or Mach (M) based on configuration and current mode selection (IAS or Mach). Unlike altitude constraints, speed constraint values always appear white (reference only).

Examples:

Cross at or Above 200 kt



Cross at or Below 200 kt



Cross at 200 kt



AIRSPEED TYPES



Published Reference Speed

Airspeed is retrieved from the navigation database. Bar above and/or below the value indicates constraint type.



Pilot-specified Constraint

Pencil icon indicates manual designation or manual data entry.



No Constraint Value

Dashes indicate when airspeed is not available, either from the navigation database or by manual entry.

Speed constraints are *not* VNAV data, nor do they affect VNAV descent calculations.

Flight Plan Map Overlays

Leg Status Indications

LEG STATUS	COLOR
Active	Magenta
Next & Future	White
Past or Inactive	Gray

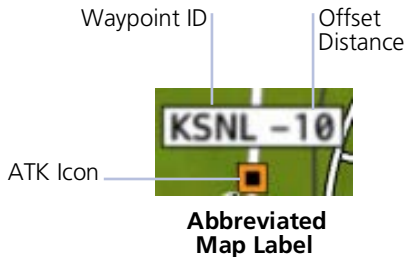
Active, next, and previous flight plan legs overlay on the Map page and HSI Map. These indications are display only.



Along Track Offset Indications

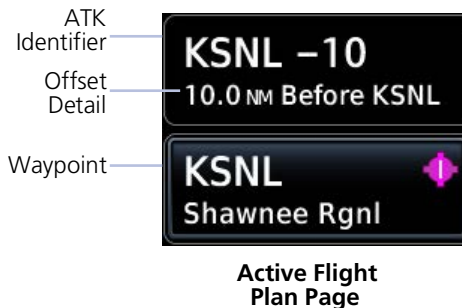


The along track waypoint (ATK) symbol represents a temporary lateral position (or checkpoint) relative to an existing waypoint in the flight plan.



Identifier labels on Map denote the adjacent waypoint's ID and offset distance from the specified ATK.

Selecting an ATK displays its bearing and distance in the info banner.



Unlike waypoints, ATKs indicate a temporary route fix in the flight plan. The GDU receives this data from the navigator, not a database.

Enabling the altitude constraint function in the Map Setup menu adds a VNAV altitude reference to the flight plan. The ATK serves as a lateral checkpoint when the feature is off.

Altitude Constraint Labels



Altitude
Constraint
Label

Altitude constraint data display as text labels on Map. Units are typically feet or meters depending on current altitude setting in the System Units app. They display as flight level altitudes if specified as such on the navigator.

If excessive labels are cluttering the map, this feature may be turned off.

Speed Constraint Labels



Altitude
Constraint
Label with
Speed
Constraint

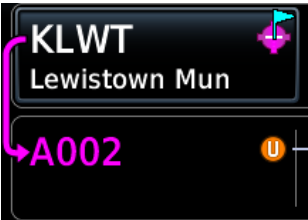
While not VNAV data, speed constraint text labels may appear on Map along with altitude constraint labels.

Units are typically knots, KPH, or MPH depending on configuration. They display as a Mach value if specified as such on the navigator.

User Airport Symbol

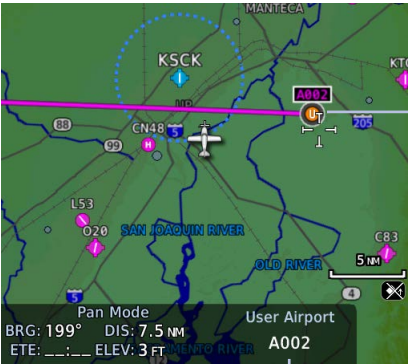


A dedicated symbol indicates user created airport waypoints received from the external GPS navigator.



User airport indications display on Active Flight Plan and Map.

Identifier Type Icon



Selected User Airport

When selected, the user identifier annunciates in the info banner.

User Identifier

Fly-over Waypoint Symbol



This symbol appears on instrument procedures when a waypoint is coded as a *fly-over* in the navigation database.

FEATURE REQUIREMENTS

- TXi software v3.50 or later
- GTN Xi v20.30 or later



Identifier Type
Symbol

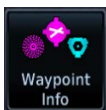
The navigator automatically creates a course that takes into account the waypoint type: fly-over or fly-by.



Fly-over Map
Symbol

For information about fly-over and fly-by waypoints, consult the AIM.

Waypoints



Dedicated information pages provide waypoint search functions and details not available on the Map page.

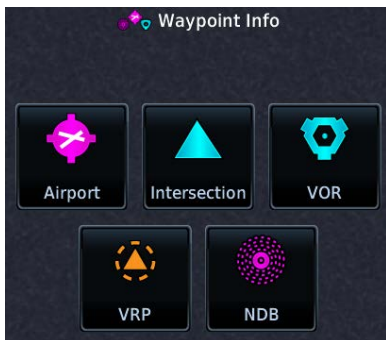
Waypoint Information

FEATURE REQUIREMENTS

- Viewing NOTAMs requires ChartView or FIS-B

FEATURE LIMITATIONS

- 2-D maps provide zoom functionality only (panning is not available)
- FIS-B transmits distant and FDC NOTAMs within 100 nm of radio station position



Waypoints are organized into five groups:

1. Airports
2. Intersection
3. VOR
4. VRP
5. NDB

The Airport page is a great place to start while performing an approach brief, checking weather, or considering a diversion.

Intersection, VOR, VRP, and NDB information pages have a uniform layout.



COMMON PAGE FEATURES

All waypoint information pages share the following features.

Data fields:

- Distance and bearing from current aircraft position
- Latitude and longitude
- Applicable country and/or region (e.g., "SW USA")
- 2-D map of the surrounding area (Map views display on Preview and Runways tabs of the Airports page)
- Identifier and type icon

Controls:

- **Waypoint Identifier** key with access to multiple search tab

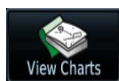
WAYPOINT SPECIFIC PAGE FEATURES

The following features are unique to the corresponding waypoint.

Airport

Selectable tabs:

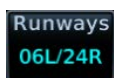
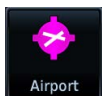
Info: Airport location, elevation, time zone, airport charts, and fuel availability.



To open the specified airport chart, tap **View Charts**.

Preview: SafeTaxi depiction of airport and surrounding area.

Runway: Identifiers, size, surface type, traffic pattern direction, and PCL frequency.



To open a list of available runways, tap **Runways**.

Frequencies: Available communication and localizer frequencies. The "c" symbol denotes frequencies that function as the CTAF. If available, for additional frequency information, tap "**More Information**."

Weather Data: METARs, city forecast, and TAF weather information for an airport.

NOTAMs: Applicable distant and FDC NOTAMs.

Airport Directory: Airport arrival information including local information based on AC-U-KWIK or AOPA data.

Intersection

Data fields:

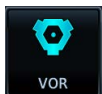
From NRST VOR: identifier, type icon, bearing, and distance



Very High Frequency Omni-directional Range

Data fields:

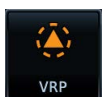
- Frequency
- To NRST APT: identifier, type icon, bearing, and distance
- Station Declination
- VOR Class



Visual Reporting Point

Data fields:

To NRST APT: identifier, type icon, bearing, and distance



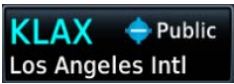
Non-Directional Beacon

Data fields:

- Frequency
- To NRST APT: identifier, type icon, bearing, and distance
- Marker Description

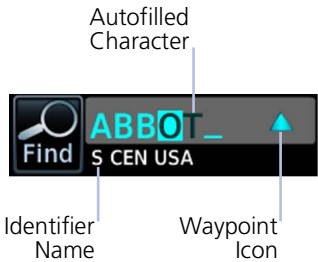


Waypoint Selection



The **WP Identifier** key provides access to different waypoint search options. Enter a specific identifier or select one from the available search tabs.

Waypoint Autofill



The autofill function assigns alphanumeric values into search entry fields.

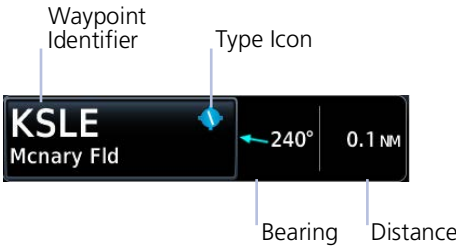
Autofill characters are cyan and display from cursor position to the right of the field.

Characters autofill based on the first alphabetical match in the navigation database. Matching entries typically change with the addition of a each typed character. “No matches found” and “Duplicate found” annunciate when applicable.

Search Tabs



The **Find** key provides access to multiple search tabs. Each tab displays a list of identifiers based on specific criteria.



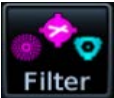
Each entry includes general information about the associated waypoint.

RECENT

Lists up to 20 of the most recently viewed waypoints.

NEAREST

Lists up to 25 waypoints within a 200 nm radius.

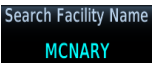


Tap **Filter** and select from the available filter options. Only waypoints belonging to the selected class appear in the list.
To list all classes, select **All**.

FLIGHT PLAN

Lists all waypoints contained in the active flight plan.

SEARCH NAME



Lists all airports, NDBs, and VORs associated with the specified facility name. Tap the key and enter the facility's name.

SEARCH CITY



Lists all airports, NDBs, and VORs found in proximity of the specified city. Tap the key and enter the city's name.

Remote Radio Frequency Entry

FEATURE REQUIREMENTS

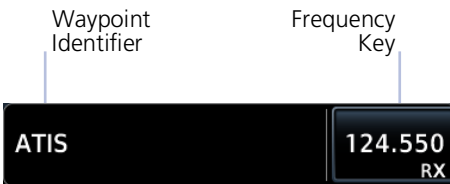
- *GTN Xi series navigator*
- *TXi software v3.21 or later*



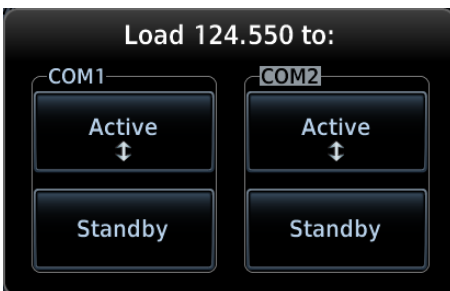
Frequencies Tab

When connected to a GTN Xi series navigator, radio frequencies are automatically available for remote radio entry.

You can load active or standby frequency values to a GTN Xi COM or NAV radio from a search tab or waypoint information page (e.g., Airport, VOR).



Selecting a frequency from one of these locations opens a pop-up.

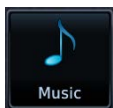


Select the **Active** or **Standby** key for the appropriate radio. The frequency transfers to the selected radio.

In the event of a failure, a pop-up informs you that frequency transfer did not occur.

COM 2 and NAV 2 are available in dual GTN Xi installations only.

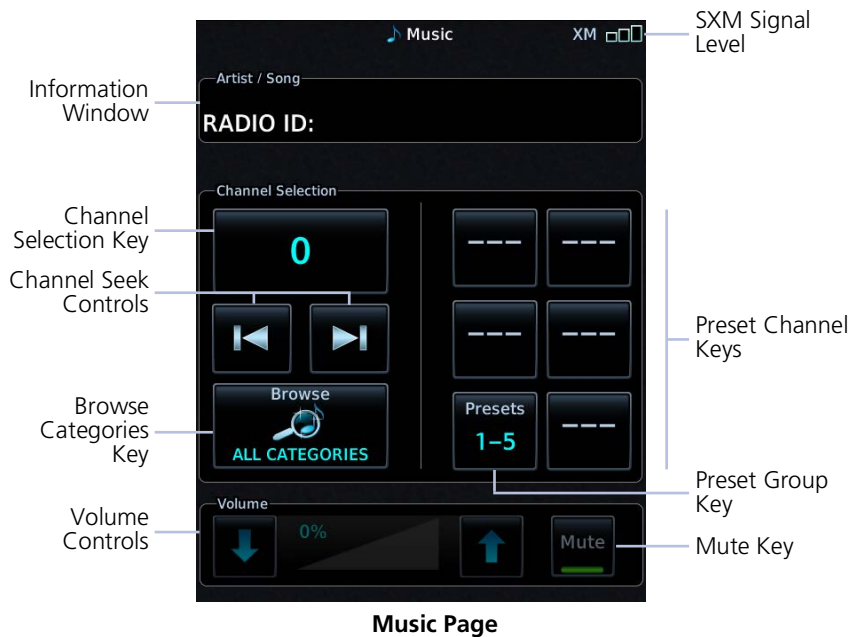
SiriusXM Audio Entertainment



The Music page provides controls for tuning and presetting satellite radio music channels.

FEATURE REQUIREMENTS

- Active SiriusXM Satellite Radio subscription
- Audio Radio and/or Data Radio ID
- Configured GDL 69/69A interface



RADIO VOLUME



Directional keys allow volume adjustments.

Mute toggles radio audio output on or off.

SiriusXM Audio Activation



The GDL 69/69A Status page provides information necessary for activating SiriusXM Satellite Radio services. To access this page, navigate to the External LRUs list, or tap **Status**.

For information, consult *GDL 69/69A SiriusXM Satellite Radio Activation Instructions*. For subscription activation instructions, visit <https://www.siriusxm.com/aviation>.

GDL 69A Status	
Data ID CCR060MZ	Audio ID JVVG60WY
Subscription Level Unknown	Version: 4.01.00 S/N: 47753142

If the Audio Radio ID and/or Data Radio ID do not display during GDL 69 operation, contact a Garmin dealer.

Browse Music Channels

SiriusXM Radio Options

- Channels 0 to 999
- Categories 0 to 63

Music provides both numeric entry and channel seek functionality.

Audio Category Selection



To filter the list of available channels according to a specific music type, tap **Category**.

Browse Music Channels

2

James Arthur

Say You Won't Let Go

Hits 1

3

Selena Gomez

Kill 'Em With Kindness

Venus

4

Zay Hilfiger/Zayion McCall

JuJu On The Beat

Pitbull

5

Tico & the Triumphs

Motorcycle

50s on 5

6

Kai Winding

More (63)

60s on 6

7

Bee Gees

More Than A Woman (78)

70s on 7

8

Huey Lewis & The News

The Power Of Love (85)

80s on 8

Active Channel

8

All Categories

Current Category Selection

Current Channel Selection

Category Key

Direct Channel Tuning



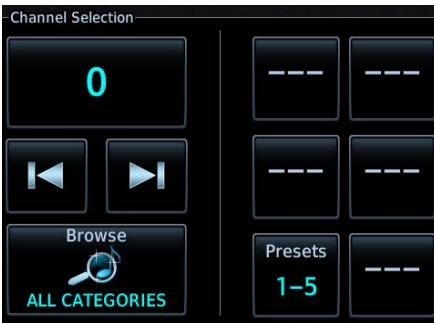
To enter a specific radio channel by number, tap **Channel**.

ASSIGNING CHANNEL PRESETS

Quick Tuning Options

- Six preset groups
- Up to three preset channels per group

For quick tuning, use presets to store multiple channels in the unit's memory.



Assign a Channel Preset

1. Select a channel.
2. Tap **Presets** until desired preset group displays.
3. Tap and hold any preset key for three seconds.

Select a Preset

1. Select a preset group.
2. Tap any preset key.

External Video

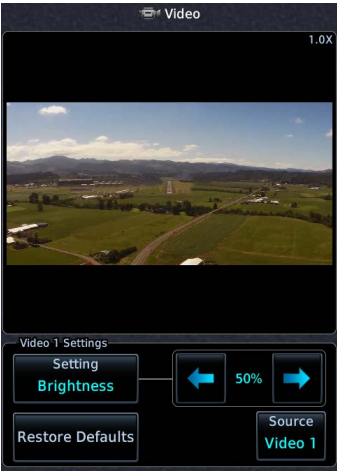


The Video app displays live video from up to four mounted cameras.

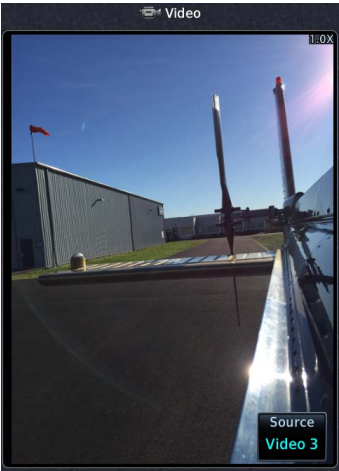
FEATURE REQUIREMENTS

- Composite or HD video camera input
- Purchased External Video feature enabled on GDU

SD Display (Control Window Active)



HD Display



For GDU 1060/1210 with dual MFDs, external video can be viewed on only one MFD at a time.

INPUT SOURCES

FORMAT	LIMIT
SD	2
HD	2

Two inputs are available for each video format. If an input source is not available, “No Video Input Signal” annunciates in place of a live image.



Camera designations are configured during installation. Video source can also be controlled using discrete inputs.

PAN & ZOOM



Zooming magnifies the live image up to 10x its range. Zoom settings apply only to the active camera display. Panning allows movement of the zoomed image in any direction.

SD Video Setup

SD Display Settings

- Brightness
- Contrast
- Saturation

Selectable display settings are available for SD video only. HD display settings are configured at installation.

Settings Key



To open a control window, tap **Settings**.



To return to normal display mode, tap **Full**.

SETTINGS ADJUSTMENT



Active Setting

Decrease Level

Increase Level

Directional keys allow adjustment of the active setting. Setting values are adjustable for each SD source.

RESTORE DEFAULTS



- Restores the active video display settings configured at installation
- Requires pilot confirmation

6 Weather Awareness

WEATHER DISPLAY 6-2

DATALINK WEATHER 6-5

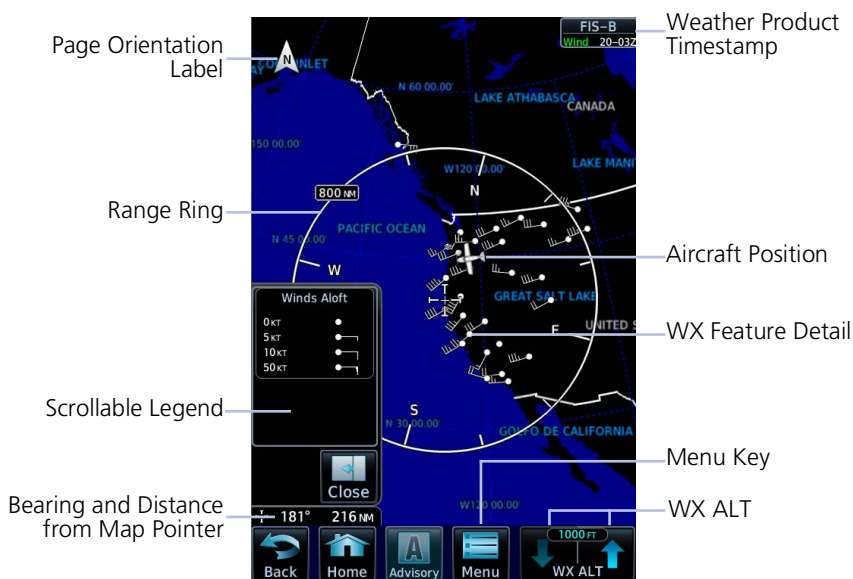
STORMSCOPE..... 6-43

AIRBORNE WEATHER RADAR 6-51

Weather Display



Weather data displays as overlays on the dedicated weather page, Map page, and HSI Map. Available functions depend on the active weather source(s).



Common Weather Elements

Weather Products

WEATHER PRODUCT	WEATHER DISPLAY				
	MFD WX PAGE	MFD WX POP-UP TEXT	HSI MAP	MFD MAP PAGE	WAYPT INFO (ARPT)
AIRMET (Textual)	X, C	X, C			
AIRMET (Graphical)	F, X, C	F, X, C			
Cell Movement/SCIT	X	X		X	
Center Weather Advisory	F	F			
City Forecast	X	X			X
Cloud Top/IR Satellite	F, X, C			X, C	
County Warnings	X	X			
Cyclone Track	X				
Echo Tops	X			X	
Freezing Levels/Zero Degree Isotherm	X				
Icing Potential/SLD	F, X				
Lightning (Datalink)	F, X, C		F, X, C	F, X, C	
Lightning (Stormscope)	S		S	S	
METAR w/Decoding	F, X, C	F, X, C		F, X, C	F, X, C
NEXRAD/PRECIP	F, X, C		F, X, C	F, X, C	
NOTAMs					F
AIREP	X	X			
PIREP	F, X, C	F, X, C			
SIGMET	F, X, C	F, X, C			
Surface Analysis	X				
TAF	F, X, C	F, X, C			F, X
Temps Aloft	F				
TFR	F, X, C	F, X, C	F, X, C	F, X, C	

WEATHER PRODUCT	WEATHER DISPLAY				
	MFD WX PAGE	MFD WX POP-UP TEXT	HSI MAP	MFD MAP PAGE	WAYPT INFO (ARPT)
Turbulence	F, X				
Weather Radar (on-board) ¹	W			W	
Winds Aloft	F, X, C				

¹ Not available on GDU 700() MFD/EIS.

LEGEND

Connex Product	C
FIS-B Product	F
On-board WX Radar Equipment	W
SiriusXM Product	X
Stormscope Product	S

Datalink Weather



WARNING

Do not rely solely on datalink weather for weather information. Datalink weather provides a snapshot in time. It may not accurately reflect the current weather situation.



NOTE

Datalink weather is not intended to replace weather briefings or in-flight weather reports from FSS or ATC.

FEATURE REQUIREMENTS

- FIS-B, GSR 56 transceiver, GDL 69/69A, or GDL 69/69A SXM receiver

Datalink weather services are available through FIS-B Weather, SiriusXM Weather, and Connex WX that transmit from a ground station or satellite.

Weather Page Interactions

ALTITUDE ADJUSTMENT KEYS

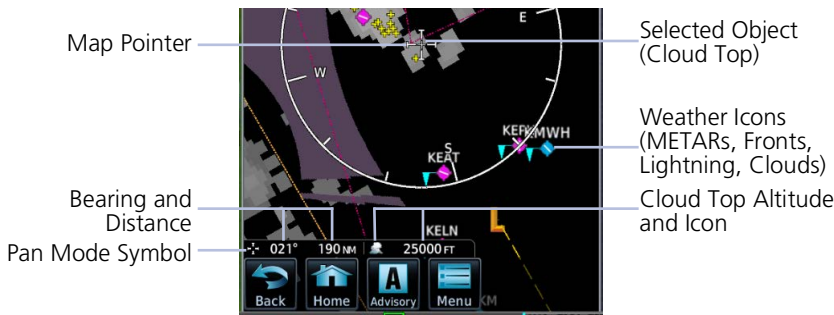


Plus and minus keys adjust altitude in increments for altitude-based weather products. Selectable altitude values vary by weather service and product. Includes:

- Icing • Turbulence • Winds Aloft

WEATHER ICONS

Tapping any weather icon displays an information banner. When applicable, a pop-up window displays additional data.



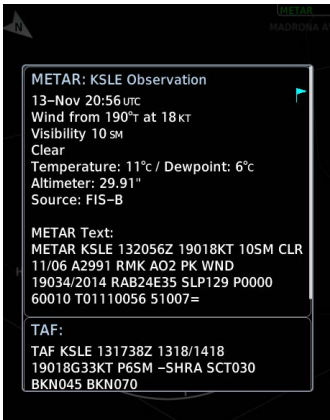
WX Info Banner Features

- Pan mode symbol
- Bearing and distance to map pointer from aircraft's current position
- Cloud Top altitudes
- Pan location
- Report data

AIRPORT ICONS



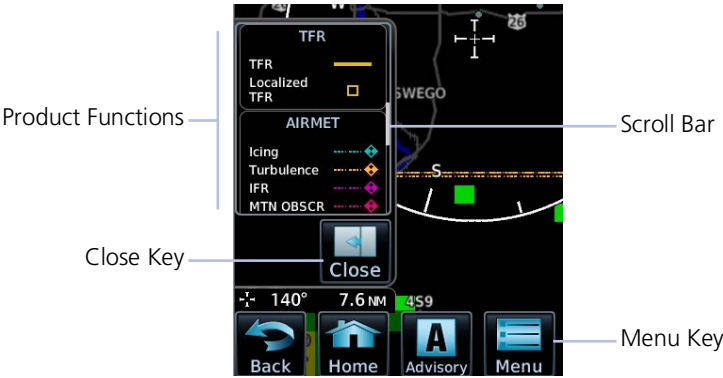
Tapping an airport icon displays information about local weather conditions.



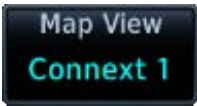
PRODUCT LEGENDS

A scrollable legend provides relevant product colors and symbols via the datalink weather menu page. Definitions are organized by product function.

To view a product legend, tap the **Menu** key and select **Legend**.



Weather Map Views



The **Map View** key in a weather menu may be used to select a user-configured view.



Three different views may be individually configured for each weather product. All settings configured in the weather product’s menu are saved.

View configurations may be changed at any time.

Selecting any view loads previously saved view settings.

Tap **Customize** to change a view name or reset both the name and all menu settings to their default values.

WX Display/Map Settings

FEATURE LIMITATIONS

Shared map settings are per MFD instance, meaning that if you have GDU 1060/1210 with MFD 1 and MFD 2, each MFD has its own map settings. These settings are shared only with the datalink weather apps on that specific MFD. Map settings are not shared across multiple MFDs.

Several weather map display settings are based on selections made in the setup menu of the primary Map application.

From the Home page:

Tap **Map > Menu > Map Setup**, and select the appropriate tab. Applicable map settings are as follows:

Map Tab	Aviation Tab	Land Tab
Settings: • North Up Above range • LAT/LON Lines	Settings: • Runway Extensions • Airport Range • Intersection Range • NDB Range • VOR Range • VRP Range • User Waypoint Range	Settings: • Road Detail • City Detail • State/Province Borders • River/Lake Detail

Changes to these map settings take effect immediately on the configured datalink weather app(s).

Map Settings and Multiple Map Views

Shared map settings are independent of map views. Map settings will transfer to a datalink weather app regardless of which map views you select.

Weather Product Age

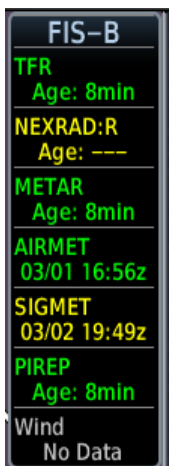


NOTE

Data contained within a composite weather product may be older than its weather product age and should never be considered current.

A timestamp identifies the approximate time of data collection for each weather product. For quick reference, the age of each active weather product is calculated and shown in a color-coded side bar on the Weather and Map pages.

TIMESTAMP COLOR DEFINITIONS



Green

Weather product is considered current. Its age is newer than half its expiration time.

Tapping the Timestamp window displays time for all green colored weather products.

Yellow

Weather product is considered stale. Its age is older than half its expiration time.

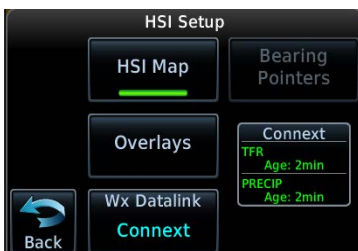
A weather product may be yellow when its issue date and time occurs in the future by more than the complete expiration time for the requested weather product. (e.g., some TFRs).

Gray

Weather product data is one of the following:

- Expired
- Not received
- Not supported at the selected altitude

“No Data” or “ALT UNAVBL” displays next to the weather product title.



On the PFD:

Timestamps for NEXRAD/PRECIP, lightning, and TFR products display in the HSI Setup menu.

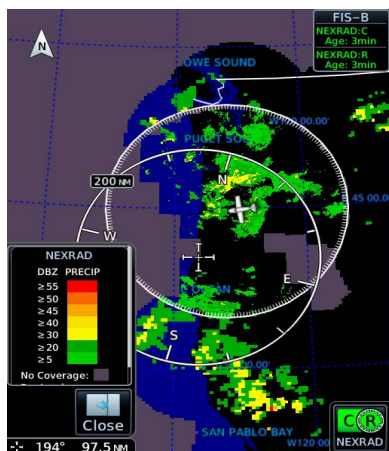
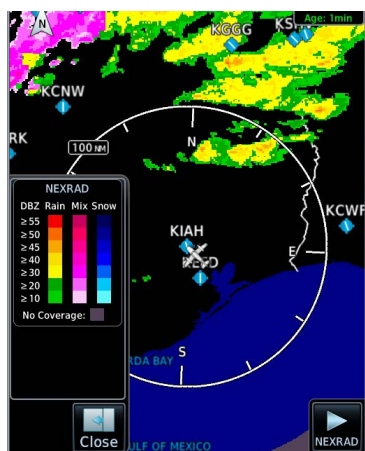
Precipitation

AVAILABLE WITH:
FIS-B | SIRIUSXM | CONNEXT



WARNING

Never use NEXRAD weather for maneuvering in, near, or around areas of hazardous weather. NEXRAD images are snapshots of past weather data. They are not safe for use as real time depictions of nearby weather activity.



NEXRAD weather radar displays a mosaic of precipitation data, colored according to reflectivity. Composite reflectivity images depict the highest radar energy received from multiple antenna tilt angles at various altitudes. Base reflectivity images depict radar returns from the lowest antenna tilt angle.

The precipitation intensity level reflected by each pixel represents the highest level of composite radar reflectivity data sampled in that location.

A clear understanding of ground-based Doppler weather radar capabilities will allow you to interpret the NEXRAD weather imagery in the safest way possible. The National Oceanic and Atmospheric Administration hosts a description of the technology on its website: <https://www.noaa.gov/jetstream>

RADAR DATA ANIMATIONS



To depict trending weather movements over time, an animation function stitches the last three to six received radar images together in sequence, from oldest to newest, and replays them on a continuous loop. Play and stop controls are active when three or more NEXRAD images are available for playback.

Connexx PRECIP

Precipitation products are available for areas around the world. For current coverage areas and product information, visit:

<https://fly.garmin.com/fly-garmin/connexx/worldwide-weather/>

FIS-B NEXRAD

FIS-B NEXRAD is uplinked to the aircraft as two separate weather products: CONUS and Regional NEXRAD. Both products display individually or simultaneously, separated by a white hash-marked boundary. CONUS and Regional NEXRAD are selectable separately via the NEXRAD button on the bottom right of the weather map. FIS-B NEXRAD does not differentiate between liquid and frozen precipitation types.

REGIONAL

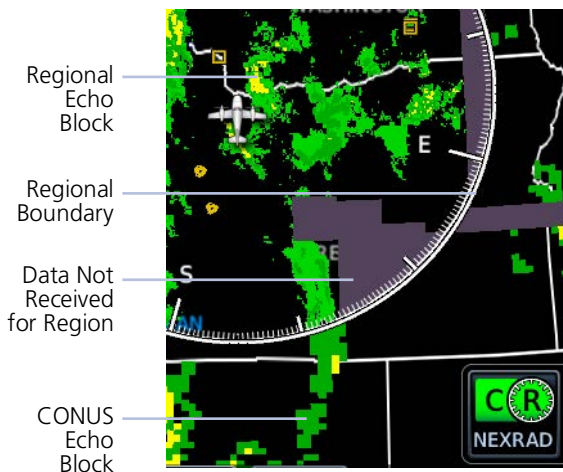
FIS-B Regional NEXRAD is a high-resolution weather image with a limited range, centered around each broadcasting ground station. The (high resolution) Regional NEXRAD pixels are 1.5 minutes (1.5 nautical miles = 2.78 km) wide by 1 minute (1 nautical miles = 1.852 km) tall. Each weather pixel varies with latitude. Above 60° latitude, pixel block widths double to 3 minutes/nautical mile for regional maps.

Depending on the locations of received FIS-B ground stations, Regional NEXRAD coverage can extend more than 250 nm around an aircraft's position. Aircraft flying at higher altitudes typically receive data from more ground stations than aircraft flying at low altitudes.

CONUS

FIS-B CONUS NEXRAD is a large, low-resolution weather image for the entire continental U.S. CONUS NEXRAD pixels are 7.5 minutes (7.5 nm = 13.89 km) wide by 5 minutes (5 nautical miles = 9.26 km) wide.

FIS-B Weather Display

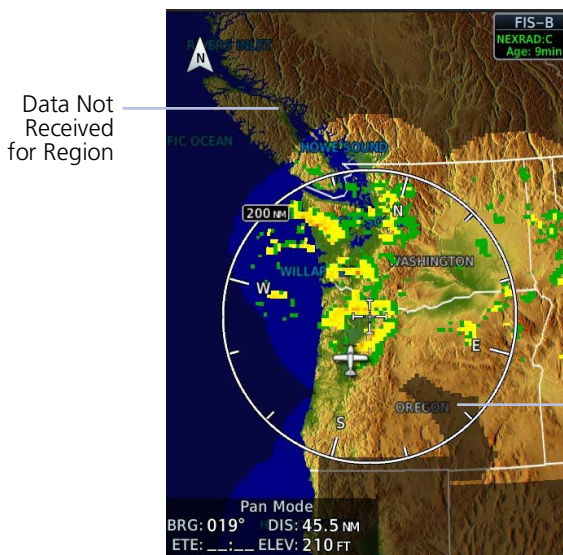


Missing Data Indications

A semi-transparent mask indicates regions where data may be missing or incomplete. Color is dependent upon the presence of topographical data.

The mask is purple when there is no underlying topographical data, such as on the FIS-B weather display, or on Map when the TOPO overlay is off.

MFD Map Display



On Map, the mask is dark gray when the TOPO overlay is on, so that the underlying topographical features can still be interpreted.

Data Not Received for Region

SiriusXM NEXRAD

SiriusXM offers NEXRAD radar imagery from CONUS and non-CONUS sources. Weather menu options are Composite and Base.



Depending on the SXM service and installed GDL hardware, radar base reflectivity imaging may have broader coverage within North America or be limited to only Canada.

Source Options:

Composite (CONUS)

Composite reflectivity image of radar for the conterminous U.S.

Base

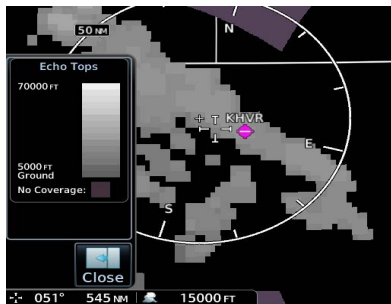
Base reflectivity image. Coverage and availability are dependent upon subscription.

Precipitation above 52° N may display as mixed precipitation regardless of actual precipitation type. Precipitation of an unknown type always displays as rain.

Echo Tops



AVAILABLE WITH: SIRIUSXM



Depicts the location, elevation, and direction of NEXRAD radar echoes. This information is useful for gauging storm intensity in a relative sense. A higher radar echo means a stronger storm updraft.

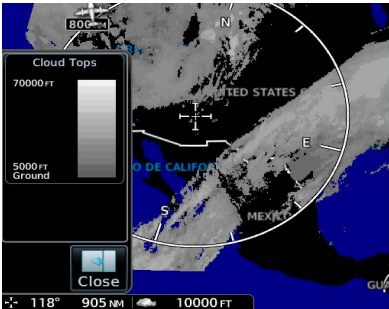
Due to similarities in color schemes, this product is mutually exclusive with Cloud Tops.

Clouds

Cloud Tops



AVAILABLE WITH:
FIS-B | SIRIUSXM



Indicates the altitude of the highest visible portions of a cloud at the time of measurement.

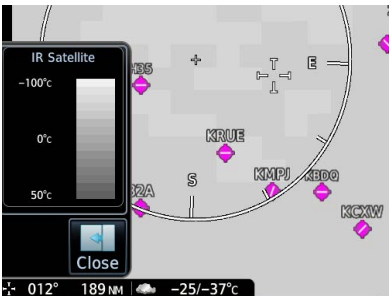
SXM cloud top altitude data is estimated from IR satellite imagery.

FIS-B cloud top data is generated by a computer model and has limited accuracy compared to actual conditions.

Due to similarities in color schemes, this product is mutually exclusive with Echo Tops.

IR Satellite

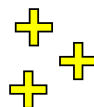
AVAILABLE WITH: CONNEXT



Infrared Satellite weather information is available for North America and Europe.

- Lighter grays depict colder temperatures
- Darker grays depict warmer temperatures

Lightning



AVAILABLE WITH:
FIS-B | SIRIUSXM | CONNEXT



SiriusXM Weather:

SXM displays a yellow cross icon to indicate a strike event occurring within a two kilometer region. Strike location is an estimate of its center.

Unlike Stormscope, the icon does not change shape or color as data ages.

Connex Weather:

Only cloud-to-ground strikes are reported through the Connex weather service.

FIS-B Weather:

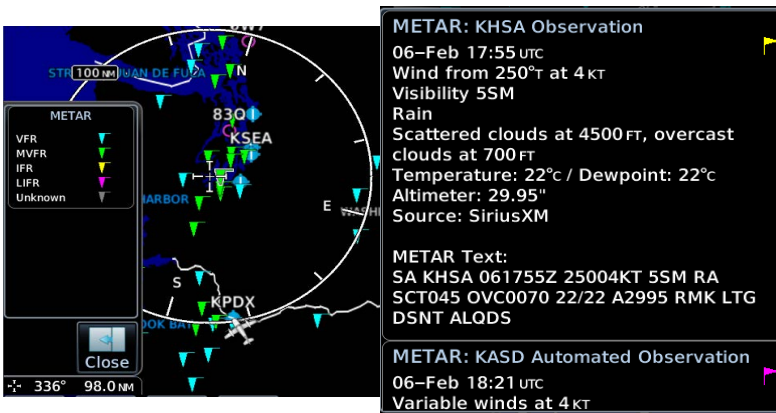
Lightning strikes display as a bolt or cluster of bolts. The bolt color indicates the strike polarity.

METARs & TAFs



AVAILABLE WITH:
FIS-B | SIRIUSXM | CONNEXT

METAR and TAF reports provide information on current and forecast conditions. Colored METAR flags display for all TXi weather sources when a METAR reporting station is matched with a corresponding TXi navigation database identifier. When selected, the METAR is decoded and shows above the original METAR text. METARs are provided only in areas covered by the navigation database currently loaded.



METAR SYMBOLS



VFR

Ceiling greater than 3,000 ft AGL and visibility greater than five miles.



Marginal VFR

Ceiling 1,000 to 3,000 ft AGL and/or visibility three to five miles.



IFR

Ceiling 500 to 1,000 ft AGL and/or visibility one to three miles.



Low IFR

Ceiling below 500 ft AGL and/or visibility less than one mile.



Unknown

Ceiling and/or visibility data unavailable.

Cell Movement



AVAILABLE WITH: SIRIUSXM

Information derived from NEXRAD displays the location of storm cells within a strong storm. Cell direction, speed, and storm top altitude is provided with a corresponding visual storm cell vector (arrow) indication.



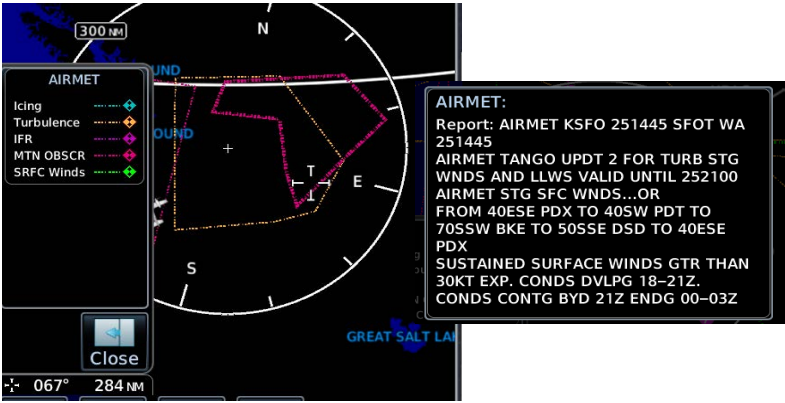
AIRMETS

Textual AIRMETS



AVAILABLE WITH:
SIRIUSXM | CONNEXT

The AIRMET overlay draws the geographical boundaries of received AIRMETS on the dedicated weather page. AIRMETS include advisories of weather that may be particularly hazardous to single engine, light aircraft and VFR pilots. Tap an AIRMET graphic area to view the detailed AIRMET report text.



AIRMET TYPES



Icing
Areas of moderate airframe icing including areal extent.



Turbulence
Areas of moderate turbulence including vertical extent.



IFR
Weather conditions with ceilings <1,000 ft and/or visibility <3 mi.



MTN OBSCR
Widespread mountain obscuration due to clouds, precipitation, smoke, haze, mist, or fog.



Surface Winds
Sustained surface winds >30 kt and/or potential non-convective wind shear below 2,000 ft AGL.

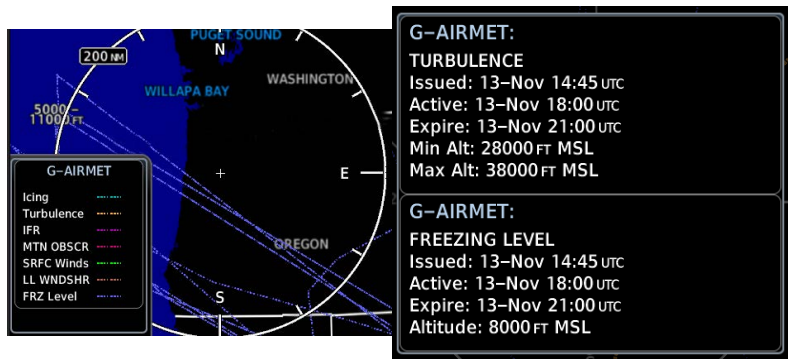
Graphical AIRMETs



AVAILABLE WITH: FIS-B

Graphical AIRMETs (G-AIRMETs) display more weather phenomena than textual AIRMETs, while eliminating the need to interpret raw text.

Tapping a G-AIRMET displays textual details. Updates occur four times daily.



Filtering options allow you to mitigate page clutter. Forecast and filter selections are accessible from the FIS-B Weather menu.

Forecast Settings:

Off	Current (Auto)	All Forecasts
Disables automatic functionality. Filter options not available.	Displays active graphical records based on the current UTC. Function automatically switches from 0 hr to 3 hr forecasts.	Displays the most recent, non-expired graphical records.

Forecast periods include 0 hr, 3 hr, and 6 hr.

G-AIRMET FILTERS

	Freezing Level		MTN OBSCR
	Icing		Surface Winds
	IFR		Turbulence
	Low-level Wind Shear		

To view graphical AIRMETs:

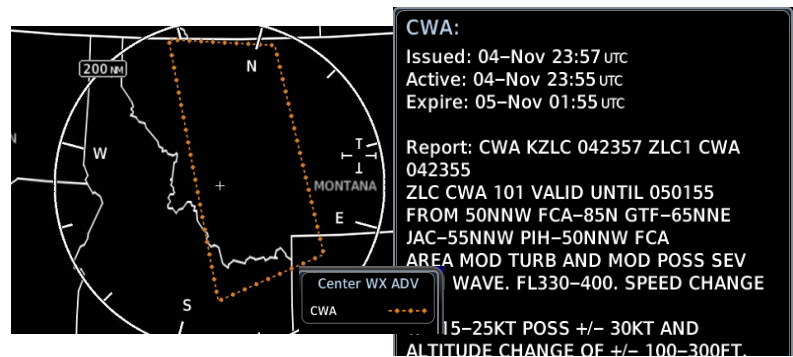
1. Home > **Weather** > **FIS-B Weather** > **Menu**.
2. Toggle **G-AIRMET** on.
3. Tap **Settings**.
4. Select between **Current (Auto)** and **All Forecasts**.
5. Select one or more filters.
6. Return to the FIS-B Weather page.
7. Select any G-AIRMET line to view details.

Center Weather Advisory



AVAILABLE WITH: FIS-B

These advisories communicate en route and terminal weather conditions expected to occur within the next two hours. Information is valid for up to 2 hours.

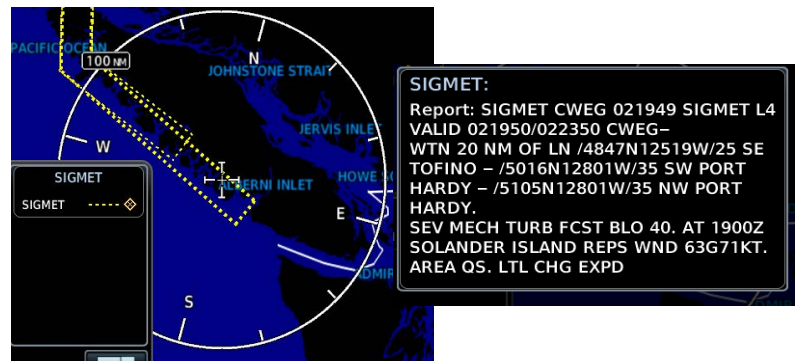


SIGMETs



AVAILABLE WITH:
FIS-B | SIRIUSXM | CONNEXT

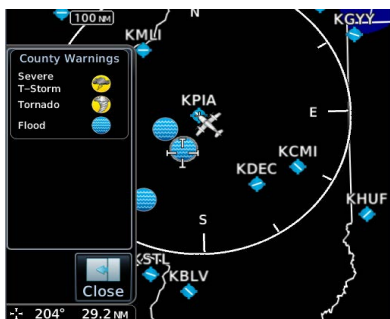
The SIGMET overlay draws the geographical boundaries of received SIGMETs on the dedicated weather page. Convective and non-convective SIGNificant METeological Information combine in a textual report and display graphically for the observed or forecast region. The SIGMET text displays when the graphical SIGMET is selected. Touch the graphical depiction for additional information.



County Warnings



AVAILABLE WITH: SIRIUSXM



County warnings are provided for the counties within the continental United States. Specific public awareness and protection alerts such as fires and natural disasters are included with warnings for tornadoes, thunderstorms, floods and flash floods.

Filtering options include Severe T-Storm, Tornado, and Flooding.

Cyclone/Hurricane Track



AVAILABLE WITH: SIRIUSXM



Provides hurricane and tropical storm information, showing location, forecast track and track errors as issued by the National Hurricane Center.

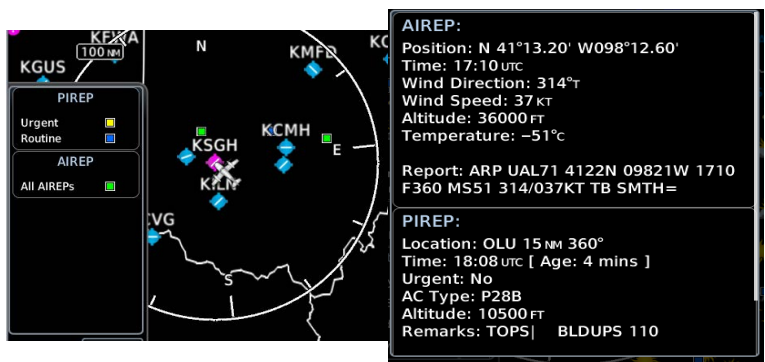
AIREP/PIREPs



AVAILABLE WITH:
FIS-B | SIRIUSXM | CONNEXT

AIREPs are routine, automated reports of in-flight weather conditions such as wind and temperature. AIREPs display with a green map icon. Selecting the icon displays a report window.

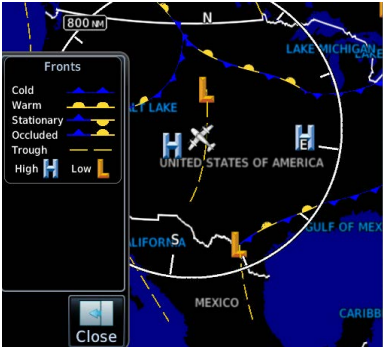
PIREPs are pilot-generated weather reports that may contain non-forecast adverse weather conditions, such as low in-flight visibility, icing conditions, wind shear, and turbulence. PIREPs are issued as either Routine (UA) or Urgent (UUA). Routine display with a blue map icon and urgent display with an amber icon. Selecting either icon displays a report window.



Surface Analysis



AVAILABLE WITH: SIRIUSXM



Displays weather fronts, high (H) and low (L) pressure centers, and isobars.

Forecast Period Options:

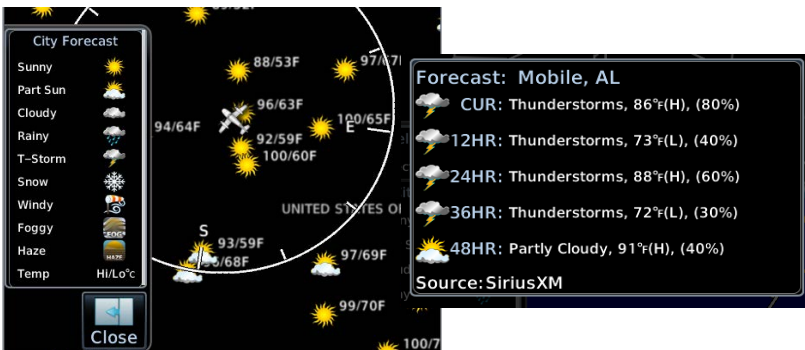
- Current • 12 hour • 24 hour
- 36 hour • 48 hour

City Forecast



AVAILABLE WITH: SIRIUSXM

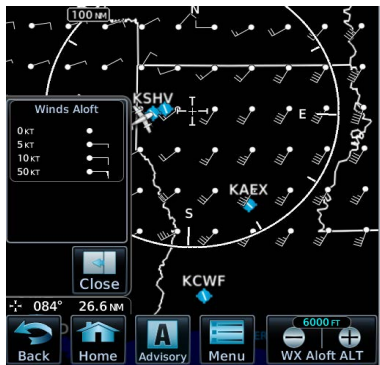
Four-day forecasts for major U.S. cities, including daily forecast high and low temperatures, chance of precipitation, and sky conditions.



Winds Aloft



AVAILABLE WITH:
FIS-B | SIRIUSXM | CONNEXT



Indicates wind speed and direction, and temperature forecast for the specified altitude.

The wind barb extends outward from the reporting location dot in the direction of wind origination.

Reporting altitudes vary depending on the weather provider.

FIS-B Winds & Temperatures Aloft:
FIS-B wind forecasts also provide temperature data for the selected altitude. These forecasts update every 12 hours.

Altitude Range:

FIS-B	SIRIUSXM & CONNEXT
1,000 ft	Surface to 45,000 ft (at 3,000 ft intervals)
1,500 ft	
2,000 ft	SiriusXM data not available at 45,000 ft
3,000 ft	
6,000 ft	
9,000 ft	
12,000 ft	
15,000 ft	
18,000 ft	
24,000 ft	
30,000 ft	
34,000 ft	
39,000 ft	
45,000 ft	
53,000 ft	

Not all altitudes provide winds and temperatures aloft forecasts for all regions.

Icing

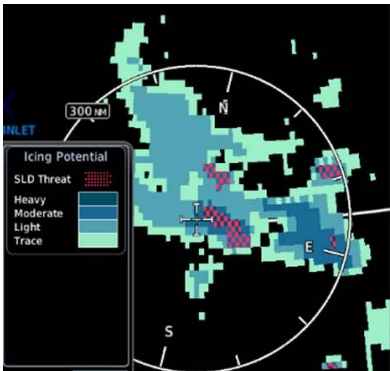


AVAILABLE WITH:
FIS-B | SIRIUSXM



NOTE

Due to the incremental and overlapping nature of the FIS broadcast, timestamps, regional coverage, and map data availability may vary with altitude for computer generated FIS-B icing forecasts.



The Icing potential shows a graphic view of the current icing conditions. SLD threat areas are depicted as black and pink blocks over the icing colors. Directional keys allow you to adjust the reporting altitude within product-specific limits.

	FIS-B	SIRIUSXM
Timestamp shows:	Valid time in UTC	Time of data compilation
Altitude Range:	2,000 to 24,000 ft (at 2,000 ft intervals)	1,000 ft 3,000 ft 6,000 ft 9,000 to 30,000 ft (at 3,000 ft intervals)

ICING CATEGORIES

	SLD Threat
	Heavy
	Moderate
	Light
	Trace ¹
	No Coverage ²

Categories vary depending on the configured weather source.

Icing potential is not a forecast, but a presentation of icing potential at the time of analysis. Supercooled Large Droplet (SLD) icing conditions are characterized by the presence of relatively large, super cooled water droplets indicative of freezing drizzle and freezing rain aloft.

¹ Available only from certain weather sources. ² FIS-B Weather only.

Turbulence



AVAILABLE WITH:
FIS-B | SIRIUSXM



NOTE

Due to the incremental and overlapping nature of the FIS broadcast, timestamps, regional coverage, and map data availability may vary with altitude for FIS-B turbulence forecasts.



Turbulence data identifies the potential for erratic movement of high-altitude air mass associated with winds. This information is intended to supplement AIRMETs and SIGMETs.

Turbulence is classified as light, moderate, severe or extreme.

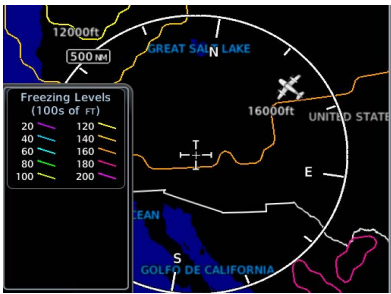
Directional keys allow you to adjust the reporting altitude within product-specific limits.

	FIS-B	SIRIUSXM
Timestamp shows:	Valid time in UTC	Time of data compilation
Altitude Range:	2,000 to 24,000 ft (at 2,000 ft intervals)	21,000 to 45,000 ft (at 3,000 ft intervals)

Data may not be available for all regions at all altitudes.

Freezing Levels

AVAILABLE WITH:
FIS-B | SIRIUSXM



Color-coded contour lines indicate the altitude of the freezing level.

FIS-B freezing levels display as a G-AIRMET. Updates occur four times daily.

TFRs



AVAILABLE WITH:
FIS-B | SIRIUSXM | CONNEXT



WARNING

Do not exclusively use datalink services for TFR information. TFR depictions on GDU may not be a complete listing and may vary between cockpit devices. Always confirm TFR information with official sources such as Flight Service Stations or Air Traffic Control.

TFRs identify areas of airspace where aircraft are temporarily restricted from operating. They are routinely issued for occurrences such as dignitary visits, military activities, and forest fires.

Tapping a TFR displays textual details.



FIS-B Weather:

Latest TFR data transmit every 10 to 20 minutes.

For FIS-B datalink weather, TFRs and NOTAMs that do not include geographical locations are viewable from the FIS-B Raw Text Reports page.

SiriusXM Weather



SiriusXM satellites deliver high bandwidth (S-band) data to provide high-resolution images. Available data is received within a few minutes after the aircraft satellite receiver comes into view of the SiriusXM satellite network.

FEATURE REQUIREMENTS

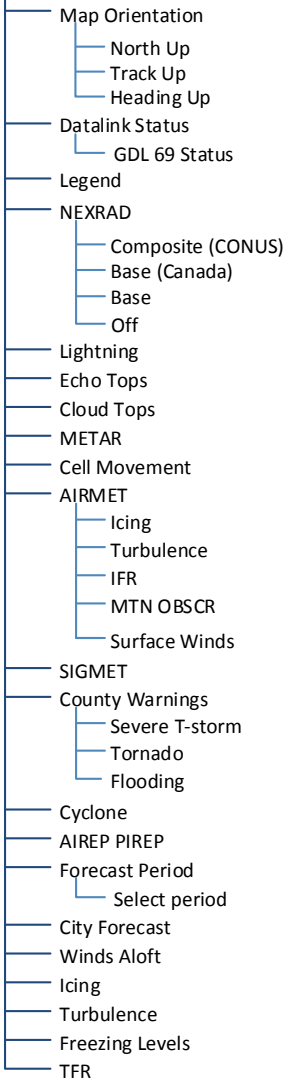
- *Active SiriusXM Aviation Weather subscription*
- *Data Radio ID*
- *Configured GDL 69/69A interface*

FEATURE LIMITATIONS

- *For Icing Potential, Winds Aloft (U.S.), and Turbulence, the valid time displays instead of product age*
- *For Freezing Level and Winds Aloft (Canada), the generation time displays instead of valid time*

SiriusXM Weather Setup

SiriusXM Weather Menu



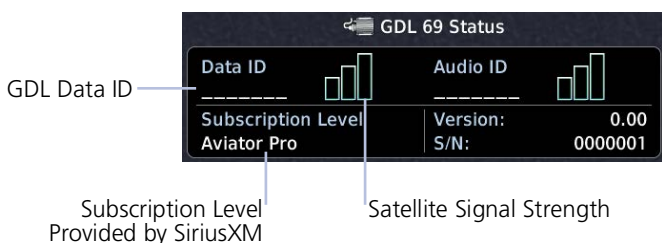
Activating Services

Before SiriusXM Weather is accessed, service must be activated by the SiriusXM customer service. This is accomplished by providing either one or two coded IDs, SiriusXM Satellite Weather (data) and SiriusXM Satellite Radio (audio). The coded IDs are used to send an activation signal, which allows the use of weather and entertainment products received by the GDL 69/69A.

For subscription activation instructions, visit <https://www.siriusxm.com/aviation>.

To complete activation:

1. Contact SiriusXM using the customer service phone number listed on the website.
2. Provide the customer service agent with your Data ID and Audio ID.
3. Position aircraft within satellite reception area and power unit.
4. Tap **Weather > SiriusXM Weather > Menu > Datalink Status**.
5. Verify activation. Status page displays Data ID, Subscription Level, and weather product listings.
6. Tap **Menu > Lock activation > OK**.



GDL 69 Status Page

SiriusXM Weather Products

AIREP	METAR
AIRMET	NEXRAD: Base
Canada METAR	NEXRAD: Base CVRG
Canada TAF	NEXRAD: Composite
Canada Wind	NEXRAD: CMP CVRG
City Forecast	PIREP
Cloud Top	SIGMET
County Warning	Storm Cell
Cyclone	Surface Analysis
Echo Top	TAF
Freezing Level	TFR
Icing/SLD	Turbulence
Lightning	Winds Aloft

Legend
 Unavailable Subscribed Data Available

When SiriusXM weather services are not activated:

- All weather product boxes are cleared on the GDL 69 Status page
- A yellow "Activation Required" message appears at the center of the page

Service Class refers to the groupings of weather products available for subscription.

FIS-B Weather



No pilot action is required to receive FIS-B weather information.

FEATURE REQUIREMENTS

- UAT receiver (GDL 88, GNX 375, GTX 345)

FIS-B weather products include:

- | | | |
|----------------------|--------------|---------------------|
| • NEXRAD | • Cloud Tops | • Lightning |
| • Icing | • Turbulence | • Winds/Temps Aloft |
| • METAR | • PIREP | • SIGMET |
| • Center WX Advisory | • G-AIRMET | • TFR |

The FAA provides FIS-B as a Surveillance and Broadcast Service operating on the UAT (978 MHz) frequency band. FIS-B uses a network of FAA-operated ground-based transceivers to transmit weather datalink information to the aircraft's receiver on a scheduled continuous basis.

The Flight Information Service-Broadcast (FIS-B) Weather service is freely available for aircraft equipped with a capable datalink universal access transceiver (UAT). Ground stations provide uninterrupted services for the majority of the contiguous U.S., Hawaii, Guam, Puerto Rico, and parts of Alaska. No weather subscription service is required. For coverage information, visit:

https://www.faa.gov/air_traffic/technology/adbsb

FIS-B weather product update and transmission intervals are published in the SBS Description Document associated with TSO-C157b. This information is available electronically via the FAA's Dynamic Regulatory System: <https://drs.faa.gov>

FIS-B Data Transmission Limitations

FIS-B broadcasts provide weather data in a repeating cycle which may take several minutes to completely transmit all available weather data. Therefore, not all weather data may be immediately present upon initial FIS-B signal acquisition.

LINE OF SIGHT RECEPTION

To receive FIS-B weather information, the aircraft's datalink receiver must be within range and line-of-sight of an operating ground-based transceiver. Reception may be affected by altitude, terrain, and other factors. Per the FAA, much of the United States has FIS-B In airborne coverage at and above 3,000 feet AGL. Terminal coverage is available at altitudes below 3,000 feet AGL and is available when flying near approximately 235 major U.S. airports. Surface coverage allows FIS-B ground reception at approximately 36 major U.S. airports.

PER FAA TSO-C157B

FIS-B information may be used for pilot planning decisions focused on updating the pilot's awareness of the dynamic flight environment; including avoiding areas of inclement weather that are beyond visual range and pilot near-term decisions where poor visibility precludes visual acquisition of inclement weather. FIS-B weather and NAS status information may be used as follows:

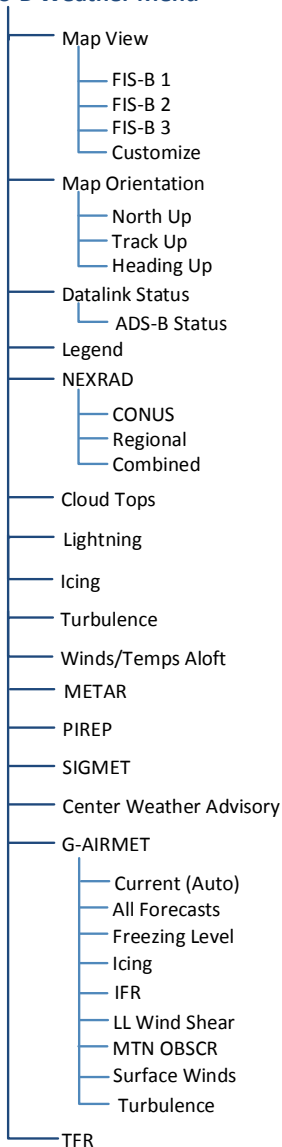
- a. To promote pilot awareness of ownship location with respect to reported weather, including hazardous meteorological conditions; NAS status indicators to enhance pilot planning decisions; and pilot near-term decision-making.
- b. To cue the pilot to communicate with Air Traffic Control, Flight Service Station specialist, operator dispatch, or airline operations control center for general and mission critical meteorological information, NAS status conditions, or both. FIS-B information, including weather information, NOTAMs, and TFR areas, are intended for the sole purpose of assisting in long-/near-term planning and decision making. The system lacks sufficient resolution and updating capability necessary for aerial maneuvering associated with immediate decisions. In extreme scenarios, the oldest weather radar data on the display can be up to 15 to 20 minutes older than the display's age indication for that weather radar data. Therefore, do not attempt to use FIS-B weather information to maneuver the aircraft at minimum safe distances from hazardous weather. FIS-B information must not be used in lieu of a standard preflight briefing.
- c. [Displaying FIS-B weather using GDL 88, GNX 375, or GTX 345 Class 1 transceivers] FIS-B uplink is a subscription-free FIS broadcast managed by FAA SBS. It provides an FAA approved source for METAR, TAF, WINDS, PIREPs, NEXRAD, AIRMET, SIGMET, and TFR information, and is subject to the broadcast range limits for these products. FIS-B uplink is not an FAA approved source for NOTAMs.

NOTAM 30-DAY LIMITATION

NOTAMs received via FIS-B may not be a complete listing. Active NOTAMs are removed from the FIS-B data stream 30 days after issuance. Before flight, review all necessary aeronautical and meteorological information from official sources. For more information, consult AC 00-63.

FIS-B Weather Setup

FIS-B Weather Menu



Raw Text Reports



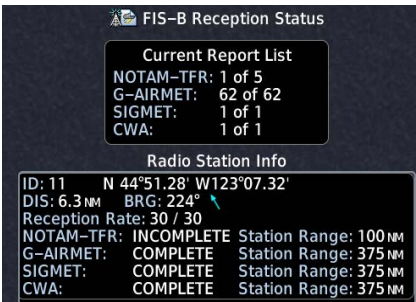
Access FIS-B textual data by tapping the **Raw Text Reports** key on the ADS-B Status page. Select a FIS-B datalink textual product to view the latest uploaded data.

To display raw text reports from the FIS-B Weather menu, tap **Datalink Status** > **Raw Text Reports**.

FIS-B Ground Reception Status

FIS-B ground station transmission status may be monitored when the UAT transceiver is powered and the FIS-B weather option is selected. A current report lists completeness of NOTAM-TFR, G-AIRMET, CWA, and SIGMET data for all received ground stations.

For the received ground station, the station range field indicates complete/incomplete status only for those products within the ground station's range.



To access the FIS-B Reception Status page from the FIS-B Weather menu, tap **Datalink Status** > **Radio Stations**.

Connex Weather



Garmin provides Connex datalink, on-demand weather products through the Iridium satellite system via a GSR 56 transceiver and L-band frequencies. An Iridium satellite data subscription service is required for worldwide Connex weather coverage.

FEATURE REQUIREMENTS

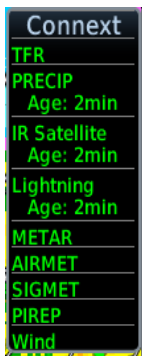
- *GSR 56 transceiver*
- *Data plan*

Connex Weather coverage is available throughout most of Europe, Canada, and the U.S. Additional radar coverage areas are added continuously. Various world-wide weather subscription package options provide weather reporting for most of Europe, Canada, Australia, and the U.S.

For the latest radar coverage information, visit:

<https://fly.garmin.com/fly-garmin/connex/worldwide-weather/>

Connex Weather Product Age



Connex weather product requests are pilot controlled. Weather products are refreshed at intervals that are defined and controlled by Connex and its data vendors.

Weather product age is based on the time difference between when the data was assembled by Connex and current GPS time. Weather products expire at intervals based on each product. When the data expires, it is removed from the display. This ensures that displayed data is consistent with what is currently provided by Connex Satellite Radio services.

Connext Weather Setup

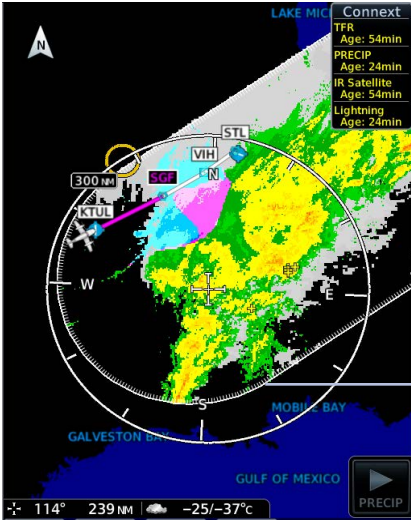
Connext Weather Menu

- Map Orientation
 - North Up
 - Track Up
 - Heading Up
- Connext Settings
 - Settings Menu
- Legend
- PRECIP
- Lightning
- IR Satellite
- METAR
- AIRMET
 - Icing
 - Turbulence
 - IFR
 - MTN OBSCR
 - Surface Winds
- SIGMET
- PIREP
- Winds/Temps Aloft
- TFR

Connext Settings Menu

- Coverage Region
 - Present Position
 - Flight Plan
 - Waypoint
 - COV Diameter
- Data Request
 - Request Data
 - Auto Request
- Datalink Status
 - GSR 56 Status

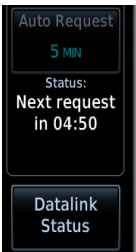
Connext Data Requests



Connext weather data may be updated on an on-request basis. This feature allows the pilot to download selected weather products for a specific waypoint, flight plan, or destination, and the time frame.

Access is via the Connext Settings menu.

Coverage Boundary



The pilot can update Connext weather data at any time regardless of predetermined automatic update frequency.

CONNEXT WEATHER REQUEST CONTROLS

Present Position	Request weather information around aircraft's present position. COV diameter determines extent of request.
Destination	Request weather information for the destination airport.
Flight Plan	Request weather information for the active flight plan route.
Distance	Enter requested weather information for a specified distance along the active flight plan. Range: 10 nm to 500 nm
Waypoint	Request weather information for a specified waypoint.
COV Diameter	Define the diameter of coverage. Range: 10 nm to 500 nm
Request Data	Request weather data at any time. Reset Auto Request timer (if active) for next request interval.
Auto Request	Select the automatic weather request period. Options: Off, 5 min, 10 min
Datalink Status	View Iridium satellite datalink status and signal strength.

Stormscope



Stormscope lightning data display on a dedicated weather page and as overlays on the MFD Map and HSI Map.



WARNING

Do not exclusively use the lightning detection system for weather avoidance. The system may display inaccurate or incomplete information. For additional information, consult the lightning detection system documentation.

FEATURE REQUIREMENTS

One of the following:

- WX-500 Stormscope Weather Mapping Sensor
- WX-1000E Stormscope Weather Mapping System (with TXi software v3.61 or later)

FEATURE LIMITATIONS

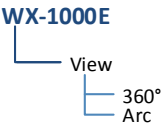
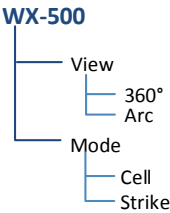
- Stormscope lightning information cannot display concurrently with a datalink lightning weather product (SiriusXM, Connex, or FIS-B)

Stormscope Features

- Detects electrical discharges from thunderstorms within 200 nm of current position
- Plots strike count and relative bearing location every two seconds
- Depicts distance from aircraft
- Provides arc and 360° viewing options
- Indicates the selected display range, with range options including 25 nm, 50 nm, 100 nm, and 200 nm

For more information, consult the appropriate third-party documentation.

Stormscope Setup



Tap **Menu** to access Stormscope setup options. Available features are dependent upon Stormscope type.

MFD Map Overlay

Control for the Stormscope map overlay resides in the Map setup menu.

Home > **Map** > **Menu**, and select **Stormscope**.

HSI Map Overlay

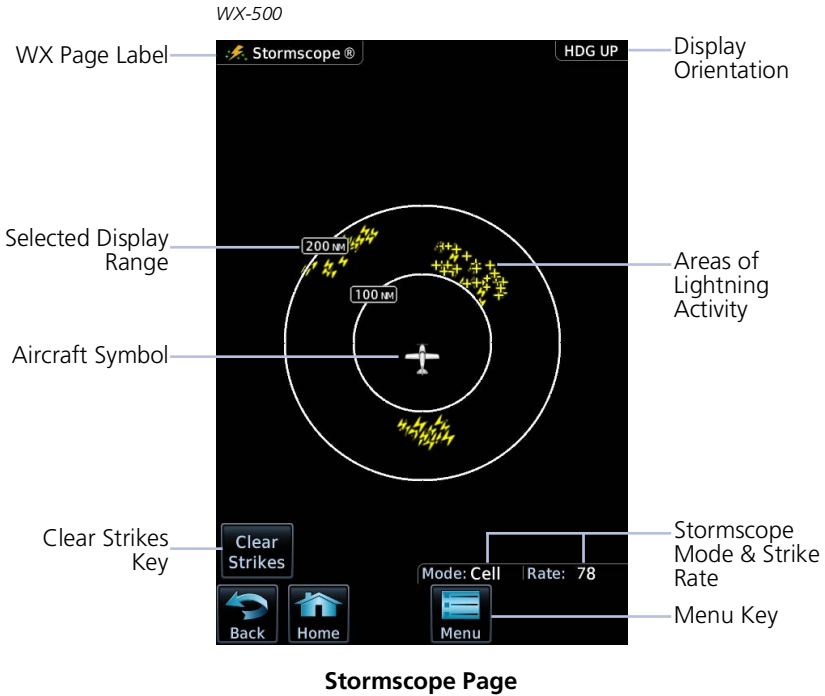
Control for the Stormscope HSI Map overlay resides in the PFD's HSI Setup menu.

Tap **Menu** > **HSI Setup** > **Overlays**, and select **Stormscope**.

Stormscope Types

WX-500

Strikes are initially indicated at representative range and bearing relative to the aircraft. To maintain bearing through ownship turns, the Stormscope data display rotates with changes to heading or GPS track.



WX-500 MODES

Changes in mode selection also reflect on the associated map overlay (MFD Map, HSI Map).

Cell Mode	Strike Mode
Identifies clusters of electrical activity, grouping individual strikes together.	Displays individual discharge points in relation to where they are detected.
When to use: During heavy storm activity to identify where storm cells are located.	When to use: During periods of light electrical activity to plot the initial strikes associated with a building thunderstorm.

WX-500 SYMBOLS

Symbols denote time since last lightning strike.

	Less than 6 seconds (initial strike)		Less than 60 seconds
	Less than 120 seconds		Less than 180 seconds

STRIKE RATE

Strike Rate reports the approximate number of strikes per minute for a given view range. Refer here when trying to determine if storm cells are building or decaying.

Mode: Strike | Rate: 47

Strikes occur at a higher rate as a storm matures. They are less frequent as the storm dissipates.

CLEAR STRIKES



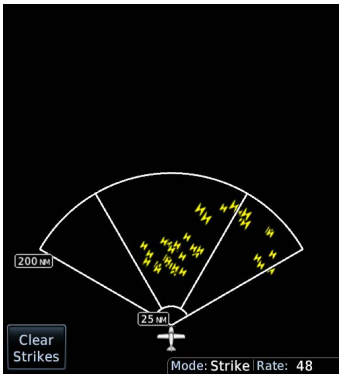
Tapping this key resets the rate value and clears the display of all strikes. Clearing strikes on the Stormscope display also clears them on Map.

If neither heading nor GPS track is available, “Turn COMP Fail” annunciates on the Stormscope display and data will no longer rotate to compensate for ownship turns. In such cases, you must clear all strikes following each heading change.

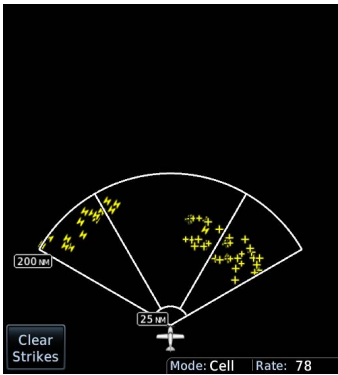
WX-500 VIEWS

Stormscope provides both arc and 360° views of lightning data for each mode.

Arc View

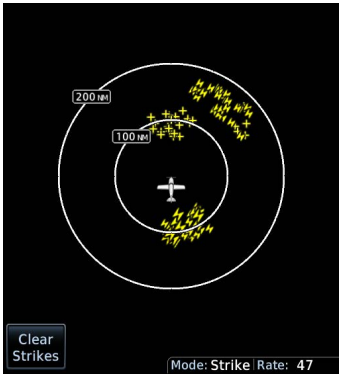


Strike Mode

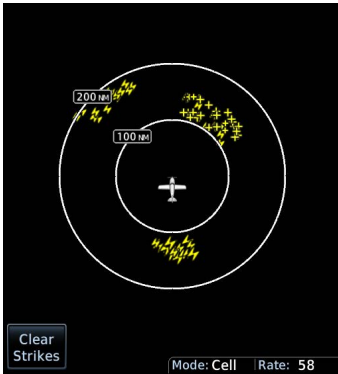


Cell Mode

360° View



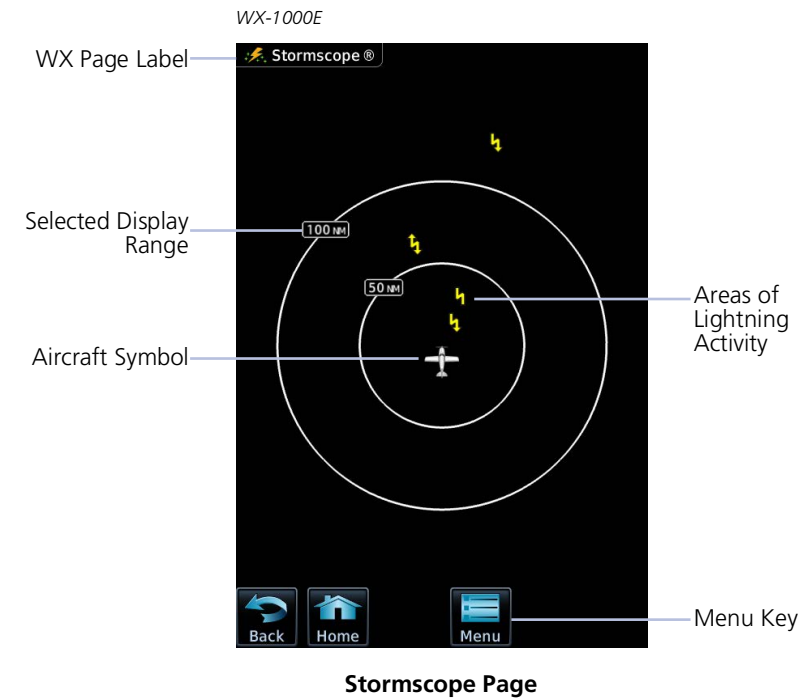
Strike Mode



Cell Mode

WX-1000E

Strikes are accumulated and initially indicated as *cells* at representative range and bearing relative to the aircraft. To maintain bearing through ownship turns, the Stormscope data display rotates with changes to heading.



If Stormscope does not receive valid heading data, then the system cannot properly infer cells of electrical activity, thus resulting in a failure. In such cases, "Lightning Failed" annunciates on the Stormscope display. Read more about this alert annunciation in *Stormscope Alerts*.

WX-1000E SYMBOLS

Symbols denote lightning cell intensity and approximate strikes per minute.



Light

Up to 8 strikes per minute



Moderate

Between 9 and 25 strikes per minute

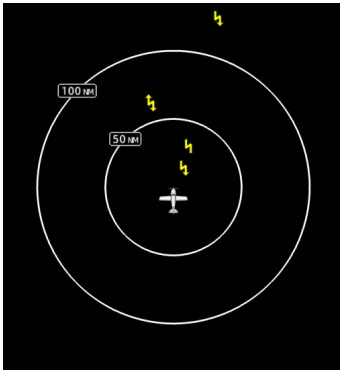


Heavy

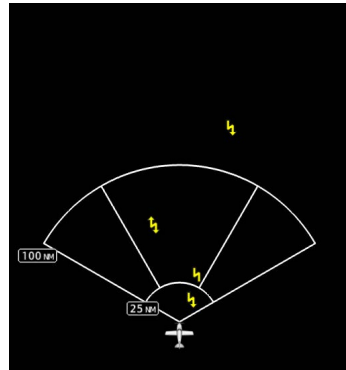
26 strikes or more per minute

WX-1000E VIEWS

Stormscope provides both arc and 360° views of lightning data.

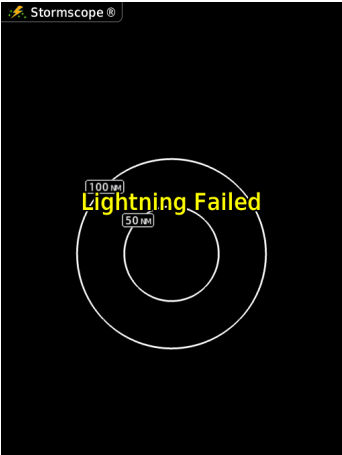


360° View



Arc View

Stormscope Alerts



A caution message alerts you when the Stormscope system is offline or reports a failure. On WX-1000E, this message also appears if you experience a loss of valid heading data.

If Stormscope is enabled on a navigation map (MFD Map, HSI Map), the associated overlay status icon is crossed out.



Stormscope Alert Annunciation

Airborne Weather Radar



Weather radar information displays on a dedicated weather page and as overlays on Map.



WARNING

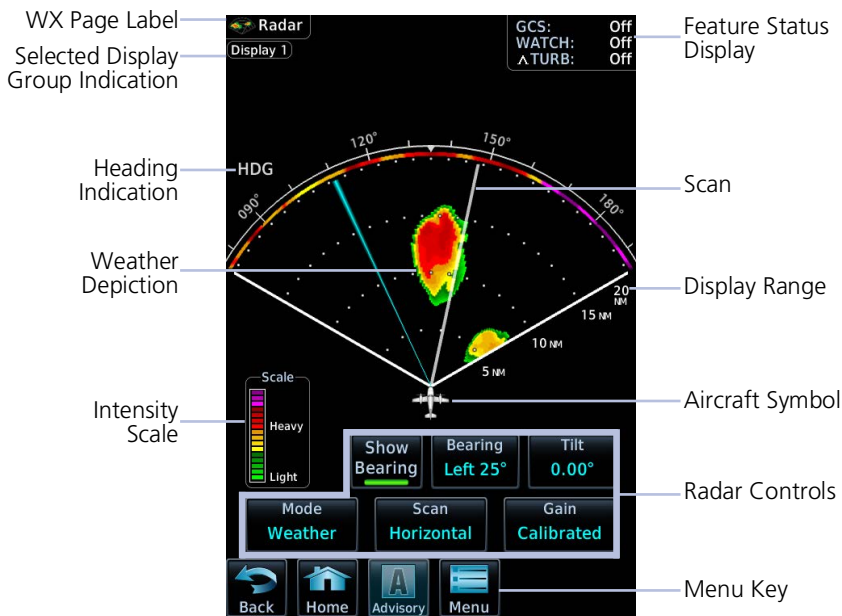
Use weather radar to avoid severe weather, not to enter it.

FEATURE REQUIREMENTS

- GDU 1060/1210 or full screen MFD on GDU 700P

FEATURE LIMITATIONS

- Available modes and intensity levels depend on the installed radar type
- Not available on GDU 700() EIS/MFD



Weather Radar Page

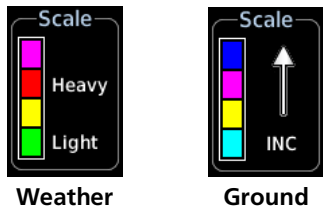
RADAR UNITS

Distances on the Weather Radar page display in nautical miles.

DISPLAY ADJUSTMENTS

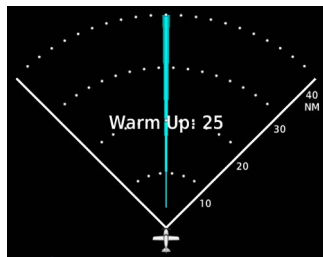
To change radar display range, adjust the inner control knob.
To adjust tilt and bearing angle, tap and drag the line indicator as necessary, or use the associated control keys.

INTENSITY SCALE



Each operating mode uses a unique color palette to depict increasing intensity levels.

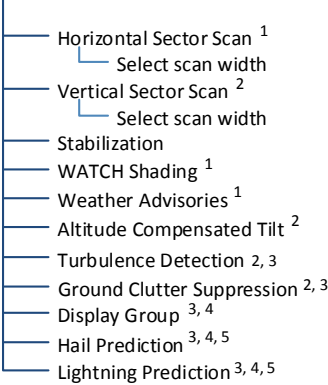
WARM-UP COUNTDOWN TIMER



A warm-up period initiates upon power up (GWX 68 only). Duration varies depending on how long the radar is off. The radar begins transmitting once warm-up is complete.

Weather Radar Setup

Radar Menu



¹ Not available for third-party radars.
² Available with GWX 70/75 only.
³ Requires feature enablement.
⁴ Available with GWX 8000 and TXi software v3.61 or later.
⁵ Available only during automatic mode.

Tap **Menu** to access weather radar setup options. Available features are dependent upon configuration.

With the exception of Sector Scan and Display Group, all selections are on/off only.

Map WX Overlays

The Radar overlay control resides in the Map Setup menu.

From the Home page, tap **Map > Menu > Radar**.

Radar, NEXRAD, and terrain overlays are mutually exclusive. Selecting one automatically turns the other off.

Sector Scan

FEATURE LIMITATIONS

- Allowable horizontal swath widths are dependent upon the installed radar

The sector scan option focuses the scan on a smaller segment of the radar sweep. This is useful when monitoring priority weather targets.

SCAN	INCREMENTS
Horizontal	20°, 40°, 60°, 90°, Full
Vertical	10°, 20°, 30°, Full
The 90° horizontal scan option is available only for radars with extended scan capabilities (120° sweep).	

Increments vary according to scan type.

Horizontal sector scans center on the bearing line, vertical sector scans center on the tilt line.

Stabilization

FEATURE LIMITATIONS

- Manual control not available for GWX 8000

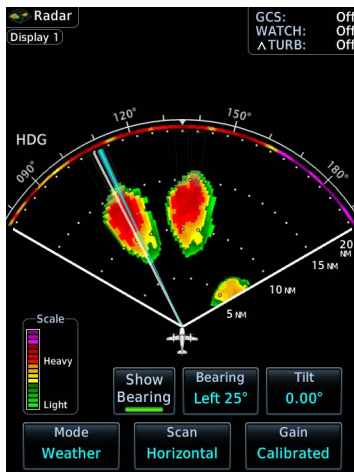
This feature helps stabilize the antenna so the scan is parallel to the ground when active. When off, the antenna scan is relative to the aircraft lateral axis. Status displays in the upper right corner of the Radar page.

WATCH

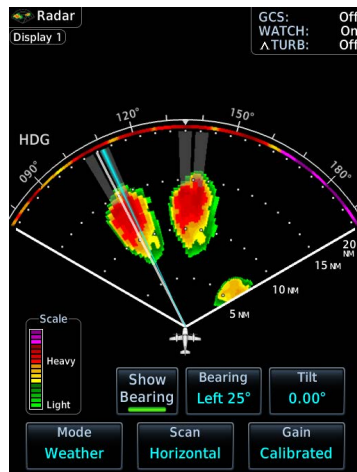
FEATURE LIMITATIONS

- Not available for third-party radars

Use WATCH to determine where the displayed intensity may be understated beyond a high-intensity area. Adjust tilt to determine the extent of attenuation in a shaded area. WATCH only displays in horizontal scans.



WATCH Off

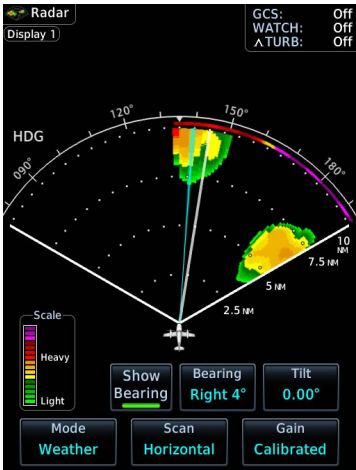


WATCH On

Weather Alert

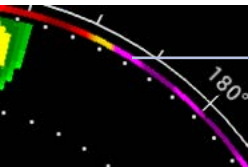
FEATURE LIMITATIONS

- Not available for third-party radars



These alerts indicate the presence of heavy precipitation beyond the current display range.

Multicolored bands on the outer range ring display at the approximate azimuth of severe weather targets.



Weather Alert Band

If the antenna tilt is adjusted too low, a weather alert is generated by ground returns. If a weather alert is detected within ± 10 degrees of the aircraft heading, an advisory displays on the MFD in the advisory window.

Altitude Compensated Tilt

AVAILABLE WITH:
GWX 70/75

FEATURE LIMITATIONS

- Not available for third-party radars



This feature adjusts the tilt to compensate for altitude changes as the aircraft climbs and descends. Status displays above the radar controls.

Turbulence Detection

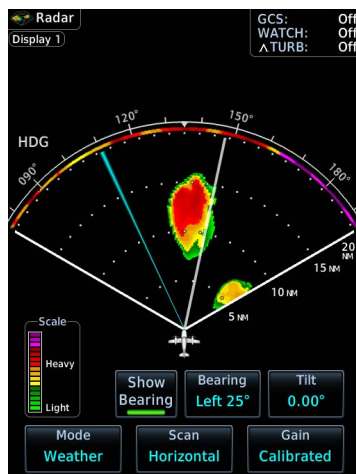
AVAILABLE WITH:
GWX 70/75/8000

FEATURE REQUIREMENTS

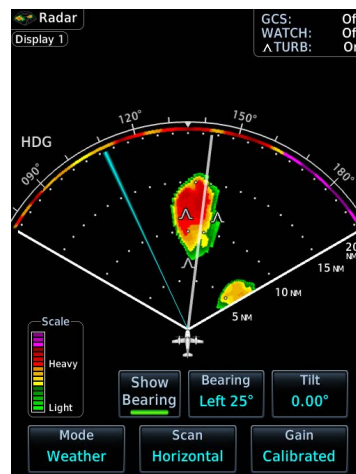
- Purchased GWX Turbulence Detection feature enabled on GDU

This feature detects and displays severe turbulence.

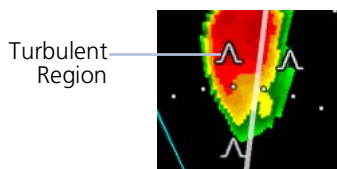
Example of Turbulence Detection on the GWX 8000 radar display.



Turbulence Off



Turbulence On



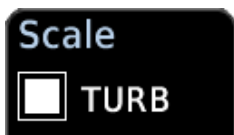
GWX 8000 uses symbols to denote regions of severe turbulence. These areas appear white on GWX 70 and GWX 75 radar displays.

The decision to fly into an area of radar targets depends on target intensity, spacing between targets, aircraft capabilities, and pilot experience.

Some weather radars detect only precipitation, not clouds or turbulence. While GDU may indicate clear areas between intense returns, this does not mean it is safe to fly between them.

A legend shows when turbulence detection is active.

GWX 70/75



Legend appears above the intensity scale when the feature is on. It is absent when the feature is off.

GWX 8000



On/off status annunciates in the feature status display.

Turbulence Detection reports as “Off” when:

- Current scan range is greater than 160 NM
- Radar is not in weather mode
- Vertical scan is active

Ground Clutter Suppression

**AVAILABLE WITH:
GWX 70/75/8000**

FEATURE REQUIREMENTS

- *Purchased GWX Ground Clutter Suppression feature enabled on GDU*

FEATURE LIMITATIONS

- *TXi software v3.61 and later: Not available for third-party radars, except RTA-800 which supports Ground Clutter Suppression.*
- *Enhanced functions available only with GWX 8000*
- *Horizontal scans only*

This feature reduces the amount of returns of highly reflective objects on the ground, while maintaining the intensity and size of weather returns.

Enhanced Ground Clutter Suppression with GWX 8000

During manual mode, the system uses Doppler radar to suppress most clutter to 40 nm. Ground clutter shows for one sweep following any adjustment to the tilt setting, allowing the pilot to determine the proper tilt angle.

During automatic mode, the system employs the terrain database to provide data for ground clutter reduction.

Predictive Hail & Lightning

AVAILABLE WITH: GWX 8000

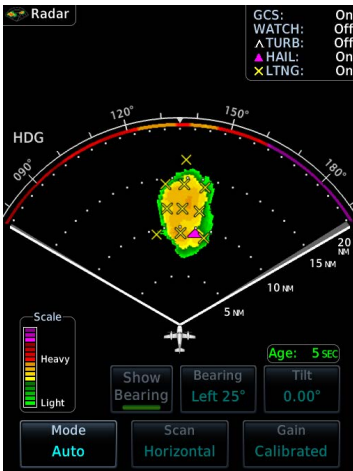
FEATURE REQUIREMENTS

- Purchased GWX 8000 feature enabled on GDU
- TXi software v3.61 or later

FEATURE LIMITATIONS

- Available only when automatic scanning mode is active

Predictive Hail & Predictive Lightning On



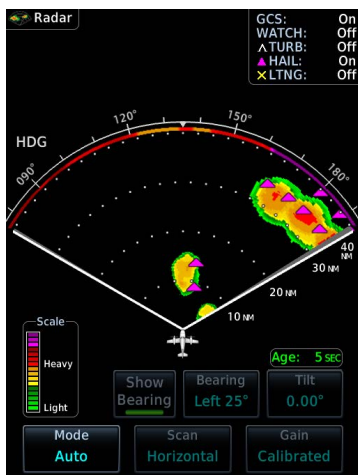
A legend indicates which predictive features are active.

The system searches volumetric data for areas that are conducive to the formation of hail or lightning. TXi depicts these regions using dedicated symbols. The circular footprint of each symbol represents an approximate area of concern.

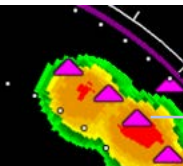
You can enable predictive features while the radar is in automatic mode.

Enable predictive hazard features for initial awareness of possible adverse weather ahead.

Predictive Hail Depiction

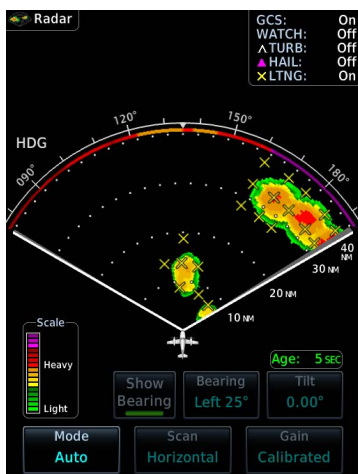


A magenta triangle indicates regions of hail.

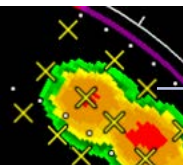


Hail Region

Predictive Lightning Depiction



A yellow "X" indicates regions of lightning.



Lightning Region

Display Groups

AVAILABLE WITH: GWX 8000

FEATURE REQUIREMENTS

- Purchased GWX 8000 feature enabled on GDU
- TXi software v3.61 or later

FEATURE LIMITATIONS

- GWX 8000 allows up to two display groups for selection
- Radar display groupings pertain to GWX 8000 only
- Not available when automatic scanning mode is active



Enable modes and features for all configured GDU TXi and/or GTN Xi units within an assigned group.



To select a display group, tap **Display Group** and choose between Group 1 and Group 2.

Group 1 is the default group selection.



The selected display group appears in the upper left of the radar display.

This indication is absent during automatic mode when the display grouping function is inactive.

Radar Modes



WARNING

Do not transmit when personnel or objects are within 16 ft of the antenna.



CAUTION

Place radar in standby mode before taxiing to prevent damage to the radar assembly.

Mode Key



Tapping this key opens a list of available mode options.



A confirmation pop-up alerts the pilot when attempting to scan while on the ground.

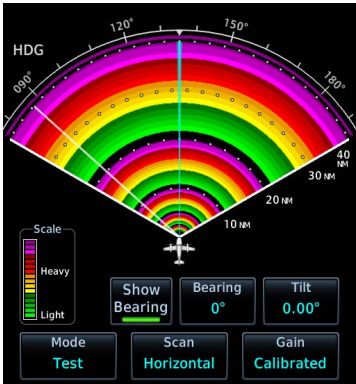
STANDBY



- Parks the antenna at the centerline
- Automatic standby occurs during power up and landing

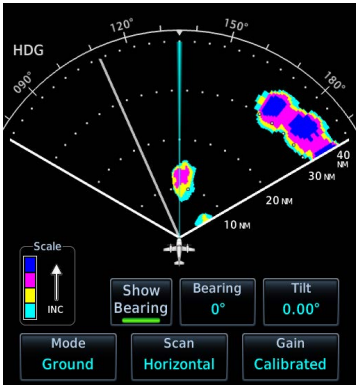
Enable predictive hazard features for initial awareness of possible adverse weather ahead.

TEST



- Places transmitter in standby as the display simulates a radar sweep
- Test data verifies communication between the weather radar and display
- Radar pulses do not transmit from the antenna during this mode

GROUND

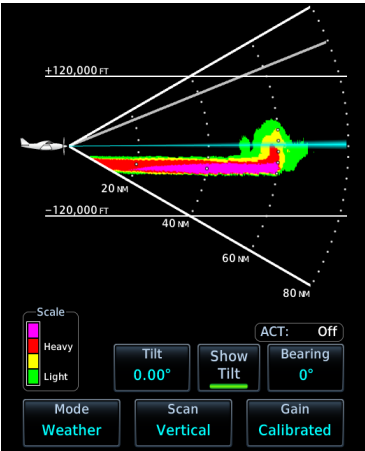


- Presents a depiction of terrain similar to that of a topographical map
- Useful when trying to verify aircraft position

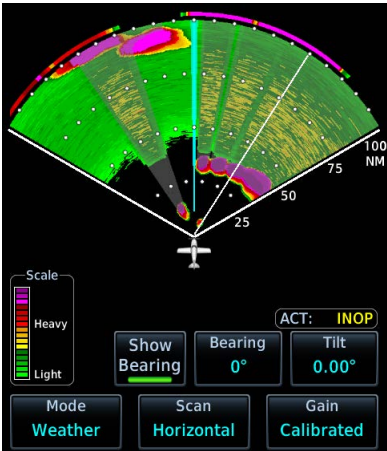
GROUND MAP MODE COLOR	GROUND TARGET INTENSITY LEVELS	
	GWX RADAR INTENSITY	THIRD-PARTY RADAR LEVEL
Black	0 to 2 dB	0
Cyan	3 dB to < 13 dB	1
Yellow	13 dB to < 21 dB	2
Magenta	21 dB to < 29 dB	3 and above
Blue	29 dB and greater	Not used

WEATHER

- Presents an airborne depiction of precipitation
- Colors represent approximate rainfall intensity and rates for weather radar targets



GWX 68/70



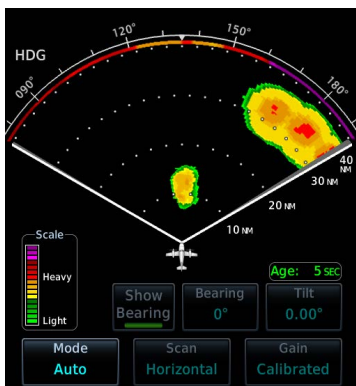
GWX 75

WEATHER MODE COLOR	GWX 68/70/75/8000 RADAR		THIRD-PARTY RADAR
	APPROXIMATE INTENSITY	APPROXIMATE RAINFALL RATE (IN/HR)	RADAR RETURN LEVEL ¹
Black	< 23 dBZ	< .01	0
Green	23 dBZ to < 33 dBZ	.01 to < 0.1	1
Yellow	33 dBZ to < 41 dBZ	0.1 to < 0.5	2
Red	41 dBZ to < 49 dBZ	0.5 to < 2.0	3
Magenta	49 dBZ and greater	2.0 and greater	4
White	Turbulence Detection ²		

¹ Consult the appropriate third-party radar documentation. ² GWX 70 and GWX 75 only.

AVAILABLE WITH: GWX 8000

- *TXi software v3.61 or later*

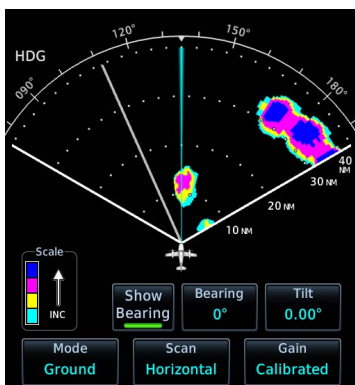


- Provides automatic 3D volumetric scanning of the area ahead of the aircraft (weather information only)
- Bearing/Tilt, Gain, Scan, and enhanced Ground Clutter Suppression functions occur automatically
- Hail Prediction, Lightning Prediction, and Turbulence features are available in the setup menu
- Manual range adjustments are still available

During automatic mode, the radar antenna performs multiple scans and tilt angles to update the overall weather solution. The system collects and processes this data to provide a display of the strongest returns in the volume of air ahead. In some ways, this is similar to a NEXRAD composite image.

In addition to automatic mode, GWX 8000 retains the ability to use manual mode for investigating returns. To perform manual scanning, place the radar in Weather or Ground mode. If multiple radar displays are present, ensure that at least one of the radar displays is not in standby.

Automatic Mode Status Indications



Upon activation, GWX 8000 requires 20 seconds to render its initial image.

"Initializing" annunciates on the radar display to inform you that initialization is in progress (i.e., the radar is functioning but scan depictions are not yet available).

If automatic mode is inoperative, a yellow "Auto INOP" message replaces the "Initializing" annunciation. Select a different scanning mode to resume radar operation.

Automatic Mode & Display Groups

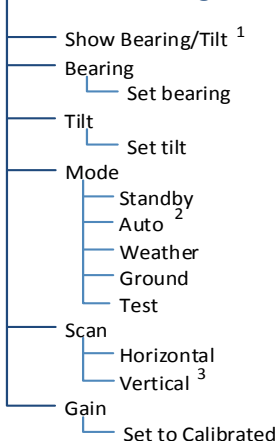


Selecting **Auto** on one radar display places all configured radar displays in automatic scanning mode, thus disabling the display grouping function and removing the display group indication from the radar display.

Changing from auto to any other mode resumes the previous display group setting and places all affected radar displays in the selected mode. The display group indication returns to the radar display to show the active group designation. Displays not included in the group are automatically placed in standby mode.

Radar Controls

Weather Radar Page



Some radars allow independent sweeps when connected to multiple displays.

GWX 68 synchronizes the controls from all connected displays. GWX 70 and GWX 75 receive commands from each display independently and perform a separate sweep for each. GWX 8000 allows control from two display groups.

¹ Dependent upon scan type selection. ² GWX 8000 only. ³ Not available for RS 181 and RS 811 radars.

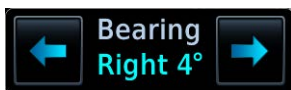
Show Bearing/Tilt

Displays the tilt or bearing line depending on current scan selection. This feature is useful when making horizontal or vertical scan adjustments.

Bearing

FEATURE LIMITATIONS

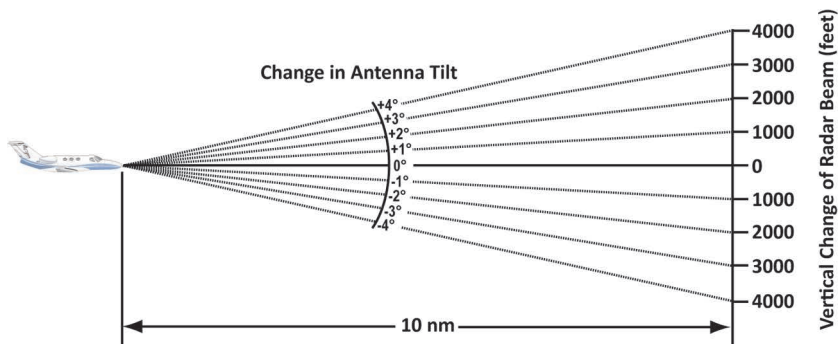
- Not available on RS 181 and RS 811 radars



Directional keys allow fine adjustment of the bearing line angle. This method is optional to tapping and dragging.

Tilt

Use antenna tilt to locate the top and bottom of storm cells and to increase intensity of ground target returns. One degree of tilt equals 100 feet of altitude per nautical mile.



Basic Antenna Tilt Setup



In a typical tilt setup, the bottom of the radar beam is adjusted to 4° below parallel with the ground. To achieve this:

1. Fly the aircraft level.
2. Adjust antenna tilt so ground returns display at a distance equal to the aircraft's current altitude (AGL) divided by 1,000.

Remember the following points when flying with the antenna at this basic angle setting.

Altitudes between 2,000 and 30,000 ft AGL:

- Avoid target returns at 5 nm. These may be either weather or ground returns that are 2,000 ft or less below the aircraft.
- Raise the antenna tilt 4° to separate ground returns from weather returns in flat terrain. Return the antenna to the previous setting after a few sweeps of the radar.
- Set the display range to 60 nm for aircraft flying at 15,000 ft or lower. Monitor ground returns for possible threats.

Altitudes above 29,000 ft:

Be cautious of targets 30 nm or closer. This may indicate a thunderstorm that the aircraft cannot fly over safely.

For a more accurate view of target coverage and intensity, center the tilt angle on the strongest return area, aiming below the freezing level of the storm.

In areas of multiple heavy cells, use the vertical scan feature along with antenna tilt to examine the cells. Avoid shadowed areas behind targets.

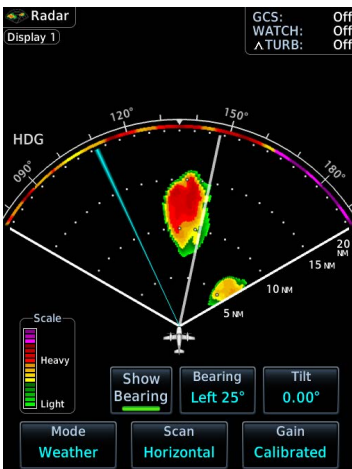
Scan

FEATURE LIMITATIONS

- The following third-party radars do not support vertical scan functionality:
RS 181, RS 811, and RTA-800

Horizontal scans provide a visual depiction of weather in front of the aircraft. Vertical scans focus the radar on a particular vertical target.

HORIZONTAL



For bearing line visibility, tap **Show Bearing**.

To adjust the horizontal scan bearing:

- Tap and drag the bearing line left or right

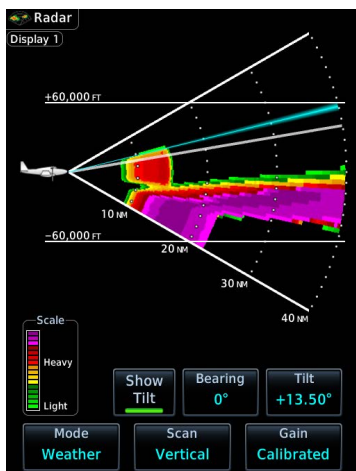
Or

- Adjust the angle using the **Bearing** control

To adjust vertical angle for the horizontal scan, tap **Tilt** and then adjust the angle using directional keys.

VERTICAL

When vertically scanning with stabilization on, the physical area that the radar is sweeping may not match the displayed vertical scan. This occurs when the aircraft pitch is not at 0°. To compensate, the GDU does not draw the unscanned portion.



For tilt line visibility, tap **Show Tilt**.

To adjust the tilt of vertical scan:

- Tap and drag the tilt line up or down

Or

- Adjust the angle using the **Tilt** control

To adjust horizontal angle for the vertical scan, tap **Bearing** and then adjust the angle using directional keys.

Gain



NOTE

Precipitation intensity may not be accurate if the gain is changed.

FEATURE LIMITATIONS

TXi software v3.61 and later: The Gain control is active only during ground mode on third-party radars, except RTA-800 which supports all modes.

This feature controls the sensitivity of the radar receiver. Adjustments to receiver sensitivity automatically change the intensity of radar targets.



Directional keys allow sensitivity adjustments.

Tapping **Set to Calibrated** restores the calibrated gain setting and returns intensity depictions to their actual colors.

Radar Alerts



Caution messages alert you when there is a weather radar failure. For a list of possible radar alerts, refer to the annunciations table in this segment.



Radar Alert Annunciation

When a radar alert occurs:

- A textual annunciation appears over the radar sweep
- Scan depictions do not display
- Weather radar overlay detail is absent from Map if heading input is lost

ALERT ANNUNCIATIONS

ANNUNCIATION	CONDITION
Radar Fault	Condition: Data contains a fault unrelated to attitude
Radar Fail	Condition: - Weather radar product status is timed out - GWX only
Radar Controls Disagree	Condition: - Radar's actual state does not match the commanded state - Third-party radars only
Radar Active	Alert Type: - Radar is in standby, but remains active due to another interfacing controller - GWX 70/75 and third-party radars

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7 Traffic Awareness

- TRAFFIC DISPLAY 7-2
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Traffic Display

FEATURE LIMITATIONS

- *Symbols depicted depend on traffic source (e.g., TIS-A, TAS, ADS-B)*
- *Intruding aircraft without altitude reporting capabilities do not display altitude separation data or climb/descent indications*
- *Available display ranges and vector types depend on traffic source*

Collision avoidance and traffic surveillance data display on the Traffic page or as overlays on the Map page and HSI Map. Synthetic vision traffic depictions display on PFD when SVT is enabled.

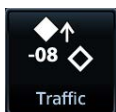
Available Traffic Sources

- TIS-A
- TAS/TCAS I
- ADS-B with or without TCAS
- TCAS II

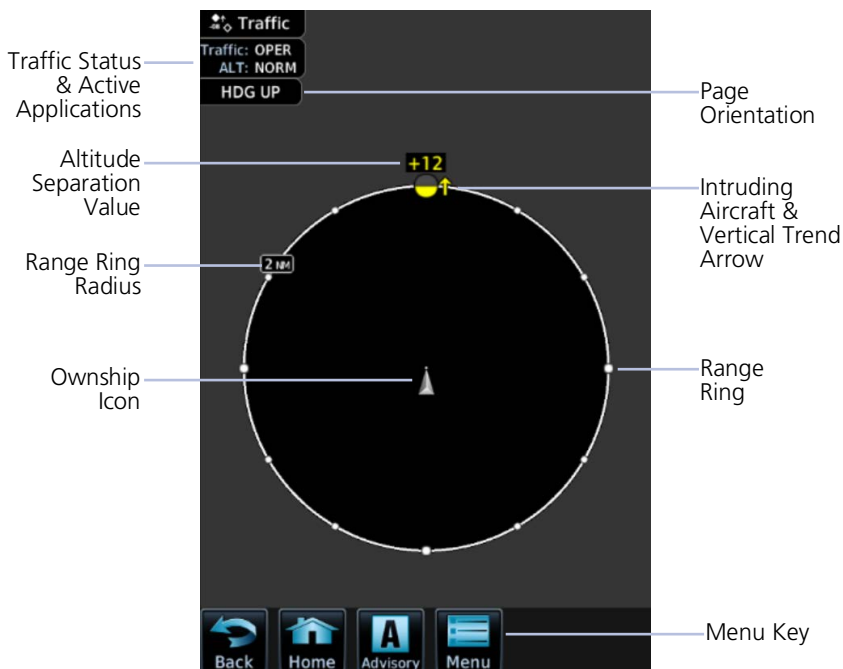
The availability of functions, alerting features, and options depend on the traffic system source.

ADS-B controls are accessible from the Traffic menu. Controls for other traffic systems are located on the Traffic page.

Traffic Page



The Traffic page displays intruding traffic in relation to the current position and altitude of the aircraft without clutter from other data.



Traffic Page Features

TRAFFIC DISPLAY OBJECTS	
Ownship Icon	• Depicts current aircraft position • Nose of the ownship is the actual ownship location • Unlike the configured aircraft symbol on map displays, ownship icon is always a directional arrow
Page Orientation Label	Orientations: 1. Heading up (HDG UP) during normal operation. 2. Track up (TRK UP) if there is no valid heading.
Range Rings	• Ranges vary according to traffic system type • Outer ring represents selected range
Altitude Separation Value	Indicates when an intruder is above or below the ownship. Value placement is based on intruder location. • Above traffic symbol if intruder is above • Below symbol if intruder is below • Plus or minus sign denotes higher or lower altitude
Vertical Trend Arrow	• Active when an intruder is climbing or descending at a vertical speed greater than 500 fpm
Intruding Aircraft Symbol	Color may be cyan or white depending on configuration. Actual intruder location varies according to symbol type. • Tip of directional symbols • Center of non-directional symbols

INVALID ALTITUDE DEPICTIONS



If an intruder does not have a valid altitude, its symbol displays without the altitude tag.

OFF SCALE TRAFFIC ALERTS



Off scale (out of range) traffic alerts are depicted as half yellow symbols on the outermost range ring at the correct bearing.

TRAFFIC OVERLAY STATUS ICONS

ICON	DESCRIPTION
	Map is displaying traffic.
	Map is not displaying traffic.

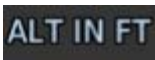
Overlay icons indicate the status of displayed traffic on HSI Map, Rotorcraft PFD VFR Map, and the MFD Map page.

TRAFFIC UNITS

PARAMETER	UNITS
Altitude	ft
Distance	nm

Traffic units are always uniform.

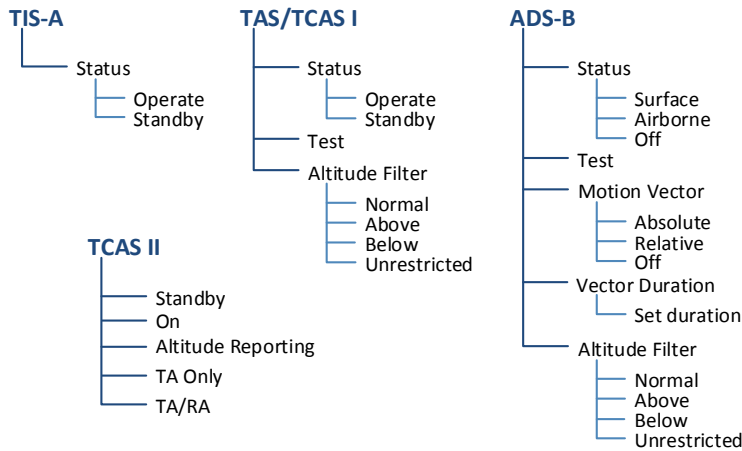
System Units page selections do not affect the traffic display.



If the system units for altitude are set to meters, an annunciation reminds the pilot that all displayed altitude values remain in feet.

Traffic Setup

Traffic settings are synchronized between all MFDs.

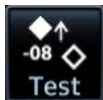


Traffic Test

FEATURE LIMITATIONS

- *ADS-B, TAS, and TCAS applications only*
- *Available only when the aircraft is on ground (i.e., standby mode)*
- *Test pattern depiction is dependent upon the active traffic system*

TEST KEY



The test function displays a test pattern on the Traffic page. For ADS-B, this option is accessible via the Traffic menu.

The system automatically returns to normal operating mode once the test is complete.

Altitude Filtering

FEATURE LIMITATIONS

- *ADS-B, TAS, and TCAS applications only*

Pilot selectable filters limit the display of traffic to a specific altitude range relative to the altitude of the ownship.

Filter selections apply to both the Traffic page and the traffic overlay on the Map page.

SELECTION	ALTITUDE RANGE
Normal	-2,700 ft to 2,700 ft
Above	-2,700 ft to 9,900 ft
Below	-9,900 ft to 2,700 ft
Unrestricted	-9,900 ft to 9,900 ft

On a standalone PFD, HSI Map always presents traffic at the Normal filter setting. On GDU 1060/1210 with PFD and MFD, it uses the same filter selections on MFD.

Traffic Types

TIS-A

FEATURE REQUIREMENTS

- Aircraft location is within the service volume of a Mode S terminal radar site
- TIS-capable Mode S transponder is interfaced to the GDU

FEATURE LIMITATIONS

- For information about TIS-A functions and limitations, consult the AIM

TIS-A Features

- 2-D graphical depiction of proximate traffic based on data received from terminal radars
- Track vector depicts intruder trajectory over the ground
- Vertical coverage: +3,500/ -3,000 ft from ownship
- Eight intruder maximum within an 8 nm radius
- Five second update cycle
- Normal and standby mode options

TIS-A Setup Selections

Status Select operating mode. Options are Operate and Standby.

TIS-A Traffic Symbols

Traffic information is for advisory use only. The pilot is responsible for identifying and avoiding traffic.



Non-threat Traffic



Traffic Advisory (TA)



TA Off-scale

TIS-A Status Indications

ANNUNCIATION	DESCRIPTION
Data Failed	GDU receives data from the transponder, but a failure exists in the data stream.
Failed	Transponder failure.
No Data	GDU is not receiving valid data from the transponder.
TA X.X NM ±XX ↕	System cannot determine TA bearing. Intruder depiction not available. Annunciation includes: • Distance in nautical miles • Altitude separation in hundreds of feet • Altitude trend arrow (climbing/descending)
TA OFF SCALE	TA is outside the selected display range.
Traffic Coast x SEC	Intruder depictions are more than six seconds old.
Traffic Removed	Intruder depictions are more than 12 seconds old. Traffic symbols no longer display.
Unavailable	Transponder not receiving TIS-A data from ground station.

TAS/TCAS I

TAS/TCAS I Features

- Airborne traffic system independent of ground radar
- 2-D graphical depiction of traffic relative to aircraft position and altitude

TAS/TCAS I Setup Selections

Status	Select operating mode. Options are Operate and Standby.
Test	Initiate a test of the traffic system.
Altitude Filter	Select filter range. Options include: • Normal • Below • Above • Unrestricted

TAS/TCAS I Traffic Symbols

Traffic information is for advisory use only. The pilot is responsible for identifying and avoiding traffic conflict.

	Other Traffic
	Proximity Advisory
	Traffic Advisory
	TA Off-scale

TAS/TCAS I Status Indications

ANNUNCIATION	DESCRIPTION
Data Failed	GDU receives traffic data, but the TAS unit is reporting a failure.
Failed	Traffic data failure.
No Data	GDU is not receiving valid data from the TAS unit.
TA X.X NM ±XX ↕	System cannot determine TA bearing. Intruder depiction not available. Annunciation includes: • Distance in nautical miles • Altitude separation in hundreds of feet • Altitude trend arrow (climbing/descending)

TCAS II



WARNING

Traffic information shown on system displays is provided as an aid in visually acquiring traffic. Traffic avoidance maneuvers are based upon TCAS II Resolution Advisories, ATC guidance, or positive visual acquisition of conflicting traffic.



NOTE

If the installed TCAS II traffic system is not a GTS 8000, consult the applicable documentation for system-specific information.

FEATURE REQUIREMENTS

- GTS 8000 or third-party TCAS II system

TCAS II Features

- Monitors nearby airspace for aircraft flying with operating transponders
- Issues TAs to assist in visual identification of traffic
- Issues RAs to provide recommended vertical guidance maneuvers to resolve traffic conflicts

TCAS II Status Indications

MFD TRAFFIC PAGE INDICATIONS



TCAS II status is shown in the MFD Traffic page data window.

SELECTION	TCAS STATUS
TCAS II Self-Test Initiated (TEST)	TEST
Traffic and Resolution Advisory (TA/RA)	TA/RA
Traffic Advisory Only	TA ONLY
TCAS II Standby	STBY
TCAS II Failed	FAIL

PFD INDICATIONS

SELECTION	ANNUNCIATION
TCAS II Self-Test Initiated (TEST)	TCAS TEST
Traffic Advisory Only	TA ONLY
TCAS II Standby	TCAS STBY
TCAS II Failed	TCAS FAIL
TCAS Resolution Advisories Unavailable	TCAS VS RA FAIL
Traffic and Resolution Advisory (TA/RA)	TRAFFIC
RA Display Unavailable (Expand the PFD display area to resolve)	VS RA N/A

TCAS annunciations will appear on the right side of the PFD. Information annunciations are in black text on white backgrounds. Caution annunciations are in black text with yellow backgrounds. Warning annunciations are in white text with red backgrounds.

When the TCAS II unit issues a TA or RA, "TRAFFIC" appears and flashes for 5 seconds, and remains until no TAs or RAs are detected.

- RA "TRAFFIC" annunciations are white text with red backgrounds
- TA "TRAFFIC" annunciations are black text with yellow backgrounds
- If a TA and RA occur simultaneously, only the red and white RA "TRAFFIC" annunciation displays

If the GDU is not displaying the Traffic page, the system displays a traffic alert pop-up.

- During a TA event, the system issues a single "Traffic, Traffic" voice alert each time the system detects a new TA threat
- During an RA event, voice alert(s) provide vertical guidance to resolve the traffic conflict
 - The VSI displays a range of vertical speeds to fly to or avoid as applicable
 - Additional voice alerts occur if the RA status changes and when the aircraft is clear of the conflict

TRAFFIC ALERT BANNER

If the traffic system cannot determine the bearing of a traffic or RA, the alert displays as a traffic alert banner.

Warning Banner

RA X.X ± XX <UP> OR <DN>

- Displays when the system is unable to determine the bearing of an RA and extreme pilot vigilance is required
- Indicates distance in nm and altitude separation in hundreds of feet
- Indicates altitude trend up <UP> for climbing and down <DN> for descending traffic¹

Caution Banner

TA X.X ± XX <UP> OR <DN>

- Displays when the system is unable to determine the bearing of a TA and pilot vigilance is required
- Indicates distance in nm and altitude separation in hundreds of feet
- Indicates altitude trend up <UP> for climbing and down <DN> for descending traffic¹

¹ If altitude trend is available.

TCAS II Traffic Symbols

Traffic information is for advisory use only. The pilot is responsible for identifying and avoiding traffic conflict. TCAS II categorizes detected traffic into four groups of increasing collision threat potential. Lowest threat is OT and the highest is RA.



OT

Other non-threatening traffic.



PA

Traffic is not currently a threat, but is within 6 nm and $\pm 1,200$ ft of the own-aircraft altitude.



TA

Indicates traffic is within 20-48 seconds of a potential collision area.



Off-scale TA

Traffic is beyond the selected map range and the system displays a half-TA symbol at the edge of the map at the approximate relative bearing of the TA traffic. If TA traffic subsequently meets the criteria for an RA, the system issues an RA.



RA

Indicates traffic is within 15-35 seconds of a potential collision area.



Off-scale RA

If RA traffic is beyond the selected map range, the system displays a half-RA symbol at the edge of the map positioned at the approximate relative bearing to the RA traffic.

TCAS II with ADS-B



NOTE

Aircraft that are surveilled by ADS-B In only will not trigger a TCAS resolution advisory.

TCAS II interrogates Mode-S transponder data while automatically receiving ADS-B position and velocity information directly from a comparably equipped aircraft target. For preciseness, the system correlates between two data sources and the system displays the traffic information for the source determined to be the most accurate. Traffic that is not correlated (i.e., only detected by one system but not the other) is also displayed for the flight crew. This may occur, for example, if another aircraft is beyond the surveillance range of the TCAS II, but it is still receiving position and velocity information from other ADS-B equipped aircraft. The traffic correlation feature improves the accuracy of the traffic displayed, while reducing the occurrence of displaying a single target twice.

ADS-B

ADS-B Features

- Runway and taxiway depiction during SURF mode (< 2 nm range scale)
- Selectable traffic icons display intruder and vector information
- Customizable motion vectors (type, duration)
- Airborne and surface mode options
- On-scene mode option (rotorcraft only)

ADS-B Setup Selections

ADS-B Status	Enable automatic mode selection. System selects between Surface and Airborne depending on the state of the aircraft. Selecting Off disables the function.
Test	Initiate a test of the traffic system.
Motion Vector	Select motion vector type. Selecting Off removes all motion vectors from the display. • Absolute • Relative • Off
Vector Duration	Select the amount of time represented by the endpoint. Options range from 0 seconds to 5 minutes. A longer duration results in a longer vector.
Altitude Filter	Select filter range. Options include: • Normal • Above • Below • Unrestricted

ADS-B Traffic Symbols

Traffic Information is for advisory use only. The pilot is responsible for identifying and avoiding traffic conflict.

	Basic Directional		Proximate Directional
	Basic Non-directional		Proximate Non-directional
	Basic Off-scale Selected		Proximate Off-scale Selected
	Directional (On-Ground)		Non-directional Alerted
	Non-directional (On-Ground)		Off-scale Non-directional Alerted Traffic
	Proximate Directional (On-Ground)		Directional Alerted Traffic
	Proximate Non-directional (On-Ground)		Off-scale Directional Alerted Traffic
	Directional Surface Vehicle		Non-directional Surface Vehicle

ADS-B Traffic Applications

FEATURE LIMITATIONS

- *On-scene mode functionality is available only for rotorcraft*

MODE	FEATURES
AIRB	• Airborne traffic application • Active in the en route environment (>5 nm and >1,500 ft above the nearest airport)
SURF	• Airborne and ground traffic application • Active on ground or within the terminal environment (<5 nm and <1,500 ft above field elevation) • Runway and taxiway depictions when the zoom scale is <2 nm (Traffic page only) • Ground vehicle depictions
On-scene	• Rotorcraft nuisance alert suppression option • Reduces alerts caused by proximity rotorcraft operating in a scene environment (e.g., EMS landing zone, news gathering events) • Mode activation and control by way of navigator • Look-ahead time: 3 seconds • Traffic alarm vertical threshold: 200 ft • Incremental protected traffic volume: 1,000 ft

Motion Vectors

FEATURE LIMITATIONS

- Motion vectors display on the Traffic page only

A motion vector is a line extending from the nose of an intruder icon. Its orientation represents the intruder’s direction and movement. A yellow vector indicates when traffic meets intruding TA criteria (i.e., closing rate, distance, vertical separation).

MOTION VECTOR TYPES	
Absolute	• Cyan or white vector depending on configuration • Depicts intruder ground track • Calculations based on intruder direction and ground speed • Endpoint depicts intruder’s position over the ground at the end of the selected duration • Airborne and ground functionality
Relative	• Green vector • Depicts intruder movement relative to the ownship • Calculations based on track and ground speed of both intruder and ownship • Endpoint depicts intruder’s location relative to the ownship at the end of the selected duration • Airborne functionality only • “Relative Motion - Unavailable” annunciates during ground operations

ADS-B Status Indications

ANNUNCIATION	DESCRIPTION
Absolute Motion - XX SEC/MIN	Active motion vector type is absolute.
Failed	ADS data failure.
No Data	GDU is not receiving valid traffic data.
No GPS Position	ADS-B LRU detects that GPS unit is initializing.
Relative Motion - XX SEC/MIN	Active motion vector type is relative.
Relative Motion - UNAVAILABLE	Aircraft is not airborne and the active motion vector type is relative.
Unavailable	Necessary traffic data not available.

ADS-B Traffic Interactions

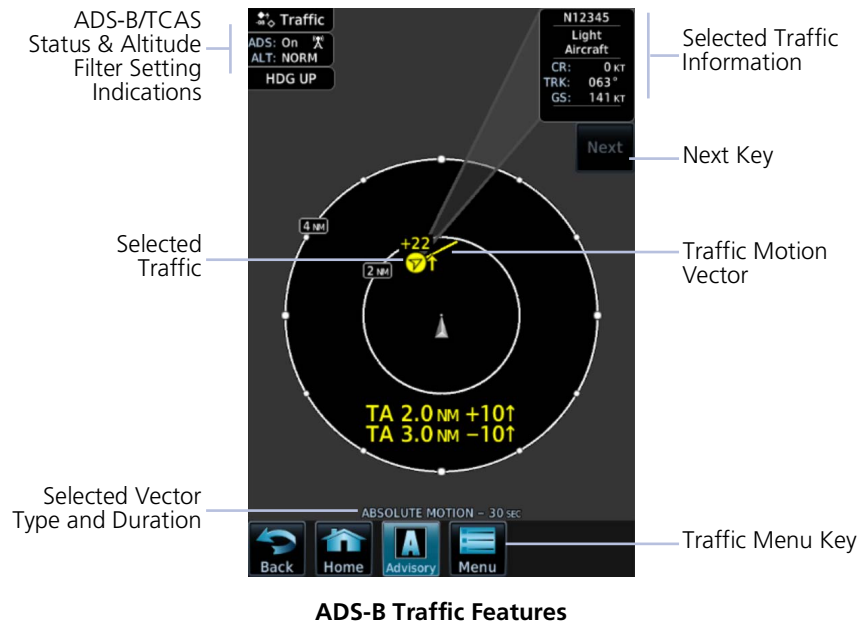
Selected Traffic Information

- Registration/call sign
- Vehicle type
- Closure rate
- Track
- Ground speed
- AIRB/SURF eligibility

Selecting a traffic symbol displays information about the aircraft in the upper right corner of the page.

Selections remain active through altitude filtering, zoom scale, and page changes.

Tapping **Next** repeatedly steps through multiple symbols spaced closely together.



Traffic Alerting

Traffic alerts occur anytime there is an increase in the number of traffic advisories. They remain active until the area is clear of all TAs.



MFD Pop-up

Traffic alerts appear as textual annunciations on the PFD, and as a pop-up window on the MFD, when applicable.

Aural traffic alerts are not a function of the GDU.



PFD Annunciation

ANNUNCIATION	CONDITION POP-UP AURAL MESSAGE
TRAFFIC	Condition:
	The traffic system reports a traffic advisory.
	Pop-up Alert: Yes
	Aural Message:
	Message content depends on current traffic system configuration.

For installations with GTX 345 and ADS-B software v3.20 or later, tapping the **Mute Alert** key silences the active traffic alert voice message. This function is applicable only to the active aural alert (does not mute future alerts). For more information, read *Aural Alerts* in section 2.

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8 Terrain Awareness

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Terrain Configurations

Available Terrain Configurations

- (H)Terrain Proximity
- (H)Terrain-FLTA
- TAWS-B
- External TAWS/HTAWS
- TAWS-A

Terrain, obstacle, and wire data display as overlays on the Terrain page, Map page, and HSI Map. Alerting functions are dependent upon the aircraft type and configured terrain alerting options.

Terrain controls are accessible from the Terrain menu. Map overlay keys are accessible from the associated map menu.

EXTERNAL TAWS

Depending on system configuration, the GDU may be capable of remotely indicating TAWS information from the navigator.

GPS Altitude for Terrain

FEATURE REQUIREMENTS

GPS altitude is derived from satellite measurements. To acquire an accurate 3D fix (latitude, longitude, altitude), a minimum of four operating satellites must be in view of the GPS receiver antenna.

The terrain system uses GPS altitude and position data to:

- Create a 2-D image of surrounding terrain and obstacles relative to the aircraft's position and altitude
- Calculate the aircraft's flight path in relation to surrounding terrain and obstacles
- Predict hazardous terrain conditions and issue alerts

GSL Altitude & Indicated Altitude

The GDU converts GPS altitude data to GSL altitude (i.e., the geometric altitude relative to MSL) for use in terrain functions. All Terrain page depictions and elevation indications are in GSL.

Variations between GSL altitude and the aircraft's corrected barometric altitude (or indicated altitude) are common. As a result, Terrain page altitude data may differ from current altimeter readings. Both GSL altitude and indicated altitude represent height above MSL, but differ in accuracy and reliability.

GSL Altitude

- Highly accurate and reliable geometric altitude source
- Does not require local altimeter settings to determine height above MSL
- Not subject to pressure and temperature variations
- Affected primarily by satellite geometry

Indicated Altitude

- Barometric altitude source corrected for pressure variations
- Requires frequent altimeter setting adjustment to determine height above MSL
- Subject to local atmospheric conditions
- Affected by variations in pressure, temperature, and lapse rate

Database Limitations



NOTE

Garmin cross-validates terrain and obstacle data in accordance with TSO-C151c. However, the information should never be considered all-inclusive. Database inaccuracies or omissions may exist.

Terrain and obstacle data are not available when the aircraft is operating outside of the installed database coverage area.

Garmin obtains terrain and obstacle data from government sources and cannot independently verify the accuracy and completeness of the information. Pilots must familiarize themselves with the appropriate charts and other data for safe flight.

DATABASE	COVERAGE LIMITATIONS
Terrain	• Not available north of 89° N latitude and south of 89° S latitude
Obstacle	• Coverage areas vary according to database type • Power line indications for the contiguous United States and small parts of Canada and Mexico • Regional definitions may change without notice • May not contain uncharted obstacles • May include power lines or only HOT lines depending on database type¹

¹ HOT lines are power lines that share location with other obstacles identified by the FAA.

Terrain Display



Unlike the Map page and HSI Map, the Terrain page does not provide controls for toggling overlays on or off.



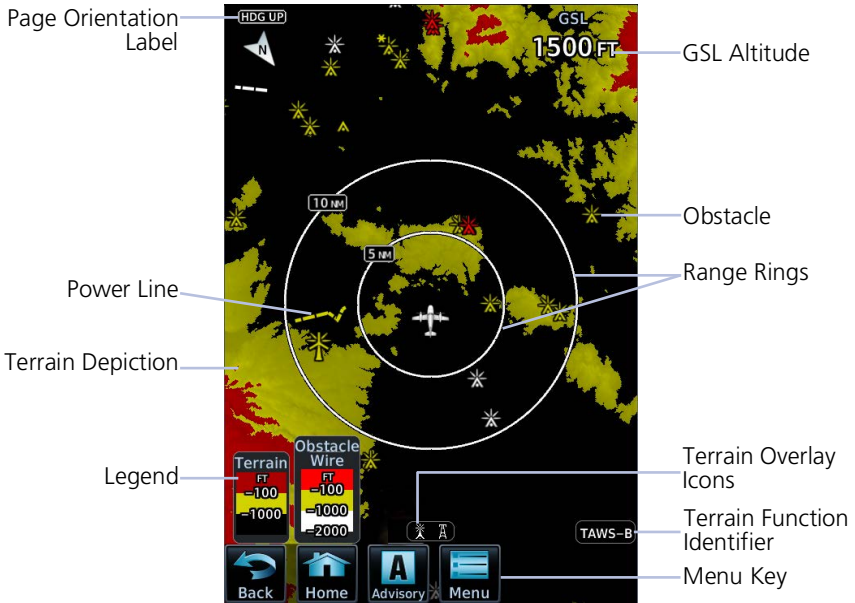
WARNING

Do not use Terrain and obstacle data to navigate or maneuver around terrain. They are an aid to situational awareness only.

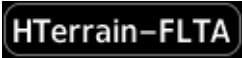
FEATURE REQUIREMENTS

All terrain functions require the following components to operate properly.

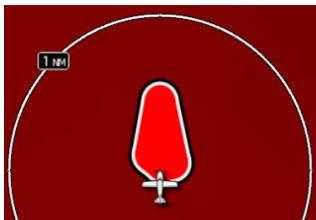
- Valid 3D GPS position for terrain and obstacle data display
- Valid terrain/obstacle database



Terrain Display Features

TERRAIN DISPLAY OBJECTS	
Page Orientation Label	Normally, Heading Up orientation. Changes to Track Up in the event of a heading failure.
GSL Altitude	Displays current GPS height above mean sea level. Pilot selectable units are available on the System Units page.
Legend	Displays color designations for terrain and obstacle relative altitude ranges.
Terrain Function Identifier	Indicates the active terrain function. Includes:  TAWS-A, TAWS-B, (H)Terrain-FLTA, Terrain Proximity
Terrain Overlay Icon	Indicates when power lines or obstacles are present at the current zoom scale.  Automatic removal of obstacle and power line data occurs at range scales greater than 10 nm.
Menu Key	Provides access to pilot selectable settings, self-test, and alert inhibit functions.

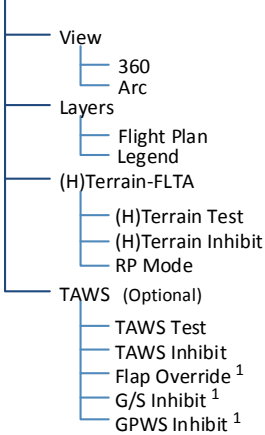
AUTOMATIC ZOOM



In the event an alert occurs, the page automatically zooms to provide the best depiction of that alerted terrain, obstacle, or power line.

Terrain Setup

Terrain Page Menu



Tap **Menu** to access pilot selectable terrain settings, including self test and alert inhibit functions.

¹ TAWS-A only.

SELECTION		FUNCTION
View	360	Changes view format to a 360° ring encircling the aircraft (default view)
	Arc	Changes view format to a forward-looking 120° arc
Layers	Flight Plan	Toggles the active flight plan display over terrain map on or off
	Legend	Toggles the Terrain and Obstacle/Wire legend on or off
(H)Terrain FLTA	(H)Terrain Test	- Performs terrain alerting system test - Verifies the validity of required databases
	(H)Terrain Inhibit	Inhibits the FLTA aural and visual alerts
	RP Mode ¹	Reduces alerting thresholds for low-level operations

SELECTION		FUNCTION
TAWS-A	Flap Override	• Overrides flap-based FIT alerting while other FIT alert functions remain in effect • Inhibits nuisance FIT alerts where flap extension is not desired
	G/S Inhibit	• Inhibits glideslope or glidepath alerts depending on current state • Use to prevent glideslope/glidepath deviation alerts (e.g., when flying a localizer backcourse approach) • Active only for a single approach
	GPWS Inhibit	• Inhibits GPWS audible and visual alerts (i.e., EDR, ECR, FIT, and NCR)
TAWS-A & B	TAWS Test	• Performs TAWS alerting system test • Verifies the validity of required databases
	TAWS Inhibit	• Inhibits PDA/FLTA, EDR, and NCR audible and visual alerts

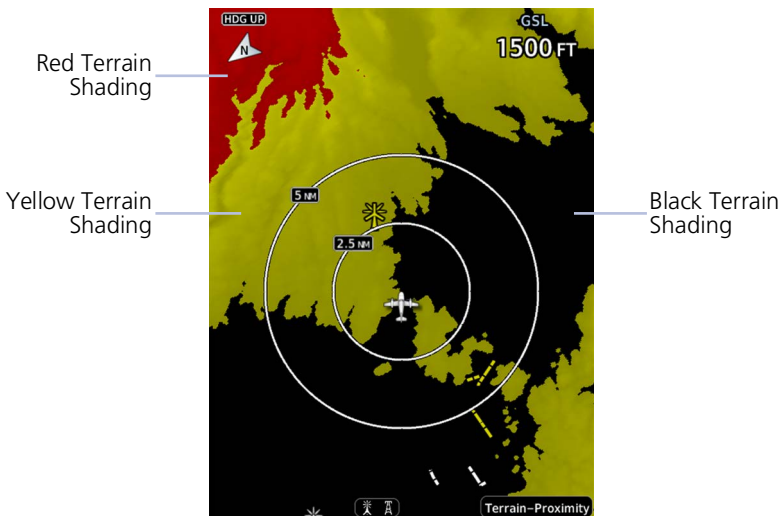
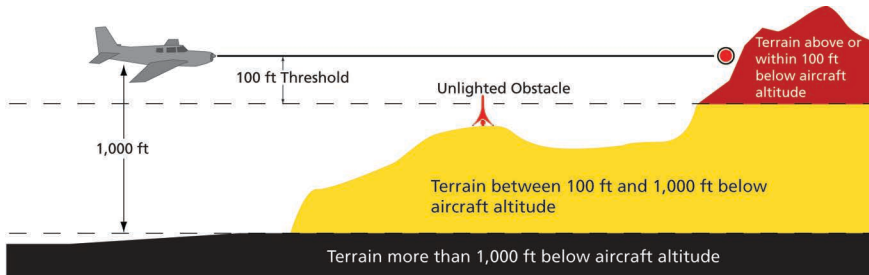
¹ Rotorcraft only.

Terrain & Obstacle Depictions

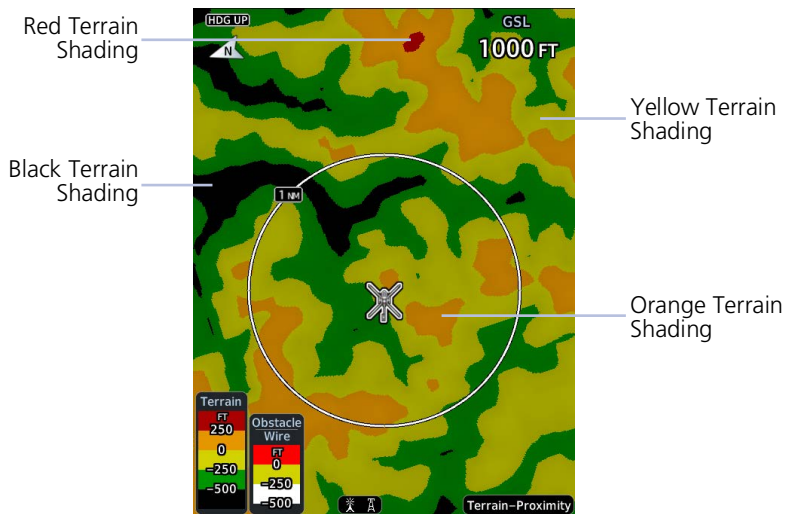
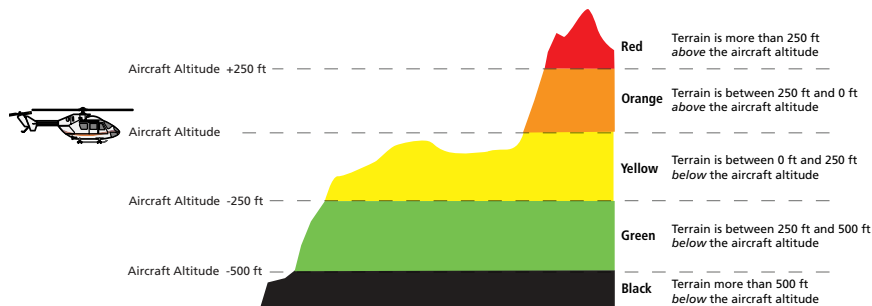
Terrain Elevation Depictions

Color shading depicts terrain elevations relative to the aircraft's position and altitude. Colors automatically adjust as the aircraft's altitude changes.

TERRAIN SHADING, FIXED WING



TERRAIN SHADING, ROTORCRAFT



Obstacle Elevation Depictions

FEATURE LIMITATIONS

- Fixed-wing aircraft: obstacles more than 2,000 ft below current altitude do not display
- Rotorcraft: obstacles more than 500 ft below current altitude do not display
- Power line depictions in synthetic vision are currently not available

TOWER OBSTACLES

UNLIGHTED OBSTACLE		LIGHTED OBSTACLE		OBSTACLE LOCATION	
<1000' AGL	>1000' AGL	<1000' AGL	>1000' AGL	FIXED WING	ROTORCRAFT
				Red obstacle is above or within 100 ft below current altitude.	Red obstacle is at or above current altitude.
				Yellow obstacle is between 100 ft and 1000 ft below current altitude.	Yellow obstacle is within 250 ft below current altitude.
				White obstacle is between 1,000 ft and 2,000 ft below current altitude.	White obstacle is between 250 ft and 500 ft below current altitude.

WIND TURBINE OBSTACLES

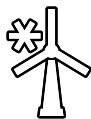
UNLIGHTED WIND TURBINE OBSTACLE	LIGHTED WIND TURBINE OBSTACLE	OBSTACLE LOCATION	
		FIXED WING	ROTORCRAFT
		Red obstacle is above or within 100 ft below current altitude.	Red obstacle is at or above current altitude.
		Yellow obstacle is between 100 ft and 1,000 ft below current altitude.	Yellow obstacle is within 250 ft below current altitude.
		White obstacle is between 1,000 ft and 2,000 ft below current altitude.	White obstacle is between 250 ft and 500 ft below current altitude.

POWER LINE OBSTACLES

OBSTACLE	OBSTACLE LOCATION	
	FIXED WING	ROTORCRAFT
	Red power line is above or within 100 ft below current altitude.	Red power line is at or above current altitude.
	Yellow power line is between 100 ft and 1,000 ft below current altitude.	Yellow power line is within 250 ft below current altitude.
	White power line is between 1,000 ft and 2,000 ft below current altitude.	White power line is between 250 ft and 500 ft below current altitude.

OBSTACLE GROUPS

With the exception of power lines, obstacles within close proximity of each other may display as a group. When this occurs:



- An asterisk indicates that the obstacle belongs to a group
- The relative altitude of the highest obstacle determines color
- The depiction is of the most immediate threat

In most cases, however, TXi displays these obstacles individually.

Terrain Proximity

FEATURE LIMITATIONS

- *Terrain and obstacle depictions are relative to aircraft altitude*
- *Obstacle depictions are dependent upon database*
- *Setup does not provide visual or aural alerts*

Terrain Proximity Features

- Non-TSO C151c certified terrain display system
- 2-D graphical representation of surrounding terrain, obstacles, and power lines relative to aircraft position and altitude
- Declutter automatically removes obstacle and power line data at large ranges
- Continuous monitoring of database validity, GPS and hardware status
- Displays when higher level terrain functions are active

Terrain Alerts

The behavior of an alerting function is determined at installation. Installer configurable settings allow:

- Alert suppression for specific runway types
- Gender selection for voice messages
- Volume level

Alert Types

TERRAIN-FLTA
• Imminent Impact • Reduced Clearance
TAWS-A
• Imminent Impact • Reduced Clearance • Premature Descent • Excessive Descent Rate • Excessive Closure Rate • Negative Climb Rate • Flight Into Terrain • Excessive Below Glideslope/Glidepath Deviation
TAWS-B
• Imminent Impact • Reduced Clearance • Premature Descent • Excessive Descent Rate • Negative Climb Rate

Available alerting functions depend on the installed terrain system.

Imminent Impact and Required Clearance alerts are functions of Forward Looking Terrain Avoidance (FLTA).

ALERT TYPE		CONDITION
FLTA	Imminent Impact¹	Aircraft reaches the minimum clearance altitude of any obstacle (IOI), terrain (ITI), or power line (ILI) in the projected flight path.
	Reduced Clearance¹	Aircraft's vertical flight path is projected to be within the minimum clearance altitude of an obstacle (ROC), terrain (RTC), or power line (RLC).
Premature Descent²		Aircraft is significantly below the normal approach path for the nearest runway. - Altitude is <700 ft above terrain - Distance from destination airport is 15 nm or less
Excessive Descent Rate		Aircraft descends toward terrain at an excessive rate.
Excessive Closure Rate³		Aircraft closes upon terrain at a rate excessive for gear and flaps.
Negative Climb Rate		Aircraft loses altitude following takeoff. - Altitude is <700 ft above terrain - Distance from departure airport is 5 NM or less - Deviation from departure heading is <110°
Flight Into Terrain		Aircraft is too low with respect to terrain. Based on landing gear status, flap position, and ground speed.
Excessive Below Glideslope or Glidepath Deviation		Aircraft is significantly below the glidepath for the selected approach. Active only after departure and when the following conditions are met. - Altitude is <1,000 ft AGL - Flaps are in landing configuration - Gear is configured for landing - ILS, LPV, or LNAV/VNAV approach is active and the unit is indicating vertical navigation

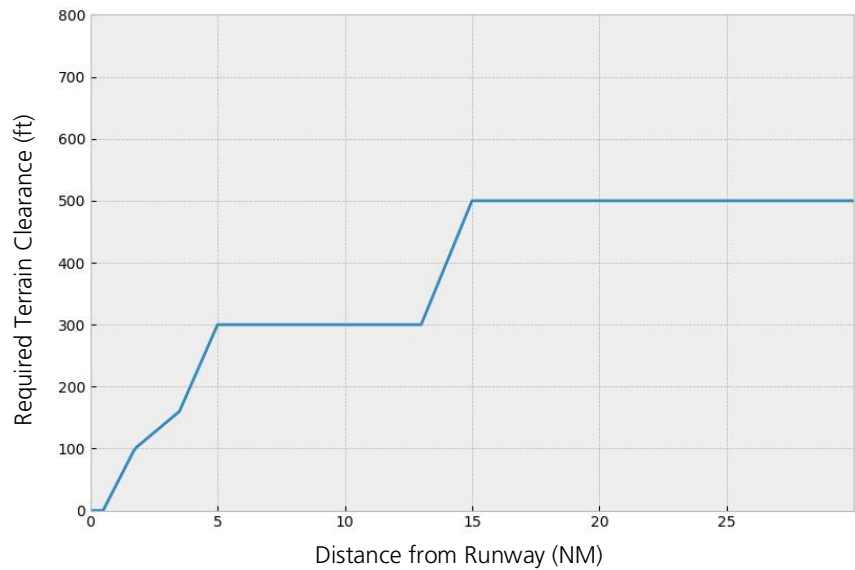
¹ Alerting inhibited <200 ft AGL within 0.5 nm of approach runway or <125 ft AGL within 1.0 nm of runway threshold.

² Alerting inhibited within 0.5 nm of approach runway or <125 ft AGL within 1.0 nm of runway threshold. Alerting thresholds for final descent are based on current position, speed, and flight path data.

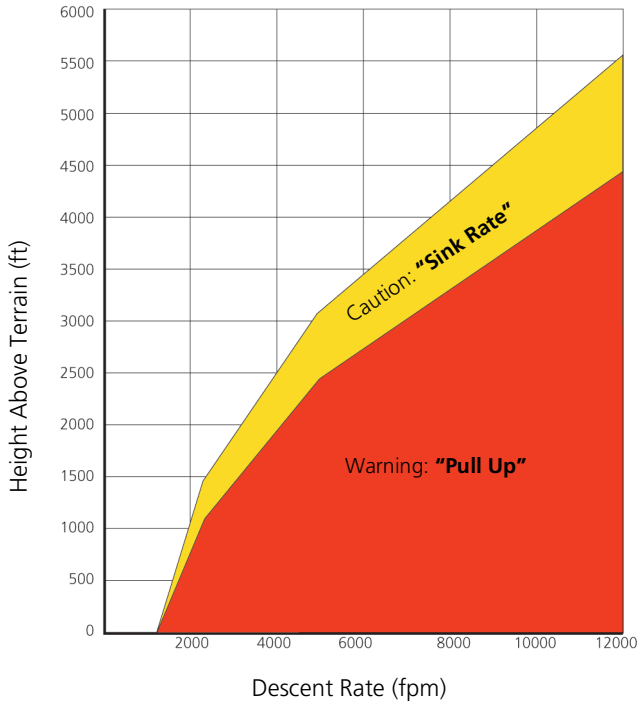
³ Alerting inhibited within 5 nm of nearest airport, except when FLTA is not available. In such cases, "TAWS N/A" or "TAWS FAIL" annunciates and ECR alerting remains active until landing.

Alerting Thresholds

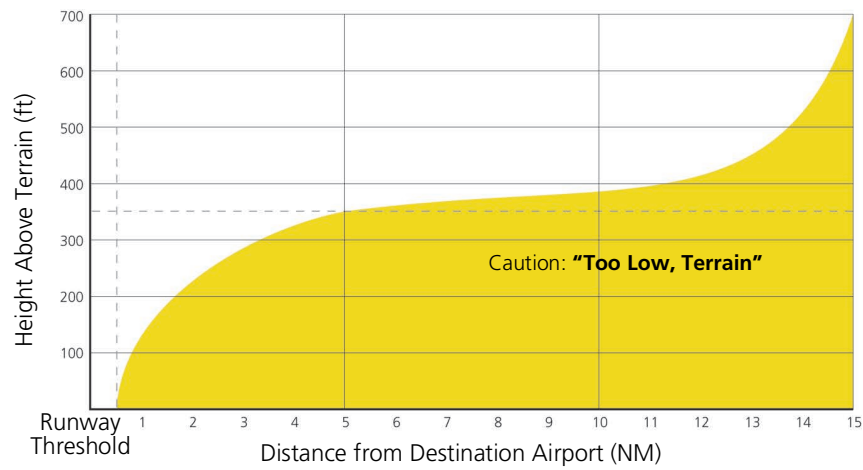
FLTA SEARCH VOLUME - REQUIRED TERRAIN CLEARANCE



EDR THRESHOLDS

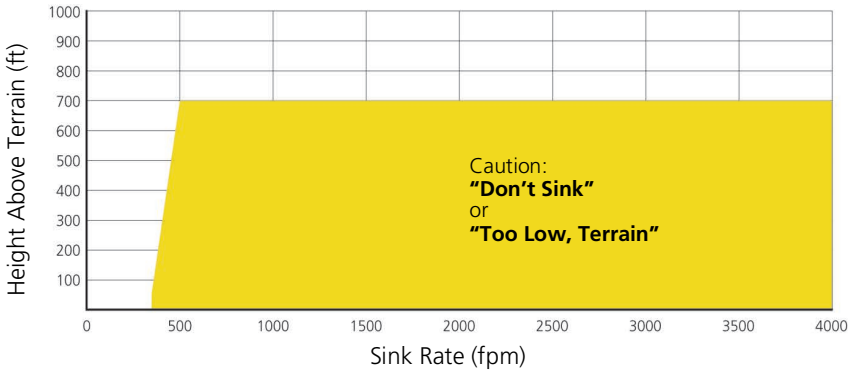
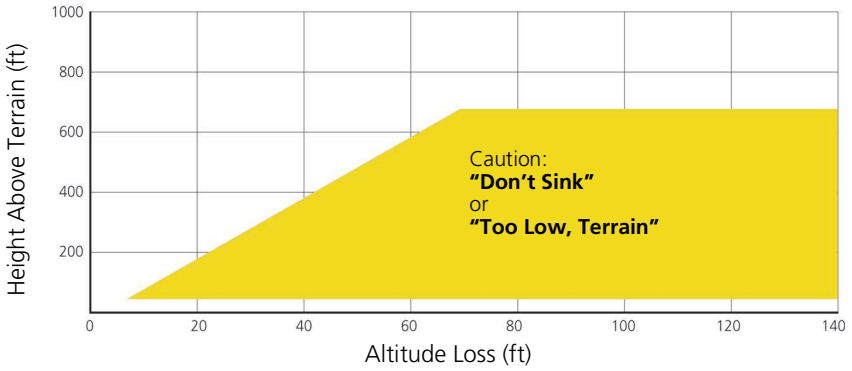


PDA THRESHOLD



NCR THRESHOLDS

Alert triggers: altitude loss, sink rate



Alert Inhibit



The **Terrain Inhibit** control is accessible via the terrain pop-up alert or Terrain page menu.



NOTE

Always use discretion when inhibiting TAWS or Terrain-FLTA alerts. Re-activate the alert function when appropriate.

TERRAIN INHIBIT FUNCTIONS	
INHIBIT	Manually inhibits TAWS or Terrain-FLTA aural and visual alerts for low altitude approaches or rotorcraft operation.
AUTOMATIC INHIBIT	Automatically inhibits TAWS and Terrain-FLTA alerts when the aircraft meets the following approach criteria. TAWS: • GPS/SBAS approach • Position inside FAF TAWS & TERRAIN-FLTA: • Altitude <200 ft above runway elevation • Position <0.5 nm of approach end or between each runway end

TAWS-A INHIBIT ANNUNCIATIONS

Terrain Page



TAWS-A alert inhibit annunciations appear at the bottom right of the display.

- “FLAP OVRD” does not annunciate if GPWS Inhibit is already active, as both functions inhibit FIT alerts.
- A plus sign indicates multiple alerts (e.g., “TAWS INHB+”)

Terrain-FLTA

Terrain-FLTA Features

- Non-TSO C151c certified terrain alerting system
- Aural and visual alerts for potential flight path conflicts involving terrain, obstacles, or power lines
- FLTA functions: RTC, RLC, ROC, ITI, ILI, and IOI
- Terrain depictions and display overlays
- Cautions and warnings indicate alert severity and threat type
- Textual annunciations on PFD and MFD
- Pop-up alerts on MFD
- Threat location indication on map displays (Map page, HSI Map, Terrain page) and SVT
- Self-test and alert inhibit functionality. FLTA is active whether synthetic vision is on or off
- Reduced protection mode (rotorcraft only)

Reduced Protection Mode



ROTORCRAFT ONLY

FEATURE LIMITATIONS

- *Alerting times are significantly less than during normal operating mode*

During reduced protection mode, alerting times are significantly less than during normal operating mode. Use this function only when terrain is within sight.



PFD Annunciation

- Allows rotorcraft-configured systems to operate using reduced terrain alerting system thresholds
- Suppresses caution alerts during low level operations and off airport landings while providing warning alert protection from terrain and obstacles



MFD Annunciation

Terrain-FLTA Alerts

TERRAIN-FLTA ALERT INDICATIONS	
ANNUNCIATION	CONDITION POP-UP AURAL MESSAGE
TERRAIN	Condition: FLTA Terrain Warning (RTC-W, ITI-W) Pop-up Alert: Yes Aural Message: "Warning, Terrain, Terrain"
OBSTACLE	Condition: FLTA Obstacle Warning (ROC-W, IOI-W) Pop-up Alert: Yes Aural Message: "Warning, Obstacle, Obstacle"
WIRE	Condition: FLTA Wire Warning (RLC-W, ILI-W) Pop-up Alert: Yes Aural Message: "Warning, Wire, Wire"
TERRAIN	Condition: FLTA Terrain Caution (RTC-C, ITI-C) Pop-up Alert: Yes Aural Message: "Caution, Terrain, Terrain"
OBSTACLE	Condition: FLTA Obstacle Caution (ROC-C, IOI-C) Pop-up Alert: Yes Aural Message: "Caution, Obstacle, Obstacle"
WIRE	Condition: FLTA Wire Caution (RLC-C, ILI-C) Pop-up Alert: Yes Aural Message: "Caution, Wire, Wire"

TERRAIN-FLTA SYSTEM STATUS

TERRAIN-FLTA SYSTEM STATUS INDICATIONS	
ANNUNCIATION	CONDITION AURAL MESSAGE
None	Condition: Terrain system transitions from unavailable to available Aural Message: "Terrain System Available"
TER TEST	Condition: Terrain system test in progress Aural Message: None
None	Condition: Terrain system test is successful Aural Message: "Terrain System Test Okay"
TER INHB	Condition: Terrain alerting is disabled Aural Message: None
TER N/A	Condition: Terrain is not available due to one of the following: • No certified GPS position • Degraded GPS signal • Outside of the terrain database regional coverage Aural Message: "Terrain System Not Available"
TER FAIL	Condition: • Terrain system test fails • The database is missing or corrupt Aural Message: "Terrain System Failure"

TAWS-B

FEATURE REQUIREMENTS

- *Valid 3D GPS position*
- *TAWS-B feature enablement*
- *External annunciator for displaying alert information (non-PFD GDUs only)*

TAWS-B Features

- Optional TSO C151c Class B terrain alerting system
- All Terrain-FLTA functions plus: premature descent, excessive descent rate, negative climb rate, and altitude voice callout (500 ft) alerts

TAWS-B Alerts

TAWS-B ALERT INDICATIONS	
ANNUNCIATION	CONDITION POP-UP AURAL MESSAGE
PULL UP	Condition: FLTA Terrain Warning (RTC-W, ITI-W) Pop-up Alert: Yes Aural Message: “Terrain Ahead, Pull Up; Terrain Ahead, Pull Up” or “Terrain, Terrain; Pull Up, Pull Up”
PULL UP	Condition: FLTA Obstacle Warning (ROC-W, IOI-W) Pop-up Alert: Yes Aural Message: “Obstacle Ahead, Pull Up; Obstacle Ahead, Pull Up” or “Obstacle, Obstacle; Pull Up, Pull Up”
PULL UP	Condition: FLTA Wire Warning (RLC-W, ILI-W) Pop-up Alert: Yes Aural Message: “Wire Ahead, Pull Up; Wire Ahead, Pull Up” or “Wire, Wire; Pull Up, Pull Up”
PULL UP	Condition: Excessive Descent Rate Warning (EDR-W) Pop-up Alert: Yes Aural Message: “Pull Up”
TERRAIN	Condition: FLTA Terrain Caution (RTC-C, ITI-C) Pop-up Alert: Yes Aural Message: “Terrain Ahead; Terrain Ahead” or “Caution, Terrain; Caution, Terrain”

TAWS-B ALERT INDICATIONS	
ANNUNCIATION	CONDITION POP-UP AURAL MESSAGE
OBSTACLE	Condition: FLTA Obstacle Caution (ROC-C, IOI-C) Pop-up Alert: Yes Aural Message: "Obstacle Ahead; Obstacle Ahead" or "Caution, Obstacle; Caution, Obstacle"
WIRE	Condition: FLTA Wire Caution (RLC-C, ILI-C) Pop-up Alert: Yes Aural Message: "Wire Ahead; Wire Ahead" or "Caution, Wire; Caution, Wire"
TERRAIN	Condition: Premature Descent Alert Caution (PDA-C) Pop-up Alert: Yes Aural Message: "Too Low, Terrain"
TERRAIN	Condition: Excessive Descent Rate Caution (EDR-C) Pop-up Alert: Yes Aural Message: "Sink Rate"
TERRAIN	Condition: Negative Climb Rate Caution (NRC-C) Pop-up Alert: Yes Aural Message: "Don't Sink" or "Too Low, Terrain"
None	Condition: Voice Call Out (VCO-500) Aircraft is descending through 500 ft AGL Aural Message: "Five Hundred"

TAWS-B SYSTEM STATUS

TAWS-B SYSTEM STATUS INDICATIONS	
ANNUNCIATION	CONDITION AURAL MESSAGE
None	Condition: TAWS transitions from unavailable to available after: - GPS signal integrity returns or - The aircraft is back within the database coverage area Aural Message: "TAWS Available"
TAWS TEST	Condition: TAWS system test in progress Aural Message: None
None	Condition: TAWS system test is successful Aural Message: "TAWS System Test Okay"
TAWS INHB	Condition: TAWS alerting is disabled Aural Message: None
TAWS N/A	Condition: TAWS is not available due to one of the following: - GDU is using backup GPS - The navigation solution has become degraded - Aircraft is beyond the database coverage area Aural Message: "TAWS Not Available"
TAWS FAIL	Condition: - TAWS system test fails - Incorrect TAWS configuration - A necessary database is missing or corrupt - TAWS audio fault Aural Message: "TAWS System Failure"

External TAWS/HTAWS

FEATURE REQUIREMENTS

- *TAWS/HTAWS function provided by a source external to the GDU*

FEATURE LIMITATIONS

- *TAWS/HTAWS controls are available only on the external TAWS/HTAWS device*

Garmin External TAWS/HTAWS function

- Terrain function relies on an external Garmin GPS device for all TAWS/HTAWS data¹
- GDU determines terrain map coloration based on the external GPS position and GPS altitude
- **On the PFD:** Changes in CDI source selection have no effect on TAWS/HTAWS alerting functions

¹ For more information, consult the applicable GNS 500W series or GTN series pilot’s guide and/or addendum.

TAWS/HTAWS SOURCE	TXI FEATURES
GTN	• TAWS/HTAWS alert depictions on the Terrain page, Map page, and in synthetic vision • Terrain depictions and display overlays • Textual annunciations on PFD • Pop-up alerts on MFD
GNS	• Terrain depictions and display overlays • Textual annunciations on PFD
Third-party TAWS	• GDU display provides only Terrain Proximity functionality

TAWS-A

FEATURE REQUIREMENTS

- *Valid 3D GPS position*
- *GTN with TAWS-A feature enablement*
- *Valid flap and landing gear status inputs into the GTN*
- *Valid radar altimeter*
- *External annunciator for displaying alert information (non-PFD GDUs only)*

TAWS-A Features

- Optional TSO C151c Class A terrain alerting system
- All Terrain-FLTA functions plus: premature descent, excessive descent rate, negative climb rate, excessive glideslope/glidepath deviation, flap override, flight into terrain, and altitude voice callout (500 ft) alerts
- Terrain depictions and display overlays
- Textual annunciations on PFD
- Pop-up alerts on MFD

On the PFD: Changes in CDI source selection have no effect on TAWS alerting functions.

For more information, consult the applicable GTN series pilot's guide.

TAWS-A Alerts

TAWS-A ALERT INDICATIONS	
ANNUNCIATION	CONDITION POP-UP AURAL MESSAGE
PULL UP	Condition: FLTA Terrain Warning (RTC-W, ITI-W) Pop-up Alert: Yes Aural Message: “Terrain Ahead, Pull Up; Terrain Ahead, Pull Up” or “Terrain, Terrain; Pull Up, Pull Up”
PULL UP	Condition: FLTA Obstacle Warning (ROC-W, IOI-W) Pop-up Alert: Yes Aural Message: “Obstacle Ahead, Pull Up; Obstacle Ahead, Pull Up” or “Obstacle, Obstacle; Pull Up, Pull Up”
PULL UP	Condition: FLTA Wire Warning (RLC-W, ILI-W) Pop-up Alert: Yes Aural Message: “Wire Ahead, Pull Up; Wire Ahead, Pull Up” or “Wire, Wire; Pull Up, Pull Up”
PULL UP	Condition: Excessive Descent Rate Warning (EDR-W) Pop-up Alert: Yes Aural Message: “Pull Up”
PULL UP	Condition: Excessive Closure Rate Warning (ECR-W) Pop-up Alert: Yes Aural Message: “Pull Up”
TERRAIN	Condition: FLTA Terrain Caution (RTC-C, ITI-C) Pop-up Alert: Yes Aural Message: “Terrain Ahead; Terrain Ahead” or “Caution, Terrain; Caution, Terrain”

TAWS-A ALERT INDICATIONS	
ANNUNCIATION	CONDITION POP-UP AURAL MESSAGE
OBSTACLE	Condition: FLTA Obstacle Caution (ROC-C, IOI-C) Pop-up Alert: Yes Aural Message: "Obstacle Ahead; Obstacle Ahead" or "Caution, Obstacle; Caution, Obstacle"
WIRE	Condition: FLTA Wire Caution (RLC-C, ILI-C) Pop-up Alert: Yes Aural Message: "Wire Ahead; Wire Ahead" or "Caution, Wire; Caution, Wire"
TERRAIN	Condition: Premature Descent Alert Caution (PDA-C) Pop-up Alert: Yes Aural Message: "Too Low, Terrain"
TERRAIN	Condition: Excessive Descent Rate Caution (EDR-C) Pop-up Alert: Yes Aural Message: "Sink Rate"
TERRAIN	Condition: Excessive Closure Rate Caution (ECR-C) Pop-up Alert: Yes Aural Message: "Terrain, Terrain"
TERRAIN	Condition: Negative Climb Rate Caution (NRC-C) Pop-up Alert: Yes Aural Message: "Don't Sink" or "Too Low, Terrain"
TERRAIN	Condition: Flight Into Terrain High Speed Caution (FIT-C) Pop-up Alert: Yes Aural Message: "Too Low, Terrain"
TERRAIN	Condition: Flight Into Terrain Gear Caution (FIT-C) Pop-up Alert: Yes Aural Message: "Too Low, Gear"

TAWS-A ALERT INDICATIONS	
ANNUNCIATION	CONDITION POP-UP AURAL MESSAGE
TERRAIN	Condition: Flight Into Terrain Flaps Caution (FIT-C) Pop-up Alert: Yes Aural Message: "Too Low, Flaps"
TERRAIN	Condition: Flight Into Terrain Takeoff Caution (FIT-C) Pop-up Alert: Yes Aural Message: "Too Low, Terrain"
GLIDESLOPE	Condition: Glideslope Deviation Caution (GSD-C) Pop-up Alert: Yes Aural Message: "Glideslope"
None	Condition: Voice Call Out (VCO-500) Aircraft is descending through 500 ft AGL Aural Message: "Five Hundred"

TAW-A SYSTEM STATUS

TAW-A SYSTEM STATUS INDICATIONS	
ANNUNCIATION	CONDITION AURAL MESSAGE
None	Condition: TAWS transitions from unavailable to available after: • GPS signal integrity returns or • The aircraft is back within the database coverage area Aural Message: "TAWS Available"
TAWS TEST	Condition: TAWS system test in progress Aural Message: None
None	Condition: TAWS system test is successful Aural Message: "TAWS System Test Okay"
TAWS INHB	Condition: TAWS alerting is disabled Aural Message: None
TAWS N/A	Condition: TAWS is not available due to one of the following: • GDU is using backup GPS • The navigation solution has become degraded • Aircraft is beyond the database coverage area Aural Message: "TAWS Not Available"
TAWS FAIL	Condition: • TAWS system test fails • Incorrect TAWS configuration • A necessary database is missing or corrupt • TAWS audio fault Aural Message: "TAWS System Failure"
GPWS N/A	Condition: GPWS is not available due to one of the following: • Incorrect TAWS configuration • Radar altimeter unavailable • GPS position unavailable/degraded • TAWS audio fault Aural Message: None

TAWS-A SYSTEM STATUS INDICATIONS	
ANNUNCIATION	CONDITION AURAL MESSAGE
GPWS FAIL	Condition: • Incorrect TAWS configuration • Radar altimeter unavailable • GPS position unavailable/degraded • TAWS audio fault Aural Message: “GPWS System Failure”
G/S INHB	Condition: Glideslope Inhibit function active Aural Message: None
FLAP OVRD	Condition: Flap Override function active Aural Message: None

9 Fuel & Engine Indication System

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Reciprocating Engines

EIS Display

The EIS displays engine, electrical, and fuel information using parameter specific indicator types (e.g., analog indicators, digital data fields, bar graphs, horizontal and vertical bar gauges).

The EIS instruments replace traditional analog gauges used for monitoring engine parameters. Always consult the AFM or POH for engine operating limitations.

Layout varies according to:

- Display type
- Number and type of engines in aircraft
- Number and type of installed sensors

GDU 700() EIS

GDU 700P EIS



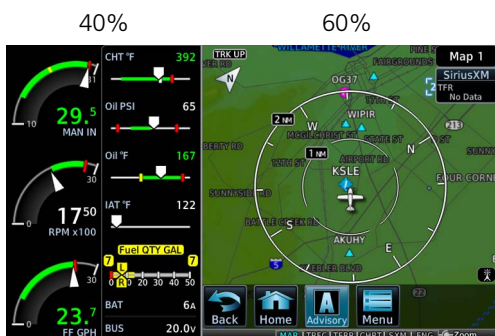
Multi-engine

When configured for EIS only, GDU 700() is a dedicated full-screen display of engine instrumentation.

GDU 700L EIS



Single Engine



GDU 700L with EIS/MFD

When configured for EIS/MFD, GDU 700L dedicates 40% of its screen to a full-time display of primary EIS information, with the rest of the screen serving an MFD function. This configuration supports single-engine aircraft.

GDU 1060/1210 EIS



With the display of EIS enabled, GDU 1060/1210 dedicates 20% of its screen to a full-time display of primary EIS information. In addition, a dedicated Engine page is accessible from the MFD Home page. This page is available even if the EIS display is not enabled.

Tap the **Engine** icon to open the corresponding page (Home > **Engine**).

The MFD Engine page provides expanded information for select configured gauges. Other features include pilot selectable user fields, a graphical depiction of engine CHT/EGT/TIT values¹, and a **Menu** access key.

MFD Engine Page, GDU 1060/1210



Gauge Strip, GDU 1060/1210

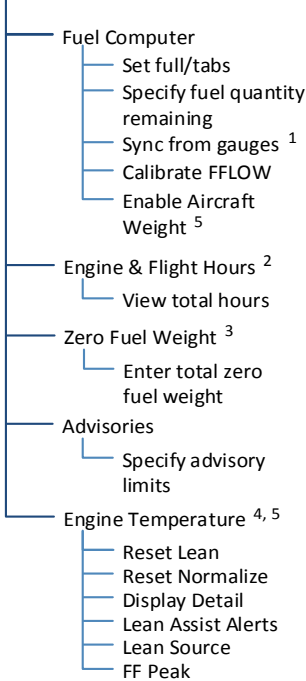


The gauge strip presents an unobstructed compact view of primary engine information. Depending on configuration, the gauge strip may reside at the left or right edge of the GDU 1060/1210 display. Unlike the Engine page, the gauge strip has no selectable features.

¹ Primary EGT and TIT dependent upon aircraft type.

EIS Setup

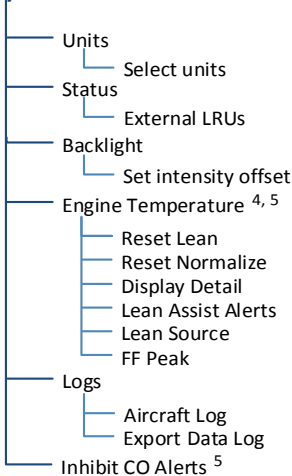
Engine Menu



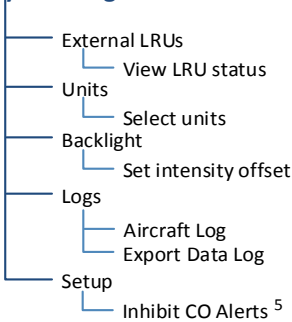
The **Menu** key provides access to the fuel computer, engine and flight hours, engine advisories, and lean modes. On GDU 700(), the Engine menu also provides controls for customizing system settings.

GDU 700() EIS setup functions are accessible from the System menu. On GDU 1060/1210, they reside on the MFD System page.

System	GDU 700()
--------	------------



System Page	GDU 1060/1210
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¹ Available with TXi software v3.21 and later. ² This page is informational only.
³ Available with TXi software v3.61 and later. Availability dependent upon configuration.
⁴ Mode selections dependent upon aircraft configuration.
⁵ Available with TXi software v3.80 and later. Availability dependent upon configuration.

EIS SETUP SELECTIONS	
Units	Specify units of measure for fuel computer and pilot selectable gauges. Selections synchronize across all configured GDUs.¹ • Distance • Fuel Computer • Temperature
Status	• View unit and software information • Check status of all configured LRUs
Backlight	• Adjust display brightness
Aircraft Log	• View engine and airframe cycle counters
Export Data Log	• Save logged data to SD card
Inhibit CO Alerts	• Toggle CO caution alerts on or off
Engine Temperature	Access available CHT/EGT mode options.² • Reset Lean • Reset Normalize • Display Detail • Lean Assist Alerts • Lean Source • FF Peak

¹ Engine gauge units are not adjustable. ² Options dependent upon configuration.

Gauge & Indicator Types

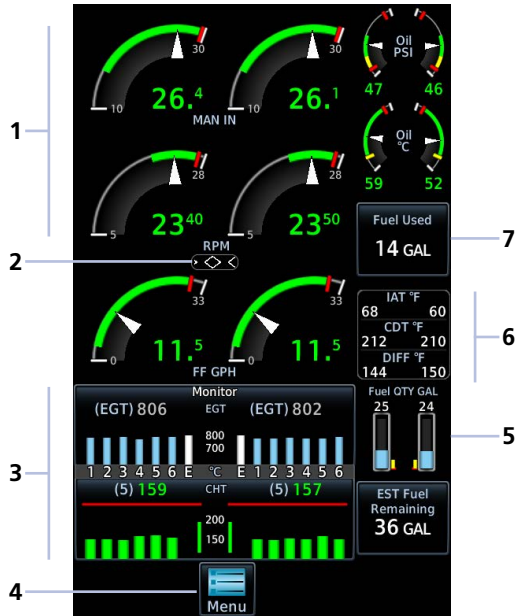
GDU 700P EIS Configured for Dial Gauges



Multi-engine

1	Standard Dial Gauges	5	Fuel Quantity Gauge
2	Propeller Sync Indicator	6	Dual Digital Data Fields
3	Engine Temperature Graph	7	Selectable User Field
4	Menu Key		

GDU 700P EIS Configured for Arc Gauges



Multi-engine

1	Arc Style Gauges	5	Fuel Quantity Gauge
2	Propeller Sync Indicator	6	Dual Digital Data Fields
3	Engine Temperature Graph	7	Selectable User Field
4	Menu Key		

COMMON EIS DISPLAY ELEMENTS	
Engine Gauges	Display a visual representation and a digital readout of the specified engine parameter value. ¹ The style of gauge (arc or standard dial) depends on configuration. ⁴
Digital Data Fields	Display a digital readout value of the engine parameter. ¹ These include single or dual input values depending on system configuration. Placement corresponds to engine location.
Fuel Quantity Gauges	Display fuel amounts for the specified fuel tanks. ³
Selectable User Fields	Opens a menu of the selectable data fields available for display. The type of parameters available for selection is determined during installation. A white border differentiates selectable user fields from non-selectable digital data fields.
Engine Temperature Graph	Provides digital EGT, CHT, and TIT² cylinder readings in three graphical views: monitor, lean, and normalized. • EGT, CHT, and TIT² digital values are selectable by cylinder • After 10 seconds, values default to the hottest cylinder on each engine • Alternate views via knob turn or swipe
Menu Key	Provides access to the fuel computer, pilot selectable settings, and alert inhibit functions.

Read more about EIS display features in their respective sections.

¹ Limits are configured according to system design or the AFM/POH. They are not pilot selectable.
² Primary EGT and TIT indications are dependent upon aircraft type.
³ Fuel gauges are configured and calibrated during installation.
⁴ Piston arc and standard dial gauge formats are mutually exclusive.

PROP SYNC INDICATOR

On multi-engine aircraft, a prop sync indicator provides a visual reference for synchronizing multi-engine RPM. This indicator uses inputs received from the left and right engine tachometer to show a comparison of the matched RPM.

The style of indicator depends on the configured gauge format: arc or standard dial.

Arc Gauge Format

The speed at which the indicator moves left or right is determined by the RPM differential between the two engines.



Prop Sync
Indicator

- If RPM for both engines is the same, diamond-shaped symbols appear and motion stops.
- If RPM varies between engines, arrow-shaped symbols point and move in direction of the faster engine.

Standard Dial Gauge Format

The speed at which the indicator rotates is determined by the RPM differential between the two engines.



Prop Sync
Indicator

- If RPM for both engines is the same, indicator remains stationary.
- If RPM varies between engines, the indicator rotates in direction of the faster engine.

Gauge Types

Gauge labels are customizable to match aircraft configuration. Labeling presented in the following table may differ from actual labeling on the unit.

LABEL	UNITS	FUNCTION
RPM	RPM	Engine tachometer
MAN	inHg, psi	Engine manifold pressure
PWR	%	Percent power¹
Fuel or Fuel Pressure	psi, bar	Fuel pressure
FF	gph, lt/hr, PPH, kg/hr	Fuel flow
CARB or CARB Temperature	°C, °F	- Carburetor air temperature - Blue arc indicates temperature range where carburetor icing is likely to occur (-15 to 5°C)
Oil or Oil Pressure	psi, HPa, bar	Oil pressure
Oil or Oil Temperature	°C, °F	Oil temperature
CHT	°C, °F	- Cylinder head temperature for hottest cylinder (bar indicator) - Graphical representation of CHT for each cylinder² - Hottest cylinder and cooling rate indications - Selectable temperature values on full EIS layouts³
CT	°C, °F	Coolant temperature⁴
EGT	°C, °F	- Graphical representation of exhaust gas temperature (EGT) for each cylinder² - Peak EGT - Selectable temperature values^{5, 6} - Average exhaust gas temperature for engine (primary EGT)⁷
TIT	°C, °F	Turbine inlet temperature
IAT	°C, °F	Inlet air temperature
CDT	°C, °F	Compressor discharge temperature

LABEL	UNITS	FUNCTION
IAT CDT Diff	°C, °F	IAT, CDT, and IAT/CDT temperature differential
ISA	°C, °F	Degrees deviation from the International Standard Atmosphere model⁸
RAT	°C, °F	Outside ram air temperature⁸
SAT	°C, °F	Outside static air temperature⁸
TAT	°C, °F	Outside total air temperature⁸
Fuel or Main Fuel	lb, lt, gal, kg	Current fuel quantity in main tanks
Tip Fuel	lb, lt, gal, kg	Current fuel quantity in tip tanks
Aux Fuel	lb, lt, gal, kg	Current fuel quantity in auxiliary tanks
Bus Volts	V	Bus voltage
Bat Volts	V	Battery voltage
ALT AMPS or ALT %	A, %	Alternator load
BAT Amps	A	Battery load
Vac or Vacuum	in Hg, PSI	Vacuum/pressure
Rudder or Rudder Trim	° (degrees)	Rudder trim

¹ Cirrus aircraft only. ² Due to limited space, not all cylinders display on EIS/MFD layouts.

³ GDU 700() or GDU 1060/1210 Engine page.

⁴ Replaces CHT when coolant temperature sensor is configured.

⁵ Graph labels are dependent upon aircraft configuration (e.g., “E” indicates Primary EGT; “T” indicates TIT).

⁶ EIS gauge strip presents Primary EGT (if installed) or hottest of the individual EGTs.

⁷ Range and units are dependent upon configuration. ⁸ TXi software v3.61 only.

ENGINE GAUGES

TXi offers two styles of piston gauges: arc and standard dial¹. Both may be configured to include additional features, such as digital readouts, custom labels, and dynamic gauge markings.

Piston arc gauges are truncated, round, and have fewer graduations.

Tachometer



Oil Pressure



Standard round gauges completely replace analog gauges.

Tachometer



Oil Pressure



For aircraft equipped with a starting vibrator, tachometer RPM readings are not accurate during engine cranking. For aircraft that measure engine RPM using P lead sensors, the readings may momentarily fluctuate when selecting operation on a single magneto.

¹ Gauge formats are mutually exclusive and dependent upon configuration.

BAR GAUGES

These gauges display engine parameter information on a single horizontal or vertical bar. They have single or dual pointers depending on engine type.



Linear indications move from left to right on horizontal bar gauges. They move up and down on vertical bar gauges.

Digital Readout



If configured, a corresponding digital readout displays to the right of the gauge.

NON-LINEAR GAUGES



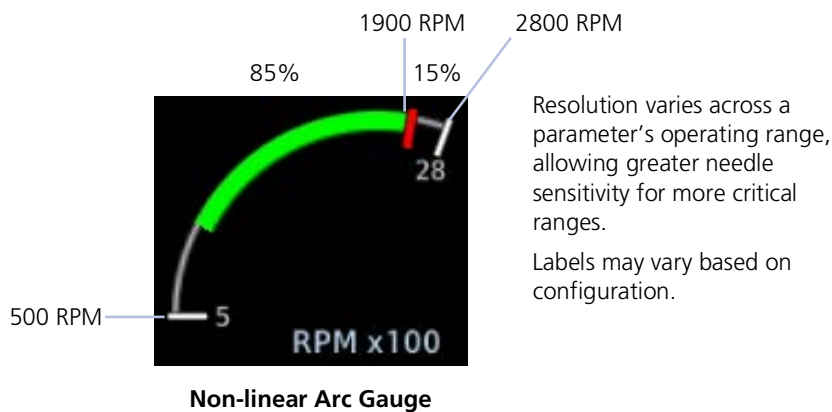
NOTE

Prior to operation, pilots must familiarize themselves with the gauge configuration provided as part of the Configuration Summary. The only other reference to non-linear scaling is current needle position with respect to the digital readout value.

FEATURE LIMITATIONS

- Non-linear scaling not available for piston bar gauges

Some reciprocating engine installations may be configured for non-linear gauges. These are arc gauges that provide non-linear scaling for up to six consecutive segments.



Markings & Indications



NOTE

It is the responsibility of the pilot in command to know and abide by all published limitations and operating ranges in the POH/AFM.

GAUGE MARKING COLOR

Gauge marking colors are in accordance with the criticality of operating limits. Units of measure, limits, and gauge colors are configured during installation.

- Green arc denotes a normal operating range
- Yellow radial line or arc denotes a caution range
- Red minimum/maximum line or arc denotes a limitation

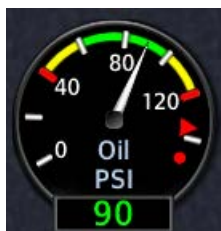
When an indication is within a colored gauge range, the color of the digital readout will be the same as the range.

STATIC REFERENCE MARKINGS

Gauge markings may be configured to denote special gauge values defined in the FMS or placard. They may include: • Arc • Dot¹ • Line/Radial • Minimum Line • Maximum Line • Triangle¹

Reference markings are non-alerting.

Standard Dial Gauge



Arc Gauge



¹ Available with TXi software v3.61 and later.

RANGE INDICATIONS & ALERTING

Alert Suppression

The following gauges do not alert when the tachometer indicates less than 100 RPM and the aircraft is on ground.

- Fuel flow
- Fuel pressure
- Manifold pressure
- Oil pressure

TXi EIS gauges may be configured to alert the pilot when an engine indication:

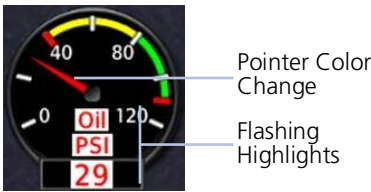
- enters a caution or warning range
- reaches a specified alerting value

When alerted, gauge features change color, and the gauge label and digital readout (if applicable) are highlighted in the appropriate color. Color highlighting may flash depending on configuration.

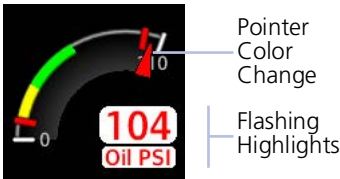
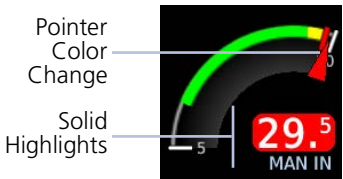
Power Gauge



Performance Gauge



Standard Dial Gauge



Arc Gauge

For alerting purposes, gauges are split into two groups: engine power and engine performance. Alerting functions differ for each group.

GROUP	GAUGE	FUNCTION
Engine Power	Manifold Pressure Tachometer Fuel Flow	• Pointer changes color to reflect non-safe operating status. • Readout field turns solid amber or red depending on alert type.
Engine Performance	Oil Pressure ¹ Oil Temperature Primary EGT CHT CDT Amps/Volts Fuel Pressure Fuel Quantity	• Pointer changes color to reflect non-safe operating status. • Gauge label, readout field, and units (if displayed) flash amber or red depending on alert type. • On-screen ACK key flashes. All flashing indications turn solid once the pilot acknowledges the alert.²

¹ To minimize the number of nuisance alerts, oil pressure gauges do not flash when within the caution band.
² GDU 1060 PFD/MFD/EIS with TXi software earlier than v3.50: On-screen acknowledge key not available. Unless configured for external acknowledge switches, alerts automatically acknowledge after 10 seconds.

DYNAMIC GAUGE MARKINGS

FEATURE LIMITATIONS

- *It is the responsibility of the installer to configure dynamic gauge markings in accordance with the aircraft-specific engine operating limitations listed in the AFM/POH*
- *Dynamic gauge markings may not be able to represent all operating limitations and scenarios depicted in the AFM/POH*
- *Dynamic gauge markings for multi-engine piston EIS and time-delayed dynamic gauge markings are available only with TXi software v3.21 and later*

This feature changes gauge markings based on configured conditions to comprehensively depict safety-critical ranges and markings. For example, manifold pressure gauge markings may be configured to change dynamically based on altitude.

Safety-critical ranges may be configured with time delays for conditions with time dependent limitations.

The EIS is useful for managing engine and aircraft systems within safe operating limits. Dynamic gauges further assist you by presenting the gauge limits and markings appropriate for the current aircraft conditions. This allows for easier gauge interpretation as well as increased safety and efficiency.

Marking Sets

The following examples depict the implementation of dynamic markings in a manifold pressure gauge. Markings are configured to adjust the green arc and red line maximum values based on pressure altitude.

Dynamic Markings (Example)

Standard



The standard gauge markings are configured by the installer to match the existing manifold pressure gauge in the aircraft. This is the default marking set for the gauge.

Standard markings will display if the conditions for dynamic marking are not met.

20,000 ft



Gauge markings adjust to match the maximum manifold pressure for the pressure altitude.

In this example, the maximum manifold pressure decreases as the pressure altitude increases.

24,000 ft



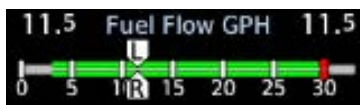
Gauge markings dynamically adjust as pressure altitude thresholds are reached.

Unlike analog gauges, which require you to memorize limitations or interpret multiple markings, dynamic gauges present only the limitations applicable to specific situations or flight conditions.

**NOTE**

Prior to operation, pilots must familiarize themselves with all configured dynamic markings and their associated conditions. These are described in the Configuration Summary for your specific installation.

For multi-engine aircraft, the system is configured such that left and right engine gauges have identical marking logic. The manner in which markings display is determined by the system for each engine independently. As a result, paired left and right engine gauges may show different markings based on current operational differences between engines.

Twin Engine Dynamic Markings (Example)**Dual Pointer Dial Gauge****Dual Bar Gauge****Selectable User Fields**

These fields allow you to select engine and fuel parameters for display on the EIS. Available user fields and selectable parameters are determined during installation.

Tapping any user field opens a list of available engine parameters and their current values. Selecting a parameter displays its value on the EIS and closes the menu.

- Fields may be modified to display alternate data at any time
- Several parameter units are pilot adjustable

EST Fuel Remaining Estimated amount of total fuel remaining ^{1, 3}	Outside Air TEMP (EIS) Outside air temperature as measured by the EIS OAT sensor ⁷
Fuel Used Total fuel used since last update to estimated fuel remaining quantity ^{1, 3}	Outside Air TEMP (ISA) Degrees deviation from the International Standard Atmosphere model ⁴
Range Total range based on remaining fuel and current ground speed ^{1, 3}	Outside Air TEMP (RAT) Outside ram air temperature ^{4, 8}
Endurance (HH+MM) Fuel endurance time in hours+minutes ¹	Outside Air TEMP (SAT) Outside static air temperature ⁴
Fuel at Destination Estimated fuel amount at current flight plan destination ^{1, 2, 3}	Outside Air TEMP (TAT) Outside total air temperature ⁴
Endurance at DEST Remaining fuel endurance time (hours+minutes) at destination ^{1, 2}	CHT DIFF Difference between hottest and coldest CHT for each engine ⁶
Efficiency Fuel efficiency ³	EGT DIFF Difference between hottest and coldest EGT for each engine ⁶
Flight/Hobbs/Tach Hours Total flight, Hobbs, and tach hours ⁵	EST Current Weight Estimated current aircraft weight ^{8, 9}
Clock Current time ⁸	EST Weight at DEST Estimated aircraft weight at flight plan destination ^{8, 10}
Percent Power Percent of maximum rated engine power calculated from manifold pressure, RPM, fuel flow, and outside air temperature	

Read about Aircraft Weight feature enablement and zero fuel weight entry options in *Zero Fuel Weight*.

¹ Values are based on fuel computer calculations. ² Requires an active flight plan from the navigator.

³ Parameter units are pilot selectable. ⁴ Units are independent of PFD units.

⁵ Hobbs hours accumulate when an engine is running. Flight hours accumulate when the aircraft is in air. Tach hours increment at a rate proportional to the configured cruise RPM.

⁶ Units are dependent upon configuration. ⁷ Available with TXi software v3.21 and later.

⁸ Available with TXi software v3.61 and later. ⁹ Requires valid estimated fuel remaining and zero fuel weight values. ¹⁰ Requires valid fuel remaining at destination and zero fuel weight values.

Counters



NOTE

Do not reboot the GDU during flight unless operational procedures dictate. Doing so may register an additional engine or airframe cycle. Contact a Garmin dealer to adjust counters to match other tracking records.

Counter Types

- Hobbs and flight hours
- Tach hours
- Takeoffs and landings

Counters record various types of cycle data (e.g., Hobbs and flight hours, takeoffs and landings). GDU stores and displays this information in the aircraft log.



The aircraft log also stores any maintenance timers set up during configuration.

Home > **System** > **Logs** > **Aircraft Log**

AIRFRAME CYCLE COUNTERS

The system records parameters separately on individual counters. Takeoffs and Landings increment based on GPS, air data, or the weight on wheels discrete.

ENGINE AND FLIGHT HOURS



The system records Hobbs, tach, and flight hours. These parameters may display as a user-selectable field. They are also viewable in the aircraft log.

For convenience, these counters also display at startup.



For twin piston engine aircraft, dual tach timers are available.

CHT/EGT

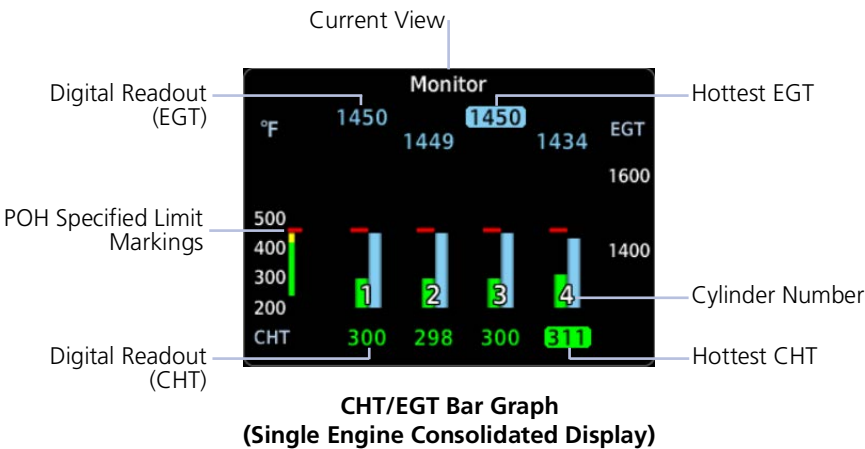
The EIS graphically displays CHT and EGT data for each cylinder.

Each bar graph contains the following components.

- Dynamic chart depicting cylinder temperature status
- Digital temperature reading of the highlighted cylinder
- Limit markings

The size of each graph varies according to aircraft type (i.e., single or twin engine aircraft), the number of cylinders per engine, and gauge configuration (i.e., TIT or primary EGT). Graduations are scaled and sized during installation.

Displayed markings and digital CHT/EGT values depend on view selection. The EIS provides three graphical views: monitor, lean assist (or *lean*), and normalized.



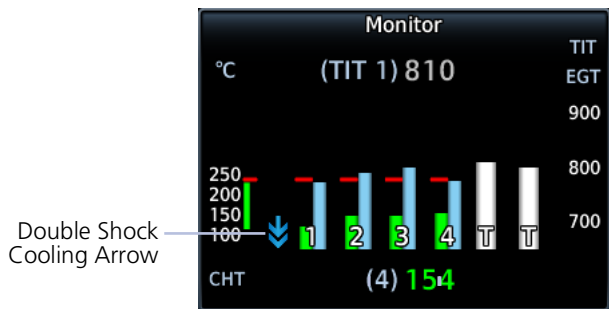
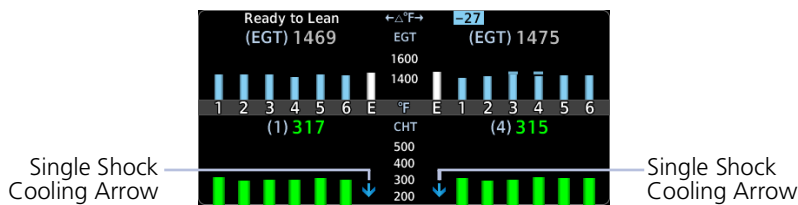
In layouts without space constraints, temperature graphs are customizable to show all digital readouts or only those corresponding to the hottest cylinders. Read more about these display options in *CHT/EGT Control Options*.

SHOCK COOLING INDICATIONS

A blue arrow indicates shock cooling on the temperature graph. The arrow type represents cooling rate.

EIS provides shock cooling indications in all views.

SHOCK COOLING ARROWS	
Blue Single Arrow	CHT is cooling at a rate faster than 30°F (16.7°C) per minute
Blue Double Arrow	CHT is cooling at a rate faster than 60°F (33.3°C) per minute



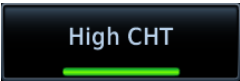
Shock Cooling Indications

VIEW-SPECIFIC COLORS

COLOR DEFINITIONS	
Green	CHT
Blue	EGT
Gray	PEGT/TIT

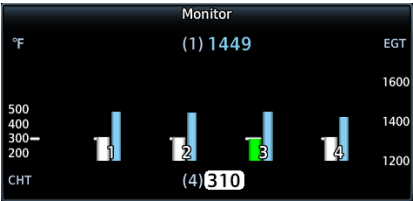
Temperature bar and digital readout colors denote the current view selection.

HIGH CHT ADVISORY



Set the temperature at which the system issues the high CHT advisory.

- 1. Tap **Menu > Advisories > High CHT**.
- 2. Tap the corresponding data entry key and specify the temperature limit value.



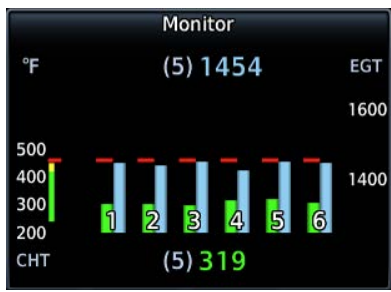
A horizontal white line appears on the CHT graph, indicating the alert temperature threshold.

When the temperature limit is exceeded, all affected cylinders turn white.



On GDU 1060/1210, a vertical white line appears on the CHT bar of the EIS gauge strip (if configured).

LAYOUTS WITH SPACE CONSTRAINTS

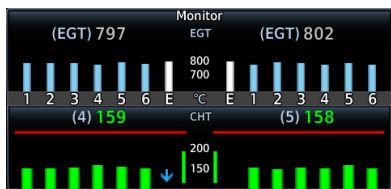


Graphs for single-engine configurations consolidate EGT and CHT cylinders on a single graph.

Display detail varies for engine temperature graphs based on space constraints and aircraft configuration.

If space is limited such that the graph cannot provide digital readouts for every cylinder, then only the values for hottest or selected cylinder are shown.

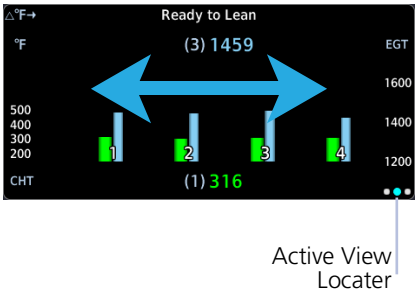
Tapping the graph displays values for the selected cylinder. After 10 seconds, values default to the hottest cylinder on each engine.



For multi-engine configurations, EGT graphs display in the upper half of the gauge while CHT graphs display in the lower half.

Graphs for multi-engine configurations show a split view of EGT and CHT cylinders.

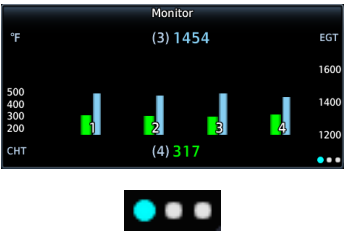
CHT/EGT View Selection



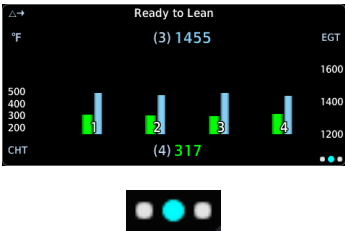
Swipe the graphical display or turn the inner knob to switch between CHT/EGT views.

Locator dots provide a momentary indication of current knob focus as you transition between views. Monitor is the default view.

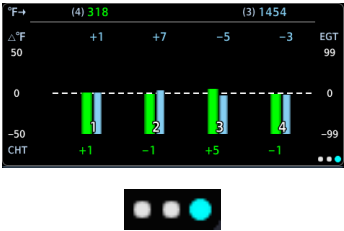
Monitor



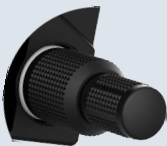
Lean



Normalized



CHT/EGT View Selection via Control Knob

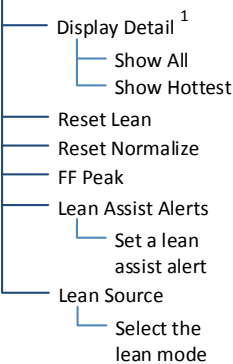


CHT/EGT views are selectable from the locator bar. Turning the inner knob switches between monitor (MON), lean (LEAN), and normalized (NRM).



CHT/EGT Control Options

Engine Temperature Menu



CHT/EGT control options reside in the Engine Temperature menu (**Menu > Engine Temperature**). From here you can:

- Select the display detail level¹
- Reset lean and/or normalized temperature values
- Disable peak fuel flow indication
- Set lean assist alerts
- Select the lean mode

Display Detail ¹	Select the CHT/EGT display level. For a simplified view, select Show Hottest.
Reset Lean	Reset and start a new Lean Assist process. Requires pilot confirmation.
Reset Normalize	Reset the normalized temperature baseline values to the current values. Requires pilot confirmation.
FF Peak	Enable or disable peak indications on the fuel flow gauge. Pertains to lean view only.
Lean Assist Alerts	Set a change in temperature and/or ROP/LOP alert.
Lean Source	Select the lean mode. GDU retains lean mode selections through power cycles. Read about available options in <i>Lean Assist</i> .

You can reset lean or normalized temperature values by pushing the applicable control knob while the view is active. This method is immediate as it requires no confirmation. Read more about each reset function in the respective engine temperature display section.

Peak fuel flow indications correspond to EGT/TIT peaks on the lean display. Read more about these indications in *Lean Assist*.

¹ Not applicable to layouts with spatial constraints. These displays show only the hottest cylinders.

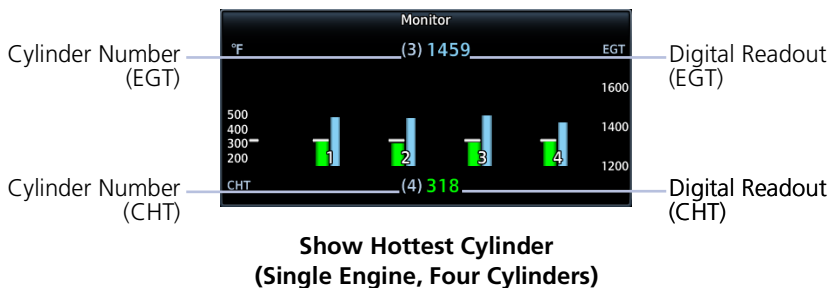
SHOW HOTTEST

Selecting the Show Hottest detail setting provides a simplified view of the hottest CHT and EGT/TIT cylinders on the monitor and lean engine temperature displays.¹

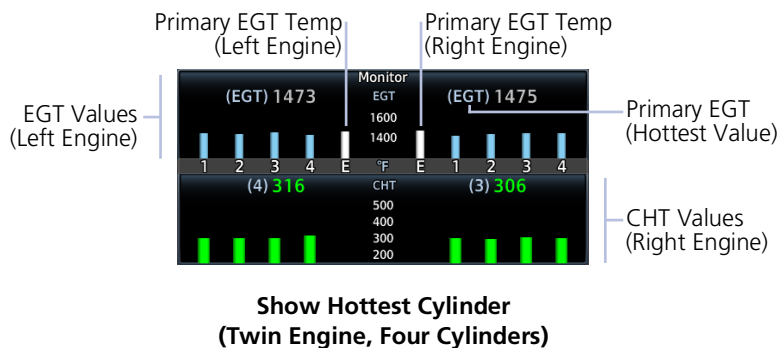
Once detected, the hottest cylinder temperatures appear as follows:

- *Single engine*: along the top (EGT/TIT) and bottom (CHT) of each graph
- *Twin engine*: along the top of the graph

Individual cylinders are selectable at this detail level. Tap a cylinder to display its temperature value.



“EGT” annunciates in place of a cylinder number when Primary EGT is the hottest value.



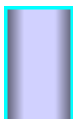
¹ Not applicable to normalized view, which displays all temperature deltas by default.

Cylinder Selection

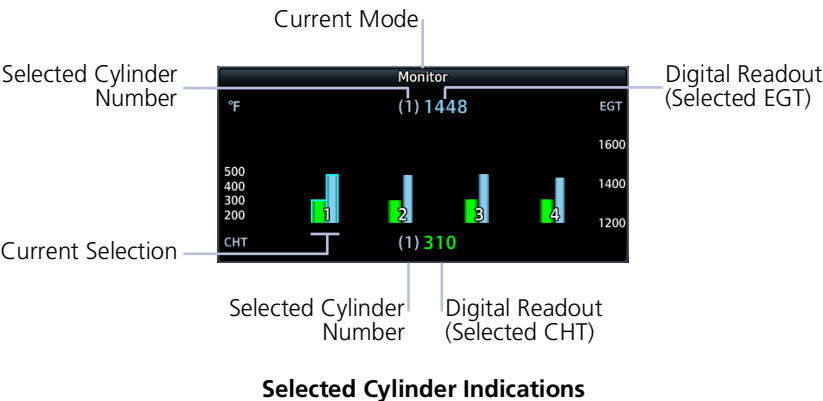
Tapping a graph manually advances the selector to the next temperature value (e.g., CHT, EGT, Primary EGT, or TIT).

- *Single engine*: Tapping the graph cycles through each numbered pair of CHT and EGT cylinders.
- *Twin engine*: Tapping the graph cycles through each quadrant (e.g., Left EGT, Right CHT)

Selections revert to auto mode after 10 seconds of inactivity.



A solid cyan border highlights the current selection. A digital readout displays the cylinder's current temperature value.



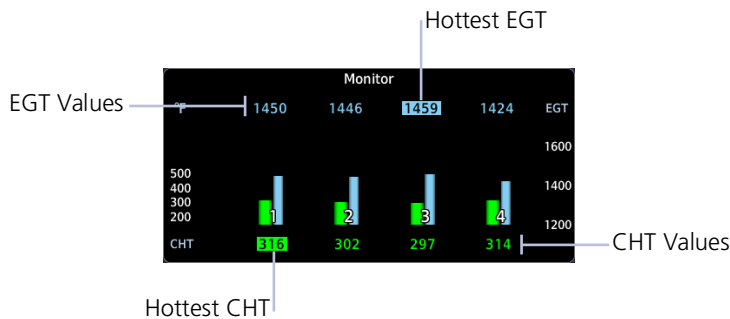
SHOW ALL CYLINDERS

Selecting the Show All detail setting¹ displays:

- Cylinder CHT value in green below each bar
- Cylinder EGT value in blue above each bar
- TIT or Primary EGT label and value (if available)

Once detected, the hottest cylinder appears as follows:

- *Cylinder CHT*: Black text on green background
- *Cylinder EGT*: Black text on blue background



**Show All Cylinders
(Single Engine, Four Cylinders)**

¹ Not applicable to layouts with spatial constraints. These displays show only the hottest cylinders.

Monitor View

FEATURE LIMITATIONS

- *Manual cylinder selection available only when monitor mode is active and the Display Detail setting is Show Hottest*

Monitor is the default engine temperature display view. When active, individual cylinder temperature values annunciate based on the selected CHT/EGT display detail setting: Show All or Show Hottest (default).

Menu > Engine Temperature > Display Detail

Lean Assist

FEATURE REQUIREMENTS

- *For specific engine leaning procedures and temperature targets, consult the AFM*

Selectable lean modes allow you to identify peak EGT/TIT/Primary EGT temperatures and temperature differential values associated with the leaning process.

Available mode options are listed in the Lean Source menu. GDU retains lean source mode selections through power cycles.

LEAN SOURCE MENU OPTIONS	
EGT - All Peaks	Used for leaning off the first peak (rich of peak operations) and last peak (lean of peak operations). Available for all installations (default selection for non-TIT installations).
EGT - First Peak	Optional. Used only for leaning off the first peak when normal operation is to always lean rich of peak.
TIT	Available only for single turbocharger installations (default selection).
TIT - All Peaks	Available only for twin turbocharger installations (default selection).
TIT - First Peak	Optional. Used only for leaning off the first TIT peak. Available only for twin turbocharger installations.
PEGT	Available only for Primary EGT installations.

EGT graph features are not selectable when in lean view.

PEAK TEMPERATURE INDICATIONS

During Lean Assist mode, the system waits for the EGT or TIT temperature to rise 14 °F and then drop 7.2 °F (or equivalent).¹ When this occurs:

- A white (PEGT/TIT) or blue (EGT) saddle indicates the maximum temperature value recorded for the sensor.
- Saddle position and deviation values automatically adjust to reflect any increases in peak temperature.

If configured for a fuel flow sensor, a corresponding decrease in fuel flow is also required for peak to occur.¹

Temperature and fuel flow values may be customized by the installer.

LEAN DELTA DIGITAL READOUTS

A numeric field displays the difference between the peak and current operating temperatures. This value changes to reflect all subsequent temperature fluctuations. A negative value denotes an operating temperature lower than the maximum recorded value.

READOUT STYLE	MEANING	PURPOSE	AVAILABLE LEAN MODES
	Cylinder EGT First Peak	Cylinder EGT rich of peak operations	EGT - First Peak EGT - All Peaks
	Cylinder EGT Last Peak	Cylinder EGT lean of peak operations	EGT - All Peaks
	Primary EGT or TIT First Peak	Primary EGT and single TIT rich of peak and lean of peak operations	Primary EGT TIT TIT - First Peak TIT - All Peaks
	TIT Last Peak	Dual TIT lean of peak operations	TIT - All Peaks

¹ Temperature and fuel flow values may be customized by the installer.

PEAK FUEL FLOW INDICATIONS

FEATURE LIMITATIONS

- Fuel flow indications available only during lean view on arc style or standard dial gauges

Magenta indications on the fuel flow gauge correspond to the first EGT or TIT peak unless/until a last peak occurs, at which point the indications update to reflect the last peak fuel flow.



Once a peak occurs, the fuel flow gauge displays a magenta triangle indicating the fuel flow when the first or last peak occurred. A numeric field displays the difference in the current fuel flow and the peak fuel flow.

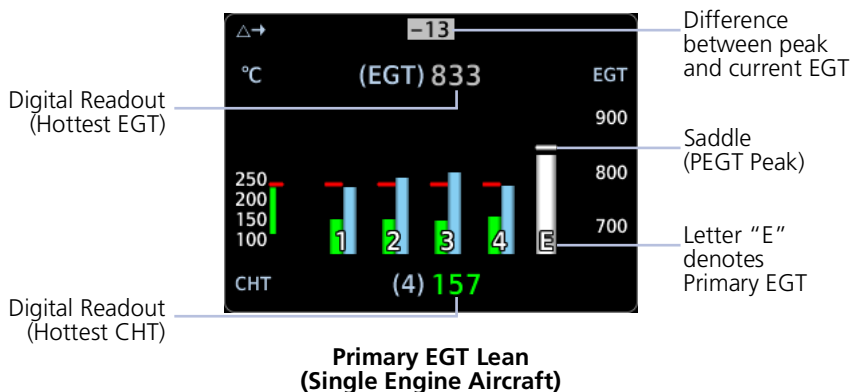
This function is active by default on the MFD and on EIS only displays.

To disable peak fuel flow indications: Tap **Menu > Engine Temperature**, and toggle off the **FF Peak** key.

EGT Lean

LEAN SOURCE MODE SELECTION	FUNCTION
EGT - All Peaks Exhaust Gas Temperature All Peaks	- Indicated the peak temperature of all cylinders - Displays temperature deltas for the first peak cylinder and last peak cylinder upon detection - Recommended for rich of peak or lean of peak operations (for rich of peak, reference the first peak delta; for lean of peak, reference the last peak delta) - Default indication for non-TIT installations
EGT - First Peak Exhaust Gas Temperature Lean First Peak	- Indicates the peak temperature of the first cylinder to peak - Displays a temperature delta based on the first peaked cylinder (no other cylinders are detected) - Recommended for aircraft that always operate rich of peak
PEGT Primary EGT Lean	- Indicates the peak temperature of the primary EGT during the leaning process - Displays a temperature delta based on the Primary EGT - Available only when Primary EGT is the configured gauge type

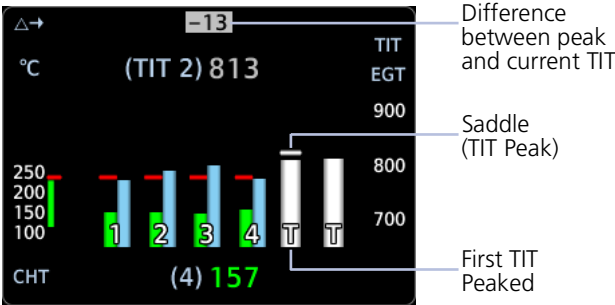
First peak temperature deltas always display with inverse highlighting. Last peak temperature deltas always appear inside a box.



TIT Lean

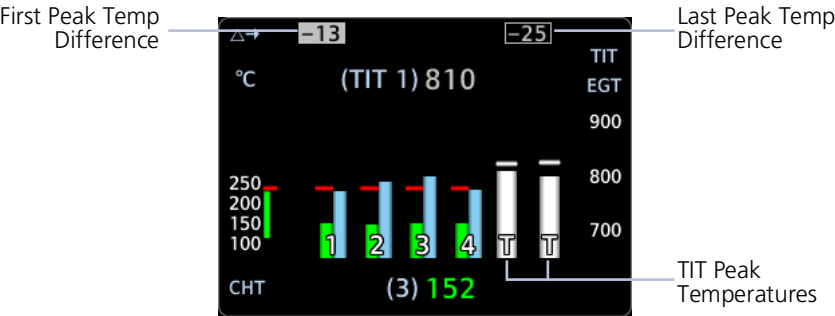
TIT leaning is available for turbocharged aircraft configured for single or dual TIT measurements.

LEAN SOURCE MODE SELECTION	FUNCTION
TIT Turbine Inlet Temperature Lean	• Indicates the peak TIT during the leaning process • Displays a temperature delta based on the peak TIT • Available only on aircraft with one TIT sensor per engine (default)
TIT - First Peak Turbine Inlet Temperature Lean First Peak	• Indicates the first temperature sensor to reach peak TIT • Displays a temperature delta based on the first TIT peak (only one TIT peak temperature is detected) • Available only on single engine aircraft equipped with dual turbochargers • Recommended for dual TIT aircraft that always run rich of peak TIT
TIT - All Peaks Turbine Inlet Temperature All Peaks	• Indicates the first and second temperature sensors to reach peak TIT • Displays two temperature deltas based on the first peak TIT and last peak TIT (use the first peak TIT when operating rich of peak; use the last peak TIT when operating lean of peak) • Available only on single engine aircraft equipped with dual turbochargers (default)



**TIT Lean - First Peak
(Single Engine Aircraft)**

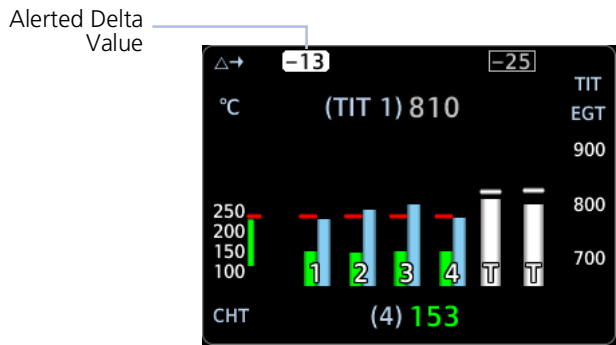
First peak temperature deltas always display with inverse highlighting. Last peak temperature deltas always appear inside a box.



**TIT Lean - All Peaks
(Single Engine Aircraft)**

LEAN ASSIST ALERTS

The system provides alerts based on pilot configured temperature and rich of peak or lean of peak state. Once an alert is triggered, the delta value for the applicable peak annunciates above the graph (black text on white background) for 5 seconds.



Lean Assist Alert Indication
(ROP Alert)

Available alert options are accessible from the Lean Assist Alerts page. Not all alert types may apply.

ALERT TYPES	CONDITION
Lean of Peak Rich of Peak	The system is configured for a fuel flow sensor.
First Peak Last Peak	A fuel flow sensor is not configured and the selected lean source is either EGT - All Peaks or TIT - All Peaks (multi-sensor leaning).
No Specific Type (General Delta Temperature Alert)	A fuel flow sensor is not configured and the selected lean source is one of the following: - EGT - First Peak - TIT - First Peak - TIT - Primary EGT (single sensor leaning)

ADJUST LEAN ASSIST ALERT SETTINGS

Lean alert settings are always adjustable.

1. Tap **Menu** > **Engine Temperature** > **Lean Assist Alerts**.
2. Select **Degrees** and enter the degree difference from peak for the alert setting. Doing this automatically enables the alert function.
3. Select **Type** to toggle between alert types: ROP and LOP or First Peak and Last Peak (if available).

DISABLE A LEAN ASSIST ALERT

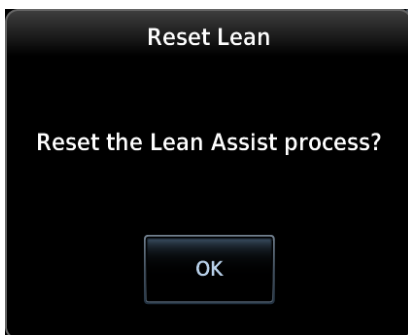
Tapping an **Alert** key enables or disables the alert depending on its status.

Example of Lean Assist Alert Settings



RESET & RESTART LEAN PROCESS

Reset and start a new Lean Assist process.



From the Engine menu:

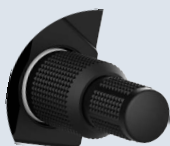
Tap **Menu** > **Engine Temperature** > **Reset Lean**.

A pop-up message requests confirmation.

Tapping **OK** confirms the request.

Tapping **Cancel** in the control bar closes the pop-up and aborts the request.

Reset & Restart Lean via Control Knob



Pushing the applicable control knob while viewing the lean display resets and restarts the Lean Assist process. This action requires no confirmation.

Push Reset MON | **LEAN** | NRM

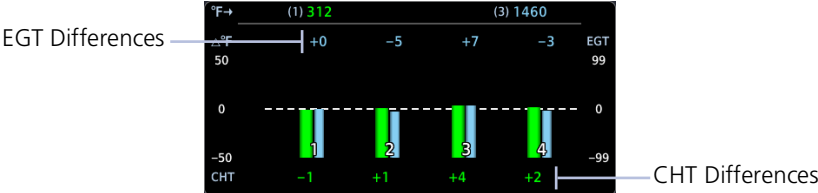
Normalized View

Selecting this mode displays the following temperatures relative to a normalized value.

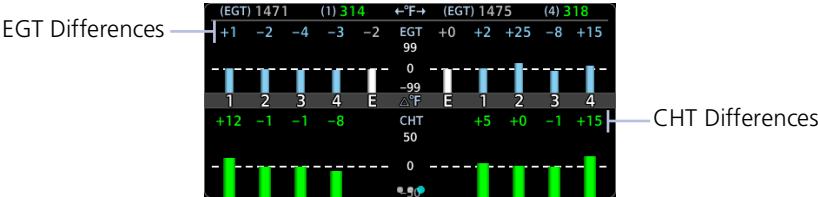
- CHT
- EGT
- TIT
- Primary EGT

On the CHT/EGT graph:

- Normalized values become active
- Bars and digital values depict the temperature difference from the normalized value
- Gauge color changes to alert you of a threshold exceedance



Single Engine Aircraft



Twin Engine Aircraft

Due to space constraints in some layouts, digital readout values may be limited to the bar scale range. Readouts beyond these values display as a series of dashes (---). To bring normalized values back into readout range, reset the normalized temperature baseline as described in the next segment.

RESET NORMALIZED VALUES

Reset the normalized temperature baseline values to the current values.



From the Engine menu:

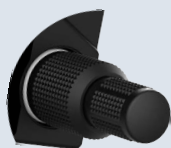
Tap **Menu** > **Engine Temperature** > **Reset Normalize**.

A pop-up message requests confirmation.

Tapping **OK** confirms the request.

Tapping **Cancel** in the control bar closes the pop-up and aborts the request.

Reset Normalized Values via Control Knob



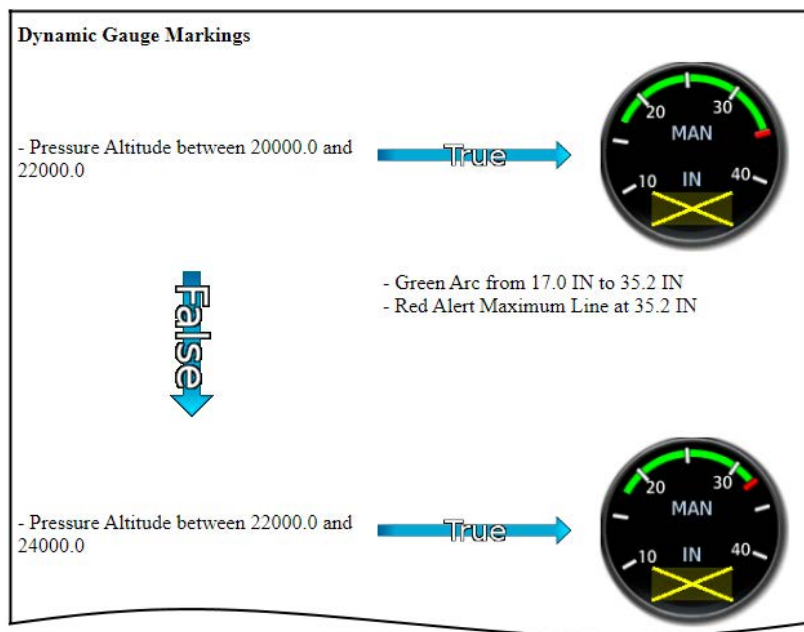
Pushing the applicable control knob while viewing the normalized display resets the normalized temperature baseline. This action requires no confirmation.

Push Reset MON | LEAN | **NRM**

The system does not retain normalized temperature values through power cycles.

Configuration Report

A report detailing all system configuration details, including the different dynamic marking states for each individual gauge, can be provided by your dealer. This information is unique to the aircraft and its configuration.



Configuration Report

Use this report to familiarize yourself with the different marking states that are possible with each gauge. To obtain a report, contact your dealer for assistance.

Turbine Engines

EIS Display

FEATURE REQUIREMENTS

Purchased Turbine Unlock feature enabled on one of the following display units:

- GDU 700P configured for EIS only
- GDU 1060/1210 with EIS display enabled on unit

FEATURE LIMITATIONS

- Turbine EIS not available on GDU 700L

The EIS displays engine, electrical, and fuel information using parameter specific indicator types (e.g., analog indicators, digital data fields, dynamic markings, horizontal and vertical bar gauges).

The EIS instruments replace traditional analog gauges used for monitoring engine parameters. Always consult the AFM or POH for engine operating limitations.

Layout varies according to:

- Display type
- Number and type of engines in aircraft
- Number and type of installed sensors

GDU 700P EIS

When configured for EIS only, GDU 700P is a dedicated full-screen display of engine instrumentation.

GDU 700P EIS



Single Engine Turboprop

Twin Engine Turboprop

GDU 1060/1210 EIS

With the display of EIS enabled, GDU 1060/1210 dedicates 20% of its screen to a full-time display of primary EIS information. In addition, a dedicated Engine page is accessible from the MFD Home page. This page is available even if the EIS display is not enabled.

Tap the **Engine** icon to open the corresponding page (Home > **Engine**).

Turboprop Aircraft

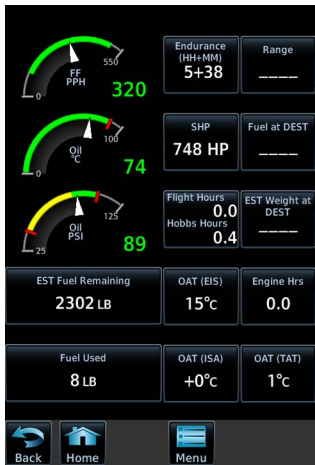


Turbofan Aircraft



The MFD Engine page provides expanded information for select configured gauges. Other features include pilot selectable user fields and a **Menu** access key.

MFD Engine Page, GDU 1060/1210



Single Engine Turboprop

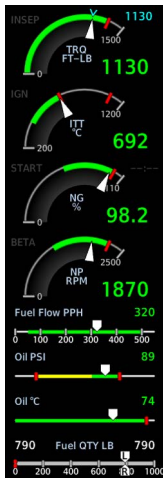


Twin Engine Turboprop

Pilot
Selectable
User Fields

Menu Key

Gauge Strip, GDU 1060/1210



Single Engine Turboprop



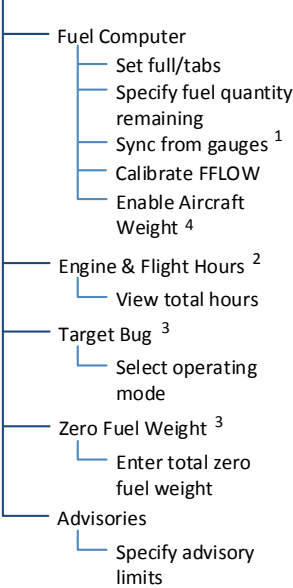
Twin Engine Turboprop

The gauge strip presents an unobstructed compact view of primary engine information. Depending on configuration, the gauge strip may reside at the left or right edge of the GDU 1060/1210 display.

Unlike the Engine page, the gauge strip has no selectable features.

EIS Setup

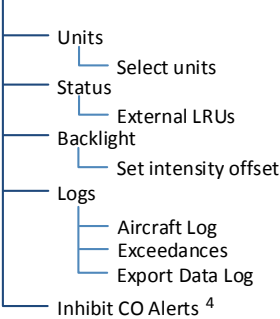
Engine Menu



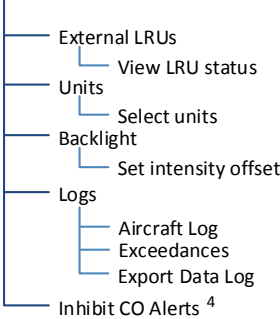
The **Menu** key provides access to the fuel computer, engine and flight hours, and engine advisories. On GDU 700P, the Engine menu also provides controls for customizing system settings.

GDU 700P EIS setup functions are accessible from the System menu. On GDU 1060/1210, they reside on the MFD System page.

System GDU 700P



System Page GDU 1060/1210



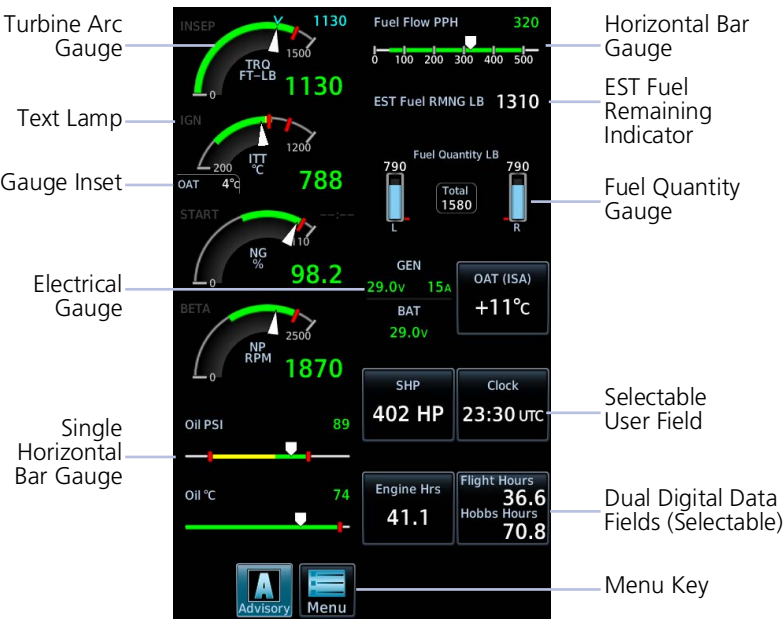
¹ Available with TXi software v3.21 and later. ² This page is informational only.
³ Available with TXi software v3.61 and later. Availability dependent upon configuration.
⁴ Available with TXi software v3.80 and later. Availability dependent upon configuration.

EIS SETUP SELECTIONS

Units	Specify units of measure for fuel computer and pilot selectable gauges. Selections synchronize across all configured GDUs. ¹	
	• Distance	• Temperature
	• Fuel Computer	
Status	• View unit and software information	
	• Check status of all configured LRUs	
Backlight	• Adjust display brightness	
Aircraft Log	• View engine and airframe cycle counters	
Exceedances	• View and acknowledge exceedance advisories	
Export Data Log	• Save logged data to SD card	
Inhibit CO Alerts	• Toggle CO caution alerts on or off	

¹ Engine gauge units are not adjustable.

Gauge & Indicator Types



Common EIS Display Elements

Turbine Arc Gauges	Display a visual representation and a digital readout of the specified engine parameter value. If configured, these gauges may be accompanied by a text lamp and/or gauge inset. ¹
Digital Data Fields	Display a digital readout value of the engine parameter. ¹ These include single or dual input values depending on system configuration.
Fuel Quantity Gauges	Display fuel amounts for the specified fuel tanks. ²
Horizontal and Vertical Bar Gauges	Display engine parameter information on a single horizontal or vertical bar. These have single or dual pointers depending on engine type. May include digital readout fields depending on display layout.
Selectable User Field	Opens a menu of the selectable data fields available for display. The type of parameters available for selection is determined during installation. A gray border differentiates selectable user fields from non-selectable digital data fields.

¹ Limits are configured according to system design or the AFM/POH. They are not pilot selectable.

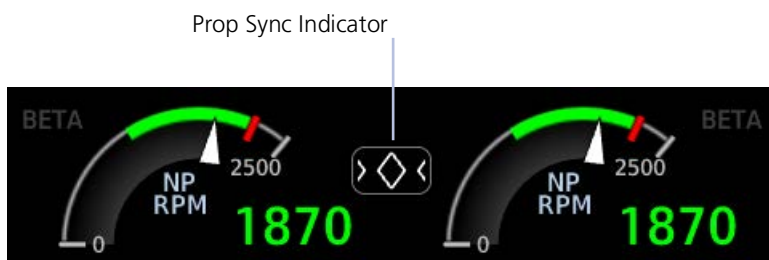
² Fuel gauges are configured and calibrated during installation.

Prop Sync Indicator

On multi-engine aircraft, a prop sync indicator provides a visual reference for synchronizing multi-engine RPM. This indicator uses inputs received from the left and right engine to show a comparison of the matched RPM.

- If RPM for both engines is the same, diamond-shaped symbols appear and motion stops.
- If RPM varies between engines, arrow-shaped symbols point and move in direction of the faster engine

The speed at which the indicator moves left or right is determined by the RPM differential between the two engines.



Gauge Types

Gauge labels are customizable to match aircraft configuration. Labeling presented in the following table may differ from actual labeling on the unit.

LABEL ¹	UNITS	FUNCTION
N1	%, RPM	Low pressure turbine RPM^{2, 3}
N2	%, RPM	High pressure turbine RPM^{2, 3}
NG	%, RPM	Engine gas producer speed
NP	%, RPM	Propeller RPM
TRQ	%, psi, ft-lb	Engine torque
Engine TEMP	°C, °F	- Turbine engine temperature - Typically interturbine temperature (ITT) or exhaust gas temperature (EGT)
IAT	°C, °F	Inlet air temperature
ISA	°C, °F	Degrees deviation from the International Standard Atmosphere model²
RAT	°C, °F	Outside ram air temperature²
SAT	°C, °F	Outside static air temperature²
Fuel or Fuel Pressure	psi, bar	Fuel pressure
FF	gph, lt/hr, PPH, kg/hr	Fuel flow
Oil or Oil Pressure	psi, HPa, bar	Oil pressure
Oil or Oil Temperature	°C, °F	Oil temperature
Fuel or Main Fuel	lb, lt, gal, kg	Current fuel quantity in main tanks
EST Fuel RMNG	lb, lt, gal, kg	Estimated fuel based on pilot specified value and fuel flow
Tip Fuel	lb, lt, gal, kg	Current fuel quantity in tip tanks
Aux Fuel	lb, lt, gal, kg	Current fuel quantity in auxiliary tanks
Fuel or Fuel Temperature	°C, °F	Fuel temperature
Bus Volts	V	Bus voltage
Bat Volts	V	Battery voltage
ALT/Gen AMPS or ALT/Gen %	A, %	Alternator/generator load
BAT Amps	A	Battery load

LABEL ¹	UNITS	FUNCTION
Vac or Vacuum	in Hg, PSI	Vacuum/pressure
Rudder or Rudder Trim	° (degrees)	Rudder trim

¹ Actual gauge labels are dependent upon installer setup. ² TXi software v3.61 only. ³ Turbofan EIS only.

TURBINE ARC GAUGES

Arc gauges are truncated, round, and have fewer graduations, similar to typical analog turbine gauges. Labels may vary based on configuration.

Gas Generator Turbine Speed (N1, NG)



- RPM or % RPM values

Engine Temperature



- Units are °F or °C

Torque



- Units are %, ft-lb, or psi

Propeller RPM



- RPM or % RPM values

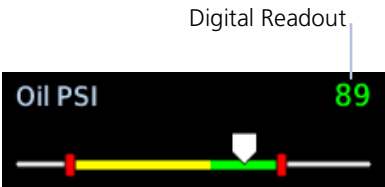
Propeller RPM Minimum Threshold

Depending on aircraft, sensor type, and configuration, this gauge may be configured with a minimum threshold value. During start and shutdown, RPM values below this threshold display as a series of dashes (---). Values above the threshold display as normal.

BAR GAUGES



Bar gauges have linear indications that move from left to right.



If configured, a corresponding digital readout displays to the right of the gauge.

DIGITAL ONLY GAUGES

Some engine gauges may be configured to provide a digital readout only.



On single engine aircraft, the readout displays to the right of the gauge label. On multi-engine aircraft, dual readouts display to the left and right of the label.

NON-LINEAR GAUGES



NOTE

Prior to operation, pilots must familiarize themselves with the gauge configuration provided as part of the Configuration Summary. The only other reference to non-linear scaling is current needle position with respect to the digital readout value.

Some turbine engine installations may be configured for non-linear gauges. These are arc gauges and bar gauges that provide non-linear scaling for up to six consecutive segments.



Non-linear Arc Gauge

Resolution varies across a parameter's operating range, allowing greater needle sensitivity for more critical ranges.

Labels may vary based on configuration.

Markings & Indications



NOTE

It is the responsibility of the pilot in command to know and abide by all published limitations and operating ranges in the AFM/POH.

GAUGE MARKING COLOR

Gauge marking colors are in accordance with the criticality of operating limits. Units of measure, limits, and gauge colors are configured during installation.

- Green arc denotes a normal operating range
- Yellow radial line or arc denotes a caution range
- Red minimum/maximum line or arc denotes a limitation

STATIC REFERENCE MARKINGS



Gauge markings may be configured to denote special gauge values defined in the FMS or placard. They may include:

- Arc • Dot¹ • Line/Radial
- Minimum Line • Maximum Line
- Triangle¹

Reference markings are non-alerting.

¹ Available with TXi software v3.61 and later.

RANGE INDICATIONS & ALERTING

Alert Suppression

To avoid nuisance alerts, flashing is suppressed for fuel pressure, oil pressure, and oil temperature when the engine is off or starting.

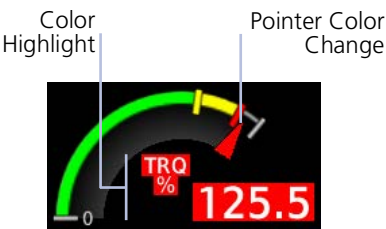
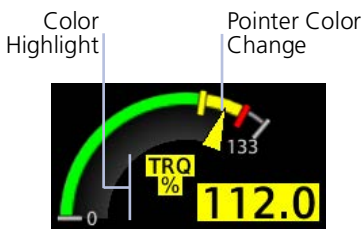
TXi EIS gauges may be configured to alert the pilot when an engine indication:

- enters a caution or warning range
- reaches a specified alerting value
- enters a state of exceedance

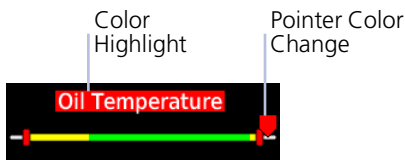
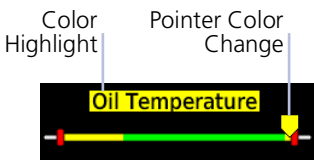
When alerted, gauge features change color, and the gauge label and digital readout (if applicable) are highlighted in the appropriate color. Color highlighting may flash depending on configuration.

Caution Indications

Warning Indications



Dynamic Gauge



Horizontal Bar Gauge



Digital Only Gauge

Flashing alert indications continue to flash until you acknowledge the condition by tapping the **ACK** key. Read more about acknowledging alerts in *Engine Alerts*.

DYNAMIC GAUGE MARKINGS

FEATURE LIMITATIONS

- *It is the responsibility of the installer to configure dynamic gauge markings in accordance with the aircraft-specific engine operating limitations listed in the AFM/POH*
- *Dynamic gauge markings may not be able to represent all operating limitations and scenarios depicted in the AFM/POH*
- *Dynamic gauge markings for multi-engine turbine EIS and time-delayed dynamic gauge markings are available only with TXi software v3.21 and later*

This feature changes gauge markings based on configured conditions to comprehensively depict safety-critical ranges and markings. For example, torque gauge markings may be configured to change dynamically based on propeller RPM, while engine temperature markings may be configured to change dynamically based on engine operating conditions.

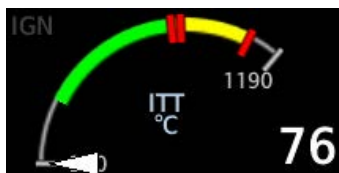
Safety-critical ranges may be configured with time delays for conditions with time dependent limitations.

The EIS is useful for managing engine and aircraft systems within safe operating limits. Dynamic gauges further assist you by presenting the gauge limits and markings appropriate for the current aircraft conditions. This allows for easier gauge interpretation as well as increased safety and efficiency.

Marking Sets

The following examples depict the implementation of dynamic markings in an ITT gauge. Two dynamic marking conditions have been configured: one when the engine is starting, and one when the engine is running.

Standard



Standard gauge markings are configured by the installer to match the existing ITT gauge in the aircraft. This is the default marking set for the gauge.

Standard markings will display if the conditions for dynamic marking are not met.

Starting (Condition)



Once the system determines that the engine is starting, the gauge markings change to show the starting temperature limits.

In this example, the absolute maximum temperature is absent and the highest red line limit indicates the maximum temperature for engine start.

Running (Condition)



Once the system determines that the engine is running, the gauge markings change to show the red line limit for cruise operation.

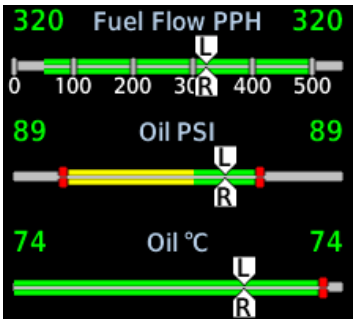
Starting temperature markings and limit indications are absent.

Unlike analog gauges, which require you to memorize limitations or interpret multiple markings, dynamic gauges present only the limitations applicable to specific situations or flight conditions.



NOTE

Prior to operation, pilots must familiarize themselves with all configured dynamic markings and adjustable gauge ranges, and their associated conditions. These are described in the Configuration Summary for your specific installation.

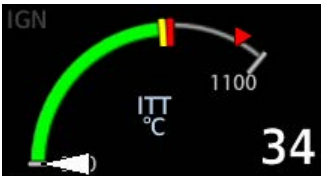


Dual Bar Gauges

For multi-engine aircraft, the system is configured such that left and right engine gauges have identical markings. The manner in which markings display is determined by the system for each engine independently. As a result, each engine gauge may show different markings based on its current parameters.

Adjustable gauge ranges may be configured as part of the dynamic markings feature for specific installations. The overall gauge range automatically adjusts for greater precision across various operating conditions.¹

Starting (Condition)



Running (Condition)



During a range transition, the minimum range value moves to the bottom of the arc as you transition to the next range segment.

¹ Available only with TXi software v3.61 and later.

TARGET BUG



NOTE

It is the responsibility of the pilot in command to know and abide by all published limitations and operating ranges in the AFM/POH.

FEATURE REQUIREMENTS

- TXi software v3.61 or later

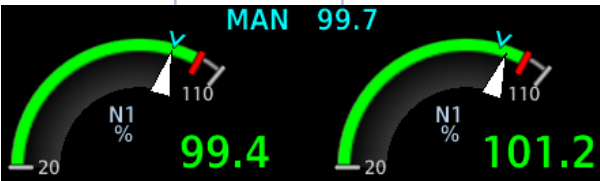
FEATURE LIMITATIONS

- Available mode selections dependent upon aircraft type and unit configuration
- Bug type dependent upon aircraft type: torque bug for turboprop aircraft; N1 bug for turbofan aircraft
- Computed targets available for Cessna 525(A) aircraft only
- On-target indications available with TXi software v3.80 and later

This indicator allows the display of the engine target value on the configured gauge.

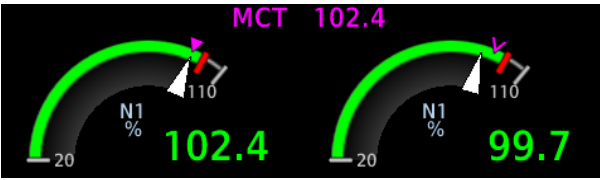
A chevron depicts the target value on the gauge.

Target value appears in cyan for pilot-entered targets.



For supported aircraft, the system automatically computes and sets the engine parameter target value for each mode.

Mode Annunciation
& Target Value



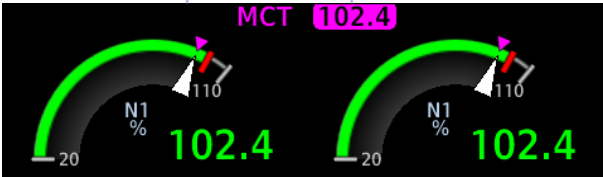
Computed Target & Mode Indications

COLOR DEFINITIONS	
Cyan	Manual target
Magenta	Computed target ¹

Marking color is based on the indicator's operating mode.

On-target indications change appearance when the computed target value is within the configured range.

Chevron changes to a solid triangle. Target value appears in solid highlight.²



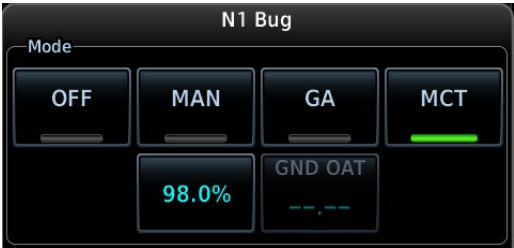
On-target Indications

TARGET THRESHOLDS	
N1 Bug	1% below target to 0.5% above target
Torque Bug	4% below target to 0.5% above target

The on-target threshold is dependent upon aircraft type and gauge configuration.

¹ Cessna 525(A) aircraft only. ² Only when all operational engines reach the target.

Example of Available N1 Bug Mode Settings



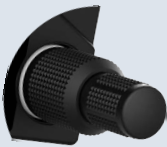
Target bug units always match the configured gauge units.

Target bug settings reside in the engine menu (**Menu > Target Bug**). From here, you can:

- Enable specific operating modes
- Specify target values if required
- Turn on/off the target bug

Bug settings for the primary power indicator are accessible from the EIS start-up page during power up.¹

Operating Mode Knob Selection



Computed operating modes are selectable from the locator bar. Turning the outer knob displays the target bug value for each available mode.

MAN | GA | MCT

Push OFF

During manual mode, turning the inner knob adjusts target value and current bug position.

MAN

Push OFF

¹ Parameter varies according to aircraft and configuration: torque bug for turboprop aircraft; N1 bug for turbofan aircraft.

MODE ABBREVIATIONS	
CLB	Climb
CRZ	Cruise
GA	Go around
MAN	Manual
MCT	Max continuous thrust
TO	Takeoff

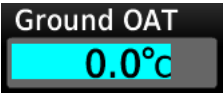
Available mode selections vary by aircraft type.

Set a Target Bug for Takeoff Power

CESSNA 525(A) AIRCRAFT ONLY



Specify a target OAT value and enable takeoff mode while the aircraft is on-ground. You may disable the bug at any time from the Engine menu.



During power up:

1. Tap **GND OAT**. GDU displays the measured RAT value by default.
2. Toggle the **TO** mode key on.

The temperature reference transitions to RAT and uses sensed temperature inputs after the takeoff phase is complete and you select another mode or turn off the target indicator.

Ground OAT bug setting is no longer available once the aircraft is in-air.

Set the Torque Bug

TURBOPROP AIRCRAFT ONLY

Specify a target torque value and enable manual mode while the aircraft is on-ground. You may disable the bug at any time from the Engine menu.

Torque Bug Manual Mode Settings



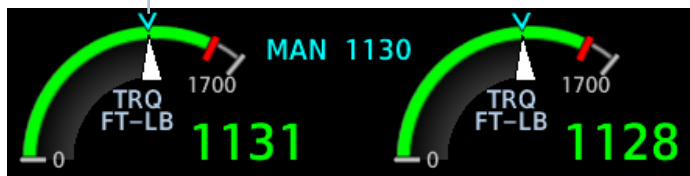
During power up:

- 1. Tap the data entry key and specify the target torque value on the provided keypad.
- 2. Toggle the **MAN** mode key on.

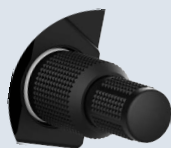
Target torque bug indications appear on the configured engine torque gauge(s).

Torque Bug Depiction

Manual Mode Annunciation & Target Torque Value



Torque Bug Knob Adjustments



During manual mode, turning the inner knob adjusts the target torque value and current bug position.



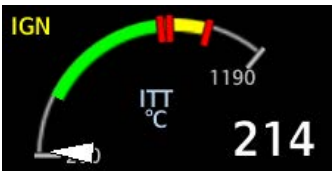
Text Lamps

Arc gauges may be configured to feature a text lamp—a window that displays a single abbreviated message when the related system is in operation.

TEXT LAMP AVAILABILITY	
TURBINE GAUGES	TURBOFAN GAUGES
• Torque • ITT • NG • NP • Fuel Flow • Oil Temperature • Oil Pressure	• N1/N2 • ITT

Text lamps reside at the upper outside edge of the gauge. For multi-engine aircraft, left/right location corresponds to the location of the associated engine:

- on the left for left engine gauges
- on the right for right engine gauges



Text lamp color and behavior (i.e., solid or flashing) are dependent upon configuration.

Starter text lamps may display a timer.



When the system is inactive, the text lamp is grayed out.

TEXT	MEANING
START	Starter ON
IGN	Ignition ON
BETA	Propeller beta valve OPEN
BLEED	Bleed valve CLOSED
INSEP	Inertial separator OPEN
SRL	SRL computer OFF

When the starter lamp is configured, GDU uses the photocell for display brightness during engine start. This is to ensure EIS visibility as power is drawn to start the engine.

Gauge Insets

FEATURE LIMITATIONS

- OAT gauge insets available only with TXi software v3.21 and later

GAUGE INSET AVAILABILITY	
TURBINE GAUGES	TURBOFAN GAUGES
• Torque • NG/N1 • NP • Fuel Flow • Engine Temperature	• N1/N2 • ITT

Arc gauges may be configured to feature a gauge inset—a window that displays a digital readout of secondary information related to the primary gauge for reference. Gauge insets reside at the bottom inside edge of the gauge.



Caution Indication

Depending on configuration, these insets may provide various supplemental data, such as calculated engine power or OAT. For example, the gauge inset digital readout may be configured to alert yellow as a means of cautioning the pilot that the outside air temperature is below the configured threshold. If connected to the Pitot heat system, the OAT gauge inset can be configured to provide a reminder to turn on Pitot heat during potential icing conditions.

TEXT	MEANING
%PWR	Percent power
ISA	Difference from Standard Air Temperature
OAT	Outside Air Temperature
SAT	Static Air Temperature
SHP	Shaft horsepower
TAT	Total Air Temperature

Always follow AFM/POH procedures when responding to potential icing conditions.

Timers

FEATURE LIMITATIONS

- *Timer type, quantity, and availability are dependent upon configuration*
- *Maximum allowable exceedance time (or grace period) dependent upon configuration*

Timers are available for the following gauges:

- Torque
- Prop RPM
- NG/N1
- N2
- Turbine Engine Temperature
- Oil Temperature
- Oil Pressure
- Fuel Pressure

Select EIS gauges may be configured to display one or more timers. For single engine aircraft, these reside at the bottom right of the gauge, adjacent to the parameter field. For multi-engine aircraft, they reside at the bottom inside edge of the gauge:

- on the right for left engine gauges
- on the left for right engine gauges

While a single gauge may contain multiple timers running simultaneously, only one timer may be displayed at a time.

The timer field displays the parameter with the shortest time remaining. The other timers continue to count down in the background.

Timer Types

EIS provides two types of timers:

- Exceedance
- Engine start/cooldown

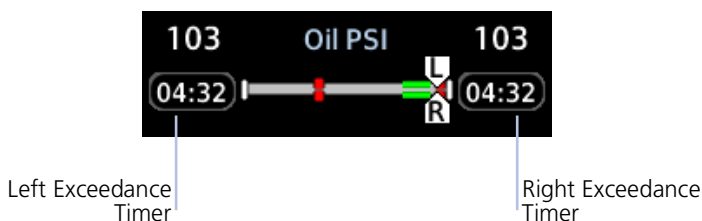
EXCEEDANCE TIMERS



Exceedance timers indicate the maximum allowable time that the parameter can be above specified values. These values and time limits are configured by the installer.

Depending on the parameter type, the time limit may be based on aircraft or engine manufacturer specifications.

Exceedance Timers, Twin Engine Turboprop



For some gauges, an exceedance timer may be configured but not display the timer value (e.g., oil temperature, oil pressure).

Exceedance Timer Behaviors

The timer begins counting once the indication exceeds the configured timer threshold. It resets/disappears if the indication falls below the configured timer threshold prior to timer expiration.

When the timer expires:

- Gauge label, digital readout, and pointer turn red, overriding any existing coloration
- Gauge label and digital readout flash until acknowledged
- System generates an exceedance log entry for the parameter

Exceedance timers may be configured to have zero seconds of allowable time above a threshold. In such cases, the system will immediately generate an exceedance log entry when the indication exceeds the configured threshold value.

Exceedance timers are very useful for recording temperature exceedances during engine starts. You can view timer exceedances in the system logs. For more information, read *Exceedance Logging* in section 2.

Within Range



Timer does not display during normal operating conditions.

Above Threshold



Timer displays once the parameter exceeds threshold. Automatic countdown begins.

Approaching Time Limit



Timer begins to flash during the last 5 seconds of the time limit.

Warning/Expired



Timer changes color once it reaches zero. It flashes in unison with the alerted gauge pointer, value, and label, indicating a warning condition. Exceedance recording begins.

While the timer counts down, attempt to remedy the exceeding parameter by managing aircraft systems or reducing engine power.



Flashing alert indications continue to flash until you acknowledge the condition by tapping the **ACK** key. Read more about acknowledging alert conditions in *Engine Alerts*.

Exceedance timers rely on the installer to configure specific aircraft/engine data and it is possible not all operating limitations are entered into the EIS.

The pilot in command is responsible to know and abide by all published limitations and operating ranges in the POH/AFM.

ENGINE START/COOLDOWN TIMERS

FEATURE LIMITATIONS

- Availability dependent upon configuration

Depending on configuration, engine start and cooldown timers may display as a text lamp with an arc gauge (e.g., NG), a fixed indication with a digital readout field (N2 only), or as a user-selectable field on the MFD page.

Start Timers with N2 Digital Data Field



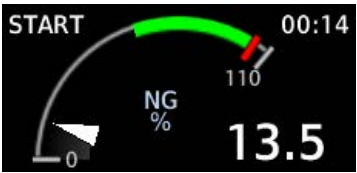
The following examples depict the implementation of engine start and cooldown timers on an arc gauge.

Starter Off (Neutral)



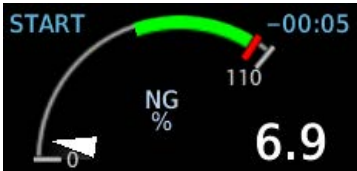
Start timers are grayed out when the starter is off.

Starter Running



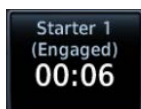
Start timers count up from zero when the starter engages.

Cooldown Timer



Starter cooldown timers change to blue and count down from zero if the starter engages then disengages without a successful engine start.

On the MFD: The start timer is available as a user select gauge on the Engine page.



When the starter is active:

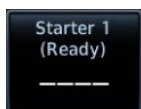
- Label changes to show that the starter is "Engaged"
- Timer begins counting up from zero to indicate the amount of time that the starter has been active

When the starter is inactive, the gauge displays underscores.



When the starter switches from active to inactive, but the engine is off:

- Timer text turns blue and the label changes to show the starter is in "Cooldown"
- Timer counts down from zero (negative value) to indicate the time since the starter was turned off



When the starter goes from active to inactive, and the engine is in the RUNNING or STARTING state, the annunciation changes to "Ready."

Selectable User Fields

These fields allow selection of engine and fuel parameters for display on the EIS. Available user fields and selectable parameters are determined during installation.

Tapping any user field opens a list of available engine parameters and their current values. Selecting a parameter displays its value on the EIS and closes the menu.

- Fields may be modified to display alternate data at any time
- Several parameter units are pilot adjustable

EST Fuel Remaining Estimated amount of total fuel remaining ^{1, 3}	% Power Engine percent power
Fuel Used Total fuel used since last update to estimated fuel remaining quantity ^{1, 3}	Outside Air TEMP (ISA) Degrees deviation from the International Standard Atmosphere model ⁴
Range Total range based on remaining fuel and current ground speed ^{1, 3}	Outside Air TEMP (SAT) Outside static air temperature ⁴
Endurance (HH+MM) Fuel endurance time in hours+minutes ¹	Outside Air TEMP (TAT) Outside total air temperature ⁴
Fuel at Destination Estimated fuel amount at current flight plan destination ^{1, 2, 3}	Outside Air TEMP (EIS) Outside air temperature as measured by the EIS OAT sensor ⁸
Endurance at DEST Remaining fuel endurance time (hours+minutes) at destination ^{1, 2}	Cabin Altitude/Rate Cabin altitude and calculated cabin rate ^{6, 7}
Flight/Hobbs Hours Total flight and Hobbs hours ⁵	Shaft Horse-power Engine shaft horsepower
Clock Current time ¹⁰	Efficiency Fuel efficiency ³
EST Current Weight Estimated current aircraft weight ^{10, 11}	Fuel TEMP Fuel temperature inside left and right tanks ⁹
EST Weight at DEST Estimated aircraft weight at flight plan destination ^{10, 12}	

Read about Aircraft Weight feature enablement and zero fuel weight entry options in *Zero Fuel Weight*.

¹ Values are based on fuel computer calculations. ² Requires an active flight plan from the navigator.

³ Parameter units are pilot selectable. ⁴ Units are independent of PFD units.

⁵ Hobbs hours accumulate when an engine is running. Flight hours accumulate when the aircraft is in air.

⁶ Units are dependent upon configuration. ⁷ Requires cabin altitude data from GFC 600. Rate calculations performed by GDU. ⁸ Available with TXi software v3.21 and later.

⁹ Available for aircraft without a dedicated fuel temperature indicator. For reference only. Requires TXi software v3.50 or later. ¹⁰ Requires TXi software v3.61 or later. ¹¹ Requires valid estimated fuel remaining and zero fuel weight values. ¹² Requires valid fuel remaining at destination and zero fuel weight values.

Counters



NOTE

Do not reboot the GDU during flight unless operational procedures dictate. Doing so may register an additional engine or airframe cycle. Contact a Garmin dealer to adjust counters to match other tracking records.

Counter Types

- Full cycles
- Hobbs and flight hours
- Takeoffs and landings
- Starts and shutdowns (based on engine data)

Counters record various types of cycle data (e.g., Hobbs and flight hours, takeoffs and landings). GDU stores and displays this information in the aircraft log.



The aircraft log also stores any maintenance timers set up during configuration.

Home > **System** > **Logs** > **Aircraft Log**

ENGINE & AIRFRAME CYCLE COUNTERS

The system records parameters separately on individual counters.

CYCLE COUNTER TYPES

Full Cycles	Increment when a takeoff is recorded between an engine start and an engine shutdown.
Starts & Shutdowns	Increment based on engine data.
Takeoffs & Landings	Increment based on GPS, air data, and the weight on wheels.

Turbine engine components are life limited by hours of operation and the number of cycles. In the past, aircraft operators would typically record the number of flight hours and operations in a paper logbook. GDU records this information automatically.

ENGINE AND FLIGHT HOURS



The system records Hobbs and flight hours. These parameters may display as a user-selectable field. They are also viewable in the aircraft log.

For convenience, these counters also display at startup.



Configuration Report

A report detailing all system configuration details, including the different dynamic marking states for each individual gauge, can be provided by your dealer. This information is unique to the aircraft and its configuration.

Dynamic Gauge Markings

- Engine Operating Status is Starting

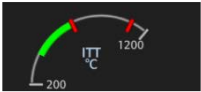
True

False

- Red Alert Maximum Line at 1090 C

Min: 200 C, Max: 1200 C





Configuration Report

Use this report to familiarize yourself with the different marking states that are possible with each gauge. To obtain a report, contact your dealer for assistance.

9-76

Pilot's Guide

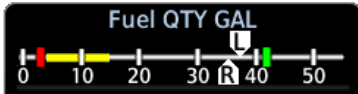
190-01717-10 Rev. U

Fuel

Fuel Gauges

Fuel gauges display the quantity of fuel available in the fuel tanks. EIS provides three basic types of fuel gauge: horizontal bar, vertical bar, and digital. The type of gauge depends on GDU and aircraft configuration.

HORIZONTAL BAR GAUGES



These gauges have one or two moving pointers to indicate fuel quantities. On gauges with two pointers, each pointer is labeled to identify the corresponding tank location.

Horizontal bar gauges may be configured with or without additional digital readouts.

DIGITAL GAUGES



These gauges have up to three simple digital readouts to indicate fuel quantities. Two digital gauges may be combined to provide up to six indicators in a single gauge.

On the narrow bar gauges available for turbine aircraft using the alternate 11-indicator layout, the digital gauge is limited to two indicators.

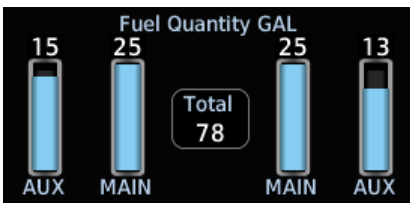
VERTICAL BAR GAUGES

These gauges have up to three filled vertical bars to indicate fuel quantities, with a digital readout located above each bar. Vertical bar gauges can provide up to six indications.

TOTAL FUEL ON BOARD INDICATION

FEATURE REQUIREMENTS

- *TXi software v3.21 or later*



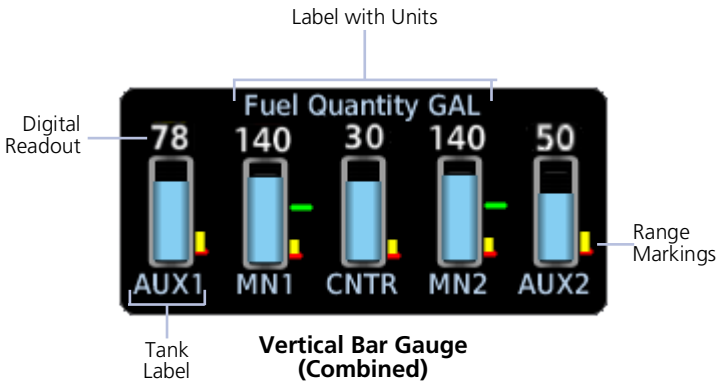
Fuel gauges may be configured to display the sum of all fuel tank quantities as an additional readout.

Gauge-driven Discretes

Gauge-driven discretes based on fuel quantity gauge values can be configured by the installer. For a list of all features that can be linked to discrete outputs, read *Gauge-driven Outputs* in this section.

Gauge Markings

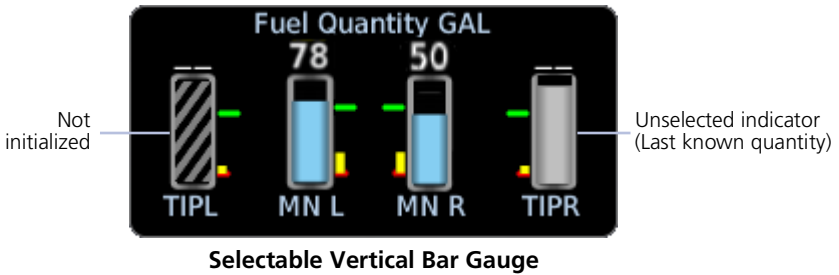
A label at the top of each gauge indicates the type of gauge (Fuel QTY) and units of measurement (GAL, LB, LT, or KG). In addition, each fuel tank indicator in a gauge may be customized with a custom tank label up to four characters long. The fuel tank indicators can be arranged in any order. Gauges can also be configured with range markings.



Selectable Fuel Gauges

For aircraft with four fuel tanks but only two selectable fuel quantity signals, selectable fuel gauges are available. Selectable gauges may be either vertical bar or digital types.

Vertical bar selectable gauges display the current fuel quantities for selected tanks and the last known fuel quantities for unselected tanks simultaneously. Unselected tank indicators have gray fill color to indicate that the last known fuel quantity is being displayed, and the digital readout is replaced by a dashed line after ten seconds. If a selectable fuel tank has not been selected since the system was started, the indicator will be filled with gray hash marks to indicate that fuel quantity for that tank has not yet been initialized.



Digital selectable gauges display only the fuel quantities for the currently selected fuel tanks. The digital indicator tank labels change to either MAIN or AUX depending on which tanks are selected. When an indicator is set to an AUX tank, the tank label will be highlighted.



Fuel Alerting

Fuel Quantity Alert Indications



Gauges with yellow caution range and red minimum alert markings display visual alerts when fuel quantities are within the markings. The gauge label, affected tank label, and digital readout are highlighted in the active range color.

On vertical bar gauges, the fill color also changes to the range color.

When the gauge enters the range, these visual indicators flash until the condition is acknowledged.¹ Any future range alerts for the same fuel tank and condition do not flash again for the remainder of the flight (i.e., until after the fuel computer is reset).

¹ GDU 1060 PFD/MFD/EIS with TXi software earlier than v3.50: On-screen acknowledge key not available. Unless configured for external acknowledge switches, alerts automatically acknowledge after 10 seconds.

Fuel Imbalance Indications

FEATURE LIMITATIONS

Optional Balance On and Fuel Imbalance Warning indications are dependent upon installer configuration and available only for installations that incorporate an automatic fuel balancing capability.

Fuel gauges can be configured to provide visual indications of a fuel imbalance. Caution and warning indications result from a detected fuel quantity imbalance.

When the system detects a fuel imbalance condition:

- A colored line indication links the affected tanks
- Tank label color changes to yellow or red depending on the exceeded threshold
- “IMBALANCE” or “IMBAL” annunciates depending on gauge type



**Imbalance
Caution Indications**

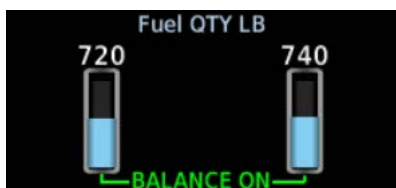


**Imbalance
Warning Indications**

Inadvertent activation can be caused by:

- Fuel quantity sensor malfunction
- Extended flight maneuvers in non-level flight

Failure of any fuel quantity interface or component will result in a fuel balancing system failure.



Balance On Indications

Optional Balance On indications may be configured for installations that incorporate an automatic fuel balancing capability.

A green "BALANCE ON" annunciation and line indication display when automatic fuel balancing is active and neither an imbalance caution nor imbalance warning is active.



Location of the annunciation varies according to gauge style.

Indicators flash upon initial detection of an imbalance. They remain flashing until the condition is acknowledged.¹



For single-engine turboprop aircraft using the alternate 11-indicator layout, colored boxes outline both digital fuel readouts on narrow strip gauges.

The system issues an advisory message when the fuel imbalance monitor is unavailable. This may be due to failed fuel quantity signals or failed fuel temperature sensors.

For more about the fuel imbalance alerting function, read *Fuel Balance Monitoring* in *Advanced EIS Configurations*.

¹ GDU 1060 PFD/MFD/EIS with TXi software earlier than v3.50: On-screen acknowledge key not available. Unless configured for external acknowledge switches, alerts automatically acknowledge after 10 seconds. Once acknowledged, visual indications remain until the imbalance is corrected.

Fuel Computer



CAUTION

Ensure that estimated fuel quantity values in the fuel computer are accurate before flight. Fuel quantity gauge indications may not provide the accuracy required for determination of on board fuel during flight.

The fuel computer calculates and displays fuel parameter values when a flight plan is active in a connected and compatible navigator.

Calculations are based on GPS ground speed, the pilot specified destination airport, estimated fuel remaining, and inputs received from the engine fuel flow sensor.

Fuel computer parameters are available for display on the EIS in a selectable user field. They include:

- Aircraft endurance
- Aircraft weight
- Efficiency
- Endurance at destination
- Estimated current weight
- Estimated weight at destination
- Fuel at destination
- Fuel used
- Range

FUEL COMPUTER CONTROLS

EST Fuel
Remaining

Opens a keypad for specifying the estimated fuel remaining value. The fuel computer bases its calculations on the value entered here.

To account for added fuel and maintain accurate fuel flow measurements over time, always update this value upon refueling.

The keypad provides additional controls for specifying preset fuel values.

Calibrate
FFLOWS

Opens the Fuel Flow Calibration page. Controls for calibrating the fuel flow meter reside here.

This key remains inactive until a new value is entered for EST Fuel Remaining.

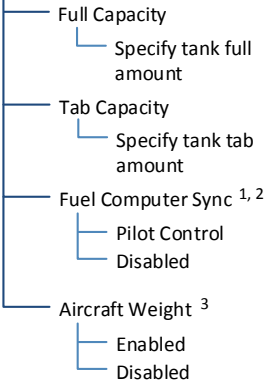
Points About Estimated Fuel Remaining Data

- Based on estimated fuel remaining value entered by the pilot
- Initial value automatically reduced based on current fuel flow
- Not limited to the capacity of the aircraft fuel system
- **Sync from Gauges**,¹ **Full**, and **Tab** preset keys aid in fuel data entry

¹ Available with TXi software v3.21 and later.

Fuel range rings are a pilot-selectable map overlay indicating an estimate of remaining flight distance based on fuel onboard, fuel consumption rates, and current ground speed. Map depicts two separate rings: an outer yellow ring displaying total endurance range, and an inner dashed green ring displaying range to reserve fuel.

Fuel Computer Setup Options



Setup options allow you to customize features on the fuel computer.

From here you can:

- Specify tank full and tab amounts
- Enable the Sync from Gauges function
- Enable the Aircraft Weight feature

Full Capacity

Specify the tank full amount.

Tab Capacity

Specify a tank tab amount.

Fuel Computer Sync

Select Pilot Control to add the Sync from Gauges option to the fuel computer. To remove this option, select Disabled.^{1, 2}

Aircraft Weight

Enable or disable the Aircraft Weight feature.³

Fuel Computer Sync option availability is dependent upon fuel computer unit selection. If the fuel computer units do not match the configured fuel gauge units, the **Fuel Computer Sync** key is unavailable. Fuel computer units are set on the System Units page.

¹ Available with TXi software v3.21 and later.

² Unavailable if fuel computer units do not match the configured fuel gauge units.

³ Available with TXi software v3.80 and later.

Preset Fuel Quantities



The fuel computer stores preset fuel amounts for estimated full and tab amounts. For the operating limitations of a specific aircraft, consult the POH.

Full	Sets the fuel remaining value to the specified full amount.
Tab	Sets the fuel remaining value to the specified tab amount.
Sync from Gauges	Performs a summation of all configured fuel tanks and inputs the amount as the fuel remaining value. ¹
Setup	Opens the Fuel Capacity Setup menu.

¹ Available with TXi software v3.21 and later.

In certain installations, fuel temperature is used to adjust the indicated weight of fuel for changes in density. An advisory alerts the pilot when fuel quantity is unavailable due to invalid fuel temperature data. To determine if temperature compensation is configured in the aircraft, consult the AFMS.

SYNC FROM GAUGES

FEATURE REQUIREMENTS

- TXi software v3.21 or later

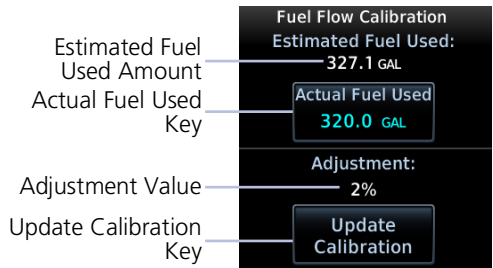


This function provides a convenient way to set the Fuel on Board value during preflight setup of the EIS.

The pilot is responsible for determining the amount of fuel on board and verifying the input value entered into the fuel computer.

Fuel Flow Calibration

A calibration function compares the estimated fuel used amount (calculated by the system) with the actual fuel used amount (determined by the pilot upon refueling). The system calculates a fuel flow correction factor based on these inputs. This adjustment value is applied to future fuel flow calculations.



Calibrating the fuel flow meter helps ensure the accuracy of all fuel computer calculations.

The adjustment maximum is 15 percent. Contact your Garmin dealer if further adjustment is required.

Actual Fuel Used	Allows you to specify the actual amount of fuel used since the last adjustment to the EST Fuel Remaining quantity.
Update Calibration	Updates the current calibration value. Confirming the request re-calibrates the fuel flow meter and saves the new calibration value.

Zero Fuel Weight

Quick Entry Options

- Up to three pilot-specified zero fuel weight preset values
- **Previous** key recalls the last manually entered zero fuel weight value

This function provides a convenient way to enter the calculated zero fuel weight of the aircraft. You can manually specify a value or use one of the available quick entry options.

FEATURE REQUIREMENTS

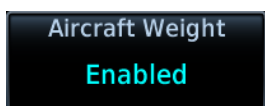
- *Aircraft Weight feature enabled by pilot on GDU*

FEATURE LIMITATIONS

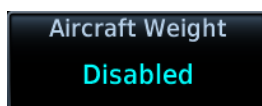
- *Zero Fuel Weight automatically disables when fuel computer is not available*

AIRCRAFT WEIGHT FEATURE

You may disable and re-enable the Aircraft Weight feature at any time.



Tapping **Aircraft Weight** toggles the function between Enabled and Disabled. The feature is enabled by default.



Selecting **Disabled** removes all aircraft weight related controls and indications from the fuel computer.

The Aircraft Weight option resides in the Fuel Computer Setup menu. You can access this menu multiple ways.

From the EIS start-up page:

Tap **EST Fuel Remaining** > **Setup** > **Aircraft Weight**.

From the EIS display:

Tap **Menu** > **Fuel Computer** > **Setup** > **Aircraft Weight**, or

Tap **Menu** > **Fuel Computer** > **EST Fuel Remaining** > **Setup** > **Aircraft Weight**.

When this feature is enabled, you can:

- Manually enter the zero fuel weight value or select an assigned preset value
- Assign or delete preset values
- Recall the last manually entered zero fuel weight value

ENTER ZERO FUEL WEIGHT

1. Tap **Zero Fuel Weight**.
2. Type the zero fuel weight value and then tap **Enter**.

Calculate zero fuel weight by adding the basic operating weight, total passenger weight, and cargo weight.

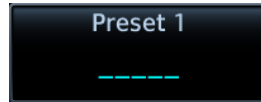
ZERO FUEL WEIGHT PRESET OPTIONS

Tap **Zero Fuel Weight > Setup** and select a preset key. Preset keys are unassigned by default.

Assigned



Unassigned



Selecting a preset key opens a keypad. From here, you have the following options.

- *Assign a preset:* Type the zero fuel weight value and then tap **Enter**.
- *Enter the assigned preset value:* Tap **Enter**.
- *Delete the assigned preset:* Tap **Delete**.

RECALL THE PREVIOUSLY ENTERED ZERO FUEL WEIGHT VALUE



For convenience, you may recall the last manually entered zero fuel weight value.

Tap **Zero Fuel Weight > Previous > Enter**.

The **Previous** key appears only after an initial zero fuel weight value is entered into the system.

Advanced EIS Configurations

Fuel Balance Monitoring

FEATURE REQUIREMENTS

- Engine adapter installed (GEA 110 or GEA 71x)
- Installer configuration

FEATURE LIMITATIONS

- Not compatible with selectable fuel gauges

Out of balance fuel loads can trigger visual indications on the affected fuel gauges and activate other aircraft systems.

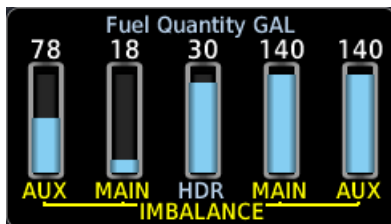
Fuel Imbalance System Installer Settings

- Activation threshold
- Activation time delay
- Imbalance monitoring on/off
- Maintenance log entry
- Maximum imbalance threshold
- Left and right heavy wing discrete outputs (includes discrete inhibit signal input and fuel imbalance discrete output activation/deactivation conditions)

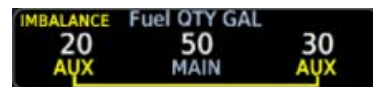
The fuel imbalance system is capable of monitoring up to three pairs of fuel tanks as well as left and right fuel totals. Installer configurable settings are available for each pair.

The system activates when a monitored pair is out of balance by more than its activation threshold for longer than its time delay. When this occurs:

- Fuel gauges provide a visual indication
- Installer configured aircraft systems activate if the imbalance is greater than the balancing or activation threshold depending on the preferred settings



Vertical Bar Gauge Imbalance



Digital Gauge Imbalance

Fuel transfer pumps, external annunciators, or other aircraft systems can be activated based on a separately configurable balancing threshold, allowing automatic fuel balancing to be performed before a full imbalance occurs. The installer can also configure a switch or other signal to inhibit the activation of these aircraft systems by the fuel imbalance monitor.

Fuel Imbalance Alerting

FEATURE LIMITATIONS

Optional Balance On and Fuel Imbalance Warning indications are dependent upon installer configuration and available only for installations that incorporate an automatic fuel balancing capability.

Timer duration and availability are dependent upon installer configuration and aircraft type.

Heavy wing discretes can be configured to activate before reaching an imbalance threshold. Additionally, a timer (e.g., 60 seconds) can be configured to count down prior to activation, when the difference in tank quantities reaches a predefined imbalance range. This delay between state transitions helps prevent inadvertent activation when turbulence or aircraft maneuvers may cause temporary fuel shifts (e.g., fuel sloshing, noise in fuel signals).

An alert annunciates once the timer expires. The type of alert is dependent upon the difference between left and right tank quantities.

FUEL IMBALANCE ANNUNCIATIONS

Balance On^{1, 2}

BALANCE ON

- Fuel imbalance amount is above the configured balancing limit and below the caution threshold
- Automatic fuel balancing system is active
- System attempts to restore balance until the difference in left and right tank quantities is less than or equal to the deactivation threshold

Fuel Imbalance Caution

IMBALANCE

- Fuel imbalance amount is above caution threshold
- Automatic fuel balancing system (if installed) is active
- Consult AFMS emergency procedures for details

Fuel Imbalance Warning

IMBALANCE

- Fuel imbalance amount is above critical level
- Automatic fuel balancing system (if installed) may be configured to turn off at this level^{1, 2}
- Consult AFMS emergency procedures for details

¹ Optional with TXi software v3.40.

² Available only for installations that incorporate an automatic fuel balancing capability.

Automatic Ignition

TURBINE AIRCRAFT ONLY



NOTE

Always monitor engine and igniter status. Control ignition manually if necessary.

FEATURE REQUIREMENTS

- Engine adapter installed (GEA 71x)
- Applicable sensor interfaces configured
- Installer configuration

FEATURE LIMITATIONS

The auto ignition feature does not trigger a unique indicator on the GDU. If an ignition text lamp is configured, the ignition indicator will be shown, but does not indicate if the system was activated by the auto ignition system.

This feature is only available with GDU 700P/1060/1210.

This feature automatically switches turbine engine ignition systems on and off based on engine parameters and/or aircraft configuration.

Engine ignition systems that can be switched on and off with an external input can be triggered by a signal from the GDU based on installation-specific configurations.

If a trigger parameter is not available, the automatic ignition feature is disabled and an advisory is displayed. If a parameter becomes unavailable on the ground, a "Service Required" advisory will be displayed. When the aircraft is in the air and the feature is disabled, an "Automatic Ignition Unavailable" advisory will be displayed.

Activation and deactivation conditions are configured separately, and may include multiple AND/OR conditions. Trigger parameters include ITT, NG, NP, torque, condition lever position, and air/ground state. Conditions may also include a delay time.

Example 1:

Auto ignition is switched on when ITT is less than 500°C AND the condition lever is not in cutoff/feather.

Example 2:

Auto ignition is switched off when ITT is more than 500°C for more than 10 seconds.

Automatic Starting

TURBOPROP AIRCRAFT ONLY



NOTE

Always respond to starting issues (e.g., hot starts, hung starts) as directed in the POH. In the case of an EIS malfunction, maintenance is recommended to troubleshoot the associated components and connections.

FEATURE REQUIREMENTS

- TXi software v3.21 or later
- Installer configuration

For approved installations that require an automatic starting system, GDU outputs can be interfaced with aircraft engine start relays to facilitate engine starting.

GDU monitors inputs, applies logic and delays, then sets discrete outputs to the necessary state to control or signal other LRUs (typically non-Garmin) in the engine starting system.

An “Automatic Starting Inhibited” advisory alerts you when the automatic starting feature is inhibited by a problem with the system.

Three engine states are used in the starting process.

Starting	Starter Cut-off ¹	Reset
Pilot engages starter via switch or button press. Start sequence begins.	NG is greater than or equal to 50%. ² Or 60 second timer expires while in the “Starting” state. ²	Pilot engages the starter reset. Or Starter power becomes inactive. System responds by depowering the starter relay(s).

¹ Due to engine running or sequence timeout. ² Value varies based on aircraft specific configuration.

Always control and monitor the start sequence as directed by the POH.

Gauge-driven Outputs

FEATURE REQUIREMENTS

- *Engine adapter installed (GEA 110 or GEA 71x)*
- *Installer configuration*

Gauge-driven outputs are available for the following gauges:

- ITT
- Torque
- Propeller RPM
- NG
- Manifold Pressure
- Fuel Flow
- Oil Pressure
- Oil Temperature
- Vacuum Pressure
- Alt/Gen Amps
- Alt/Gen Volts
- Bus Volts
- Battery Amps
- Battery Volts
- Fuel Quantity
- Engine Power
- Flight Control Trim Position

This feature sends a discrete signal to other aircraft systems based on gauge values or ranges.

Gauge-driven outputs are available with a GDU that is not configured as an EIS. This feature does not require the gauge which drives an output to be displayed.

A single gauge can drive multiple outputs of different configurations.

An output may be triggered when:

- a gauge is at or above, below, or between configured values
- a parameter is within green, yellow, or red ranges

EIS Operations

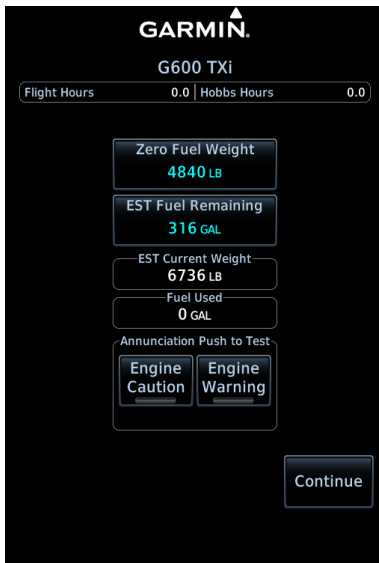
PHASE OF FLIGHT	OPERATION
System Startup & Initialization	• Customize system settings • Set fuel full and tank capacity levels • Set engine advisories • Calibrate fuel flow • Test external master caution and master warning switches (if applicable)
Preflight	• Add fuel • Set zero fuel weight • Set target fuel for takeoff²
En Route	• Lean the engine¹ • Set target fuel²
Post Flight	Record the following: • Hobbs and flight hours • Tach hours¹ • Aircraft log data • Exceedance data²

¹ Reciprocating engine only. ² Turbine engine only.

System Startup & Initialization

FEATURE REQUIREMENTS

- TXi software v3.21 or later



**EIS Start-up Page
(GDU 1060/1210 MFD)**

During power up, the EIS start-up page provides access to initial EIS setup and preflight functions.

From here you can:

- Record Hobbs, tach (piston only), and flight hours
- Enter Zero Fuel Weight (if aircraft weight is enabled)
- Perform fuel computer calculations
- View the amount of fuel used
- Test external caution/warning switches (if applicable)
- Set operating mode and target value for primary power indicator¹

Tapping **Continue** advances to the Home page.



GDU 700P

GDU 700(): Depending on configuration, EIS start-up functions may display on a dedicated page or in a slide over menu.



GDU 700L

¹ Turboprop and turbofan aircraft only.

CUSTOMIZE SYSTEM SETTINGS

Specify all necessary EIS settings. Customization options reside in either the System page (GDU 1060/1210) or the System menu (GDU 700()).

SET FUEL FULL AND TAB CAPACITY

Specify estimated fuel full and tab values using the controls in the fuel computer.

SET ENGINE ADVISORIES

Specify engine advisory parameters and limit values.

CALIBRATE FUEL FLOW



NOTE

If the calculated adjustment factor exceeds the 15 percent fuel flow calibration limit, contact your dealer for support.

Calibrate fuel flow using the controls provided in the fuel computer. Re-calibration is recommended following the first flight or any time there is a significant difference between the estimated remaining fuel quantity and the actual amount of fuel in the tanks.

1. Fuel aircraft to a known level (i.e., the bottom of the tabs inside each fuel filler port).
2. Tap **EST Fuel Remaining** and enter the amount of fuel on board.
NOTE: This value must be at least 5 liters (2 gallons) greater than the previous estimated fuel on board amount.
3. Burn no less than 10 percent of usable fuel. Burning more than 10 percent will yield a more accurate calibration.
4. Refuel to the previously entered level.
5. Tap **EST Fuel Remaining** and enter the amount of fuel on board.
6. Tap **Calibrate FFlow** and enter the amount of actual fuel used. Note the adjustment value.
7. Tap **Update Calibration**.
8. Confirm the request.
9. Acknowledge the pop-up message.

Preflight

ADD FUEL

Update the remaining fuel quantity whenever fuel is added to the aircraft.

1. Fuel the aircraft.
2. Enter the appropriate fuel amount using the fuel computer.

SET ZERO FUEL WEIGHT

Specify the total weight of the aircraft, including cargo and passenger weight, with no disposable fuel or oil on board. This setting is accessible from the EIS start-up page during power up (if configured), or from the Engine menu.

From the start-up page:

1. Tap **Zero Fuel Weight**.
2. Enter the zero fuel weight amount.

From the EIS display:

1. Tap **Menu > Zero Fuel Weight**.
2. Enter the zero fuel weight amount.

SET TARGET BUG FOR TAKEOFF

TURBINE AIRCRAFT ONLY

Select an operating mode and enter a performance target value for the configured power indicator. Primary power bug settings are accessible from the EIS start-up page during power up (if configured), or from the Engine menu.

From the start-up page:

1. Enter the target value or ground OAT (if applicable).
2. Enable the configured operating mode.

From the EIS display:

1. Tap **Menu > Target Bug**.
2. Enter the target value or ground OAT (if applicable).
3. Enable an operating mode.

En Route

LEAN THE ENGINE

PISTON AIRCRAFT ONLY

1. Select the preferred lean source from the Lean Mode menu.
2. Switch to the lean view.
3. Watch the temperatures graph for reference peak events and lean the mixture accordingly.

SET TARGET BUG

TURBINE AIRCRAFT ONLY

Select an operating mode or enter a performance target value.

Post Flight

RECORD AIRCRAFT LOG DATA

Document total Hobbs, tach (piston only), and flight time. If configured, record the number of engine and flight cycles. This information is necessary for logging and maintenance purposes.

EXPORT DATA LOGS

Export flight data and (for turbine) engine exceedance logs to an SD card for later analysis.

Engine Advisories & Alerts

APPLICABLE TO PISTON & TURBINE AIRCRAFT

Engine Advisories

You can specify values at which the system will advise of certain engine parameters. The system issues an advisory message when an engine gauge exceeds its specified value. These messages are acknowledged in the same manner as system advisories.

Available selections are based on aircraft configuration.

- Low Endurance¹
- Low Estimate Fuel Remaining¹
- High/Low Oil Temperature
- High/Low Battery Voltage
- High/Low Bus Voltage
- CHT Cooling Rate²
- EGT Differential²
- High CHT²
- High TIT²

¹ Actual amounts are based on fuel computer calculations and an active flight plan.

² Reciprocating engine only.

Engine Alerts



NOTE

Engine warnings require immediate action.

Engine alerts display as textual annunciations on the PFD. They disappear once the alert is acknowledged by the pilot.

ACKNOWLEDGE ALERT KEY

Tapping the **ACK** key acknowledges the condition and suppresses all flashing alert indications (i.e., annunciation, gauge).¹ It does not remove the annunciation from the display. Alert annunciations and non-safe range indications remain active (solid) for as long as the gauge is within the alerted range.



The key flashes yellow and black when an engine caution is present.



The key flashes red and white when an engine warning is present.

ANNUNCIATION	CONDITION
ENGINE	An engine parameter indicates a red alerting range (low or high). Note: Requires immediate action.
ENGINE	An engine parameter indicates a yellow alerting range (low or high).

¹ GDU 1060 PFD/MFD/EIS with TXi software earlier than v3.50: On-screen acknowledge key not available. Unless configured for external acknowledge switches, alerts automatically acknowledge after 10 seconds.

REMOTE ENGINE ANNUNCIATIONS

Engine alerts annunciate on the PFD when GDU 700()/1060/1210 is connected to a separate GDU configured for EIS.



Engine Annunciator on PFD

EXTERNAL ENGINE ALERT LAMPS

An external lamp is required if the only EIS display is installed outside the pilot's primary field of view. This lamp illuminates any time an engine alert is active. It turns off once the alert is acknowledged.

EXTERNAL ENGINE ALERT ACKNOWLEDGMENT

FEATURE REQUIREMENTS

- *TXi software v3.21 or later*

Depending on configuration, the system may allow you to acknowledge engine caution and warning annunciations remotely via dedicated aircraft switches. This feature replaces the on-screen Acknowledge Alert key, eliminating the need to acknowledge an engine alert in multiple locations (e.g., on GDU and on the aircraft caution/warning panel).

Annunciation Push-To-Test Options

Test keys allow you to confirm that the external master caution and master warning switches are functioning properly outside of actual alerting conditions. During unit power up, these toggle keys are available on the EIS start-up page.





Test keys are temporarily unavailable as the unit initializes. A status message informs you that the system is powering up.

After approximately 10 seconds, the keys are available for selection.

Following start-up, you can test the aircraft switches at any time from the Engine menu.

GDU 1060/1210: From the Home page, tap **Engine > Menu > Lamp Test**, and select the appropriate test key.

GDU 700(): Tap Menu > Lamp Test, and select the appropriate test key.

To test the external engine alert acknowledgment function:

1. Tap **Engine Caution** and **Engine Warning**.
2. Observe that the external alert lamps illuminate.
3. Acknowledge each alert indication by pressing the appropriate aircraft switch.
4. Observe that the color of the associated message on EIS changes to indicate acknowledgment.

INTENTIONALLY LEFT BLANK

10 Abnormal Operations

BACKUP INSTRUMENTS 10-2

EMERGENCY MODES 10-38

Backup Instruments

Display Backup

FEATURE LIMITATIONS

- *Applicable to pilot side units only*
- *Only pilot or center units trigger backup mode when offline*
- *Not available in single TXi display installations*
- *Display backup mode supports a limited set of EIS gauges*
- *Not available on GDU 700P configured as twin-engine turboprop/turbofan EIS (includes standby PFD)*

Display backup mode occurs under these three conditions

- Loss of communication with a GDU that displays EIS or PFD data
- Pilot activation through on-screen selection or toggle switch
- AHRS or ADC fault caused by miscompare or missing data (this only triggers backup mode when the unit is configured as a standby PFD)

Display backup mode attempts to match the configuration of the lost primary display unit. When transitioning, priority is given to the display of primary gauges. In some cases, menu items and lean controls are disabled in this mode. Any flight control trim gauges appear in display backup mode.

An optional external toggle switch allows manual activation of backup mode.

TCAS II RAs

When GDU 1060/1210 is in display backup mode and providing PFD/MFD/EIS functions, the format of the PFD narrows and the VSI is too small to depict RA bands.

To view TCAS II RAs, tap **Full** on the PFD to expand it to 80%.

Standby PFD Display Options

FEATURE REQUIREMENTS

Additional standby PFD functionality is available in installations that include:

- Dual external ADAHRS sensors: one external ADAHRS; one internal AHRS connected to an ADC
- Two pilot side GDU 700P units: one configured as a dedicated PFD, the other configured as either MFD, EIS, or MFD/EIS
- Non-PFD GDU configured as standby PFD

During display backup, the standby PFD allows you the option of continuing backup PFD mode or returning to the unit's previous display mode (MFD, EIS, or MFD/EIS). This option is available only if display backup is triggered by an AHRS or ADC data mismatch or by pilot activation.

DISPLAY BACKUP KEY



This key remains available during display backup for as long as the GDUs are in communication with each other. It is not available during display backup if only the standby PFD is operational.

Deselecting **Display Backup** toggles the function off and returns the standby PFD to its previous display mode. Methods for accessing this option vary according to unit type and configuration.

GDU 700()/1060



Display options are accessible via the **Power** key.

If configured for emergency functions, **Display Backup** is also accessible from the Emergency page.

Disable display backup one of two ways:

From the Display Options menu:

Push the **Power** key and deselect **Display Backup**.

From the MFD Home page:

Tap **Emergency** and deselect **Display Backup**.

GDU 1210



A banner in the Emergency Mode Status window indicates when display backup mode is active.

The display backup option resides on the Emergency page.

From the MFD Home page:

Tap **Emergency** and deselect **Display Backup**.

Composite Display Backup Mode

When a GDU 700P enters a PFD/EIS composite display backup mode, certain aspects of the normal PFD layout change to accommodate the display of EIS data.

PFD when in composite display backup mode:

- A limited, 90° field of view HSI replaces the Standard HSI or HSI Map
- **CDI** source selection key replaces air temperature and wind data fields
- Attitude background defaults to a blue over brown artificial horizon display (non-SVT)

Removed from display in composite mode:

- Air temperature field
- Clock/timer field
- Wind data fields
- PFD **Menu** key
- **Advisory** key

Unchanged from display in composite mode:

- Airspeed
- Barometric altitude
- Attitude

Unavailable PFD functions operating in composite mode:

- Changing airspeed references
- Setting BARO or RA Minimums
- Selection of AHRS or ADC sources
- Manual adjustment of backlight levels
- Viewing external LRU status



GDU 700P Composite Display in Backup Mode - Reciprocating Engines



GDU 700P Composite Display in Backup Mode - Single Engine Turboprop

DISPLAY BACKUP MODES



Normal Mode EIS



PFD/MFD



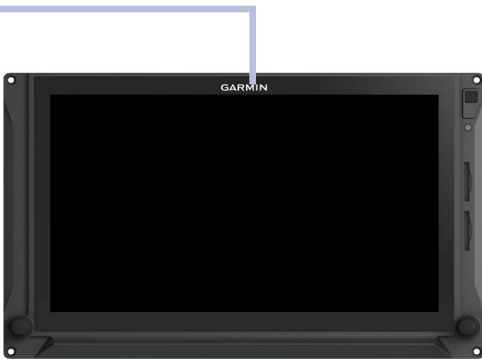
Display Failure



Backup Display Mode

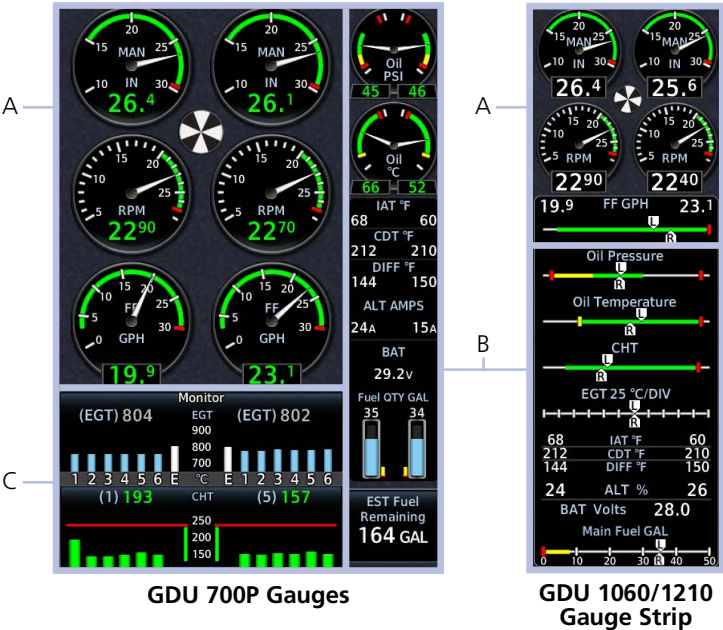


Backup Display Mode



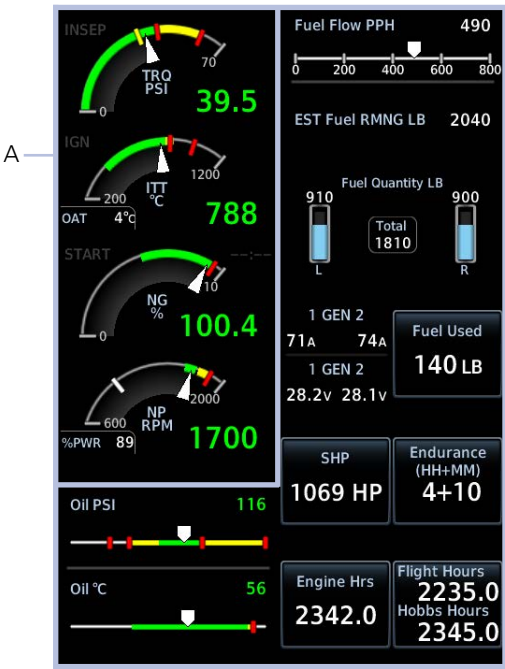
Display Failure

EIS GAUGE REVERSION



A	Gauges display with little or no change.
B	Gauges transfer depending on priority and open gauge slots.
C	Gauges simplify and transfer depending on available display area.

TURBOPROP GAUGE AREAS



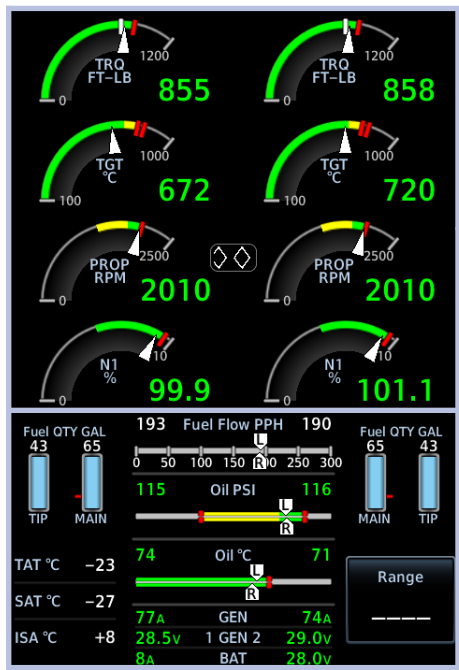
GDU 700P Gauges



GDU 1060/1210 Gauge Strip

A	Gauges display with little or no change.
B	Gauges transfer depending on priority and open gauge slots.

TWIN TURBOPROP GAUGE AREAS



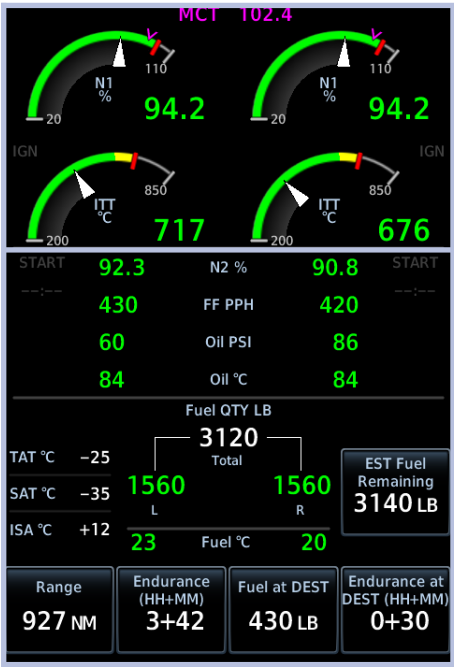
GDU 700P Gauges



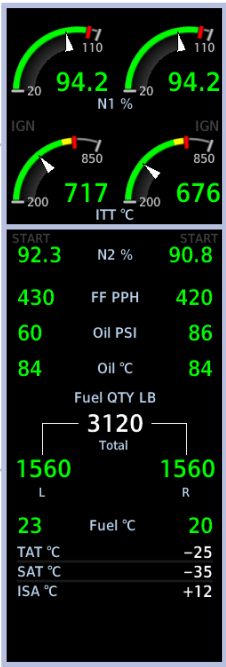
GDU 1060/1210 Gauge Strip

A	Gauges display with little or no change.
B	Gauges transfer depending on priority and open gauge slots.

TWIN TURBOFAN GAUGE AREAS



GDU 700P Gauges



GDU 1060/1210 Gauge Strip

A	Gauges display with little or no change.
B	Gauges transfer depending on priority and open gauge slots.

Backup Battery

FEATURE LIMITATIONS

- *Applicable to GDU 700() only*
- *During backup battery mode, display brightness is set for minimum sunlight readability*

Battery Supported LRUs

- Backup GPS
- GDU 700()
- GEA 110
- GMU 44
- GTP 59
- Integrated ADAHRS

GBB 54 Features

- Lithium-ion battery unit
- Integrated heater ensures performance in low temperatures
- Charged by GDU during normal operations
- Status indications denote battery state and availability
- Pop-up alerts with estimated time remaining
- System advisories with timer display



GBB 54

Icon not present when backup battery is at full charge.



Battery Icon

GBB 54 Backup Battery

Battery Status Indications

Status indications include a battery icon and timers.

BATTERY ICON

- Displays when the GBB 54 is powering the GDU, battery charge state is insufficient, or a battery fault exists
- Displays in the annunciator bar and on External LRUs page
- Removed when the battery is at full charge

ICON	STATE	CONDITION
Absent ¹	Charged	• Backup battery at full charge (>30 minutes) and available for use.
	Charging/Warming Up	• Battery has less than 30 minutes remaining charge.² • Battery temperature is below -20°C.^{2, 3}
	Discharging	• GDU on backup battery power. 15 to 30 minutes remaining.
	Discharging/Caution	• GDU on backup battery power. 0 to 15 minutes remaining.
	Discharging/Warning	• GDU on backup battery power for longer than 30 minutes. Loss of power may be imminent.⁴ • Charge state uncertain (i.e., GDU switches to backup battery power before battery reaches full charge).
	Unavailable/Error	• Battery error. Backup battery power is unavailable. • Capacity test is overdue.

¹ A green check on the External LRUs page denotes LRU availability.

² Aircraft power is required. ³ The warm up process can take up to 30 minutes.

⁴ A total power loss occurs when aircraft power fails and the backup battery is unavailable or depleted.

BATTERY TIMERS

Battery timers alert the pilot of:

- Transitions from aircraft power to backup battery power
- Total time on battery power

TIMER	ALERT TYPE	CONDITION
Elapsed time on battery power (count-up)	Advisory	Aircraft is in flight.
60 second countdown	Pop-up	Aircraft is on ground or in an unknown state.

Automatic Unit Shutdown

To prevent battery depletion, the unit automatically powers off once the 60 second countdown timer expires.



Tapping **Stay On** cancels the automatic shutdown sequence, allowing the unit to continue operation on backup battery power.

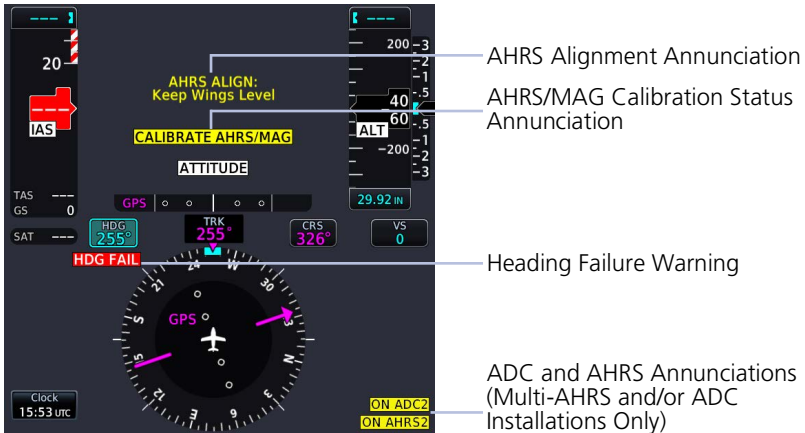
Battery Alerts

Caution messages alerting to backup battery status may display while in flight. On GDU 700L, these alerts are shortened to accommodate space limitations.

ANNUNCIATION	CONDITION POP-UP
Backup battery in use. Estimated time remaining: 15 minutes.	Condition: GDU on backup battery power, discharging for 15 minutes. Pop-up Alert: Yes
Backup battery has low charge or fault. See flight manual for operating limitations.	Condition: • Battery is charging; estimated charge is below 30 minutes • GDU detects a battery fault or warning state • Loss of communication • Battery power output failure • Capacity test failure or testing overdue Pop-up Alert: Yes

AHRS & ADC Failures

The AHRS uses aiding data from the GPS navigator, air data computer, and magnetometer to provide accurate attitude data. Failure of these inputs may cause attitude data to become unavailable. Additional annunciations indicate current alignment and calibration status.



In multi-AHRS/ADC installations, use the onside AHRS/ADC during normal operations and the offside AHRS/ADC only when necessary. Under normal conditions, sensor assignment should be as follows.

Pilot PFD: AHRS 1, ADC 1

Copilot PFD: AHRS 2, ADC 2

Standby PFD (pilot side): AHRS 2, ADC 2

AHRS & ADC Automatic Source Selection

FEATURE REQUIREMENTS

- *Three independent sources for automatic switching based on miscompare events*
- *TXi software v3.61 or later*

FEATURE LIMITATIONS

- *Miscompare-based automatic switching not applicable to ADC sources. ADC miscompares will annunciate, but ADC sources will not automatically switch.*

When configured by the installer, AHRS and ADC automatic source selection automatically switches to the next best ADC or AHRS in the event of an abnormality or failure. By default, this function is off.

AHRS AUTOMATIC SOURCE SELECTION

AHRS automatic source switching occurs when the following conditions are true.

- Automatic source switching is enabled
- On-ground state is inactive
- AHRS 1/AHRS 2 avionics discrete input is not configured
- The AHRS source to be switched *from* is in any of the following states:
 - Fail
 - Aligning
 - Is a standby source when the system transitions from on-ground to in-air
 - Individually flagged as in a miscompare state
- The AHRS source to be switched *to* satisfies all of the following:
 - Available
 - Not aligning
 - Has not already been automatically switched to within the current GDU power cycle (unless the previous switch occurred due to a standby source on-ground to in-air transition)
 - Not in a miscompare state

Aircraft with GFC 600 Extended Availability

When configured for automatic sensor switching, TXi selects valid sensors as requested by the GMC.

- If TXi does not have a correlated miscompare/no compare, a system advisory informs you that the autopilot is reporting a miscompare with the indicated sensor.
- If a sensor is invalid, TXi does not switch to the sensor and the autopilot disconnects after one second.

The miscompare state can be generic (two sources disagree) or specific (the source was individually flagged in a three-source system). In either case, the system will not automatically switch to a source that is a possible cause for disagreement.

ADC AUTOMATIC SOURCE SELECTION

ADC automatic source switching occurs when the following conditions are true.

- Automatic source switching is enabled
- On-ground state is inactive
- ADC 1/ADC 2 avionics discrete input is not configured
- The ADC source to be switched *from* is in any of the following states:
 - Fail
 - Is a standby source when the system transitions from on-ground to in-air
- The ADC source to be switched *to* satisfies all of the following:
 - Available
 - Has not already been automatically switched to within the current GDU power cycle (unless the previous switch occurred due to a standby source on-ground to in-air transition)

SOURCE PRIORITY

Source switching happens in order of greatest priority to least priority.

Source priority is determined in the following order:

1. Default sources have highest priority (i.e., when a default source is not currently active, switching back to default will be the first consideration).
2. Primary sources have priority over standby sources (e.g., ADC 2 has priority over STBY ADC).

Each GDU individually evaluates automatic source selection.

AHRS & ADC Sensor Selection

FEATURE LIMITATIONS

- Standby sensor options available with TXi software v3.61 and later
- AHRS GPS aiding option available with TXi software v3.85 and later

OPTIONAL COMPONENTS

- If connected via HSDB, GI 275 may serve as a selectable source

Source selection controls are available for up to three ADC sensors and three AHRS sensors.

SELECT SENSORS ON PILOT/COPILOT PFD

Default Selection
(Active)



Alternate Selection
(Active)

AHRS and ADC selection keys reside on the Sensors page of the PFD menu.

Alternate attitude or air data sources annunciate on the PFD. Default selections do not annunciate.

ON ADC2

SELECT SENSORS ON A STANDBY TXI PFD



Default Selections
(Active)

By default, the pilot side standby PFD uses ADC 2 and AHRS 2 data when the unit transitions to backup PFD mode. This occurs during any one of the following conditions.

- Primary PFD failure
- Miscompare state
- No compare state

Pilot Selected
Sensors (Active)



Standby PFD allows you to specify alternate data sources if the default selections are not preferred.



PFD



MFD (Standby PFD)



Display Failure



Sensor Options Available

During backup PFD mode, sensor select options remain available.

Standby PFD Sensor Options

SOURCE FAILURE INDICATIONS

FEATURE LIMITATIONS

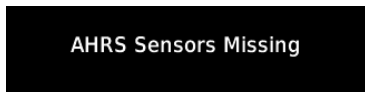
- Autopilot is not available while a standby (STBY) sensor source is active



A red "X" appears over the associated selection key when a source fails. A yellow "X" appears over the selection if the source is individually flagged as miscomparing or is currently in an aligning state (AHRS only). Sources are still selectable regardless of red or yellow "X" indications.



Standby sources (e.g., GI 275 via HSDB) are selectable when primary sources fail or while the aircraft is on-ground.



Relevant advisory messages appear at the bottom of the menu.

Standby Source Selection Availability

Standby source selection is generally unavailable as selecting a standby source will decouple the active autopilot. Specifically, standby source selection is unavailable while in-air, when primary sources are available and functioning. For testing purposes, standby source selection is available while on-ground.

Caution should be exercised to ensure that primary sources are selected again prior to takeoff. If automatic source switching has been configured by the installer, then the standby source selections that are active while the aircraft transitions to in-air automatically switch back to the default source selection if that source is functioning.

AHRS/ADC SENSOR SELECTION ANNUNCIATIONS

SINGLE TXi PFD (TWO SOURCES)		
ANNUNCIATION	DISPLAY	DESCRIPTION
ON ADC 2	Pilot's PFD	Pilot selected ADC 2 or AHRS 2 as the sensor source.
ON AHRS 2		
ON ADC STBY	Pilot's PFD	Pilot selected GI 275 standby ADC or standby AHRS as the sensor source, possibly after primary source became unavailable.
ON AHRS STBY		

SINGLE TXi PFD (TWO PRIMARY SOURCES & ONE GI 275 STANDBY SOURCE)		
ANNUNCIATION	DISPLAY	DESCRIPTION
ON ADC 2	Pilot's PFD	Pilot selected ADC 2 or AHRS 2 as the sensor source.
ON AHRS 2		
ON ADC STBY	Pilot's PFD	Pilot selected GI 275 standby ADC or standby AHRS as the sensor source, possibly after both primary sources became unavailable.
ON AHRS STBY		

TWO TXi PFDs (TWO PRIMARY SOURCES)		
ANNUNCIATION	DISPLAY	DESCRIPTION
ON ADC 2	Pilot's PFD	Pilot selected ADC 2 or AHRS 2 as the sensor source.
ON AHRS 2		
ON ADC 1	Copilot's PFD	Pilot selected ADC 1 or AHRS 1 as the sensor source.
ON AHRS 1		
ON ADC 1	Standby PFD (pilot side)	Pilot selected ADC 1 or AHRS 1 as the sensor source. GDU is in display backup mode.
ON AHRS 1		
ON ADC 2	Standby PFD (pilot side)	Sensor source is the default setting (sensor 2).
ON AHRS 2		
BOTH ON ADC 2	Both PFDs	Pilot selected ADC 2 or AHRS 2 as the sensor source on both PFDs.
BOTH ON AHRS 2		
BOTH ON ADC 1	Both PFDs	Pilot selected ADC 1 or AHRS 1 as the sensor source on both PFDs.
BOTH ON AHRS 1		

TWO TXi PFDs (TWO PRIMARY SOURCES & ONE GI 275 STANDBY SOURCE)		
ANNUNCIATION	DISPLAY	DESCRIPTION
ON ADC 2	Pilot's PFD	Pilot selected ADC 2 or AHRS 2 as the sensor source.
ON AHRS 2		
ON ADC 1	Copilot's PFD	Pilot selected ADC 1 or AHRS 1 as the sensor source.
ON AHRS 1		
ON ADC 1	Standby PFD (pilot side)	Pilot selected ADC 1 or AHRS 1 as the sensor source. GDU is in display backup mode.
ON AHRS 1		
ON ADC 2	Standby PFD (pilot side)	Sensor source is the default setting (sensor 2).
ON AHRS 2		

TWO TXi PFDs (TWO PRIMARY SOURCES & ONE GI 275 STANDBY SOURCE)		
ANNUNCIATION	DISPLAY	DESCRIPTION
ON ADC STBY	Pilot's PFD	Pilot selected GI 275 standby ADC or standby AHRS as the sensor source, possibly after both primary sources became unavailable.
ON AHRS STBY		
ON ADC STBY	Copilot's PFD	
ON AHRS STBY		
ON ADC STBY	Standby PFD (pilot side)	
ON AHRS STBY		
BOTH ON ADC 2	Both PFDs	Pilot selected ADC 2 or AHRS 2 as the sensor source on both PFDs.
BOTH ON AHRS 2		
BOTH ON ADC 1	Both PFDs	Pilot selected ADC 1 or AHRS 1 as the sensor source on both PFDs.
BOTH ON AHRS 1		
BOTH ON ADC STBY	Both PFDs	Pilot selected GI 275 standby ADC or standby AHRS as the sensor source on both PFDs, possibly after both primary sources became unavailable.
BOTH ON AHRS STBY		

AHRS GPS AIDING



This option allows you to disable AHRS GPS aiding in the event that GPS data becomes unreliable. It resides on the Sensors page of the PFD menu (**Menu > Sensors > AHRS GPS Aiding**).

FEATURE LIMITATIONS

- *Availability dependent upon configuration*

When you disable AHRS GPS aiding:

- The selection automatically crossfills to all capable TXi and GI 275 units in the cockpit¹
- TXi no longer provides AHRS GPS aiding data to the GRS or GSU
- AHRS enters backup No-GPS mode
- A system advisory informs you that the function is no longer enabled (i.e., AHRS GPS alerting is suppressed)

The function defaults to enabled following a unit power cycle.

Re-enable AHRS GPS aiding once you confirm that GPS data is reliable.

¹ Automatic crossfilling functionality dependent upon configuration of each capable LRU.

AHRS Operating In Reversionary Mode

A failure of the air data input has no effect on AHRS output, while AHRS is operating in normal mode. If the air data input fails, while the AHRS is operating in backup No-GPS mode, PFD attitude and heading data are invalid.

GPS Failure

Two GPS inputs may be provided to the AHRS. If GPS information from one of the inputs fails, the AHRS uses the remaining GPS input, and an advisory message informs the pilot. If both GPS inputs fail, the AHRS continues to provide attitude and heading information to the PFD as long as true airspeed data is available and valid.

AHRS Alerts

ANNUNCIATION	CONDITION
HDG FAIL	Invalid heading.
AHRS ALIGN: Keep Wings Level	AHRS is aligning. Appears during AHRS initialization.
AHRS NOT READY, DO NOT TAKE OFF	AHRS is not initialized. Appears prior to AHRS initialization on the ground.
AHRS NOT READY, KEEP WINGS LEVEL	AHRS is not initialized. Appears prior to AHRS initialization in the air.
ATTITUDE	Flight dynamics monitor detects a possible discrepancy in AHRS outputs.
CALIBRATE AHRS/MAG	AHRS calibration is incomplete. Service required. Contact a Garmin dealer for support.
MAG ANOM¹	AHRS detects a magnetic anomaly. Consult flight manual supplement for any required action. May appear as an advisory depending on configuration.

¹ Applicability and alert type are dependent upon configuration.

LRU FAILURE EFFECT ON AHRS OPERATION

LRU	CONDITION	AHRS OUTPUT
ADC	ADC input fails during normal mode.	Valid attitude and heading information (no interruption).
	ADC input fails while AHRS is in NO-GPS mode.	Attitude and heading fails.
GPS Navigator	GPS inputs fail. AHRS continues to operate in No-GPS mode (ADC and magnetometer required).	Valid attitude and heading information (no interruption).
Magnetometer	Magnetometer input fails while aircraft is on the ground. AHRS transitions to back-up mode without magnetometer.	Valid attitude information (no interruption). Heading displays using GPS track data.
	Magnetometer input fails while aircraft is in the air. AHRS transitions to no magnetometer mode.	Valid attitude information (no interruption). Invalid heading information. Display enters reversionary track mode.
GPS and Magnetometer	Magnetometer and GPS inputs fail.	Attitude and heading fails.

Miscompare & No Compare

AHRS/ADC SENSOR COMPARISON

The GDU continuously monitors and compares sensor data when more than one AHRS or ADC is present. A miscompare state occurs when the difference between two or more sensor outputs is above the predefined threshold.

MISCOMPARE THRESHOLDS		
PARAMETER	CONDITION	VALUE
Altitude ¹	All	200 feet
Heading ²	All	6°
Indicated Airspeed ¹	Both <35 kts	Inhibited
	Either or both ≥35 kts, and both <80 kts	10 kts
	Either ≥80 kts	7 kts
Lateral Acceleration	All	0.2G
Normal Acceleration	All	0.2G
Pitch Attitude	All	5°
Pitch Attitude Rate	All	1° per second
Roll	All	5°
Roll Rate	All	3° per second

¹ ALT and IAS miscompares do not display when SSEC states do not match (i.e., one ADC, either the primary or standby (not both), is configured for SSEC while the other is not).
² Depends upon configuration.

MISCOMPARE ANNUNCIATIONS

Miscompares generate a caution annunciation over the affected PFD instrument(s).



Indicated airspeed, "IAS," annunciates below the airspeed pointer.



Altitude, "ALT," annunciates below the barometric altitude pointer.



Pitch and/or roll, "ATTITUDE," annunciates on the pitch scale.

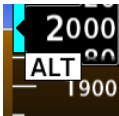
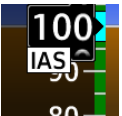


Heading, "HDG," annunciates below the current heading.

In addition to miscompare monitors, the GDU provides an independent flight dynamics monitor that compares computed cross heading, vertical accelerations, and roll angles against reported values generated by the AHRS. If there is a disagreement between the flight dynamics monitor and the AHRS parameters, the "ATTITUDE" caution annunciates.

NO COMPARE ANNUNCIATIONS

A no compare monitor determines when data to the second ADC or AHRS source is unable to be used for comparison.



No compares generate an advisory annunciation over the affected instrument(s) driven by ADC and AHRS units.



HSI Failure Modes



NOTE

TAWS/terrain alerting is unavailable in the absence of a certified GPS source.

The HSI continues to display limited navigation information during magnetometer and GPS failures. If the magnetometer input fails, the AHRS transitions to backup No-Magnetometer mode and continues to output valid attitude information.

If the aircraft is airborne, the PFD heading output reverts to GPS track. Magnetic heading is replaced by GPS track information and HSI compass card orients to GPS track. The following TXi system conditions exist:

- SVT turns off and automatically re-enables when heading is restored
- Map orientations change from HDG UP to TRACK UP
- When heading is restored orientations change back to HDG UP
- When heading fails heading bug remains
- GDU continues driving autopilot heading error output using track in place of heading



**HSI with
Magnetometer Failure**

“HDG FAIL” annunciates when there is an AHRS heading data failure.

If the magnetometer fails, the HSI compass card indicates GPS track. The numeric heading value turns magenta to indicate GPS track is displayed.



**HSI with GPS and
Magnetometer Failures**

If both the magnetometer and GPS units fail:

- HSI reverts to a non-directional CDI
- Numeric heading display is removed
- Use the selected course control to manually control OBS

GPS Failure



WARNING

Do not use projected position data as the only means of navigation.

When dead reckoning mode is active on the GPS navigator:

- System flags all external outputs dependent on GPS position data
- Map page reports “No GPS Position”; overlays are not available
- “DR Mode” annunciates on HSI, replacing ENR or OCN
- Traffic and Stormscope data continue to display on their respective pages
- Charts are available for display on the dedicated MFD page, but the ownship is no longer geo-referenced on the chart
- Terrain functionality not available
- CDI not available



Dead Reckoning

Backup GPS

Each GDU is equipped with a backup GPS receiver. When the optional Garmin backup GPS antenna is connected, and if all the certified GPS sources are lost, the system provides 2-D GPS position information from the backup GPS. No pilot action is required to enable the use of backup GPS data.

Only one GDU in a multi-GDU installation requires a connected backup GPS antenna. The system automatically cross-fills backup GPS data between GDUs.

Backup GPS data is used only to provide non-safety critical functions and only after the failure of all certified GPS position sources. Backup GPS data is only provided when a valid GPS position is received from an airborne certified GPS source that is subsequently lost. The system always utilizes time from the backup GPS when no other source is available, even prior to receipt of GPS time from a certified GPS source.

Backup GPS automatically turns off when you disable system GPS via a capable Garmin navigator.

Backup GPS data provides:

- Ownship position on the moving map
- GPS track display
- Waypoint bearing/distance information
- Ground speed
- Wind calculations

When backup GPS is in use:

- Yellow "BACKUP GPS" annunciation displays on all geo-referenced maps
- Navigation using backup GPS is not provided
- Map overlays for Stormscope, traffic, terrain, and obstacles are disabled
- Traffic page continues to display data from TIS-A and TCAS sources

Functions inhibited with backup GPS:

- Bearing pointer on Waypoint Info page
- Chart geo-referencing
- SVT
- GPS altitude displays
- GPS navigation status field
- TAWS/terrain alerting
- Terrain page display

Emergency Modes



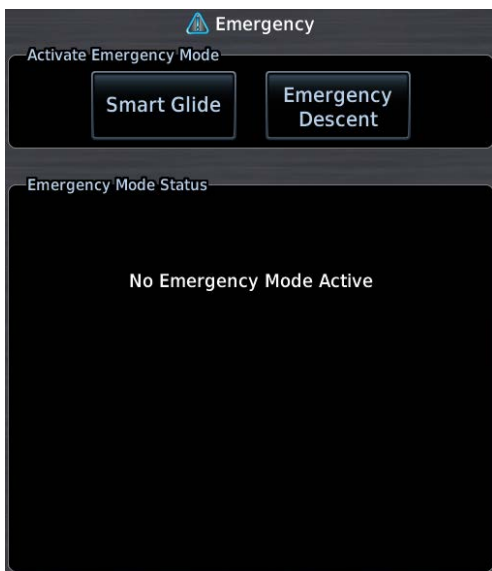
Emergency modes are available to assist you in the event of engine failure or a loss of cabin pressure.



NOTE

While emergency features can assist in workload reduction, it is the responsibility of the pilot in command to know and follow all published POH/AFM normal and emergency procedures.

MFD Emergency Page



Emergency modes are accessible via the Emergency page. Available options are dependent upon configuration.

A status window displays information related to the active emergency mode.

Contact a Garmin dealer to see if emergency features are available for your aircraft.

Smart Glide

Smart Glide

In the event of engine failure or partial power loss, this feature allows you to quickly locate and plot a direct course to the most suitable airport within glide range, avoiding terrain and obstacles along the way.



WARNING

Do not rely solely upon Smart Glide for navigation, airspeed and altitude management, or landing field selection. It is the pilot's responsibility to navigate, manage airspeed and altitude, and determine the best field for landing.



NOTE

Smart Glide is not an autonomous landing system. It indicates the latest appropriate time for the pilot to take control of the aircraft by issuing visual and aural "Maneuver and Land" alerts. It is the pilot's responsibility to disengage the autopilot (if present) and safely fly the approach and landing.

FEATURE REQUIREMENTS

- TXi software v3.30 or later
- GTN Xi series navigator with Smart Glide enabled (function host)

Smart Glide is a function of GTN Xi. For more information regarding functionality, consult *GTN Xi Series Pilot's Guide*.

Activate Smart Glide



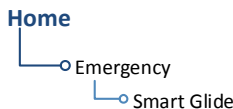
NOTE

Contact a Garmin dealer if your installation does not provide access to emergency features.

FEATURE LIMITATIONS

- Activation options are dependent upon installer configuration

WHERE TO FIND IT

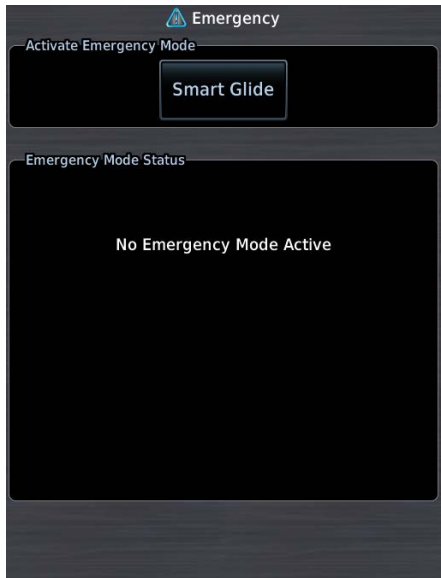


Activate Smart Glide when engine failure occurs. The **Smart Glide** activation key resides on the Emergency page.

From the MFD Home page:

Tap **Emergency** > **Smart Glide**.

MFD Emergency Page



A message in the Emergency Mode Status window informs you that no emergency modes are active.

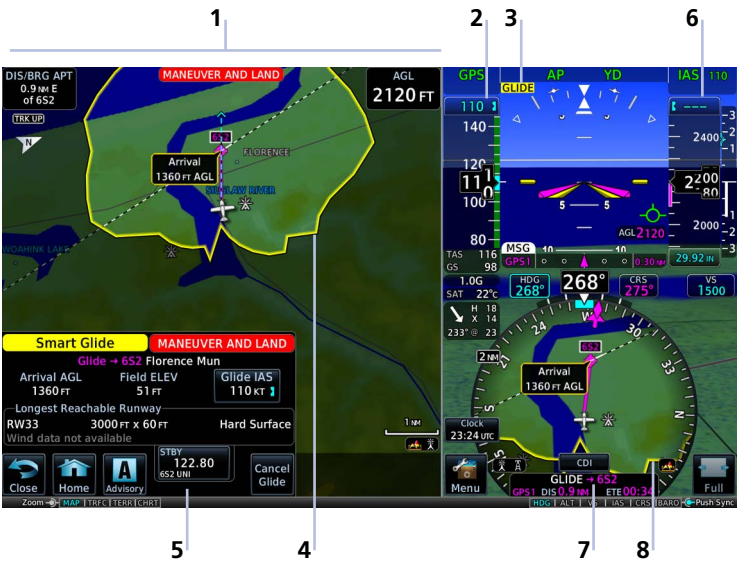
You may activate Smart Glide from any configured GTN Xi series navigator or MFD TXi series display unit. For information about all available means of activation, consult *GTN Xi Series Pilot's Guide*.

Active Mode Indications



Upon activation, Smart Glide calculates the glide route to a suitable destination airport and displays the information on the Emergency page.

Active mode indications and overlays appear on the MFD and PFD. A status information window is accessible when the Emergency page is not in use.



1	Alert Banner & User Fields	5	Status Information Window
2	Best Glide Airspeed	6	Invalid Altitude Setting
3	Active State Annunciation	7	Glide Route
4	Glide Range Ring Active on Map	8	Glide Range Ring Active on HSI Map

60% MFD

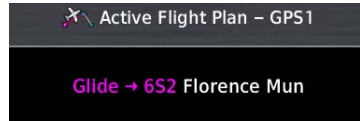


Map features automatically declutter and TOPO and Terrain overlays turn on if not already active. Map features revert to their previous settings when you cancel Smart Glide.

On the MFD:

Smart Glide replaces the active flight plan with a new direct course.

On Map, the yellow Smart Glide Range Ring reflects the measured glide ratio based on real-time data rather than the configured POH ratio.



Active flight plan leg indication depicts a direct course to the center of the airport. Runway Extensions turn on if not already active.

An altitude label indicates the estimated arrival AGL for the airport.



Glide key replaces the **Advisory** key when the status information window is not in use.

On the PFD:

The active route changes to a new direct course. Active flight plan leg indication depicts a direct course to the center of the airport.

Smart Glide Range Ring overlays on HSI Map.

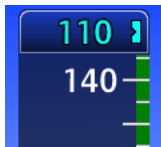
The CDI switches to GPS.¹



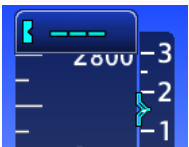
Full screen PFD



GLIDE key replaces the **Advisory** key when the status information window is not in use (full PFD only).



Selected airspeed control sets to the V_G of the GTN Xi series navigator.



Selected altitude value sets to invalid, displaying a series of dashes.

GLIDE

When GFC 500/600 is present, the "GLIDE" active status indication appears below the autopilot mode annunciation bar during IAS vertical mode.

¹ The GPS transmitting active Smart Glide status has priority.

View Smart Glide Status Information

Tapping **Glide** displays active route and longest runway details, the configured best glide speed for the aircraft, and available control options. This information is also accessible via the **Emergency** icon on the Home page.



When Smart Glide is active, this key replaces the **Advisory** key in the control bar.



Advisory key returns once the Emergency page/window opens or when you cancel Smart Glide.

Smart Glide Status Information Window

1

Smart Glide

APPROACHING AIRPORT

2

Glide → 6S2 Florence Mun

Arrival AGL 4940 FT Field ELEV 51 FT

Longest Reachable Runway

RW15 3000 FT x 60 FT Hard Surface

Wind data not available

3

STBY 122.80 6S2 UNI

Map

Cancel Glide

5

Glide IAS 110 KT

4

1	Mode Status Banner	4	Longest Reachable Runway Information
2	Active Route Display	5	Best Glide Speed Reference Indicator & Control
3	Smart Glide Controls		

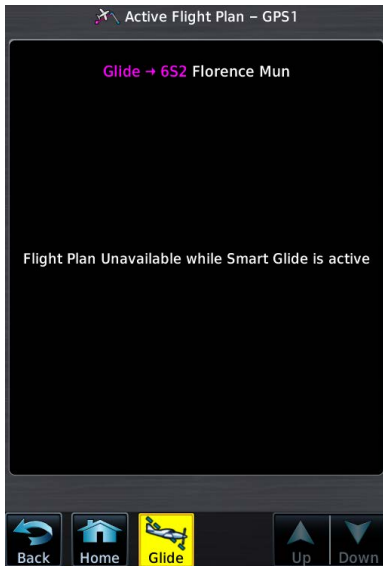
ACTIVE ROUTE DISPLAY



Shows the active direct route for gliding to the destination airport. Information includes:

- Active airport identifier
- Estimated arrival AGL
- Airport name
- Field elevation

Active Flight Plan, MFD



Active route identifiers also appear along the top of the active flight plan and in the GPS NAV status field of the PFD.

GPS NAV Status & HSI Map, PFD



Upon activation, Smart Glide replaces the active flight plan with a new direct course. A message informs you that the flight plan is unavailable while Smart Glide is active.

Arrival AGL (or *extra altitude*) is the aircraft's estimated height above ground level when crossing the center of the airport.

LONGEST REACHABLE RUNWAY INFORMATION



NOTE

Wind data display for airports without weather reporting if another airport with valid weather data is within 5 nm. The pilot is responsible for determining current wind direction and intensity.

View details about the destination runway.

Identification Number	Length and Width ¹	Surface Type (hard/soft)
RW35	5900 FT X 150 FT	Hard Surface
Headwind:	5 KT	Crosswind: 1 KT →

Headwind and Crosswind (if available)

RW13	5811 FT X 150 FT	Hard Surface
Tailwind: 19 KT	Crosswind: 8 KT ←	

Headwind data field changes color to indicate when tailwind conditions exist.

Advisory wind data received via datalink could be up to 90 minutes old.

“Wind data not available” Indication

This message displays in the absence of valid wind data.

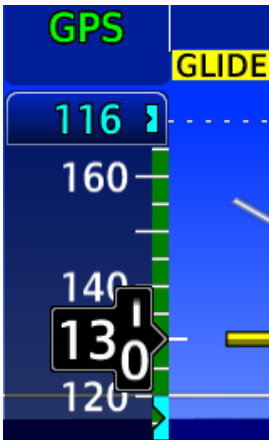
Wind data not available

¹ Runway length is always listed first followed by the runway width.

BEST GLIDE SPEED REFERENCE INDICATOR & CONTROL



Indicates the airspeed expected to follow the calculated glide route. Best glide speed is configured in the host navigator and may differ from PFD settings.



On units with a PFD, this reference field is selectable. Tapping it resets the airspeed reference control to the indicated Glide IAS value. This may be necessary if the selected airspeed is manually changed after activating Smart Glide.

On units without a PFD, this field is read-only.



Mismatch (yellow)



Glide IAS target airspeed and airspeed bug values do not match.

Synced (white)



Glide IAS and airspeed bug are in sync.

Multi-engine aircraft: The airspeed bug does not automatically set to V_G . The Best Glide Speed reference control still functions where applicable.

SMART GLIDE CONTROLS

Emergency Smart Glide Page

- COM Standby Frequency
- Cancel Glide
- Map

Available control options allow you to:

- Load the destination CTAF or tower frequency to a GTN Xi COM radio
- Cancel Smart Glide
- View Smart Glide information on Map

Monitor COM Standby Frequency



Monitor the COM standby frequency tuned in the host navigator. Information includes the station identifier and airport CTAF or tower frequency.

Tapping this key allows you to load the frequency to a GTN Xi COM radio.

For more about this feature, read *Remote Radio Frequency Entry* in section 5.

Deactivate Smart Glide



You may deactivate Smart Glide at any time by tapping **Cancel Glide**.



A pop-up message requests confirmation.

Tapping **OK** confirms the request.

Tapping **Back** closes the pop-up without deactivating Smart Glide.

Upon deactivation:

- Map features revert to their previous settings
- Active Flight Plan restores and activates the previous flight plan route
- Aural message "Smart Glide canceled" alerts flight crew of deactivation
- Autopilot (if present) remains in its current active modes or reverts to ROL lateral mode

View Smart Glide Information on Map



The **Map** key provides direct access to the moving map. Map shows a graphical representation of the active direct-to course for the suggested airport. Shading denotes areas estimated to be unreachable on glide.

Smart Glide Map Features



Tapping **Glide** opens the Smart Glide status information window.

An alert banner displays textual warning, caution, and advisory annunciations. User fields indicating the aircraft's present AGL and its distance/bearing from the destination airport display on a solid black background for greater visibility. They appear in the upper corners of each configured map display as shown.



Abnormal Operations

Other map features declutter to remove unnecessary controls and overlays.

Smart Glide Inactive



Smart Glide Active



Data depicted during Smart Glide

- Active glide route
- AGL and DIS/BRG APT user fields
- Airports
- Alert banner
- Estimated Arrival AGL label
- Glide Range Ring
- METARs
- Runway Extensions
- Terrain overlay
- TOPO overlay

Data removed during Smart Glide

- Airways
- Best Glide Airport indicator
- Charts overlay
- Fuel Range Ring
- Heliports
- Intersections
- METAR product timestamp
- NAV Range Ring
- NDBs
- TOPO scale

Emergency Descent

Emergency Descent

Emergency Descent Mode (EDM) assists pilots of pressurized aircraft in the event of cabin depressurization.

FEATURE REQUIREMENTS

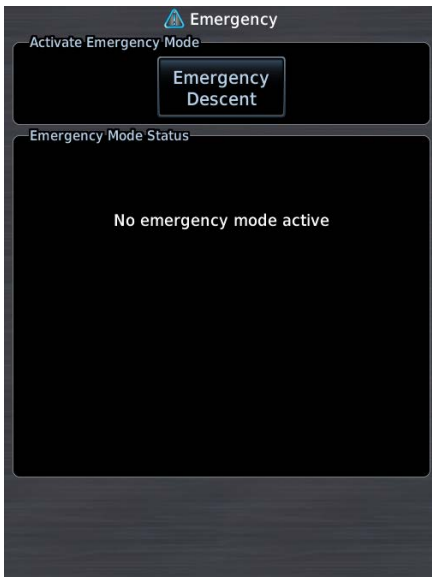
- GDU 700()/1060/1210 PFD
- GFC 600 autopilot configured for EDM
- EDM must be armed in order for activation to occur

FEATURE LIMITATIONS

Availability and the manner in which activation occurs (manual or automatic) are dependent upon configuration.

EDM Activation

MFD Emergency Page



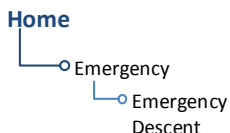
If configured for cabin pressure monitoring, the system monitors cabin pressure once EDM is armed. EDM is considered armed when:

- Autopilot is active
- Aircraft is above 15,000 ft MSL

Once armed, activation may occur manually or automatically.

MANUAL ACTIVATION

WHERE TO FIND IT



You can activate EDM one of two ways:

From the MFD Home page:

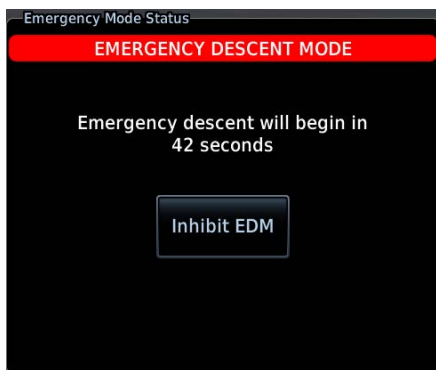
Tap **Emergency** > **Emergency Descent**.

From a dedicated external switch:

Enable the dedicated switch (if configured).

AUTOMATIC ACTIVATION

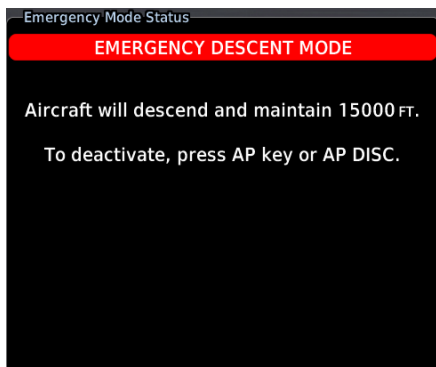
If the aircraft experiences decompression and cabin altitude increases beyond the threshold configured for the aircraft, the system automatically triggers EDM.



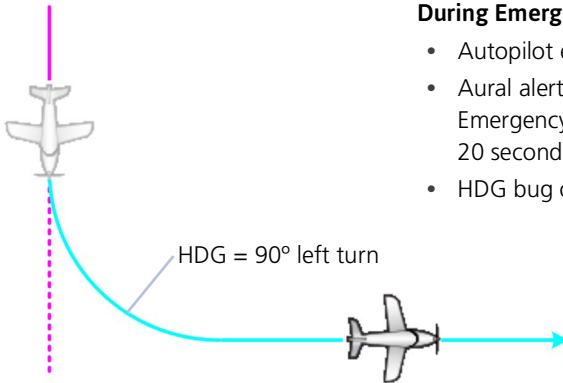
Upon Activation

The status window displays a warning message and timer.

Emergency descent occurs once the timer expires. This delay can range from 10 to 60 seconds.



If manually activated, you may skip the delay and activate EDM immediately by pressing the emergency descent switch or **Emergency Descent** key a second time.

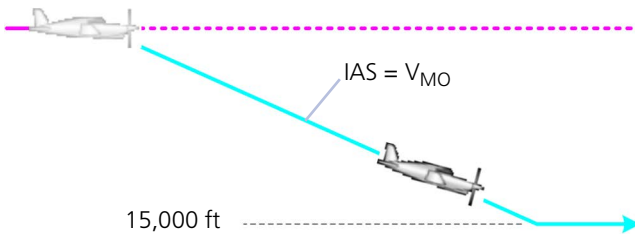


During Emergency Descent

- Autopilot enters HDG lateral mode
- Aural alert "Emergency Descent, Emergency Descent" repeats every 20 seconds
- HDG bug on PFD sets to 90° to left

Simultaneously:

- Autopilot enters IAS/FLC mode¹
- PFD sets airspeed bug at V_{MO} and altitude preselect bug to 15,000 ft



- Aircraft descends at V_{MO}
- Autopilot captures altitude hold mode at 15,000 ft

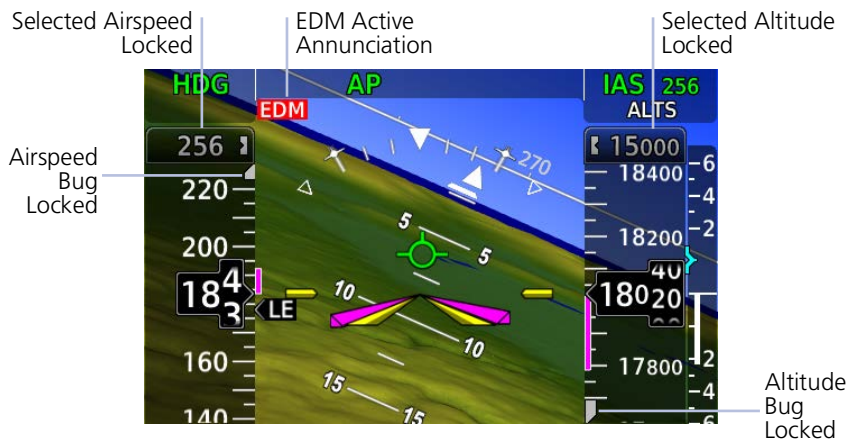
If available, the pilot may reduce power and deploy speed brakes to increase the rate of descent and reduce the time needed to reach a lower altitude. Ensure that all actions are in accordance with the POH.

¹ FLC mode applicable to TXi software v3.61 and later.

EDM ACTIVE INDICATIONS

ANNUNCIATION	STATE
EDM	Active
EDM	Override or Inhibit Mode

During EDM, airspeed and altitude bugs are locked and unchangeable. The active state annunciates in the upper left corner of the PFD.



EDM Selected Data Lockout

EDM Inhibit

Tapping **Inhibit EDM** prevents automatic activation once the delay timer expires. During inhibit mode:

- Automatic activation remains inhibited for a duration of 5 minutes
- An advisory message with timer counts down the remaining minutes/seconds
- Inhibit state annunciation appears beneath the AFCS status box





EDM Pop-up Warning, PFD

Once the 5 minute inhibit timer expires, the same pop-up warning appears regardless of whether EDM was triggered automatically or by the pilot.

IF	THEN
Pilot triggers EDM and then chooses to inhibit.	Automatic activation remains enabled.
EDM is triggered automatically and the pilot chooses to inhibit.	Manual activation remains available for as long as the aircraft is above 15,000 ft and the autopilot is engaged. ¹
The 5 minute inhibit timer expires and activation conditions are still present.	EDM automatically triggers again.
Pilot loses consciousness after choosing to inhibit the function.	Once the 5 minute inhibit timer expires, EDM automatically activates in order to bring the aircraft to a safer altitude.

¹ Dependent upon configuration.

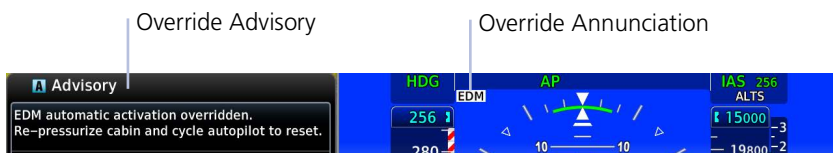
EDM Override

To deactivate EDM, disengage the autopilot. You may do this multiple ways:

- *On the PFD:* Tap **Menu > Test > AP Disconnect**.
- *On GMC 605:* Push the **AP** bezel key.
- *On the yoke:* Push the AP DISC button.

Activating autopilot will override EDM and return the autopilot to normal operation. In this state:

- EDM does not automatically reactivate until arming conditions are cleared
- An advisory message alerts you to re-pressurize the cabin and reset the autopilot
- Override state annunciation appears beneath the AFCS status box



AUTO EDM



This key allows you to manually enable or disable automatic activation. During system power up, this function is active by default.

When the function is off, manual activation of EDM is still available.

To reset EDM automatic activation following an override:

1. Descend below 15,000 ft.
2. Disengage and then re-engage the autopilot.

OR

1. Open the PFD Setup menu.
2. Toggle **Auto EDM** off and on.

Disabling the Auto EDM function may be desired if current cabin conditions are causing erroneous activation of EDM. This can occur when:

- Cabin altitude is above the configured threshold
- GDU reports incorrect cabin altitude

EDM Active Discrete Output

Active Discrete Output & Engine Torque Control

On the Socata TBM 850, the discrete output may be connected to the engine torque limiter. In such case, turning on EDM would energize the torque limiter circuit, resulting in overtorque protection while the aircraft is in emergency descent mode.

This capability is dependent upon installation and should not be relied upon to maintain the engine within torque limits.

It is the full responsibility of the pilot-in-command to operate the aircraft and engine within the limitations defined in the POH/AFM.

PFD provides a discrete output that allows synchronization of EDM with other systems in the aircraft.

- When EDM is on, the discrete output is active.
- When EDM is off, the output is inactive.

Use of the active discrete output is airframe and installation specific.

EDM Failures

ANNUNCIATION	CONDITION
EMERGENCY DESCENT MODE UNAVAILABLE. SERVICE REQUIRED.	EDM is disabled. TXi and GFC 600 V_{NE} , V_{MO} , and/or M_{MO} configurations do not match.

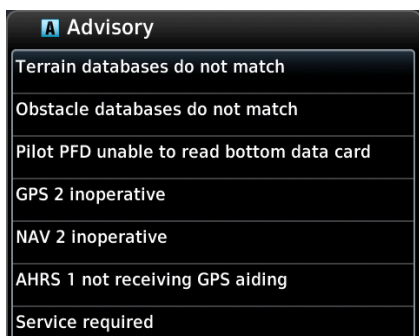
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11 Messages

ADVISORY MESSAGES..... 11-2

Advisory Messages

Advisories are system-related messages that display across all connected GDUs in the TXi system. The alert includes the ID relative to the display.



- Most recent advisories are at the top of list
- View-once advisories remain in queue until viewed by the pilot
- Persistent (or conditional) advisories remain active until the indicated condition is resolved
- Acknowledging an advisory on one GDU automatically acknowledges the message across all TXi units

All advisories are logged in the unit's internal storage. This log may be viewed in configuration mode and exported to an SD card.

Advisory Key



Tapping the **Advisory** key once displays the advisory list. Tapping it again acknowledges all active advisories and closes the list.

- Flashes in navigation bar when a new advisory is present
- Turns solid once all active advisories are acknowledged
- No longer displays after all active advisories are cleared

Audio Advisories

Contact the dealer for all audio related issues.

ADVISORY	CONDITION
Audio inoperative.	A failure occurred with audio device.
	An audio clip does not play.
	Communication with audio device is lost.

Battery Advisories

ADVISORY	CONDITION	CORRECTIVE ACTION
<GDU> battery unavailable: capacity test required.	Battery capacity test failure.	Service required. Acknowledge pop-up alert. Contact dealer for support.
	Annual capacity test overdue.	Testing required. Acknowledge pop-up alert. Contact dealer for support.
<GDU> battery unavailable: communication lost.	Battery communication is lost.	Service required. Acknowledge pop-up alert. Contact dealer for support.
<GDU> battery unavailable: fault detected.	A battery fault is present.	Service required. Acknowledge pop-up alert. Contact dealer for support.
<GDU> battery unavailable: power output failed.	Battery power output failure.	Service required. Acknowledge pop-up alert. Contact dealer for support.
<GDU> on battery power for <MM:SS>.	The indicated GDU is operating in battery mode.	Ensure that the aircraft is in a safe state. Correct the issue to restore electrical power to the GDU.

Database Advisories

ADVISORY	CONDITION	CORRECTIVE ACTION
AHRS 1/2 magnetic model database needs update.	AHRS 1/2 magnetic field model database is out of date. Occurs only when the aircraft is on ground.	Update IGRF database according to instructions in the <i>Databases</i> section.
<GDU> <Navigation/ SafeTaxi/Obstacle> database expires on <date>.	The indicated database is approaching expiration (PFD-only displays).	If available, load the next cycle of database into standby.
	The indicated database is expired (PFD-only displays).	Update indicated database, if necessary.
<GDU> <Navigation/ Terrain/Obstacle> database unavailable.	The indicated database is unavailable or corrupt.	Re-download and install the indicated database on the GDU. Contact a Garmin dealer for support.
<Navigation/ Terrain /Obstacle> databases do not match.	Two or more TXi GDUs have a different version of the indicated database.	Update the indicated database. For information about all installed databases, go to the System Status page.
<GDU> <SafeTaxi/ Basemap> database unavailable.	The indicated database is corrupt (PFD-only displays).	Re-download and install the indicated database, Contact a Garmin dealer for support.
	The indicated database is unavailable (PFD-only displays).	
Chart database expires on <date>.	The charts database is approaching expiration.	If available, load the next cycle of the charts database into standby.
	The charts database is expired Appears only on units with an MFD.	Update the database if necessary for operation.
<GDU> Chart database incomplete. Some charts unavailable.	Charts database verification failure.	Re-download and install the indicated database. Contact a Garmin dealer for support.
Chart database unavailable.	The configured charts database is unavailable (MFD displays).	Re-download and install the indicated database, Contact a Garmin dealer for support.

ADVISORY	CONDITION	CORRECTIVE ACTION
Chart streaming unavailable. Using installed chart database.	Chart streaming is not available. The GDU reverted to the currently installed charts.	Open the System Status page and check database synchronization status. If the problem persists, contact a Garmin dealer for support.
Database SYNC in progress. View System Status page for more info.	Database synchronization is in progress.	Wait for database synchronization to complete. Restart the GDU before attempting to use new databases.

Emergency Descent Mode Advisories

ADVISORY	CONDITION	CORRECTIVE ACTION
Emergency Descent Mode ACTIVATE switch inoperative.	EDM switch inoperative.	Contact a Garmin dealer for service.
Emergency Descent Mode automatic activation unavailable. Service required.	EDM cabin pressurization is enabled on GDU, but GMC 605 is not providing cabin pressure altitude.	
Emergency Descent Mode unavailable. Service required.	EDM is enabled on GDU but not supported by GMC 605.	
Emergency Descent Mode automatic activation is inhibited.	Auto EDM toggle key in the PFD Setup menu is off.	Enable the Auto EDM function. On the PFD: Tap Menu > Auto EDM .

Engine System Advisories

ADVISORY	CONDITION	CORRECTIVE ACTION
<High/Low> <gauge name> advisory (e.g., High Oil Temperature, Low Battery Voltage).	The indicated engine parameter exceeds pilot-specified value.	Verify that parameter is within the operating limitations defined in the POH.
EGT Difference advisory. ¹	Pilot-specified EGT Diff is exceeded by one or more engines.	
CHT Cooling advisory. ¹	A cylinder's cooling rate is faster than pilot-specified CHT cooling rate.	
Low EST Fuel Remaining advisory.	Estimated fuel on board is below pilot-specified limit.	
Low Endurance advisory.	Remaining flight endurance is below pilot-specified limit.	
Automatic Ignition unavailable. ²	System is unable to activate automatic ignition (in-flight only).	Monitor engine status and environmental conditions and activate igniters manually in accordance with the POH. Contact dealer for service.
Service required.	The system or unit requires service. Appears only when the aircraft is on ground.	Contact dealer for service.
Service required. Automatic Starting inhibited due to <I/O> invalid. ³	The indicated input or output is not functioning. Automatic starting is inhibited as a result.	Contact dealer for service.
Fuel Balancing System is inoperative.	The system is not capable of determining fuel imbalance, and automatic balancing systems will not operate.	Contact dealer for service.

¹ Piston installations only. ² Turbine installations with automatic ignition enabled.

³ Turbine installations with automatic starting enabled.

Terrain Advisories

ADVISORY	CONDITION	CORRECTIVE ACTION
Service TAWS. Invalid config.	Terrain/TAWS alerting configuration is invalid.	Contact a Garmin dealer for support.
	Fixed Wing/Rotorcraft mismatch. The GDU's aircraft setting conflicts with the current external TAWS configuration.	
SVT could not be enabled.	The SVT feature cannot be unlocked with the current SD card.	Verify that the SVT feature unlock card is inserted in the GDU. If necessary, contact a Garmin dealer for support.
SVT has been successfully enabled. Restart all GDUs to complete enablement.	The SVT feature is enabled during the current power cycle.	To complete feature enablement, restart all configured GDUs.
<GDU> terrain database resolution insufficient.	Terrain database resolution does not support the current terrain alerting configuration (rotorcraft only).	Download and install the 2.9 arc-second Terrain database.
SVT Disabled.	Outside terrain database coverage area.	Contact a Garmin dealer for database options.
Terrain/SVT Free Trial Flight time remaining: XX HRS	Terrain Settings Menu displays the flight hours remaining in the SVT free trial period.	If SVT is desired, do one of the following: • Contact a Garmin dealer to complete feature enablement within the 11 hour free trial period • Visit flyGarmin.com and purchase the feature enablement

Traffic System Advisories

The following advisories pertain to various traffic system failures.

ADVISORY	CONDITION	CORRECTIVE ACTION
1090ES traffic receiver fault.	ADS-B LRU reports that it is unable to receive 1090 Extended Squitter traffic.	Service required. Contact dealer for support.
ADS traffic alerting function inoperative.	The ADS-B LRU reports a traffic alerting failure.	Service required. Contact dealer for support.
ADS traffic function inoperative.	The ADS-B LRU reports a failure with the ADS-B Traffic input.	Service required. Contact dealer for support.
TAS/TCAS function inoperative.	TAS/TCAS device reports a traffic failure.	Reset the TCAS device. Contact dealer for support.
	Communication with the TAS/TCAS device is lost.	
TAS/TCAS system in standby.	The TAS/TCAS device is in standby mode for longer than 60 seconds while airborne.	If traffic alerts are desired, enable the operating mode of the traffic device.
Traffic/FIS-B functions inoperative.	The ADS-B LRU reports a critical fault and is inoperative.	Service required. Contact dealer for support.
	Communication with the ADS-B LRU is lost.	Service required. Contact dealer for support.
UAT fault.	The UAT LRU reports a low battery or fan fault.	Service required. Contact dealer for support.
UAT traffic/data receiver fault.	The ADS-B LRU reports that it is unable to receive UAT traffic and FIS-B data.	Service required. Contact dealer for service.
TCAS RA display unavailable on VSI. Expand <GDU> to full screen mode to restore. ¹	The GDU is operating in display backup mode, and the current VSI format is not able to display RAs.	Change the VSI type by expanding the PFD using the FULL key.

¹ TCAS II equipped aircraft only.

Satellite Service Advisories

The following advisories pertain to various satellite service conditions and failures. For support regarding all satellite service advisories, contact a Garmin dealer.

ADVISORY	CONDITION
Connex weather receiver inoperative.	Communication with the GSR 56 is lost. The satellite weather service is in operative.
Connex weather service not registered.	GSR 56 weather service requires registration for operation.
SiriusXM receiver inoperative.	SiriusXM receiver failure.
	Communication with the GDL 69/69A is lost.

PFD Advisories

The following advisories pertain to various conditions and devices associated with the PFD.

ADVISORY	CONDITION	CORRECTIVE ACTION
ADC 1/2 error correction inoperative.	The indicated ADC reports that airspeed error correction is unavailable.	Contact dealer for service.
	The indicated ADC reports that altitude error correction (i.e., SSEC) is unavailable.	
ADC miscompare.	ADC data from one or more units is not within the tolerances of the other ADC units.	Ensure that a valid ADC source is selected on the Sensors page of the PFD menu. Verify by manually cross-checking available sources. Contact dealer for service.

ADVISORY	CONDITION	CORRECTIVE ACTION
ADC 1/2/STBY data is missing and cannot be compared. <No-compare alert inhibited.>	ADC data from one or more units is missing. Applicable to installations with more than one ADC. Note: In systems with three ADC sources, the text “No-compare alert inhibited” is added to the advisory message. PFD no compare flags are removed once the advisory is viewed.	Ensure that a valid ADC source is selected on the Sensors page of the PFD menu. Verify by manually cross-checking available sources. Contact dealer for service.
Not enough ADC sensors to perform a comparison. <No-compare alert inhibited.>		
<GDU> ADS-B in traffic does not match configuration.	The primary ADS-B traffic source is not available. GDU is relying on data from backup source.	Contact a Garmin dealer for support.
AHRS 1/2 magnetometer inoperative.	The indicated AHRS is not receiving valid GMU data, either due to a loss of communication or a GMU fault.	Contact dealer for service.
AHRS 1/2 not receiving GPS aiding.	The indicated AHRS is not receiving GPS data from any source.	Ensure that the navigator(s) is on and receiving a GPS signal. Check the AFMS for limitations. Contact dealer for service.
AHRS GPS aiding has been disabled.	Pilot toggles the AHRS GPS Aiding function off.	Re-enable AHRS GPS aiding. From the PFD menu, tap Sensors , and toggle the AHRS GPS Aiding key on.
AHRS miscompare.	AHRS data from one or more units is not within the tolerances of the other AHRS units.	Ensure that a valid AHRS source is selected on the Sensors page of the PFD menu. Verify by manually cross-checking available sources. Contact dealer for service.
AHRS 1/2/STBY miscompare.	When tie-breaker voting is available, the message states the specific unit exceeding tolerances (AHRS 1, AHRS 2, or STBY AHRS).	

ADVISORY	CONDITION	CORRECTIVE ACTION
AHRS 1/2/STBY data is missing and cannot be compared. <No-compare alert inhibited.>	AHRS data from one or more units is missing. Applicable to installations with more than one AHRS.	Ensure that a valid AHRS source is selected on the Sensors page of the PFD menu.
Not enough AHRS sensors to perform a comparison. <No-compare alert inhibited.>	Note: In systems with three AHRS sources, the text “No-compare alert inhibited” is added to the advisory message. PFD no compare flags are removed once the advisory is viewed.	Verify by manually cross-checking available sources. Contact dealer for service.
Backup GPS not available, system GPS is disabled via navigator.	Pilot disables system GPS via the manual setting on the primary navigator(s), resulting in the loss of backup GPS.	Re-enable system GPS on all primary navigators.
<GDU> air data input failure. V_{NE} is uncorrected.	The indicated GDU is not receiving temperature or pressure altitude data. Applicable to rotorcraft only.	Refer to airspeed limitations placard for applicable V_{NE} .
<GDU> demo mode, do not use in flight.	The unit is in demonstration mode.	To exit demo mode: Cycle power to the GDU. If the problem persists, contact dealer for service. Operating the GDU in demo mode during flight is prohibited.
GPS 1/2 inoperative.	The indicated GPS source failed.	Ensure that the navigator(s) is on and receiving a GPS signal. Switch to an alternate navigation source. Contact dealer for service.
	Communication with the indicated GPS is lost.	
NAV 1/2 inoperative.	The indicated NAV source is failed, or input data is not received.	Switch to an alternate navigation source.
<LRU> reports service required.	The indicated LRU reports an internal fault and requires service. Appears only when the aircraft is on ground.	Contact dealer for service.

ADVISORY	CONDITION	CORRECTIVE ACTION
Reset altitude preselector to enable VPATH descent.	TOD ahead. Vertical deviation is available and the selected altitude is < 75 ft below current altitude. Appears only for aircraft equipped with GFC 600.	Lower the selected altitude to at least 75 ft below the current aircraft altitude.
<GDU> backlight calibration lost.	The GDU cannot read LED calibration data, or the data is corrupt. Sunlight readability may be degraded. Appears only when the aircraft is on ground.	Contact dealer for service.
<GDU> controller unavailable.	Communication with GCU 485 is lost. Appears only when the aircraft is on ground.	Operate the indicated GDU using the touchscreen controls.
Service required.	The system or unit requires service. Appears only when the aircraft is on ground.	Contact dealer for service.
Service soon.	The system or unit detects an internal fault but continues to function. Appears only when the aircraft is on ground.	Contact dealer for service.

System Hardware Advisories

ADVISORY	CONDITION	CORRECTIVE ACTION
<GDU> controller <ALT ARM/GPSS/ CDI/VIS ENG/XFR/ BANK> key stuck.	The indicated key is stuck on GCU 485.	Push the indicated key again. If the key remains stuck, use the corresponding touchscreen controls. Contact dealer for service.
<GDU> controller <HDG/BARO/VIS/ SPD> knob-push stuck.	The indicated knob is stuck on GCU 485.	Push the indicated knob again. If the knob remains stuck, use the corresponding touchscreen controls. Contact dealer for service.
<GDU> controller has multiple stuck controls.	The unit detects multiple stuck controls on the GCU 485.	Use the touchscreen controls for all manual operations. Contact dealer for service.
<GDU> cooling fan failed.	The indicated GDU detects a cooling fan fault. PFD/MFD coloration may be incorrect. The backlight may dim to reduce power and heat.	Extended operation at high temperatures is not recommended as damage to the unit may occur. Contact dealer for service.
<GDU> display hot, display dimmed.	Unit temperature exceeds the over temperature threshold.	Decrease cabin temperature and increase cabin airflow near the GDU. If the problem persists, contact the dealer for service.
<GDU> <left/right/both> knob-push stuck.	The indicated GDU knob(s) is stuck.	Use the touchscreen controls for all manual operations.
Electronic Stability Protection failed.	Garmin Mode Controller signals an ESP system fault.	Consult the pilot's guide for the Garmin Mode Controller.

ADVISORY	CONDITION	CORRECTIVE ACTION
<GDU> unable to read <top/bottom/left/right> datacard.	The SD card in the indicated slot is unreadable or corrupt (i.e., data is unavailable).	Reformat the SD card using an external computer. Re-insert the SD card in the slot. If the problem persists, contact the dealer for service.
	The SD card is missing from the indicated slot.	Re-insert the SD card in the slot.
	User ejects Flight Stream 510 wireless transceiver. Network connection lost.	Restart GDU to reconnect.
<GDU> <top/bottom/left/right> datacard full.	The SD card in the indicated slot is approaching full. May not occur with 32 GB SD cards.	Remove the SD card once the aircraft is on-ground and the unit is powered off. Insert a card with sufficient free space and power on the unit to resume writing data.

Weather Service Advisories

ADVISORY	CONDITION	CORRECTIVE ACTION
Possible severe weather ahead. Check Weather Radar.	Weather alert detected within +/- 10 degrees of the aircraft heading. This advisory does not appear when the Weather Radar page is active on any TXi unit.	To view information about severe weather in the current flight path, open the Weather Radar page.
Stormscope inoperative.	The GDU is not receiving data from the Stormscope unit.	Consult the appropriate third-party documentation.
	Stormscope reports a failure status.	
Weather radar needs service.	Weather radar reports a degraded or inoperative condition exists. Appears only when the aircraft is on ground.	Contact dealer for service.
Weather radar inoperative.	The unit is not receiving weather radar data.	Contact dealer for service.

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12 Qualification

GLOVE QUALIFICATION 12-2



NOTE

The procedures described in this section are not authorized for completion during flight. All tasks must be performed while the aircraft is on the ground.

Glove Qualification

The touchscreen uses capacitive touch technology to sense the proximity of skin to the display. A glove creates a barrier between the skin and the display glass, potentially reducing the ability of the display to detect touches.

This procedure qualifies a specific glove for use with the touchscreen. Due to differences in finger size, glove size, and touchscreen between the GDU 700(), GDU 1060, and GDU 1210 units, the qualification procedure is specific to the pilot/glove and GDU combination. Multiple units must be evaluated individually.

GLOVE SELECTION CONSIDERATIONS

- Thinner gloves perform better than thicker gloves.
- Leather gloves, and gloves designed specifically for use with capacitive touchscreen devices, are often found to be acceptable.
- To improve touchscreen sensitivity while wearing gloves, use the pad of your finger instead of the tip during touch interactions.

GLOVE QUALIFICATION GUIDANCE

- Checklist 1 contains the tasks required to qualify a glove.
- Checklist 2 contains tasks that are not required to qualify a glove, but may limit how some functions are accessed while wearing a glove.

GLOVE QUALIFICATION STEPS

Complete only the tasks for the capabilities relevant to the installed GDU(s).

1. Sit in the pilot's seat.
2. Start the GDU in normal mode.
3. Perform the tasks listed in each checklist, tapping each key with a non-gloved finger. It is not necessary to record any results for this step.
4. Repeat step 3 using a gloved hand.
5. For each task, determine whether the touchscreen response is the same or worse than without the glove.
6. Record the results in the applicable column. Items that may cause the operation to be worse include, but are not limited to:
 - Multiple attempts to select a key
 - Unintentional selection of adjacent key(s)
 - Excessive force on the touchscreen to select a key
7. If all applicable tasks produce the same response with and without a glove, the pilot may use the glove in flight.

GLOVE QUALIFICATION PROCEDURE

Pilot: _____

Glove Description: _____

Circle the applicable TXi GDU and system function(s).

700L 700P 1060 1210 / PFD MFD EIS

CHECKLIST 1 REQUIRED TASKS

TASK	OPERATION WITH GLOVE (CIRCLE ONE)	
PFD		
Change the current CDI source.	Same	Worse
Tap HDG	Same	Worse
Change AP HDG REF setting.	Same	Worse
Close control menu.	Same	Worse
Tap Barometric Setting Control	Same	Worse
Toggle STD BARO key on and off.	Same	Worse
Close	Same	Worse
Menu	Same	Worse
Minimums	Same	Worse
BARO	Same	Worse
Type a minimums altitude value.	Same	Worse
Enter	Same	Worse
Back	Same	Worse
HSI Setup	Same	Worse
Bearing Pointers	Same	Worse
Toggle Bearing Pointer 1 on and off.	Same	Worse
Back	Same	Worse
PFD Setup	Same	Worse
Clock/Timer	Same	Worse
Enable clock/timer.	Same	Worse
Exit PFD Setup.	Same	Worse

TASK	OPERATION WITH GLOVE (CIRCLE ONE)	
Close menu.	Same	Worse
Clock/Timer Control	Same	Worse
Enable timer mode (if necessary).	Same	Worse
Start and stop timer.	Same	Worse
Close control menu.	Same	Worse
Advisory	Same	Worse
Close advisory panel.	Same	Worse
MFD		
Home	Same	Worse
Waypoint Info	Same	Worse
Airport	Same	Worse
Tap WP Identifier key.	Same	Worse
Type a new airport identifier.	Same	Worse
Enter	Same	Worse
Add each of the following waypoints in the same manner.		
KPRC	Same	Worse
LOTKE	Same	Worse
EIS		
For GDUs with integrated EIS only.		
Home	Same	Worse
Engine	Same	Worse
Advance selector through each EGT/TIT cylinder.	Same	Worse
Toggle Lean Mode key on and off.	Same	Worse
Menu	Same	Worse
Advisories	Same	Worse
Type a temperature value for the High Oil TEMP advisory.	Same	Worse

TASK	OPERATION WITH GLOVE (CIRCLE ONE)	
Enter	Same	Worse
Return temperature value to its previous setting.	Same	Worse
Back	Same	Worse
Engine & Flight Hours	Same	Worse
Back	Same	Worse
Exit menu.	Same	Worse

CHECKLIST 2

NON-REQUIRED TASKS

TASK	OPERATION WITH GLOVE (CIRCLE ONE)	
PFD		
Menu	Same	Worse
HSI Setup	Same	Worse
Enable HSI Map.	Same	Worse
Exit menu.	Same	Worse
With one finger on the map, swipe up to zoom in, swipe down to zoom out.	Same	Worse
MFD		
Open the Map page.	Same	Worse
With two fingers on the map, stretch to zoom in, pinch to zoom out.	Same	Worse
With one finger on the map, swipe in any direction to pan.	Same	Worse
Menu	Same	Worse
Map Setup	Same	Worse
Drag to the bottom of the Map tab list.	Same	Worse
LAT/LON Lines	Same	Worse
Exit menu.	Same	Worse
EIS		
None.		

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13 Glossary

A

ACT	Altitude Compensated Tilt
ADAHRS	Air Data/Attitude & Heading Reference System
ADC	Air Data Computer
ADF	Automatic Direction Finding Equipment
ADI	Attitude and Direction Indicator
ADIZ	Air Defense Identification Zone
ADS-B	Automatic Dependent Surveillance Broadcast
AFCS	Automatic Flight Control System
AFM	Aircraft Flight Manual
AFMS	Aircraft Flight Manual Supplement
AGL	Above Ground Level
AHRS	Attitude Heading Reference System
AIM	Airman's Information Manual
AIRB	Basic Airborne Application
ALT	Altitude Hold
AOA	Angle of Attack
AP	Autopilot
ATC	Air Traffic Control
ATCRBS	Air Traffic Control Radar Beacon System
ATK	Along Track

C

CAT	Carburetor Air Temperature
CDI	Course Deviation Indicator
CDU	Control and Display Unit
CHT	Cylinder Head Temperature
CRS	Course
CTAF	Common Traffic Advisory Frequency
CWA	Center Weather Advisory

D

DG	Directional Gyro
DH	Decision Height
DME	Distance Measuring Equipment

E

EDM	Emergency Descent Mode
ECR	Excessive Closure Rate
EDR	Excessive Descent Rate
EGT	Exhaust Gas Temperature
EIS	Engine Indication System
ENR	En Route
ESP	Electronic Stability and Protection
EXT	External

F

FAF	Final Approach Fix
FDC	Flight Data Center
FIS-B	Flight Information Services Broadcast
FIT	Flight Into Terrain
FLC	Flight Level Change
FLTA	Forward Looking Terrain Avoidance
FMS	Flight Management System
FPM	Feet Per Minute
FS	Flight Stream

G

GCS	Ground Clutter Suppression
GDC	Garmin Air Data Computer
GDL	Garmin Data Link
GDU	Garmin Display Unit
GEA	Garmin Engine and Airframe
GFC	Garmin Flight Controller
GMC	Garmin Mode Controller
GP	Glidepath
GPS	Global Positioning System
GPSS	Global Positioning System Steering
GPWS	Ground Proximity Warning System
GRS	Garmin Reference System
GS	Glideslope
GSL	Geometric Sea Level
GSU	Garmin Sensing Unit
GTP	Garmin Temperature Probe
GWX	Garmin Weather Radar

H

HDG	Heading
HOT	Hazardous Obstacle Transmission
HPM	Heading Preset Mode
HSI	Horizontal Situation Indicator
HTAWS	Helicopter Terrain Awareness and Warning System

I

IAF	Initial Approach Fix
IAS	Indicated Airspeed
IAT	Induction Air Temperature
IFR	Instrument Flight Rules
IGRF	International Geomagnetic Reference Field
ILI	Imminent Line Impact
ILS	Instrument Landing System
ITT	Interstage Turbine Temperature
IOI	Imminent Obstacle Impact
ISA	International Standard Atmosphere
ITI	Imminent Terrain Impact

K

KIAS	Knots Indicated Airspeed
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L

LDI	Lateral Deviation Indicator
LOA	Letter of Authorization
LOC	Localizer
LRU	Line Replaceable Unit

M

MAP	Missed Approach Point
MAHP	Missed Approach Holding Point
MBR	Master Boot Record
MDA	Minimum Descent Altitude
METAR	Meteorological Terminal Aviation Routine Weather Report
MFD	Multi-Function Display
MGT	Measured Gas Temperature
MOA	Military Operations Area
MSL	Mean Sea Level

N

NAVAID	Navigation Aid
NAS	National Airspace System
NCR	Negative Climb Rate
NDB	Non-Directional Beacon
NEXRAD	Next-Generation Radar
NOTAM	Notice to Airmen
NRST	Nearest
NVIS	Night Vision Imaging System

O

OAT	Outside Air Temperature
OBS	Omni Bearing Selector
OCN	Oceanic

P

PCL	Pilot Controlled Lighting
PDA	Premature Descent Alert
PFD	Primary Flight Display
PVT	Position, Velocity, and Time

R

RAT	Ram Air Temperature
RLC	Reduced Line Clearance
ROC	Reduced Required Obstacle Clearance
RTC	Reduced Required Terrain Clearance

S

SAT	Static Air Temperature
SBAS	Satellite-Based Augmentation System
SD	Secure Datacard
SSEC	Static Source Error Correction
SSID	Service Set Identifier
SURF	Surface Situation Awareness
SVT	Synthetic Vision Technology
SXM	SiriusXM Weather

T

TA	Traffic Advisory
TACAN	Tactical Air Navigation
TAF	Terminal Aerodrome Forecast
TAS	Traffic Advisory System
TAT	Total Air Temperature
TAWS	Terrain Awareness and Warning System
TCAS	Traffic Alert and Collision Avoidance System
TERM	Terminal
TFR	Temporary Flight Restriction
TIS	Traffic Information Service
TIT	Turbine Inlet Temperature
TSO	Technical Standard Order

U

UTC Universal Time Coordinated

V

VDI Vertical Deviation Indicator

VFR Visual Flight Rules

VLOC VOR/Localizer

VNAV Vertical Navigation

VOR Very High Frequency Omni-directional Range

VRP Visual Reporting Point

VS Vertical Speed

VSI Vertical Speed Indicator

W

WAAS Wide Area Augmentation System

WATCH Weather Attenuated Color Highlight

WP Waypoint

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14 Regulatory Information

SOFTWARE LICENSE AGREEMENT 14-2

Software License Agreement

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